

Optimal sensor configuration for ultrasonic guided-wave inspection based on value of information

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Abstract

Condition-based maintenance critically relies on efficient and reliable structural health monitoring systems, where the number, position and type of sensors are determined according to rational and principled criteria. This paper proposes the use of the *value of information* and the relative expected information gain as optimality criteria to determine the best number and positions of sensors, respectively. The proposed methodology is general, but in this paper it is specialized for ultrasonic guided-wave optimal system configuration. Two case studies are used to illustrate the suitability of the proposed methodology in providing the optimal sensor configuration of an ultrasonic guided-wave based structural health monitoring system. The results confirm the value of information as an efficient and rational index to compare among different sensor positioning strategies, while accounting for the underlying modeling and measurement uncertainties. As key contribution, a novel framework that trades-off between amount and cost of information is provided. The results show that geometrically unconstrained sensor configurations are preferred, since they provide a healthier balance between the amount of information and the benefit of such information.

Keywords: Optimal sensor configuration, Value of information, Bayesian inverse problem, Ultrasound, Time of flight, SHM

1. Introduction

Recent advances in structural health monitoring (SHM) are enabling a progressive transition from scheduled preventive maintenance to condition-based predictive maintenance in safety-critical industries such as the aerospace industry [1, 2]. These maintenance activities have a profound impact not only on safety but also on cost, since the service needs to be suspended during inspection and maintenance. Therefore, higher reliability, availability, and lower operation and maintenance costs are desired outcomes of condition-based maintenance, which highly relies on the amount, coverage, and accuracy of the SHM data. Theoretically, an infinitely reliable system would require an infinite amount of information; however such a theoretical rule finds an exception in practice when features related to the

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Nomenclature

$b(\cdot)$	benefit function	n_s	maximum potential number of sensors
B'	maximum prior expected benefit	\mathcal{N}	Gaussian distribution
B''	maximum posterior expected benefit	$P(\cdot)$	probability
BIP	Bayesian inverse problem	$p(\cdot)$	probability density function
BEIG	benefit of the expected information gain	PEB	posterior expected benefit
CVI	conditional value of information	RCI	relative cost of implementation
\mathcal{C}	sensor spatial configuration	$\text{ToF}_{\mathcal{D}}^{(a-s)}$	measured ToF between actuator a and sensor s
$D^{(k)}$	mean of the hyper-robust model for the k -th sensor	$\text{ToF}_M^{(a-s)}$	modeled ToF between actuator a and sensor s
\mathbf{D}	set of data $D^{(k)}$ for all the sensors	\mathcal{U}	uniform distribution
$\mathcal{D}^{(k)}$	signal acquired in the k -th sensor	γ	threshold parameter for AIMS algorithm
e	error term in ToF model	γ_p	performance index of the sensor configuration
E	Young's modulus	ε_η	error of damage reconstruction
EVI	Expected value of information	ν	Poisson ratio
$f(n)$	inverse cost function	ρ	density
$\text{KL}(\cdot\ \cdot)$	Kullback-Leibler divergence	Θ	set of possible values of the parameters in the BIP
n'_{opt}	optimal configuration under the prior expected benefit	θ	set of ToF model parameters
n''_{opt}	optimal configuration under the posterior expected benefit		
\mathfrak{N}	space of potential configurations		

9 complexity of the monitoring system such as the cost or the weight, are taken into consideration. The latter suggests
10 a trade-off between reliability and complexity for SHM design that needs to be assessed rigorously for optimal SHM
11 functionality. In the particular case of ultrasonic guided-wave based SHM, the aforementioned trade-off is especially
12 relevant since uncertainty factors such as sensor noise, material property uncertainties, and variations in the acquisition
13 and generation equipment, are well-known sources of complexity which limit the monitoring functionality. For this
14 reason, several authors have recently adopted the concept of *value of information* [3, 4] to investigate optimal SHM
15 designs based on a healthy and principled balance between the amount and the benefit of the monitoring information.
16 For example, the value of information has been adopted to address the maintenance decision optimization in [5, 6] by
17 using SHM data [7, 8], reliability methods [9], and influence diagrams [10]. Besides, it has also been used to assess
18 the value of SHM systems based on structural risks, integrity management, service life costs, pre-posterior analysis,
19 and Bayesian decision theory [11–14]. In the context of optimal sensor placement, the value of information has been
20 adopted to optimize spatially distributed systems, which allow for information propagation [15], although with known
21 modularity issues [16]. However, none of these contributions to date has focused on ultrasonic guided-wave based

22 SHM, precisely where the benefits of this decision-theoretic concept can be fully exploited due to the high reliance of
23 ultrasonic information on both the number and position of the sensors.

24 In the context of ultrasonic SHM systems design, several approaches have been proposed in the literature to op-
25 timize the number of piezoelectric (PZT) sensors and/or their locations in plate-like structures in both *active* and
26 *passive sensing diagnosis* using ultrasonic guided-waves. Passive sensing diagnosis techniques consider an arbitrary
27 number of sensors in listening-mode which capture sudden changes in the structure, such as impacts or crack growth.
28 Alternatively, active sensing diagnosis techniques consider the interaction of the sensors with the structure by means
29 of emission and reception of guided waves. In both cases, several attempts have been proposed to optimize the
30 SHM performance (number of sensors and location) through optimization methods such as artificial neural networks
31 combined with (1) genetic algorithms [17–20], whose efficiency has been improved by using the trilateration tech-
32 nique [21], (2) simulated annealing to achieve complete coverage and discrimination [22], and (3) particle swarm
33 optimization [23]. Alternatively, the optimization has been addressed by the use of performance indexes such as
34 the *probability of detection* [24–27] and the maximization of the *area of coverage* [28–31], whereby the ultrasonic
35 guided-wave properties and geometrical complexities are taken into account, as well as features like the influence of
36 faulty sensors [32]. The vast majority of these approaches rely on deterministic assumptions, and hence do not account
37 for the inherent uncertainties associated with ultrasound-based SHM, such as signal noise and uncertain material and
38 ultrasonic parameters. These uncertainties propagate across different models leading to uncertainty in the damage
39 reconstruction, which needs to be quantified for a rigorous optimal sensor location assessment.

40 Several Bayesian probabilistic approaches for optimal sensor configuration have been proposed in the literature,
41 but for applications other than guided-wave based SHM. For example, the position and/or the number of sensors in
42 structural systems have been optimized by making use of model-based Bayesian inverse problems (BIPs) and proba-
43 bilistic metrics such as (1) the Shannon-entropy of the *posterior* distribution [33–37], (2) the Kullback-Leibler (KL)
44 divergence between the *prior* and posterior distributions [38–42], or (3) the mutual information [43]. In analogous
45 applications, a *Bayesian experimental design* approach has been proposed, which provides a case-specific utility func-
46 tion, such as the KL divergence [44], whereby the maximization of its probabilistic expectation provides the optimal
47 experimental design [45]. As a general comment, these methods provide rigorous criteria for optimal sensor configu-
48 ration from an information-theoretic point of view, so that the more sensors the more information gained and the better
49 damage identification. Note that some of these approaches, which focus on the selection of the correlated prediction
50 error models, not only allow the accountability of uncertainties but also provide indications of the optimal number
51 of sensors [36]. However, in practice, decision-theoretic instead of information-theoretic criteria would be preferred,
52 where important variables such as the cost of the monitoring system can be rigorously taken into consideration for
53 optimal SHM system design.

54 In this paper, a novel and generic methodology for optimal sensor configuration in ultrasonic guided-wave based
55 SHM is provided. To take rigorously into account both information gain and cost-related benefits while accounting
56 for the underlying modeling and measurement uncertainties, the expected value of information is proposed as an op-

57 timality criterion. The value of information is defined as the difference between the maximum *prior* and *posterior*
58 expected benefits [8], and here it is proposed to be used as an objective function for obtaining the optimal number of
59 sensors. In particular, different sensor configurations with an increasing number of sensors are proposed according to
60 a forward sequential sensor placement [35] algorithm. Then, the optimal sensor position for each of these configura-
61 tions is determined by maximizing the information gained between the prior and the posterior PDF of the damage
62 identification parameters (e.g., damage position). The computation of the information gain is carried out using a dam-
63 age localization model based on a robust Bayesian methodology proposed by the authors in [46]. Next, the expected
64 value of information for each of the optimally-located sensor configurations is obtained, so that the optimal config-
65 uration is chosen as the one which maximizes the value of information. The methodology is illustrated using two
66 case studies for two different plate-like structures, namely (1) a flat aluminium panel and (2) an aeronautical hat-type
67 stiffened aluminum plate. The optimal sensor configuration is obtained considering the prior information about the
68 possible damage occurrence within a particular bounded area, whereby the data is simulated by using a time of flight
69 model [47]. Then, the effectiveness of the optimal sensor configuration is tested in a particular damage scenario by
70 simulating the structures in Abaqus and reconstructing the damage using the robust Bayesian damage localization
71 methodology provided in [46]. In general, the results show the efficiency of the proposed methodology in obtaining
72 the optimal number and position of sensors, in comparison with pure information based approaches. In particular,
73 the results reveal that non geometrically constrained configurations are preferred from a value of information point of
74 view, since they provide the best trade-off between amount of information and benefit of such information.

75 The remainder of the paper is organized as follows. Section 2 describes the proposed optimization methodology
76 based on the value of information and the fundamentals of Bayesian damage localization. Section 3 illustrates the
77 methodology through two case studies for two different plate-like structures. A discussion is provided in Section 4 to
78 investigate the influence of the parameters involved in the optimization result and to provide insight on the extensibility
79 of the proposed approach. Finally, Section 5 provides concluding remarks.

80 2. Methodology

81 2.1. Optimal sensor configuration based on value of information

82 In this section, a methodology based on the concept of *value of information* [3, 4] is proposed to assess the
83 optimal number of sensors for an ultrasound-based SHM system. In general terms, the value of information quantifies
84 the increment of benefit as a consequence of the information gain about a set of uncertain model parameters $\boldsymbol{\theta}$ (e.g.
85 damage location parameters) when data \mathbf{D} are taken into account.

