Accepted Manuscript

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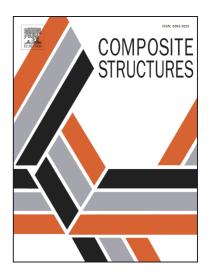
PII: S0263-8223(18)30423-9

DOI: https://doi.org/10.1016/j.compstruct.2018.06.062

Reference: COST 9856

To appear in: Composite Structures

Received Date: 27 January 2018
Revised Date: 29 April 2018
Accepted Date: 16 June 2018



Please cite this article as: Chen, S., McGregor, O.P.L., Harper, L.T., Endruweit, A., Warrior, N.A., Optimisation of local in-plane constraining forces in double diaphragm forming, *Composite Structures* (2018), doi: https://doi.org/10.1016/j.compstruct.2018.06.062

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Optimisation of local in-plane constraining forces in double diaphragm forming

S. Chen, O.P.L. McGregor, L.T. Harper*, A. Endruweit, N.A. Warrior Composites Research Group, Faculty of Engineering, University of Nottingham, UK, NG7 2RD

*Corresponding author: lee.harper@nottingham.ac.uk

Abstract

Rigid blocks (risers) were introduced in the double diaphragm forming (DDF) process to change the local in-plane strain distribution in the diaphragms, aimed at reducing wrinkling defects in the production of fabric preforms. A two-step optimisation method was developed to determine the position and dimension of each riser. In Step I, optimisation of the riser position was conducted using a simplified finite element (FE) model coupled with a genetic algorithm (GA). The height of each riser was optimised in Step II using a detailed FE model with the optimised riser positions from Step I. For demonstration, a hemisphere preform was manufactured by DDF using the optimum riser arrangement established by the optimisation routine. Results indicate that the optimum riser pattern (shape and position relative to the component boundary) can dramatically improve the preform quality through reduction of out-of-plane wrinkles, validating the feasibility of the two-step routine.

Keywords

A. Fabrics/textiles; C. Finite element analysis (FEA); E. Forming; E. Preform

1 Introduction

Double diaphragm forming (DDF) is a process suitable for production of dry preforms for manufacture of composite components employing liquid moulding. In DDF, fabric plies are sandwiched between two deformable diaphragms, which are subsequently deep-drawn. over a rigid tool by applying a pressure gradient normal to the surface [1, 2]. Compared to single diaphragm forming, both diaphragms constantly maintain contact with the fabric plies, reducing the potential for out-of-plane wrinkling and enabling rapid demoulding of the finished preform. Since only vacuum pressure is applied and only single-sided tooling of moderate stiffness is required, DDF generally requires lower capital investment compared to matched tool forming [3, 4]. Depositing stacks of fabric plies flat on the bottom diaphragm prior to forming is quick and can be easily automated. In addition, DDF is highly repeatable, and net-shaped preforms can be produced which minimises in-process waste. However, defects such as fibre bridging (poor fabric-tool conformity) and fabric wrinkling are more likely to occur in DDF than in press-tool forming or hand lay-up, in particular for complex component geometries. This is related to plies being enclosed by diaphragms, which limits the options to control the forming parameters. Bridging can effectively be reduced by optimising ply shapes [1, 2, 5], but this approach depends on the location of the defect relative to the ply perimeter. For more complex geometries it may be necessary to introduce darts (slits) in the ply to improve the formability and mitigate forming defects [1, 2, 5-8]. The downside to this approach is that fibre continuity is lost, affecting the mechanical performance, which may only be acceptable in non-critical regions of the component. The quality of formed components or preforms is also affected by the design of the single-sided tool [9], as draping the fabric over a male tool or drawing it into a female

tool may give different results. While some defects may be successfully alleviated following this method, the design variables are limited and the additional cost of tool redesign is typically high.

A general approach for preventing defect formation is the application of localised in-plane tension in the plies, using grippers to constrain the material draw-in during the forming operation. For matched tool forming, this is discussed in several published studies [10-12]. Similarly, Krebs et al. [13] showed that pre-tensioning the preform, by stretching the diaphragms prior to forming, can reduce the effect of out-of-plane fabric buckling in diaphragm formed components. In DDF, biaxial in-plane tension is applied to the plies through friction on the diaphragm surfaces, which can be controlled by adjusting the vacuum pressure between the diaphragms. The fabric is constrained much less in DDF than in matched tool forming, therefore increasing the risk of out-of-plane fabric buckling as it shears. A non-uniformly distributed constraining force is required to produce a defect-free component [14, 15], for which cost-effective solutions need to be found. One approach to achieve this is to locally increase the tension in the fabric in regions prone to defect formation.

Risers, i.e. rigid blocks, placed between the bed of the machine and the lower diaphragm (see Figure 1), can be used to locally increase the tensile strain in the diaphragms which is transferred to the fabric plies (see Figure 2). The position, shape and height of risers can be modified to control the diaphragm strain in regions where wrinkles are likely to occur in the fabric [16]. However, risers are typically designed and placed based on trial and error, and it is difficult to predict intuitively efficient riser configurations to remove wrinkles. Therefore, the development of a predictive numerical method is required.

This study addresses the optimisation of DDF processes through the introduction of risers to adjust local in-plane constraints, in order to reduce defects related to fabric wrinkling.

Numerical analysis of DDF based on detailed finite element (FE) models is computationally costly due to non-linear material behaviour, large deformations and complex surface contacts [1, 2]. Hence, a simplified FE model is developed to simulate the DDF process and to facilitate the optimisation of the riser arrangement by employing a genetic algorithm (GA). A case study is presented to demonstrate the effectiveness of applying an optimised riser pattern to remove wrinkling defects from the final formed shape, and to validate the simulation-based optimisation method through comparison with experimental results.

2 Materials

2.1 Fabric material

All preforms were made using FCIM359 biaxial carbon fibre non-crimp fabric (NCF), supplied by Hexcel, Leicester, UK. Each ply is 0.4 mm thick (uncompressed) and consists of 440 gsm of carbon fibre in 24K tow format. The fibre architecture is ±45° with a pillar stitch at 0° in the roll direction. While its dominant forming mechanism is in-plane shear deformation, similar to biaxial woven fabrics, the mesoscopic NCF architecture is based on intra-ply stitches rather than interlacing of the primary yarns. For the ±45° NCF with 0° aligned pillar stitches (FCIM359 NCF), positive shear was defined as in-plane shear inducing tension in the stitch threads, while negative shear corresponds to stitch threads being in compression [17, 18]. The shear resistance in positive shear is significantly influenced by the intra-ply stitches. However, the stitches fail at large shear angles (which induce high tensile strains in the

stitch thread), causing irreversible damage and a significant reduction in shear resistance [19]. Conversely, in negative shear, the stitches are loaded in compression and the NCF exhibits similar behaviour to a woven fabric [20], since friction between the primary yarns is the main source of the in-plane shear resistance. Once the fabric locking angle is reached due to the onset of lateral yarn compression, a rapid increase in shear resistance is induced and out-of-plane wrinkling occurs accordingly. The placement of intra-ply pillar stitches (bisecting the angle between the primary yarns) leads to asymmetric shear behaviour, increasing the difficulty in reducing forming defects [21]. The occurrence of wrinkling defects has been correlated with the shear angle, which could be used as a defect index [2, 18]. Hence, controlling the local shear angles has been identified as a strategy to minimise wrinkling defects. Only wrinkling induced by shear beyond the locking angle is considered in the current paper, as it has been previously identified to be the main wrinkling defect. An asymmetric in-plane shear behaviour was previously derived from picture frame shear tests and modelled using a non-orthogonal constitutive relation as discussed in detail in a previous paper by the authors [18]. It was implemented in a user-defined subroutine in Abaqus/Explicit, using the data shown in Table 1. The onset of forming defects was correlated to the variation in shear angle. In positive shear, the stitch yarns are in tension and successive failure starts from 28° until 43° when the stitches fail completely. In negative shear, the stitch yarns are in compression with negligible contribution to the shear modulus. Out-of-plane fabric wrinkling was observed at shear angles larger than 50° due to yarn locking.

2.2 Diaphragm material

The diaphragms were made from 1.56 mm thick Supervac silicone sheet with 50 Shore A hardness, supplied by Silex Ltd, UK [1, 2]. The density of the diaphragm is 1600 kg/m^3 . As described in detail elsewhere [2], a series of mechanical tests were performed on coupons to characterise the in-plane behaviour of the diaphragm. Uniaxial tensile tests were performed on a universal testing machine at a strain rate of 0.03 s^{-1} . The cross-section of the sample was $5.00 \text{ mm} \times 1.56 \text{ mm}$. Equibiaxial tensile tests and planar shear tests were performed at specimen dimensions of $75 \text{ mm} \times 75 \text{ mm}$. A displacement rate corresponding to a strain rate of 0.03 s^{-1} was applied. Pure shear tests were performed by keeping the specimen width constant, whilst applying an axial displacement corresponding to a strain rate of 0.03 s^{-1} .

The observed hyperelastic, non-linear stress-strain behaviour was modelled using a second-order Ogden model [22]. The Marquard-Levenberg algorithm [23] was employed to determine the Ogden constants (see Table 2), by simultaneously performing a non-linear least-squares fit on the uniaxial and equibiaxial test data. The root mean squared error (RMSE) of the obtained Ogden model is 2.2 % for the uniaxial case and 5.5 % for the equibiaxial case compared to the experimental data. A third mode of deformation (i.e. pure shear data) was also tested, and the data was used to further validate the model, showing an RMSE of 5.5 %.

2.3 Friction between materials

The friction behaviour was tested [2] for tool-diaphragm and diaphragm-fabric surface pairings employing sled tests (also fabric-fabric, but these data are not used here), according

to ASTM D1894, ISO8295. For relative surface movement at constant velocity (100 mm/min), friction coefficients were calculated from the ratio of the tangential (pulling) force and the normal force (10 N, corresponding to a normal pressure of 2 kPa), and the average value was adopted from five repeats for each surface pairing. Friction at all interfaces between different materials in DDF was assumed to be isotropic. The average coefficients used here were 0.67 for tool-diaphragm friction and 0.52 for diaphragm-fabric friction (and 0.36 for fabric-fabric friction).

This approach simplifies the friction behaviour, as both the fabric surface and the diaphragm surfaces are deformable, and the real friction coefficients may vary with applied normal force. A decrease in friction coefficient with increasing normal pressure was documented by Sachs et al. [24] for a woven material on a rigid surface and by Avgoulas et al. [25] for a NCF on a rigid surface. For fabric-diaphragm contact, geometrical features on the deformable surfaces may interlock, resulting in strongly increased friction coefficients as the pressure is increased. Although the actual pressure when vacuum is applied is higher than the normal pressure in the friction tests, and the normal force may be distributed non-uniformly during the forming process, any effect on the friction behaviour is ignored for the purpose of this study.

3 Optimisation methodology

3.1 General strategy

A hemisphere with a diameter of 100 mm was formed using the DDF process. The primary yarns of each biaxial fabric ply were orientated along the 0°/90° direction (see Figure 1), and

the fabric preform was sandwiched between two diaphragms. The initial blank size was 300 mm × 300 mm with a thickness of 0.4 mm. A pressure differential was applied normal to the diaphragm surface, and the tensile deformation of the diaphragms around the blank provides the in-plane constraining forces (transferred through friction) to control the material forming behaviour.