86 In mathematical terms, let $\mathfrak{N} = \{1, \dots, n, \dots\}$ denote a set of potential sensor configurations and
87 $b(n, \boldsymbol{\theta}) : \mathbb{N} \times \mathbb{R}^{n\boldsymbol{\theta}} \rightarrow \mathbb{R}$ a benefit function for the n -th sensor configuration, given a set of updatable model param-
88 eters $\boldsymbol{\theta} \in \Theta \subset \mathbb{R}^{n\boldsymbol{\theta}}$. Note that the n -th sensor configuration implies a layout of n sensors. Let us also denote by
89 $p(\boldsymbol{\theta})$ and $p(\boldsymbol{\theta}|\mathbf{D})$ the prior and posterior PDFs of the parameters $\boldsymbol{\theta}$, respectively, with \mathbf{D} being the dataset. In this

90 paper, $b(n, \boldsymbol{\theta})$ is defined as the product of a normalized inverse cost function (e.g. from economic or manufacturing
 91 cost sources) $f(n) : \mathbb{N} \rightarrow [0, 1]$ and the information gain between the current and prior PDF of model parameters, as
 92 follows:

$$b(n, \boldsymbol{\theta}) = f(n) \left(\alpha + \log_2 \left[\frac{\pi(\boldsymbol{\theta}, \mathbf{D})}{p(\boldsymbol{\theta})} \right] \right) \quad (1)$$

93 where $p(\boldsymbol{\theta})$ is the prior PDF of model parameters $\boldsymbol{\theta}$, $\pi(\boldsymbol{\theta}, \mathbf{D})$ is to denote the current PDF of $\boldsymbol{\theta}$, which could be
 94 either the prior or the posterior PDFs depending on the availability of the dataset \mathbf{D} , and $\log_2[\pi(\boldsymbol{\theta}|\mathbf{D})/p(\boldsymbol{\theta})]$ is the
 95 information gain between the aforementioned PDFs in terms of bits. In Equation (1), the inverse cost function $f(n)$
 96 can be defined as a generally decreasing and dimensionless function, which gives a measure on how much benefit in
 97 terms of cost savings a particular configuration of n sensors provides. The constant $\alpha > 0$ introduced in Equation (1) is
 98 to represent the basic state of information assumed in the system such that it makes $b(n, \boldsymbol{\theta}) = \alpha f(n)$ when $\pi(\boldsymbol{\theta}, \mathbf{D}) =$
 99 $p(\boldsymbol{\theta})$, which happens when there is no learning about $\boldsymbol{\theta}$ from data \mathbf{D} and therefore the benefit directly equals to the
 100 cost savings.

101 Next, the concept of *maximum prior expected benefit* B' , obtained from the adoption of the optimal configuration
 102 $n'_{opt} \in \mathfrak{N}$, is defined as follows [8]:

$$B' = \mathbb{E}_{p(\boldsymbol{\theta})} [b(n'_{opt}, \boldsymbol{\theta})] \quad \leftarrow \quad n'_{opt} = \arg \max_{n \in \mathfrak{N}} \int b(n, \boldsymbol{\theta}) p(\boldsymbol{\theta}) d\boldsymbol{\theta} \quad (2)$$

103 Analogously, the *maximum posterior expected benefit* (PEB) $B''(\mathbf{D})$ can be obtained as [8]:

$$B''(\mathbf{D}) = \mathbb{E}_{p(\boldsymbol{\theta}|\mathbf{D})} [b(n''_{opt}, \boldsymbol{\theta})] \quad \leftarrow \quad n''_{opt} = \arg \max_{n \in \mathfrak{N}} \int b(n, \boldsymbol{\theta}) p(\boldsymbol{\theta}|\mathbf{D}) d\boldsymbol{\theta} \quad (3)$$

104 where the conditioning on \mathbf{D} is to denote that B'' depends on the data obtained through the PZT sensors. Finally, by
 105 subtracting both mathematical expectations, the *conditional value of information* (CVI) given \mathbf{D} is defined as:

$$\text{CVI}(\mathbf{D}) = B''(\mathbf{D}) - B' \quad (4)$$

106 Therefore, by substituting Equations (2), (3) and (1) in Equation (4), the CVI can be expressed as follows:

$$\begin{aligned} \text{CVI}(\mathbf{D}) &= \int b(n''_{opt}, \boldsymbol{\theta}) p(\boldsymbol{\theta}|\mathbf{D}) d\boldsymbol{\theta} - \int b(n'_{opt}, \boldsymbol{\theta}) p(\boldsymbol{\theta}) d\boldsymbol{\theta} \\ &= \int f(n''_{opt}) \left(\alpha + \log_2 \left[\frac{p(\boldsymbol{\theta}|\mathbf{D})}{p(\boldsymbol{\theta})} \right] \right) p(\boldsymbol{\theta}|\mathbf{D}) d\boldsymbol{\theta} - \int f(n'_{opt}) \left(\alpha + \log_2 \left[\underbrace{\frac{p(\boldsymbol{\theta})}{p(\boldsymbol{\theta})}}_{=1} \right] \right) p(\boldsymbol{\theta}) d\boldsymbol{\theta} \\ &= \alpha \int f(n''_{opt}) p(\boldsymbol{\theta}|\mathbf{D}) d\boldsymbol{\theta} + \int f(n''_{opt}) \log_2 \left[\frac{p(\boldsymbol{\theta}|\mathbf{D})}{p(\boldsymbol{\theta})} \right] p(\boldsymbol{\theta}|\mathbf{D}) d\boldsymbol{\theta} - \alpha \int f(n'_{opt}) p(\boldsymbol{\theta}) d\boldsymbol{\theta} \end{aligned} \quad (5)$$

107 Note from the last equation that the inverse cost function is independent of the uncertain parameters $\boldsymbol{\theta}$, therefore

108 Equation (5) can be reorganized as follows:

$$\text{CVI}(\mathbf{D}) = \underbrace{\alpha f(n''_{opt}) \int p(\boldsymbol{\theta}|\mathbf{D})d\boldsymbol{\theta}}_{=1} + \underbrace{f(n''_{opt}) \int \log_2 \left[\frac{p(\boldsymbol{\theta}|\mathbf{D})}{p(\boldsymbol{\theta})} \right] p(\boldsymbol{\theta}|\mathbf{D})d\boldsymbol{\theta}}_{\text{KL}(p(\boldsymbol{\theta}|\mathbf{D})\|p(\boldsymbol{\theta}))} - \underbrace{\alpha f(n'_{opt}) \int p(\boldsymbol{\theta})d\boldsymbol{\theta}}_{=1} \quad (6)$$

109 which finally leads to

$$\text{CVI}(\mathbf{D}) = \underbrace{f(n''_{opt})\text{KL}(p(\boldsymbol{\theta}|\mathbf{D})\|p(\boldsymbol{\theta}))}_{\text{BEIG}} - \underbrace{\alpha [f(n'_{opt}) - f(n''_{opt})]}_{\text{RCI}} \quad (7)$$

110 where $\text{KL}(p(\boldsymbol{\theta}|\mathbf{D})\|p(\boldsymbol{\theta}))$ denotes the Kullback-Leibler divergence between the posterior and prior PDF of the uncer-
 111 tain parameters $\boldsymbol{\theta}$. In the last equation, the term $f(n''_{opt})\text{KL}(p(\boldsymbol{\theta}|\mathbf{D})\|p(\boldsymbol{\theta}))$ can be understood as the benefit of the
 112 expected information gain (BEIG), which is always a non-negative value, i.e. $\text{BEIG} \in \mathbb{R}_0^+$. The second part of Equa-
 113 tion (7) can be understood as the relative cost of implementation (RCI), since it accounts for the difference between
 114 the cost savings of implementing the optimal sensor configuration before and after considering the information from
 115 data \mathbf{D} . Note that, in practice, not adding any SHM system ($n = 0$) is cheaper than adding it by strictly considering
 116 economical factors, due to sensors and hardware costs saving, among others. Therefore, the RCI is positive under
 117 the assumption that $f(n)$ is decreasing, i.e. $n'_{opt} < n''_{opt}$ hence $f(n'_{opt}) > f(n''_{opt})$ as exemplified in Figure 1. Observe
 118 also that the non-negative constant α defined in Equation (1) avoids the prior expected benefit (Equation (2)) becom-
 119 ing zero, and therefore simpler configurations will be preferred under non-informative (i.e. prior equals posterior)
 120 scenarios. In addition, the value of α is defined so that both terms, i.e., BEIG and RCI, are comparable under all
 121 possible scenarios, even when the system is not very informative, so that neither one of the RCI nor the BEIG terms
 122 individually drives the optimization problem. This interpretation of Equation (7) suggests that the CVI conveys a
 123 trade-off between the global benefit of establishing a particular SHM configuration and the cost of actually imple-
 124 menting it. The CVI also provides a decision making index that ranks different strategies considering their suitability
 125 in performance and cost. The interpretation of Equation (7) is illustrated with two hypothetical examples below.

126 **Example 1.** Let us assume a non-informative SHM system where the posterior PDF is virtually equal to the prior
 127 PDF, such that the optimal number of sensors in prior and posterior states are the same, i.e. $n'_{opt} = n''_{opt}$ and thus
 128 $f(n'_{opt}) = f(n''_{opt})$, hence $\text{RCI} = 0$. Moreover, since the SHM system is non-informative, the expected information
 129 gain equals zero, and therefore the BEIG equals zero as well. In this extreme example, given that $\text{CVI} = 0$, no
 130 decision about adding or removing sensors could be made based on the current information, hence no change with
 131 respect to the prior configuration would be required. Figure 2a depicts this example.