Aimed at reducing wrinkle formation, risers were introduced to change the local strain in the diaphragms to constrain fabric deformations. Since it is difficult to predict efficient riser configurations intuitively in practice, wrinkling defects were minimised based on predictive numerical simulations. A two-step approach was developed: in Step I, the riser position was optimised; in Step II, the riser height was optimised. The optimisation in Step I was used to find appropriate locations to place risers, which was followed by a detailed design evaluation to finalise the geometry of each riser.

3.2 Step I: Optimisation of riser position

3.2.1 Simplified FE model

The high CPU cost for the DDF simulations was primarily related to the non-linear behaviour of the diaphragm material and the complex contact conditions. The maximum generated frictional force at the diaphragm-fabric interface was large enough to prevent relative displacement, when the normal pressure applied to the diaphragms was approximately 1 bar and the friction coefficient was 0.52. Thus, as shown in Figure 3, the contribution of the diaphragms was superimposed to the fabric model in regions where the diaphragm was in direct contact with the preform. In all other areas, the diaphragm was represented by 1D spring elements with equivalent in-plane stiffness. These were connected to the edge of the

fabric plies and to a rigid outer frame, thus constraining the blank. The constraining forces from the springs were therefore used to approximate the in-plane constraints obtained from the diaphragms. This dramatically decreased the CPU time from about 1 hour to less than 110 seconds.

While 1D springs are assumed fully elastic here, the approach is transferable to other diaphragm material behaviours, e.g. updating the spring properties to represent the equivalent constraining forces (even including nonlinear behaviour). This simplification of the constraints may affect the accuracy of results, but it enables a near-optimum to be determined as a starting point for further refinement in Step II, which is performed at much higher accuracy.

3.2.2 Optimisation approach in Step I

The main purpose of Step I was to determine an appropriate pattern for the riser locations relative to the component boundary. Referring to the DDF process, obtaining larger in-plane constraining forces requires larger tensile strains in the diaphragm. The optimum spring force distribution around the perimeter of the material area (where the diaphragms and the blank are superimposed) in the simplified model gives an indication to the optimum location for risers (see Figure 4a). In order to generate different in-plane constraining forces, the stiffness of each spring was defined as an optimisation variable. As shown in Figure 4b, the required distribution of in-plane constraining forces from the deformation of the diaphragm was equivalently optimised by changing the local spring stiffness. Subsequently, risers were introduced at the identified locations to implement the local adjustment of the constraining force in Step II by changing the riser height. The effect of the simplification described in

Section 3.2.1 on the precision of the simulation is acceptable, since it only provides a starting point for the next step to carry out further refinement.

In-plane constraints in the form of springs were applied to the nodes along the edge of the area with superimposed material properties at 10 mm spacing. The optimisation of the riser position was performed using a GA coupled with a FE analysis to determine the optimum stiffness value for each spring within a user-defined range. The optimisation was implemented using Matlab (see Figure 4b). Abaqus/Explicit input files were created using a group of spring stiffness arrangements called "individuals" generated from the GA for each loop or "generation". FE simulations were implemented in Abaqus/Explicit to determine the shear angle distribution after forming for each individual. FE results were then returned to the GA in Matlab to determine the corresponding fitness value for the convergence check. Iterations continued until the optimum was achieved.

The constraining force from each spring could be parameterised as

$$F_i^{ct} = k_i^{ct} d_i^{ct} \quad (i = 1, 2, ..., m)$$
 (1)

where m is the number of springs, k_i^{ct} is the stiffness of the ith spring, d_i^{ct} is the extension of the ith spring, and F_i^{ct} is the corresponding constraining force. Consequently, the optimisation variables were converted into a stiffness k_i^{ct} ($i=1,2,\ldots,m$).

The optimisation problem in Step I was expressed as

minimise
$$f\{k_1^{ct}, k_2^{ct}, ..., k_m^{ct}; \gamma_{12}(x, y, z)\}$$
 (2)

$$\begin{aligned} & \text{subject to} \begin{cases} k_i^{ct} \in \left[\left(k_i^{ct} \right)^{low}, \left(k_i^{ct} \right)^{upp} \right] & (i = 1, 2, \dots, m) \\ \gamma_{12}(x, y, z) \in [-90^\circ, 90^\circ] \\ (x, y, z) \in \Omega_M \end{cases} \end{aligned}$$

where $f\{\cdot\}$ is the GA fitness function to describe the selection criterion of the spring stiffness. The available stiffness range of each spring is selected from $(k_i^{ct})^{low}$ to $(k_i^{ct})^{upp}$. Ω_M is the spatial material region; $\gamma_{12}(x,y,z)$ is the continuous shear angle distribution in the material region, Ω_M .

In the GA, each individual constraining arrangement was evaluated using a fitness function employed as the assessment criterion. A maximum value criterion (MAXVC) was adopted here due to faster convergence compared with the Weibull distribution quantile criterion (WBLQC) previously used by the authors for a similar optimisation problem [26], whilst maintaining acceptable accuracy. The objective was therefore to keep all local shear angles below the locking angle, by minimising the maximum shear angle. The maximum was derived from the finite element approximation for $\gamma_{12}(x,y,z)$. Thus,

$$f_{MAXVC}\{k_1^{ct}, k_2^{ct}, \dots, k_m^{ct}; \gamma_{12}(x, y, z)\} = \max_{(x, y, z) \in \Omega_M} \{|\gamma_{12}(x, y, z)|\} \approx \max_{i=1, 2, \dots, N} \{|\gamma_i|\}$$
(3)

where $f_{MAXVC}\{\cdot\}$ denotes the fitness function using MAXVC, which aims to minimise the maximum shear angle; N is the total number of material points which corresponds here to the number of finite elements; $|\cdot|$ is the absolute value of the variable; γ_i is the value of the shear angle at the ith material point, (x_i, y_i, z_i) . The value of f_{MAXVC} was used for quantitative assessment of the fitness, since the spring arrangement influences the shear angle distribution.

3.2.3 Interpretation for riser position

The optimised distribution of spring forces around the blank was interpreted for each riser pattern. Once the average stiffness of each spring from the optimum generation is larger than the average stiffness, a riser is required in the corresponding spring direction. This criterion was expressed as

$$R_{i} = \begin{cases} 1 & (i.e.riser\ required), & \text{if } \overline{k_{i}^{ct}} > [\left(k_{i}^{ct}\right)^{low} + \left(k_{i}^{ct}\right)^{upp}]/2; \\ 0 & (i.e.riser\ not\ required), & \text{if } \overline{k_{i}^{ct}} \le [\left(k_{i}^{ct}\right)^{low} + \left(k_{i}^{ct}\right)^{upp}]/2. \end{cases}$$
(4)

where R_i denotes whether a riser is required or not at the \hbar th position, $\overline{k_i^{ct}}$ denotes the average stiffness of the \hbar th spring from all the individuals of the corresponding generation.

The riser shape was limited to a cuboid in the current work, where the faces were parallel to the square base of the hemisphere tool. Each selected spring represented a riser of length 10 mm, which was equal to the spring spacing. Neighbouring risers were combined to form consolidated risers. The criteria for riser consolidation should be defined according to practicalities, such as avoiding small and sharp risers. If the distance between two adjacent risers was smaller than a threshold value, they were considered to be part of the same riser. A minimum riser length was also specified, and any isolated risers were discarded. Here, these practical considerations of riser design were implemented manually, although they could be included in the optimisation as additional constraints with a significant increase of optimisation variables in the objective function. A large number of variables would be required to define the riser geometry and the additional quantity of variables could also change during optimisation. Thus, the difficulty in automatically generating FE models would dramatically increase as well as the optimisation cost.

3.3 Step II: Optimisation of riser height

While a simplified FE model was used to efficiently locate riser positions as a starting point where reduced precision was acceptable, a more detailed FE model (previously developed by the authors and validated for complex preform geometries [2], see Figure 1) was employed to finalise the optimum height of each riser in Step II. Since the position of the risers was the optimisation variable in Step I, only a single additional degree of freedom (i.e. riser height) was considered in Step II. The effect of varying riser height on the in-plane constraining force is of the same order as changing other geometric parameters of the risers, such as their shape. Although the local tensile strain in the diaphragms can also be adjusted through variation of other geometric parameters, only riser height was varied here to facilitate practical implementation of the optimisation process.

The fabric was modelled in Abaqus/Explicit using 5 mm × 5 mm square membrane elements (M3D4R), which was found to be suitable in a mesh sensitivity study presented elsewhere [18]. The diaphragms were modelled using 5 mm × 5 mm S4 shell elements, which were constrained in the x-y plane around the perimeter to replicate the constraints on the diaphragm forming rig. Risers were modelled as cuboid blocks placed between the bed of the machine and the lower diaphragm at the positions determined from Step I. All parts of the tooling (tool, machine bed and risers) were modelled as rigid bodies. A penalty contact algorithm was used to define the behaviour at the interfaces. A Coulomb friction model was adopted for all contact surfaces, using the experimentally determined values for the coefficients of friction mentioned above. Initially, pressure was applied to the upper surface of the top diaphragm and the lower surface of the bottom diaphragm, providing the clamping force on the fabric plies generated by the vacuum between the diaphragms. A

frame around the diaphragm edges was employed to drive the material stack to establish contact with the machine bed, which was modelled by applying a vertical displacement to the edge nodes of the diaphragms. Then, the pressure applied to the bottom diaphragm was reduced to create a pressure differential for drawing the diaphragm arrangement into contact with the surface of the tool. Gravity was neglected in the FE model.

In the simulations run here, discrete heights of 0 mm (i.e. no risers) to 50 mm were considered for the heights of the risers, increasing in 10 mm increments. The run time for each individual forming simulation to investigate the influence of riser height was approximately 1 hour using an Intel® CoreTM i7-3820 CPU at 3.60 GHz. Therefore the total run time for 6 simulations to optimise the riser height for the hemisphere was approximately 6 hours. The evolution of the maximum shear angle in the fabric preform using different heights of risers was plotted to seek an optimum solution that would reduce wrinkling defects under both positive and negative shear for the asymmetric material behaviour of the NCF studied here.

4 Results and discussion

4.1 Validation of simplified DDF model

In total, 961 nodal positions were selected to compare the quality of simulation results obtained using the simplified model (open red circles in Figure 5) with the detailed model (solid blue dots in Figure 5), using a reference hemisphere geometry formed without risers. The distances between the corresponding nodes were calculated to compare the difference of the two grids, represented as a grey-scale contour. The RMSE in distance between the

two grids is 0.95 mm and the maximum error is 2.74 mm, for a blank that was originally 300 mm x 300 mm. The difference in shear angle magnitude was also calculated, as shown in Figure 6. The RMSE is 1.3° and the maximum difference is less than 3.5°. The small difference in results between the simplified model (open red circles) and detailed model (solid blue dots) in terms of nodal positions reflects the inclusion of slip of the fabric on the diaphragms in the detailed model. Slip in the detailed model allows more fabric draw-in than in the simplified model, where slip between fabric and diaphragm is not considered and only the average displacement is calculated. Results indicate that the simplified FE model is able to produce a prediction of forming behaviour equivalent to that based on the detailed FE model to fulfil the requirement of Step I.