132 **Example 2.** Let us assume a non-informative SHM system such that the optimal number of sensors when data are
 133 acquired are $n''_{opt} = n'_{opt}/2$. In this example, half of the sensor network in the prior or initial state, i.e., $n'_{opt}/2$, are
 134 placed opposite to each other in the plate, as depicted in Figure 2b. In this situation, let us assume that the posterior
 135 information captured by the system is equivalent to the prior information, hence the expected information gain (KL
 136 divergence) equals zero, thus BEIG also equals zero. Henceforth, the RCI term turns negative since $f(n'_{opt}) < f(n''_{opt})$,

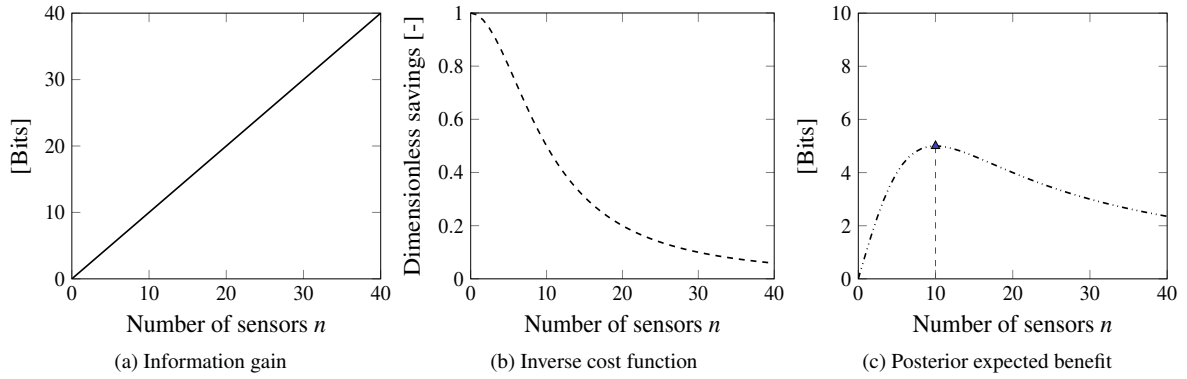


Figure 1: Example of functions of (a) information gain, (b) dimensionless inverse of cost, and (c) final combination of both functions as an example of Equation (4). It can be observed in (a) that problems that constantly gain information when the number of sensors increases, can be converted into a normal optimization problem (like in (c)) by the introduction of a dimensionless cost function (b).

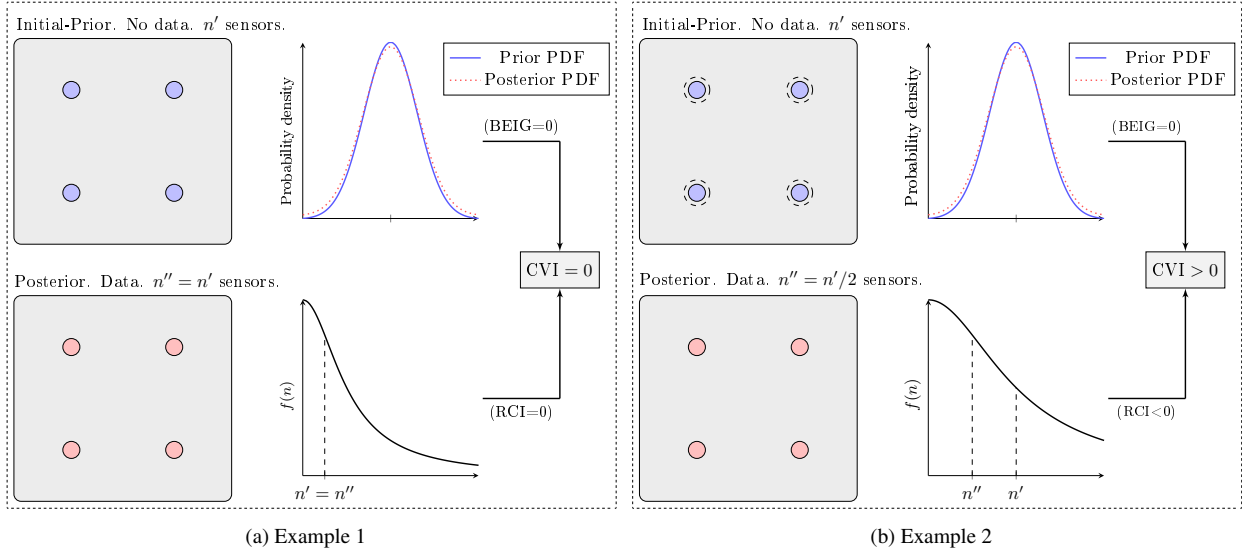


Figure 2: Graphical representation of Examples 1 and 2 about the influence of using the proposed CVI-based formulation in the decision of number of sensors. For Example 1, no action would be taken, thus the prior distribution of sensors would be maintained. For Example 2, even when the $BEIG=0$, the proposed $CVI > 0$ formulation supports the option with a smaller number of sensors. Dashed circles represent the sensors in the opposite side of the plate.

137 which leads to $CVI > 0$. This hypothetical example shows that the proposed formulation propitiates simpler and
 138 cheaper SHM systems under non-informative scenarios.

139 2.1.1. Expected value of information

140 The CVI criterion presented above formulates the optimization problem for a particular set of data \mathbf{D} , i.e., it im-
 141 plicitly implies that damage is located at a particular position. However, in practice, the optimal sensor configuration
 142 problem requires addressing a large enough set of damage scenarios, hence a space of datasets $\mathcal{D} \ni \mathbf{D}$ instead of a
 143 particular \mathbf{D} needs to be considered. Note that at the design stage, experimental data are not typically available and

144 therefore the data are simulated using a model and prior knowledge of the model parameters. Thus, a mathematical
 145 expectation is applied to Equation (7) to obtain the expected value of information (EVI) over the space of datasets \mathcal{D} ,
 146 as follows:

$$\text{EVI} = \int_{\mathcal{D}} [f(n''_{opt}) \text{KL}(p(\boldsymbol{\theta}|\mathbf{D})||p(\boldsymbol{\theta}))] p(\mathbf{D}) d\mathbf{D} - \int_{\mathcal{D}} \alpha[f(n'_{opt}) - f(n''_{opt})] p(\mathbf{D}) d\mathbf{D} \quad (8)$$

147 Observe that the first term of Equation (8) involves a double multidimensional integral that cannot be solved ana-
 148 lytically in most of the cases. Therefore, it is numerically approximated using the Monte Carlo (MC) method as
 149 follows [38, 41, 48]:

$$\text{EVI} \approx f(n''_{opt}) \frac{1}{N_{out}} \sum_{m=1}^{N_{out}} \left[\log_2 p(\mathbf{D}_{(m)}|\boldsymbol{\theta}_{(m)}) - \log_2 \left(\frac{1}{N_{in}} \sum_{k=1}^{N_{in}} p(\mathbf{D}_{(m)}|\boldsymbol{\theta}_{(k)}) \right) \right] - \alpha[f(n'_{opt}) - f(n''_{opt})] \quad (9)$$

150 where $\boldsymbol{\theta}_{(m)}$ is a sample drawn from the prior distribution $p(\boldsymbol{\theta})$ and $\mathbf{D}_{(m)}$ is a sample dataset drawn from the likelihood
 151 distribution $p(\mathbf{D}|\boldsymbol{\theta} = \boldsymbol{\theta}_{(m)})$ (refer to Equation (12) below). Thus, Equation (9) is adopted here as optimality criterion
 152 to obtain the optimal sensor configuration considering an area of possible damage locations. Further implementation
 153 details of the EVI criterion and the adopted search algorithm are provided hereinafter in Section 2.3.

154 2.2. Optimal sensor placement: Bayesian damage localization

155 The previous section presented a rational fitness function to obtain the optimal number of sensors using the
 156 EVI, which requires updated information about the model parameters (e.g. damage coordinates) given a particular
 157 dataset $\mathbf{D}_{(m)} = \mathbf{D}$. This section describes the fundamentals of the Bayesian *inverse problem* whereby model parameters
 158 are estimated.

159 The problem of damage localization is addressed by a model-based BIP using an ellipse-based time-of-flight (ToF)
 160 model [47], which was previously published by the authors in [46]. The ToF is an ultrasonic signal feature widely
 161 adopted by both practitioners and researchers due to its efficiency in providing information about material properties
 162 and damage localization by post-processing scattered signals. To estimate the ToF from scattered signals, several time-
 163 frequency (TF) representation techniques are available in the literature [49–52]. In this work, a recently published
 164 hyper-robust approach is adopted to obtain a robust ToF estimate resulting from using not just a particular TF model
 165 class based on modeler's choice, but an overall set of candidate models according to their relative plausibility [46].

166 In particular, given a TF model, the ToF can be obtained as the difference between the time to receive the first
 167 energy peak of the excitation signal and the one from the scattered signal. From a theoretical point of view, the ToF
 168 information of the scattered signals can be obtained as follows [53]:

$$\text{ToF}^{(a-s)} = \frac{\sqrt{(X_d - X_a)^2 + (Y_d - Y_a)^2}}{V_{a-d}} + \frac{\sqrt{(X_d - X_s)^2 + (Y_d - Y_s)^2}}{V_{d-s}} \quad (10)$$

169 where $(X_d, Y_d) \in \mathbb{R}^2$ are the coordinates of the damage position, $(X_a, Y_a) \in \mathbb{R}^2$ are the actuator transducer coordi-

170 nates, $(X_s, Y_s) \in \mathbb{R}^2$ are the coordinates of one particular sensor transducer, and V_{a-d} and V_{d-s} are the wave propa-
 171 gation velocities of the actuator-damage and damage-sensor paths, respectively¹. From this standpoint, the unknown
 172 model parameters of interest are $\boldsymbol{\theta} = \{X_d, Y_d, V\}$ so, under the assumption of perfect sensor, if an exact value of
 173 $\boldsymbol{\theta}^*$ is known, then the measured ToF and the modeled one using Equation (10) would be identical; mathematically:
 174 $\text{ToF}_{\mathcal{D}}^{(a-s)} \equiv \text{ToF}_M^{(a-s)}(\boldsymbol{\theta}^*)$. However, in practice there are uncertainties due to signal measurement errors, partially
 175 unknown material properties, and the uncertainty about the validity of the ToF model itself, which make the last iden-
 176 tity seldom observed in real-world scenarios. Thus, a more appropriate and rigorous approach involves assuming the
 177 existence of these modeling uncertainties and quantifying them, as follows:

$$\text{ToF}_{\mathcal{D}}^{(a-s)} = \text{ToF}_M^{(a-s)}(\boldsymbol{\theta}) + e \quad (11)$$

178 where $e \in \mathbb{R}$ is an error term enclosing the uncertainty which accounts for the discrepancy between $\text{ToF}_M^{(a-s)}$ and
 179 $\text{ToF}_{\mathcal{D}}^{(a-s)}$. By the Principle of Maximum Information Entropy [54, 55], this error term can be conservatively assumed
 180 to be modeled as a zero-mean Gaussian distribution with standard deviation $\sigma_e \in \mathbb{R}$. Thus, Equation (11) can be
 181 rewritten probabilistically as:

$$p\left(\text{ToF}_{\mathcal{D}}^{(a-s)} | \boldsymbol{\theta}\right) = (2\pi\sigma_e^2)^{-\frac{1}{2}} \exp\left(-\frac{1}{2} \left(\frac{\text{ToF}_{\mathcal{D}}^{(a-s)} - \text{ToF}_M^{(a-s)}(\boldsymbol{\theta})}{\sigma_e}\right)^2\right) \quad (12)$$

182 which provides a probabilistic measure about the similarity between $\text{ToF}_{\mathcal{D}}^{(a-s)}$ and $\text{ToF}_M^{(a-s)}(\boldsymbol{\theta})$ for a particular value
 183 of $\boldsymbol{\theta}$. The last equation is commonly known as the *likelihood function* for model parameter $\boldsymbol{\theta}$.

184 Nevertheless, our interest precisely lies in the reciprocal information, i.e., to determine the values of $\boldsymbol{\theta}$ among the
 185 set of values $\Theta \subset \mathbb{R}^{n\boldsymbol{\theta}}$ which lead to models that more likely satisfy the ideal identity $\text{ToF}_{\mathcal{D}}^{(a-s)} \equiv \text{ToF}_M^{(a-s)}(\boldsymbol{\theta})$. This
 186 *inverse problem* can be formulated by Bayes' Theorem [56, 57], as:

$$p(\boldsymbol{\theta} | \mathbf{D}) = \frac{p(\mathbf{D} | \boldsymbol{\theta})p(\boldsymbol{\theta})}{p(\mathbf{D})} \quad (13)$$

187 where $\mathbf{D} = \{D^{(1)}, \dots, D^{(k)}, \dots, D^{(n)}\}$ is the set of ToF data corresponding to the set of n sensors and a spatial confi-
 188 guration² \mathcal{C}^n , $p(\boldsymbol{\theta})$ is the prior PDF of the model parameters, and $p(\mathbf{D} | \boldsymbol{\theta})$ is the likelihood function for the set of data
 189 \mathbf{D} . Assuming stochastic independence of the measurements, the likelihood $p(\mathbf{D} | \boldsymbol{\theta})$ can be expressed probabilistically
 190 as $p(\mathbf{D} | \boldsymbol{\theta}) = \prod_{k=1}^n p(D^{(k)} | \boldsymbol{\theta})$, where each factor $p(D^{(k)} | \boldsymbol{\theta})$ is given by Equation (12). Finally, the term $p(\mathbf{D})$ refers to
 191 the *evidence* [54] of the data under the model specified by $\boldsymbol{\theta}$. This term, which acts as a normalizing constant within
 192 the Bayes' theorem, can be bypassed through sampling using Markov Chain Monte Carlo (MCMC) methods [58].

¹Note that these velocities are the same under the assumption of isotropic materials and damage concentrated within a bounded region, where $V = V_{a-d} = V_{d-s}$.

²The conditioning on \mathcal{C}^n has been omitted from $p(\boldsymbol{\theta} | \mathbf{D})$ given that it is intrinsically contained in \mathbf{D} , since a particular configuration univocally implies a dataset, i.e., $\mathcal{C}^n \rightarrow \mathbf{D}$.

Algorithm 1 Pseudo-code implementation of forward sequential search algorithm for geometrically unconstrained sensor configurations.