4.2 Riser position optimisation from Step I

In Step I, applying springs to the edges of the blanks at a spacing of 10 mm resulted in 31 constraining positions per edge, i.e. 124 variables in total. However, due to the symmetry of the blank with respect to its two diagonals, the number of independent variables was reduced to just 31.

The stiffness of each spring was normalised with respect to the initial stiffness of the corresponding representative diaphragm strip (k_i^{rep}) according to [27], where the stiffness was selected from the range of k_i^{rep} to $5k_i^{rep}$. This was then used as an optimisation constraint. The normalised average distance between individuals in the solution space is a measure of the diversity of the population. For the 31 variables in Step I, this must be less than the maximum value of 22.3 (i.e. the diagonal distance $\sqrt{31 \times (5-1)^2}$). Furthermore, this average distance should progressively decrease for subsequent generations, indicating a

reduction in search space and convergence towards the global optimum. The diversity of the population for each generation in Step I was checked by evaluating the average distance between individuals, shown in Figure 7a. For the first five generations the value is approximately 13, indicating that the initial population covers approximately 58 % (13/22.3) of the solution space. The average distance reduces by about 1.5 % (0.33/22.3) for each generation until convergence, indicating that evolution is progressive, allowing sufficient opportunity for elite genes. Once both the average distance and the fitness value are constant for at least 5 consecutive generations, convergence has been achieved and the optimum has been found.

Figure 7b shows the evolution of the fitness scores for Step I. The magnitude of the adaptive Fitness Range is similar for each generation until the Best Fitness converges, implying that a wide search range was adopted throughout. The range of the fitness score varies due to adaptive mutation. The optimum solution (i.e. convergence of the Best Fitness) is achieved during generation 41. Perturbations induced by further mutations during the next 4 generations (indicated by a non-zero fitness range) appear to have no influence on the optimum solution. Furthermore, the mutation probability reduces to zero following generation 45, after the optimum solution was determined. Therefore, Figure 7 confirms that the present diversity prevents local optimum solutions, random selection and instability. As shown in Figure 8, optimum riser positions were identified from Step I. A spring represented by a thick line indicates that a riser is required, where the spring stiffness was larger than the average of $(k_i^{ct})^{low}$ and $(k_i^{ct})^{upp}$, i.e. $3k_i^{rep}$. Springs represented by thin lines indicate that an additional in-plane constraint is not required in that location. The

maximum shear angle in the positive shear region decreases by 9.4°, from 48.3° for the

reference case with no risers (Figure 8a) to 38.9° when risers are placed at the optimised positions to locally adjust the constraining forces (Figure 8b). In the negative shear region, a significant reduction in maximum shear angle (by 10.3°) is also achieved using the optimised riser arrangement. The predictions in Step I indicate that placement of risers has the potential to reduce the generation of forming defects, especially unacceptable out-of-plane wrinkling induced by over-shearing.

Subsequently, the individually identified riser locations in Figure 8b were consolidated for practical implementation, combining neighbouring riser positions (i.e. springs marked in bold) and eliminating isolated ones. If the maximum distance between any two adjacent bold springs did not exceed 30 mm, which is equivalent to the width of 6 finite elements or 3 times the spring spacing, the springs were combined into a consolidated riser. Also, the length of each consolidated riser obtained from combining springs should not be smaller than the width (i.e. 50 mm), otherwise it was removed. As a result, only one riser was required along each edge as shown in Figure 9. Each riser was 130 mm × 50 mm in the x-y plane, with 6 mm fillets around the edges, where the longest edge was parallel to the corresponding undeformed blank. A distance of 200 mm was maintained from the longest edge of riser to the centre of the hemisphere. According to the obtained optimum riser arrangement in Figure 9, these four risers were placed symmetrically with respect to the two diagonals of the undeformed square blank, where the offset in riser position from the middle point of each edge is a reflection of the non-symmetrical fabric shear behaviour. Due to asymmetric shear behaviour of the fabric induced by the pillar stitches, the angle for onset of wrinkling is smaller in positive shear than in negative shear. Application of additional in-plane constraining forces is required to reduce the shear deformation, primarily in regions of positive shear, to avoid the onset of wrinkling. It can be concluded

that, as a general design guideline, optimum positions for risers are around regions of positive shear resulting in constraining forces being applied along both directions of the primary yarns.

4.3 Riser height optimisation from Step II

It is essential to adopt the detailed FE model in Step II to finalise the riser design, as it directly replicates the exact behaviour of the diaphragms rather than supplying an equivalent response instead. Risers were placed at the specified locations according to Figure 9. Simulations were run where the height of all risers was increased incrementally; 0 mm (i.e. no risers), 10 mm, 20 mm, 30 mm, 40 mm, 50 mm respectively. The shear angle distributions were obtained from numerical simulations using the detailed FE model to investigate the influence of riser height as shown in Figure 10. Then, the optimum height of each riser was determined from Step II to maintain the optimum formability.

The evolution of the maximum shear angle using different riser heights was plotted to determine an optimum solution, in order to reduce the number of defects under both positive and negative shear, as shown in Figure 11. It shows that the maximum shear angle in the negative shear region (black dot and dash line) decreases with increasing riser height and reaches a plateau at approximately 40 mm. On the other hand, the maximum positive shear angle has a local minimum at around 20 mm to 25 mm. The two curves in Figure 11 intersect at approximately 26.5 mm. Although lower shear angles in both positive and negative shear are preferred in order to reduce the possibility of generating forming defects, as shown in Table 1 the critical shear angle for out-of-plane wrinkling defects to occur is smaller (about 43°) in the positive shear region than in the negative shear region (around

50°) due to the non-symmetrical material behaviour of the NCF. Therefore, the optimum riser height should be selected from the range between 20 mm and 25 mm to prevent the occurrence of defects in positive shear regions. From this range, the highest possible value should be picked to simultaneously obtain a small shear angle in negative shear. Hence, a riser height of 25 mm was selected.