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1: Preamble Define  $n_s \triangleright$  {number of possible sensor locations}; Define  $n_{s,max} \triangleright$  {maximum number of sensors
   locations considered in the optimization};
2: Algorithm
3:  $n'_{opt} \leftarrow \arg \max_n f(n) \triangleright$  {Optimal number of sensors under prior information}
4:  $\{\{\boldsymbol{\theta}_{(m,h)}\}_{m=1}^N\}_{h=1}^{N_{set}} \sim p(\boldsymbol{\theta}) \triangleright$  { $N_{set}$  sets of  $N = N_{out} = N_{in}$  samples drawn from the prior PDF}
5:  $\{\{\mathbf{D}_{(m,h)}\}_{m=1}^N\}_{h=1}^{N_{set}} \sim p(\mathbf{D}|\boldsymbol{\theta} = \boldsymbol{\theta}_{(\cdot,h)}) \triangleright$  { $N_{set}$  sets of  $N = N_{out} = N_{in}$  samples drawn from the likelihood PDF}
6:  $\mathcal{C} \leftarrow \emptyset \triangleright$  {Initialize an empty vector of optimal sensors placement}
7: for  $n := 1$  to  $n_{s,max}$  do  $\triangleright$  {Forward sequential sensor loop}
8:   for  $i := 1$  to  $n_s$  do  $\triangleright$  {Exhaustive search}
9:      $\mathcal{C}^n \leftarrow \mathcal{C}^{n-1} \cup \mathcal{C}^i \triangleright$  {Define  $\mathcal{C}^n$  by concatenating the previous configuration to the  $i$ -th sensor}
10:    Obtain  $J(\mathcal{C}^n) \triangleright$  {Evaluate the objective function according to Equation (14)}
11:   end for
12:   Return:  $\mathcal{C}^n_{opt}$  and  $J(\mathcal{C}^n_{opt}) \triangleright$  {Optimal sensor layout and its corresponding objective function value}
13:    $\mathcal{C} \leftarrow \mathcal{C}^n_{opt} \triangleright$  {Store the  $n$ -th optimal sensor location for the next iteration}
14: end for
15:  $n''_{opt} \leftarrow \arg \max_n J(\mathcal{C}) \triangleright$  {Optimal number of sensors under posterior information}

```

193 2.3. Algorithmic implementation

194 In this section, a forward sequential algorithm is used to solve the optimal sensor configuration problem over
195 two main variables: (1) *number* and (2) *position* of the sensors, as proposed by [35, 36]. Therefore, the algorithm
196 provides the optimal number of sensors n''_{opt} and their optimal spatial configuration \mathcal{C}^n_{opt} , whereby the EVI is finally
197 obtained. In this algorithm, the optimal sensor placement problem is proposed to be solved through an exhaustive
198 search methodology, which is performed by exploring all the possible sensor locations within a discrete grid, given
199 that a relatively small search space is considered. The objective function $J(\mathcal{C}^n) : \mathbb{N} \rightarrow \mathbb{R}$ for the optimization problem
200 is given by the average of the EVI (Equation (9)) for each spatial configuration \mathcal{C}^n , as follows:

$$\begin{aligned}
J(\mathcal{C}^n) = f(n) \frac{1}{N_{set}} \sum_{h=1}^{N_{set}} \left\{ \frac{1}{N_{out}} \sum_{m=1}^{N_{out}} \left[\log_2 p(\mathbf{D}_{(m,h)} | \boldsymbol{\theta}_{(m,h)}, \mathcal{C}^n) - \log_2 \left(\frac{1}{N_{in}} \sum_{k=1}^{N_{in}} p(\mathbf{D}_{(m,h)} | \boldsymbol{\theta}_{(k,h)}, \mathcal{C}^n) \right) \right] \right\} - \\
- \alpha [f(n'_{opt}) - f(n)]
\end{aligned} \tag{14}$$

201 where \mathcal{C}^n is the sensor configuration variable that controls which sensors are active, their location, and therefore what
202 ToF data \mathbf{D} are used to calculate $J(\mathcal{C}^n)$. To efficiently address the expectation of the KL divergence (recall Eq. (9)), we
203 reuse the $N_{out} = N_{in}$ prior samples in both the inner and outer sums of Equation (14) at the cost of a small increment
204 in the bias of the estimator [38, 41]. Nevertheless, the computation of the KL divergence implies a numerical error
205 inversely proportional to the number of samples, leading to stochastic uncertainty in the value of the KL divergence.
206 To reduce such stochasticity, the objective function is averaged over N_{set} sets of samples for each configuration \mathcal{C}^n ,
207 as shown in Equation (14).

208 Furthermore, note that n'_{opt} in Equation (14) is obtained using Equation (2) and that the independence of the prior

209 expected benefit on the sensor configuration makes it only dependent on the inverse cost function. Thus, n'_{opt} implies
 210 the most economically beneficial amount of sensors given that $f(n)$ is decreasing, regardless of their position.

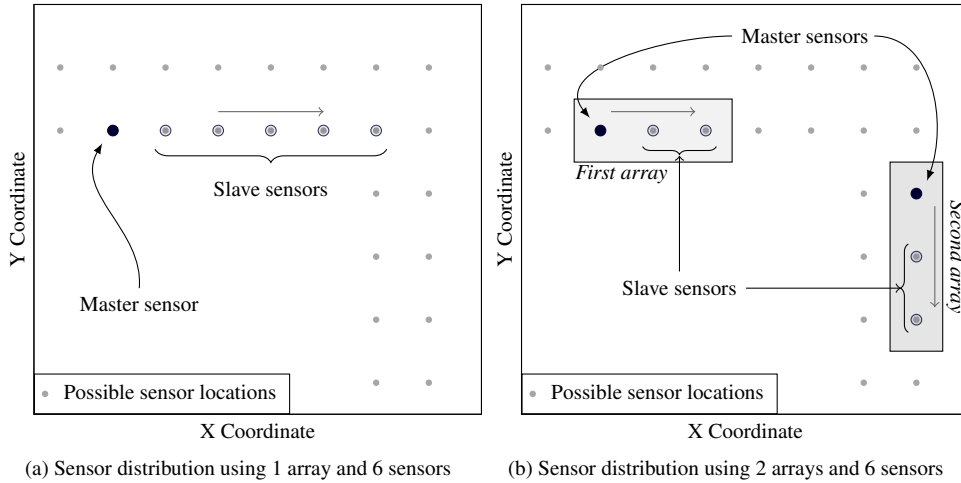


Figure 3: Illustration of sensor positioning strategies for 1 and 2 arrays. In panel (a), 1 array configurations where one master sensor is optimally placed, while the slave sensors are automatically placed next the master. Panel (b) depicts the case of 2 arrays of sensors, where the first array is optimally placed and its position stored to search then the optimal position of the second one.

211 Furthermore, observe that the search of the optimal number of sensors implies a heavy computational problem in-
 212 volving $n_s! / n!(n_s - n)!$ potential configurations to be explored. For example, for $n_s = 40$ possible sensor locations and
 213 $n = 5$ sensors, the number of possible configurations would be $40! / 5!35! = 658,008$. To avoid such computational
 214 complexity, several strategies can be applied to reduce the search space. In this work, the *forward sequential search*
 215 approach [35] is adopted, as shown in Algorithm 1, consisting of running an exhaustive search for one individual
 216 sensor (e.g., n) having stored the optimal locations of the previous $n - 1$ sensors, thus limiting the search space. Fol-
 217 lowing this methodology, the total number of configurations reduces to $n(2n_s - n + 1)/2$. Taking the same numerical
 218 example with $n_s = 40$ and $n = 5$, the number of possible configurations would be $5(80 - 5 + 1)/2 = 190$ instead of
 219 658,008 using the former method. Note that a suboptimal configuration is expected to be obtained, since the sequen-
 220 tial search method does not necessarily provide the global optimal solution. However, the suboptimal configuration
 221 can be assumed to be effective and robust [59, 60].

222 From this standpoint, it is important to remark that Algorithm 1 provides a methodology to obtain geometrically
 223 unconstrained (“open”) optimal sensor configurations. However, these may find a limitation in practice when instal-
 224 lation issues such as the routing of the wires are taken into account. Therefore, more practical sensor configurations
 225 constrained to linear arrays of sensors [61, 62] are typically preferred for their ease of installation and maintenance.
 226 These sensor configurations using one or two arrays of sensors can be assessed by slightly modifying Algorithm 1 as
 227 follows: For one array of n sensors, only the n -th sensor (master) position is optimized, considering that the remaining
 228 $n - 1$ sensors (slave) are placed either on the left or right side of the master one, as depicted in Figure 3a. Note for
 229 this configuration the search space reduces as long as n increases. In case of two arrays of n sensors, the optimization

230 is carried out in two steps, (1) the first array of $n/2$ sensors is optimally placed, as explained before, and its position
 231 is stored, and (2) the optimal position of the second array of $n/2$ sensors is addressed considering the first one fixed,
 232 as shown in Figure 3b. The search space reduces even faster for two array configurations since only even numbers of
 233 sensors n are considered to form the two arrays.

234 3. Case Studies