Using the optimised riser arrangement, the maximum positive shear angle can be reduced from 49.4° to 41.6°, whilst the maximum negative shear angle can be reduced from 49.6° to 41.1° according to Figure 11. For other geometries, it may not be possible to reduce the maximum shear angle below the critical threshold value to prevent the onset of defects in all cases. However, placing risers can be used in combination with other methods such as trimming the blank shape and placing darts [1, 2, 5]).

4.4 Experimental validation of riser optimisation

A laboratory–scale diaphragm forming machine [1, 2] was used to validate the optimum riser arrangement as shown in Figure 12. A single ply of FCIM359 NCF was formed in each forming experiment, and the biaxial preform was initially placed in the 0°/90° direction (see Figure 9). The hemisphere was formed using the optimum riser arrangement from Figure 9, which was compared to the case without risers. A small amount (6 %wt) of reactive binder (Momentive Epikote 620) was applied on both surfaces of the biaxial fabric. The fabric material was loaded onto the lower diaphragm and covered by the upper diaphragm. The edges of the diaphragms were sealed by closing a zipper to encapsulate the fabric. Then, a vacuum was generated between the diaphragms to clamp the material. The sandwiched assembly was raised to within 150 mm of infrared heaters and heated to 90 °C in order to

melt the powdered binder. Once the set-point was reached, the assembly was quickly lowered to drape over the rigid tool. Another vacuum was then generated between the tool and the lower diaphragm to provide a forming pressure for completing the forming process. The preform was cooled to below the melting point of the binder before being released from the diaphragms, in order to prevent distortion or spring back and to conserve the formed configuration for measurement of the local shear angle distribution.

A square grid with a line spacing of 10 mm was marked on the outer surface of the NCF prior to forming. The grid lines were aligned with the directions of the primary yarns. During forming, the grid deformed together with the yarns, allowing local variations in fibre direction to be tracked. The 3D coordinates of grid intersection points were digitised using a coordinate measuring machine (CMM). In total, 961 points were measured which correspond to the points taken from the detailed FE simulation. The distance between numerical and experimental coordinates were compared using a Matlab script to validate the FE simulations.

As shown in Figure 13, the simulated grid intersections are a close match to the measured points. The RMSE between the simulated grid and the experimental grid is 1.69 mm for the reference case (i.e. no risers) and 1.21 mm for the optimum case. The RMSE between two corresponding surfaces is less than 0.6 % for both cases. Also, the maximum error is 4.37 mm for the reference case (i.e. no risers) and 3.26 mm for the optimum case, which is below 1.5 % for both cases. Figure 13 shows a slightly lower draw-in in the experiment than in the numerical result, which is a result of zero bending stiffness of the membrane elements in the numerical model. Importantly, this does not induce a considerable effect on the shear angle distribution which was adopted as an index for the occurrence of wrinkling. The

maximum error of less than 1.5 % indicates that the detailed FE model is able to predict realistic forming behaviour of the NCF.

Figure 14 shows photographs of formed fabric layers. A general observation is that, compared to previous results from matched tool forming of the same fabric discussed in detail in a recent study [18], asymmetry of the formed material layer is significantly reduced due to the presence of the diaphragms which dominate the forming behaviour. As shown in Figure 14a, significant out-of-plane wrinkling occurs when forming a hemisphere using DDF with no risers. Most of the wrinkles appear in the positive shear region, which is consistent with the result in Figure 11 for the case with 0 mm risers (i.e. no risers). Using risers at the predicted optimum arrangement and optimum height (25 mm), the wrinkling defects are removed from the final preform surface (see Figure 14b). This indicates that the risers are effective in adjusting the local in-plane constraining forces transferred though the diaphragms to modify the local shear angle distribution, which can be used to reduce wrinkling defects in DDF by reducing the local shear angles. The riser arrangement can be optimised using the developed optimisation method for this particular case. However, it may require more consideration for more generic cases, starting from the first principles of the underlying mechanisms causing wrinkling.

5 Conclusions

A method was introduced to improve the performance of the double diaphragm forming (DDF) process by locally adjusting in-plane constraints using a series of rigid blocks (risers). The riser arrangement was optimised to minimise wrinkling defects in the reinforcement.

In the first of two optimisation steps, a simplified FE model was developed to efficiently simulate the DDF process for forming NCF, where springs were applied around the edge of the blank to replicate the in-plane force derived from the diaphragms. The stiffness of each spring required to constrain the blank was optimised using a GA, in order to minimise the maximum fabric shear angle for wrinkling reduction. The distribution of the resulting spring forces around the perimeter of the fabric plies was then reproduced by introducing risers at locations where adjustment of in-plane tensile force was required. The criterion for adding a riser was based on the ratio of local spring stiffness and diaphragm stiffness. Individual risers were then consolidated to enable industrial implementation according to practicalities. In the second optimisation step, simulations were run to determine the effect of riser height on the fabric forming behaviour, where different heights of risers were considered. A riser height of 25 mm was found to be the optimum solution for the riser arrangement investigated. Results indicate that the appropriate placement of risers in DDF can help to reduce wrinkling defects, which can be successfully optimised using the developed two-step simulation-based method.

6 Acknowledgements

The work presented in this paper was completed as part of the "Thermoplastic Overmoulding for Structural Composite Automotive Applications" (TOSCAA) project. The authors gratefully acknowledge the financial support of Innovate UK (Project Ref. 102663).

7 References

- [1] Chen, S., Fabric forming simulation and process optimisation for composites. 2016, University of Nottingham.
- [2] Chen, S., McGregor, O., Endruweit, A., Elsmore, M., De Focatiis, D., Harper, L., and Warrior, N., *Double diaphragm forming simulation for complex composite structures.* Composites Part A: Applied Science and Manufacturing, 2017. **95**: p. 346-358.
- [3] Pantelakis, S.G., Katsiropoulos, C.V., Labeas, G., and Sibois, H., A concept to optimize quality and cost in thermoplastic composite components applied to the production of helicopter canopies. Composites Part A: Applied Science and Manufacturing, 2009. **40**(5): p. 595-606.
- [4] Pantelakis, S.G. and Baxevani, E.A., *Optimization of the diaphragm forming process with regard to product quality and cost.* Composites Part A: Applied Science and Manufacturing, 2002. **33**(4): p. 459-470.
- [5] Alshahrani, H. and Hojjati, M., Experimental and numerical investigations on formability of out-of-autoclave thermoset prepreg using a double diaphragm process. Composites Part A: Applied Science and Manufacturing, 2017.
- [6] Hallander, P., Akermo, M., Mattei, C., Petersson, M., and Nyman, T., *An experimental study of mechanisms behind wrinkle development during forming of composite laminates.* Composites Part A: Applied Science and Manufacturing, 2013. **50**: p. 54-64.
- [7] Hallander, P., Sjölander, J., and Åkermo, M., Forming induced wrinkling of composite laminates with mixed ply material properties; an experimental study. Composites Part A: Applied Science and Manufacturing, 2015. **78**: p. 234-245.
- [8] Sjölander, J., Hallander, P., and Åkermo, M., Forming induced wrinkling of composite laminates: A numerical study on wrinkling mechanisms. Composites Part A: Applied Science and Manufacturing, 2016. **81**: p. 41-51.
- [9] Krebs, J., Bhattacharyya, D., and Friedrich, K., *Production and evaluation of secondary composite aircraft components—a comprehensive case study.* Composites Part A: Applied Science and Manufacturing, 1997. **28**(5): p. 481-489.
- [10] Haanappel, S., Ten Thije, R., Sachs, U., Rietman, B., and Akkerman, R., Formability analyses of uni-directional and textile reinforced thermoplastics. Composites Part A: Applied science and manufacturing, 2014. **56**: p. 80-92.
- [11] Kärger, L., Galkin, S., Zimmerling, C., Dörr, D., Linden, J., Oeckerath, A., and Wolf, K., Forming optimisation embedded in a CAE chain to assess and enhance the structural performance of composite components. Composite Structures, 2018. **192**: p. 143-152.
- [12] Harrison, P., Gomes, R., and Curado-Correia, N., *Press forming a 0/90 cross-ply advanced thermoplastic composite using the double-dome benchmark geometry.* Composites Part A: Applied Science and Manufacturing, 2013. **54**: p. 56-69.
- [13] Krebs, J., Friedrich, K., and Bhattacharyya, D., *A direct comparison of matched-die versus diaphragm forming*. Composites Part A: Applied Science and Manufacturing, 1998. **29**(1–2): p. 183-188.
- [14] Chen, S., Harper, L., Endruweit, A., and Warrior, N., Formability optimisation of fabric preforms by controlling material draw-in through in-plane constraints. Composites Part A: Applied Science and Manufacturing, 2015. **76**: p. 10-19.

- [15] Chen, S., Harper, L.T., Endruweit, A., and Warrior, N.A., *Optimisation of forming process for highly drapeable fabrics*, in *20th International Conference on Composite Materials*. 2015: Copenhagen Denmark.
- [16] McGregor, O.P.L., Chen, S., Haper, L.T., Endruweit, A., and Warrior, N.A., Defect reduction in the double diaphragm forming process, in 21st International Conference on Composite Materials. 2017: Xi'an China.
- [17] Creech, G. and Pickett, A., *Meso-modelling of non-crimp fabric composites for coupled drape and failure analysis.* Journal of materials science, 2006. **41**(20): p. 6725-6736.
- [18] Chen, S., McGregor, O., Harper, L., Endruweit, A., and Warrior, N., *Defect formation during preforming of a bi-axial non-crimp fabric with a pillar stitch pattern.* Composites Part A: Applied Science and Manufacturing, 2016. **91**: p. 156-167.
- [19] Dumont, F., Weimer, C., Soulat, D., Launay, J., Chatel, S., and Maison-Le-Poec, S., *Composites preforms simulations for helicopters parts*. International Journal of Material Forming, 2008. **1**(1): p. 847-850.
- [20] Lomov, S.V., Barburski, M., Stoilova, T., Verpoest, I., Akkerman, R., Loendersloot, R., and Thije, R.H.W.t., *Carbon composites based on multiaxial multiply stitched preforms. Part 3: Biaxial tension, picture frame and compression tests of the preforms.* Composites Part A: Applied Science and Manufacturing, 2005. **36**(9): p. 1188-1206.
- [21] Lee, J.S., Hong, S.J., Yu, W.-R., and Kang, T.J., *The effect of blank holder force on the stamp forming behavior of non-crimp fabric with a chain stitch.* Composites Science and Technology, 2007. **67**(3–4): p. 357-366.
- [22] Ogden, R. Large deformation isotropic elasticity-on the correlation of theory and experiment for incompressible rubberlike solids. in Proceedings of the Royal Society of London A: Mathematical, Physical and Engineering Sciences. 1972. The Royal Society.
- [23] Twizell, E. and Ogden, R., Non-linear optimization of the material constants in Ogden's stress-deformation function for incompressinle isotropic elastic materials. The Journal of the Australian Mathematical Society. Series B. Applied Mathematics, 1983. **24**(04): p. 424-434.
- [24] Sachs, U., Akkerman, R., Fetfatsidis, K., Vidal-Sallé, E., Schumacher, J., Ziegmann, G., Allaoui, S., Hivet, G., Maron, B., and Vanclooster, K., *Characterization of the dynamic friction of woven fabrics: experimental methods and benchmark results.* Composites Part A: Applied Science and Manufacturing, 2014. **67**: p. 289-298.
- [25] Avgoulas, E., Mulvihill, D., Endruweit, A., Sutcliffe, M., Warrior, N., De Focatiis, D., and Long, A., *Frictional behaviour of non-crimp fabrics (NCFs) in contact with a forming tool.* Tribology International.
- [26] Chen, S., Endruweit, A., Harper, L., and Warrior, N., *Inter-ply stitching optimisation of highly drapeable multi-ply preforms*. Composites Part A: Applied Science and Manufacturing, 2015. **71**: p. 144-156.
- [27] Ehret, A.E., *On a molecular statistical basis for Ogden's model of rubber elasticity.* Journal of the Mechanics and Physics of Solids, 2015. **78**: p. 249-268.