235 The proposed methodology to obtain the optimal sensor configuration based on EVI is exemplified herein using
 236 two case studies. Section 3.1 illustrates the methodology for a square aluminum plate. Finally, Section 3.2 provides a
 237 comparison between open and array configurations for an aeronautical hat-type stiffened plate.

238 3.1. Sensor optimization in a square metallic plate

239 This case study deals with the problem of optimal sensor configuration for a $30\text{cm} \times 30\text{cm} \times 0.2\text{cm}$ aluminum
 240 2024-T351 plate. The prior information of the model parameters θ is represented as uniform distributions for the
 241 damage position parameters (X_d, Y_d) and as a Gaussian distribution for the velocity V , as follows: $p(X_d) = \mathcal{U}(-6, 6)$,
 242 $p(Y_d) = \mathcal{U}(-6, 6)$, and $p(V) = \mathcal{N}(5400, 60)$, with position and wave velocity units expressed in [cm] and [m/s]
 243 respectively. Note that the origin of coordinates is located at the center of the plate and the wave propagation velocity
 244 refers to the symmetric-0 guided-wave mode at a frequency 300 kHz. To address the optimal sensor configuration, a
 245 grid of possible sensor locations is defined as two concentric squares of 12 cm and 18 cm of side respectively, resulting
 246 in a total of 40 possible locations. The area of possible damage occurrence is represented in blue color in Figure 4b.
 247 The actuator is considered to be fixed at the origin of coordinates, i.e., the center of the plate. The optimization is
 248 addressed by using Algorithm 1 with $N_{set} = 1000$ and $N_{out} = N_{in} = 1000$ (recall Eq. (14)).

249 The results for this case study are shown in Figure 4. As evident from Figure 4a, $J(\mathcal{C}_{opt}^n)$ reaches a global
 250 maximum using a configuration of 3 sensors, provided an inverse cost function $f_0(n)$ given by:

$$f_0(n) = 100/(n^2 + 100) \quad (15)$$

251 The corresponding EVI value is 5.4875 [bits], adopting $\alpha = 1$ in Equation (9), i.e., assuming that both the RCI and the
 252 information members in Equation (14) are equally important. The optimal sensor distribution for this configuration is
 253 shown in Figure 4b. Note that the optimal sensor positions (in red) result to be symmetric with respect to the center
 254 of the plate (where the actuator is located), and they are wholly contained within the inner square, i.e., closer to the
 255 area of possible damage occurrence.

256 3.1.1. Verification of the optimal sensor configuration

257 The performance of the optimal sensor configuration, as depicted in Figure 4b, is tested against simulated data
 258 for a specific damage within the area of possible damage occurrence. To this end, the plate is simulated in Abaqus

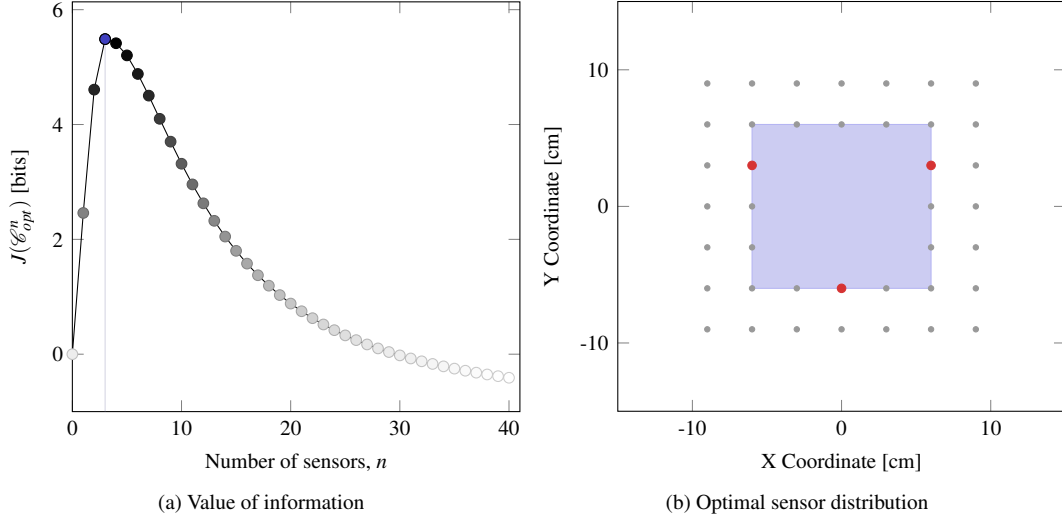


Figure 4: Estimator of the EVI (Eq. (14)) in panel (a) and the corresponding optimal sensor configuration with $n''_{opt} = 3$ sensors in (b), where the possible sensor locations (gray dots) and the optimal locations (red dots) are shown. The blue area denotes the area of possible damage occurrence.

259 (Figure 5a) using C3D8R (8-node linear brick, reduced integration, hourglass control) solid elements [63] with mesh
 260 size 0.05 cm and mechanical properties $\rho = 2780 \text{ kg/m}^3$, $E = 73.1 \text{ GPa}$, and $\nu = 0.33$ (refer to the Nomenclature). The
 261 mesh size is determined so that at least 10 nodes are contained per wavelength, and the chosen time step is obtained
 262 so that a disturbance cannot propagate through a grid spacing during one time step [64, 65]. The simulated damage is
 263 represented as a square $0.2 \text{ cm} \times 0.2 \text{ cm}$ hole, centered at $(3.5, 2.5) \text{ cm}$, considering the center of the plate as the origin
 264 of coordinates, as shown in Figure 5b. To simulate the wave propagation, an exciting force is applied at the center of
 265 the plate in the perpendicular direction, consisting of a 5 cycle sine tone burst centered at a frequency 300 kHz. The
 266 guided-waves are acquired at the three optimal sensor positions highlighted in red in Figure 4b and subtracted before
 267 and after damage to obtain the scattered signals. Then, the robust ToF estimation methodology previously developed
 268 in [46] is applied. The robust ToF data $\mathbf{D} = \{D^{(1)}, \dots, D^{(n)}\}$ obtained for each of the optimal sensors are then used
 269 as input data for the Bayesian damage localization problem described in Section 2.2, whereby the posterior PDFs of
 270 the model parameters $\boldsymbol{\theta} = \{X_d, Y_d, V\}$ are obtained. The prior knowledge of the model parameters is assumed to be
 271 equal to the one used for the optimization process, described in the Section 3.1. To numerically solve the Bayesian
 272 inverse problem of damage localization, the asymptotic independence Markov sampling (AIMS) algorithm [66, 67] is
 273 applied with a threshold value $\gamma = 1/2$, using 50,000 samples per annealing level, using a Gaussian PDF as proposal
 274 distribution, i.e. $q(\theta'|\theta) = \mathcal{N}(\theta, \sigma)$, where σ is the standard deviation of the Metropolis-Hastings random walk. In
 275 this problem, σ is selected such that the acceptance rate lies within the suggested interval $[0.2, 0.4]$ [68–70].
 276 The posterior information of the damage parameters $\boldsymbol{\theta} = \{X_d, Y_d, V\}$ is illustrated in Figures 5c and 5d. It can be
 277 observed that the damage position is remarkably well reconstructed using the optimal sensor configuration. In contrast,
 278 the uncertainty of the reconstructed wave propagation velocity is higher as shown in Figure 5d. A larger variability of

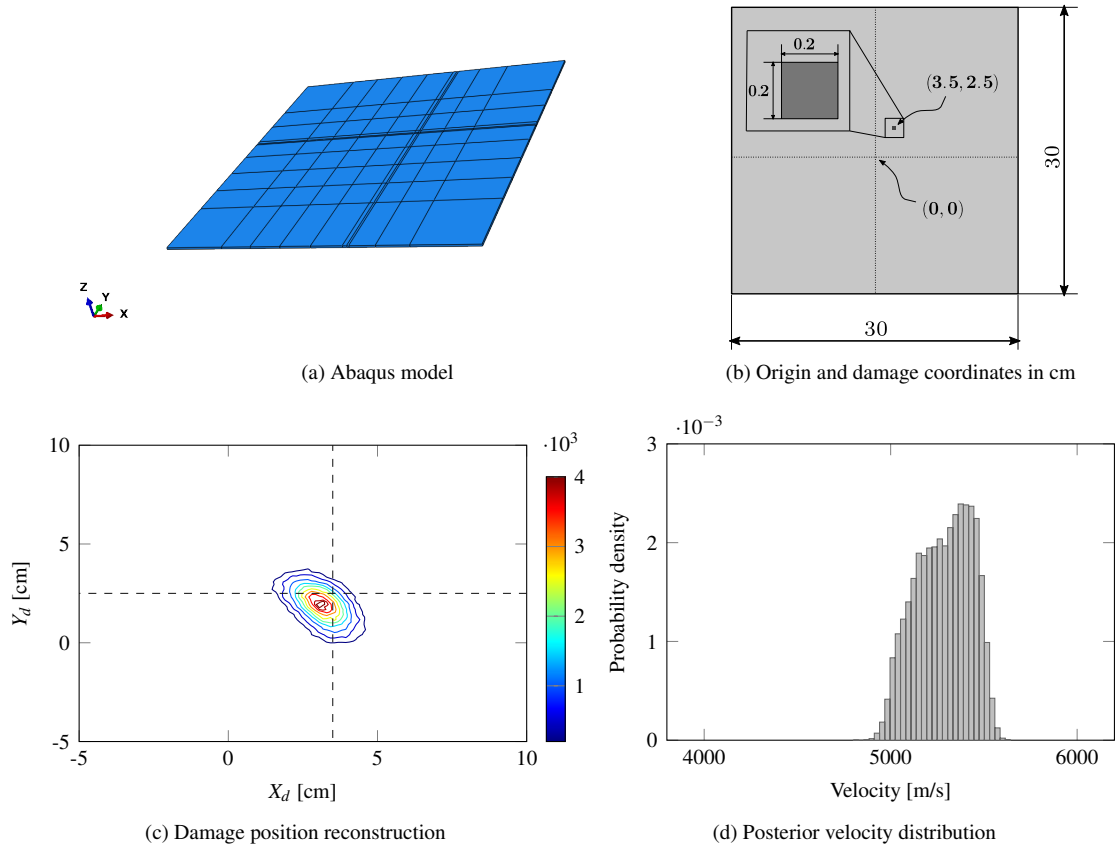


Figure 5: Abaqus model in panel (a), dimensions of the plate and location of the simulated damage in panel (b), joint posterior distribution of the damage coordinates with their associated probability densities in the color bar along with the center of the real damage represented by the intersection of both dashed lines in panel (c), and the velocity reconstruction in panel (d).

279 the ToF data at the optimal sensors is identified as the reason for such relatively high uncertainty in the velocity, which
 280 may come from potential mode mixture issues between both anti-symmetric and symmetric 0 guided waves modes,
 281 namely A0 and S0 modes, respectively.

282 3.2. Sensor optimization in aeronautical panel

283 The aim of this case study is to investigate the applicability of the proposed methodology using a more com-
 284 plex and realistic structure. In particular, a $50\text{cm} \times 50\text{cm} \times 0.2\text{cm}$ plate with top-hat section stiffeners, commonly
 285 used in aeronautical structures, is adopted for this case study. Here, the origin of coordinates is located at the center
 286 of the left plate's edge and the actuator is placed at the center of the plate, i.e., $(25, 0)$ [cm]. The prior informa-
 287 tion of the model parameters is given by $p(X_d) = \mathcal{U}(5, 45)$, $p(Y_d) = \mathcal{U}(-10, 10)$, and $p(V) = \mathcal{N}(2800, 60)$,
 288 with position and wave velocity units expressed in [cm] and [m/s] respectively. Note that the velocity prior
 289 PDF is centered at 2800 m/s, which corresponds to the group velocity of the ultrasonic guided-wave mode A0
 290 at a frequency 300 kHz. The search space is represented by eight rows of 39 possible sensor locations at

291 $Y = \{20.81, 19.81, 10.19, 9.19, -9.19, -10.19, -19.81, -20.81\}$ [cm], separated by 1 [cm] in the X direction, as
 292 depicted in Figure 6c using gray dots. The optimization is addressed considering Equation (14) with $N_{set} = 100$ sets
 293 of $N_{out} = N_{in} = 1000$ samples of the prior distribution.

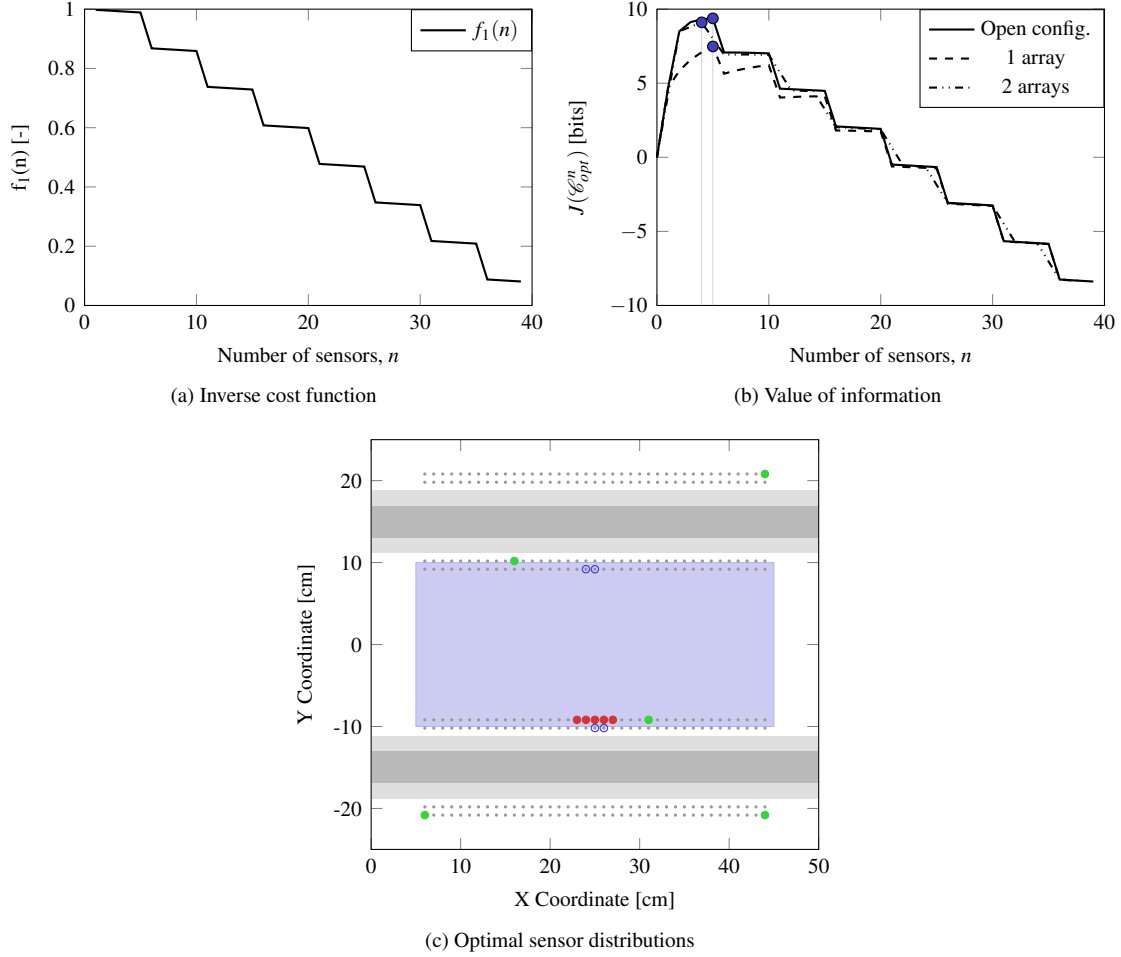


Figure 6: Inverse cost function $f_1(n)$ in panel (a). In (b), the estimator of the EVI (Eq. (14)) for the different positioning configurations is depicted. Panel (c) shows their corresponding optimal sensor layouts in green, red, blue, and gray dots for open configuration, one-array configuration, two-array configuration, and search space, respectively. The blue rectangle represents the area of possible damage occurrence with uniform probability.

294 For this case study, a comparison between different sensor placement strategies, i.e., (1) non geometrically con-
 295 strained “open” configuration, (2) sensor positioning in 1 array, and (3) sensor positioning over 2 separated arrays, is
 296 carried out. Figure 6a depicts the adopted inverse cost function, whilst Figure 6b depicts the objective function output
 297 $J(\mathcal{C}_{opt}^n)$ (recall Eq. (14)) as a function of the number of sensors and their respective optimal number of sensors for
 298 each type of configuration. The optimal number of sensors results to be $n_{opt}'' = 5$ for both the open and the one-array
 299 configurations (EVI = 9.2874 [bits] and EVI = 7.3896 [bits], respectively), whereas the two-array configuration leads
 300 to 4 sensors (EVI = 9.0416 [bits]), assuming $\alpha = 10$ to emphasize the cost over the relative gain of information. In
 301 view of Figure 6b, both the open configuration and the one using 2 arrays result to be the most valuable, and hence

302 the preferred ones under the EVI criterion. This ranking is also supported by the robust BEIG and RCI terms (Eq. (7)
 303 and (9)) in Table 1. In particular, observe that even when the RCI for the open configuration is higher than the RCI of
 304 the two-array configuration, the first one is preferred since it provides a relatively higher and robust BEIG.

Table 1: Decomposition of EVI into the robust BEIG and RCI terms (Eq. (9)) for the three positioning strategies.

Configurations	BEIG [bits]	RCI [bits]	EVI [bits]	No sensors
Open config.	9.3774	0.0900	9.2874	5
One array	7.4796	0.0900	7.3896	5
Two arrays	9.1091	0.0675	9.0416	4

305 As a general comment, the open configuration is able to reconstruct the damage more accurately due to its flex-
 306 ibility. However, in real world engineering applications, factors such as the routing options of the cables introduce
 307 limitations and complexity that may prevent us from the use of sensor open configurations. Accordingly, the two-array
 308 configuration would be the preferred one based on the value of information criterion, as shown in Table 1. Finally,
 309 Figure 6c shows the optimal sensor layouts for each of the three positioning strategies.

310 3.2.1. Verification of the optimal sensor configuration

311 The resulting optimal sensor configurations in Section 3.2 are tested using a simulated damage scenario consider-
 312 ing a hole in the metallic plate. To this end, an Abaqus model, shown in Figure 7a, is developed to obtain the simulated
 313 guided waves whereby the robust ToF is estimated and the Bayesian damage localization is addressed. Here, the stiff-
 314 eners are assumed to be perfectly bonded to the base plate whereas the damage is simulated as a $0.2\text{cm} \times 0.4\text{cm}$
 315 hole located at (35.1, 3) cm, considering the origin of coordinates at the center of one plate's edges, as shown in
 316 Figure 7b. S4R (4-node doubly curved thin or thick shell, reduced integration, hourglass control, finite membrane
 317 strains) shell elements [63] adopted with a mesh size of 0.05 cm; therefore, only anti-symmetric modes are captured,
 318 further simplifying the robust ToF identification since no mode mixture is expected to arise at a frequency 300 kHz.
 319 The guided waves are generated with a 5 cycle sine tone burst centered at the aforementioned frequency applied in
 320 the perpendicular direction at the center of the plate. The plate and stiffeners dimensions are depicted in Figure 7b.
 321 The material is aluminum 2024-T351, thus the mechanical properties (Young's modulus E and Poisson ratio ν) are
 322 the specified for the previous case study in Section 3.1.1. The prior knowledge of the model parameters remains the
 323 same than the specified in Section 3.2.

324 The reconstruction of the damage position corresponding to the optimal configurations is depicted in Figures 7c
 325 to 7e. As evident from the damage localization results, both the open configuration as well as the configuration
 326 with two arrays reconstruct the damage position remarkably well, while the configuration using one array provides
 327 higher uncertainty about the damage position. Note that the single array configuration would need comparatively more
 328 sensors than either the open or double array configurations to achieve the same level of accuracy. Such behavior is also

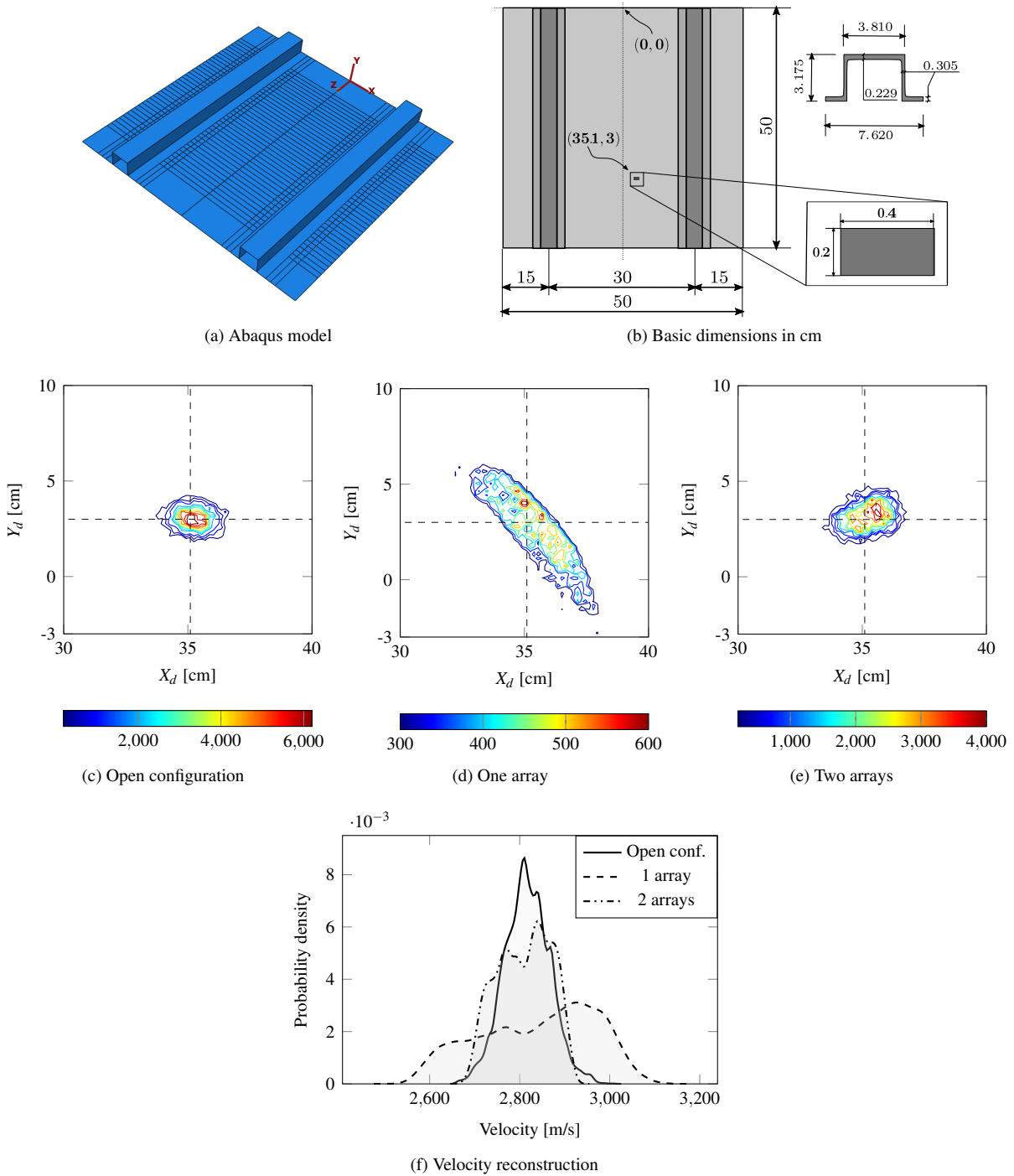


Figure 7: Abaqus model in (a) and dimensions in (b) of the aerospace panel. Panels (c) to (e) show the resulting damage position reconstruction corresponding to the optimal configurations using the open, one-array and two-array configurations, respectively. The color bars indicate the probability density of the joint distribution. Panel (f) represents the wave propagation velocity reconstruction. The irregularities of contour lines in plots (c) to (e) could be improved using more AIMS samples at the cost of heavier computation. However they are shown to be enough for illustration and validation purposes.

329 supported by the wave velocity reconstructions observed in Figure 7f, where the posterior PDFs provided by the open
330 and two-array configurations are more accurate (i.e. have less uncertainty) than the one with one-array configuration.

331 4. Discussion

332 4.1. On the case studies results

333 The proposed methodology for optimal sensor configuration in ultrasonic guided-wave based SHM has been
334 illustrated in Section 3 using two case studies. In particular, the optimal number and locations of sensors have been
335 shown to be sensitive to the choice of a particular inverse cost function. Thus, a proper inverse cost definition is key for
336 an accurate and efficient SHM design. This was an expected result as a consequence of using the *value* instead of the
337 *amount* of information as the optimality criterion. Note that the definition of the inverse cost function depends on the
338 manufacturing and maintenance costs derived from the SHM system, and thus it is case specific. For instance, if the
339 cost increases linearly with the number of sensors, a linearly decreasing inverse cost function could be assumed to be
340 appropriate. To further explore the dependence of the optimization results on the inverse cost function, the implications
341 of using different inverse cost functions in the optimal number of sensors are discussed here. In particular, Figure 8
342 shows two different but similar numerical inverse cost functions, namely $f_1(n)$ and $f_2(n)$, and their influence in the
343 value of information curve with respect to the number of sensors for case study 1. As is evident from the results in
344 Figures 8b and 8d, the optimal number of sensors varies from 5 to 9 sensors just by changing the inverse cost function
345 from $f_1(n)$ to $f_2(n)$, respectively. This simple example reveals the significance of the inverse cost function in the
346 optimal sensor configuration under the proposed methodology based on value of information.

347 The importance of the definition of the inverse cost function $f(n)$ is also revealed in case study 2. Figure 9
348 summarizes the EVI (recall Eq. (9)) for each of the optimal configurations under the consideration of both $f_0(n)$
349 (recall Eq. (15)) and $f_1(n)$. As observed, the optimal solution under $f_1(n)$ provides higher EVI values, which is
350 in agreement with the more accurate and efficient (i.e. using fewer sensors) damage reconstructions observed in
351 Figures 7c to 7e. Observe also that the EVI not only enables a rational criterion for optimal sensor configuration, but
352 also establishes a measure to compare between several candidate inverse cost functions.

353 Nevertheless, this dependence upon the inverse cost function may arguably be seen as an additional complexity,
354 which may lead to suboptimal results depending on priorities, e.g. whether or not the cost is more important than the
355 amount of information gained. However, we show here that adopting the value instead of the amount of information
356 provides a better identifiability of the optimal design point. This is manifested in Figure 10, where the KL divergence
357 criterion is compared against the EVI criterion, taking the second case study using $f_0(n)$ and $\alpha = 10$ as an example.
358 As shown in the results, an optimal SHM design point would be barely identifiable using the amount of information
359 criterion (Figure 10a) in view of the asymptotic behavior of the KL divergence. In contrast, the estimator of the
360 EVI (i.e., $J(\mathcal{C}_{opt}^n)$) shown in Figure 10b reaches a globally identifiable optimum value, which greatly facilitates the
361 optimization process.

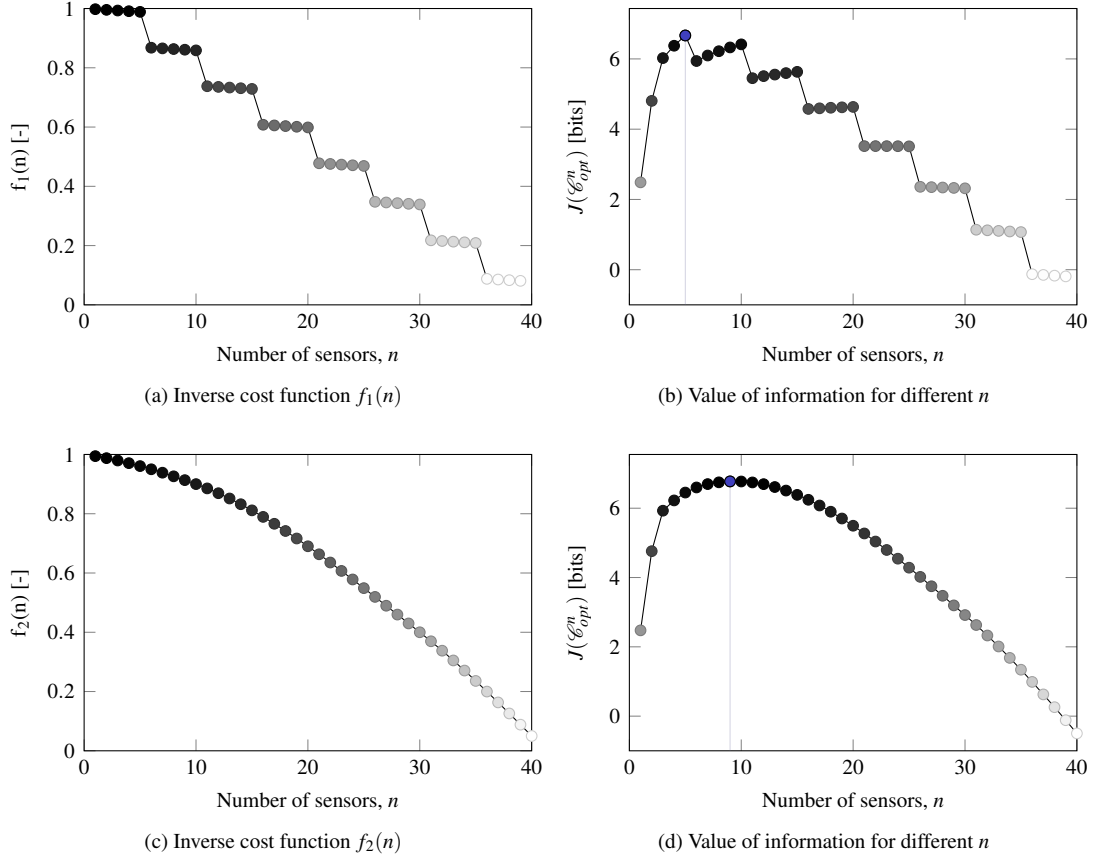


Figure 8: Alternative inverse cost functions accounting for SHM systems using slots of 5 sensors (a) along with the resulting influence in the optimal sensor configuration in (b). In (c), a different function which is continuous rather than discrete with a similar trend to (a) leads to a different optimal sensor configuration in (d).

362 Note that the constant α included in Equation (1) plays an important role in modulating the EVI since it scales
 363 the relative cost component in Equation (9). However, in view of Figure 11a the rank between the candidate sensor
 364 positioning strategies results to be unaltered by this constant, where the EVI is calculated for $\alpha \in [0, 1000]$ using $f_1(n)$
 365 and the data obtained in case study 2. Conversely, α is shown to influence the optimal number of sensors, as observed
 366 in Figure 11b. In particular, n''_{opt} tends to decrease as long as α increases as a consequence of the increasing cost of
 367 sensors in the context of the trade-off between information and cost given by Equation (9).

368 4.2. On the extensibility of the method

369 The proposed methodology provides the optimal sensor configuration for a specific area of possible damage oc-
 370 currence based on a ToF model, which takes into account the position of sensors, actuators, damage, and the wave
 371 propagation velocity. Further, the resulting optimal configurations have proven efficiency in reconstructing the dam-
 372 age position. However, the ToF model lacks of advanced physics-related information such as the interaction of the
 373 ultrasonic guided-wave with the damage or the effect of attenuation in the localization in the damage. This prevents

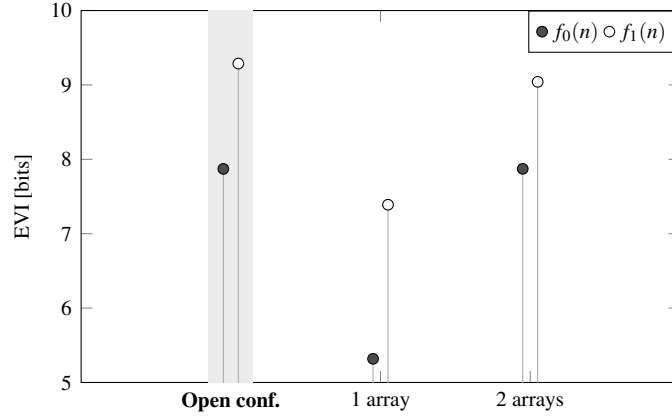


Figure 9: Results of the comparison in terms of EVI (circles) for the different sensor positioning strategies and inverse cost functions, namely $f_0(n)$ (gray symbols) and $f_1(n)$ (white symbols), for case study 2. Points on the upper part of the chart represent more informative alternatives.

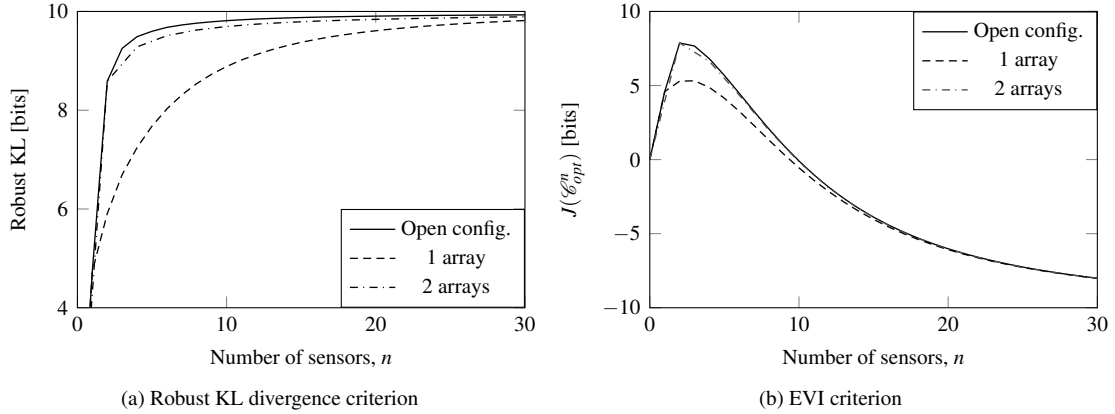


Figure 10: Comparison between the robust KL divergence over the area of possible damage occurrence and the EVI criterion using $f_0(n)$.

374 the proposed methodology from considering advanced damage features other than the position, such as damage mode
 375 and extent (e.g., delamination in composites), and from providing an optimal sensor configuration considering them.
 376 In order to consider these advanced damage features, a more complex and physics-based wave propagation model
 377 such as the hybrid wave and finite element methods [71] is required. The use of these models in the design of the op-
 378 timal sensor configuration is still unfeasible due to their considerable computational cost. Therefore, the development
 379 of methods to alleviate such computational complexity, such as surrogate methods [72–76], constitutes a desirable
 380 extension of this work.

381 Furthermore, note that the proposed method seeks the optimal sensor configuration, assuming that the actuators
 382 are fixed in the structure. However, the position and number of actuators play an important role in the optimization
 383 problem. Besides, the sequential sensor placement algorithm using an exhaustive search may find a limitation in
 384 practice when considering very large areas of possible damage occurrence. In this context, an ongoing extension of
 385 the proposed methodology is the consideration of both sensors and actuators in the optimization problem considering

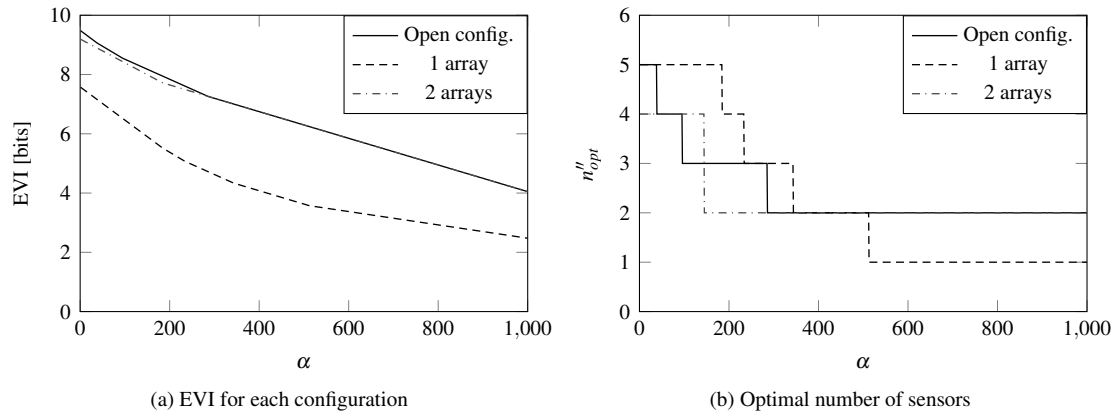


Figure 11: Effect of α values in the EVI of each sensor configuration in (a) and its effect in the optimal number of sensors in (b).

386 large structures.

387 5. Conclusions

388 A novel optimal sensor configuration approach based on value of information is presented in this paper. The
 389 methodology is based on a Bayesian damage localization framework for optimal sensor placement, while the optimal
 390 number of sensors is obtained by assessing the expected value of information; thus, uncertainties coming from several
 391 sources are taken into consideration. The effectiveness in providing an optimal design point has been illustrated using
 392 two case studies considering both flat and stiffened plate structures, respectively. The following conclusions are drawn
 393 from this paper:

- 394 • The proposed approach provides a value of information-based framework that trades-off the amount of infor-
 395 mation and the cost of monitoring, giving a globally identifiable optimal design point.
- 396 • The open configuration provides the best damage reconstruction using the lower number of sensors due to its
 397 flexibility in positioning sensors freely over the structure;
- 398 • Based on the results, the two-array configuration provides more accurate damage reconstructions using less
 399 number of sensors than the one-array configuration.
- 400 • An accurate definition of the inverse cost function has been shown to be key for an unbiased optimal sensor
 401 configuration under the proposed methodology.

402 Further research work is needed to address (1) the extensibility of the proposed method using a physics-based
 403 model instead of the time of flight one used here, (2) the efficiency of the method when considering the optimization
 404 of both actuators and sensors in large plate-like structures, and (3) the optimality of the inverse cost function in
 405 different specific cases.

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