8 Tables

Table 1: Material properties of FCIM359 used in non-orthogonal material model.

Fabric thickness	0.4 mm		
Effective density	1200 kg/m^3		
Effective modulus	138 GPa	Q-	
Wrinkling onset shear angle	Positive shear	43°	
	Negative shear	50°	
Normalised fabric shear curve	$F_{norm} = F_{norm}^{yarn\ rotation} + F_{norm}^{stitch}$ (expressed as a function of shear angle γ_{12} in radians)		
$F_{norm}^{yarn\ rotation}$	$\left(29.56\gamma_{12}^{5}-65.56\gamma_{12}^{4}+137.06\gamma_{12}^{3}+94.73\gamma_{12}^{2}+112.19\gamma_{12}\right)N/m$		
Fstitch Fnorm	$\begin{cases} (2000\gamma_{12} - 120) N/m &, & 0.06 \le \gamma_{12} < 0.50; \\ (-3520\gamma_{12} + 2640) N/m &, & 0.50 \le \gamma_{12} \le 0.75; \\ 0 N/m &, & else. \end{cases}$		

Table 2: Material properties of the silicone diaphragm material.

Thickness	1.56 mm			
Density	$1.60 \times 10^3 \mathrm{kg/m^3}$			
Constants for Ogden	μ_1	α_1	μ_2	α_2
hyperelastic model	0.15 MPa	3.09	0.81 MPa	0.18

9 Figures

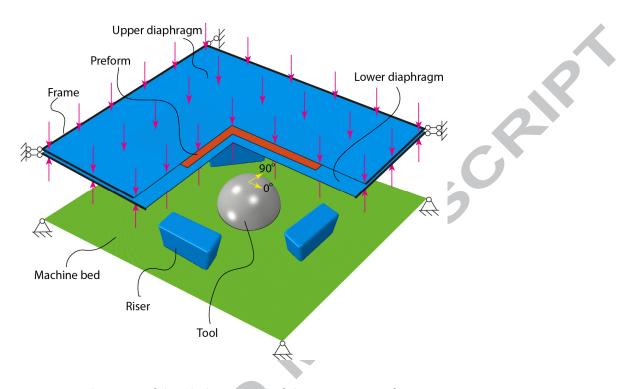


Figure 1: Schematic of detailed FE model of the DDF process for riser arrangement optimisation.

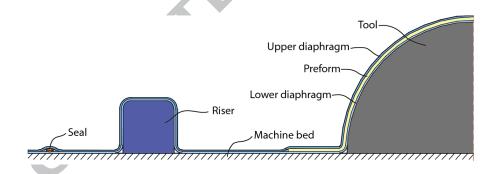


Figure 2: Cross section indicating location of riser in DDF assembly, increasing local in-plane constraining force.

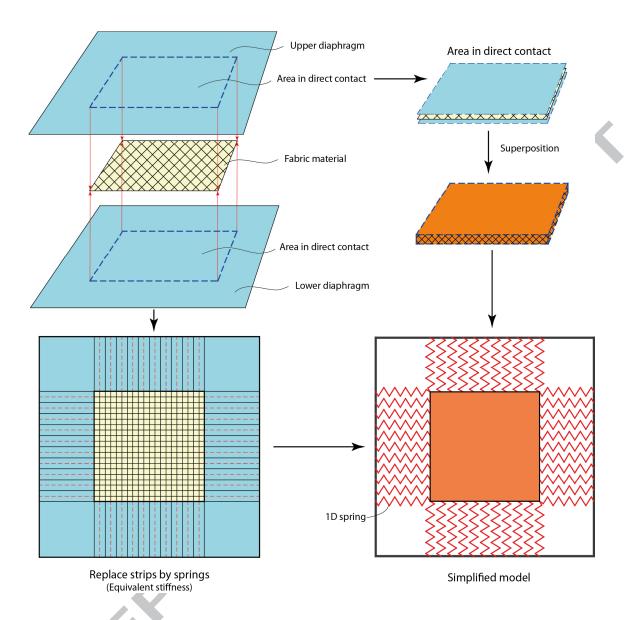
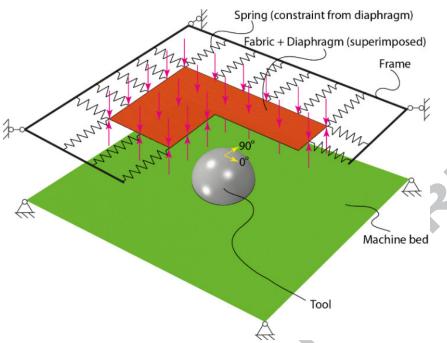


Figure 3: Procedures to simplify the DDF model for riser position optimisation in Step I.



(a) Schematic of simplified FE model for Step I of optimisation proceedure

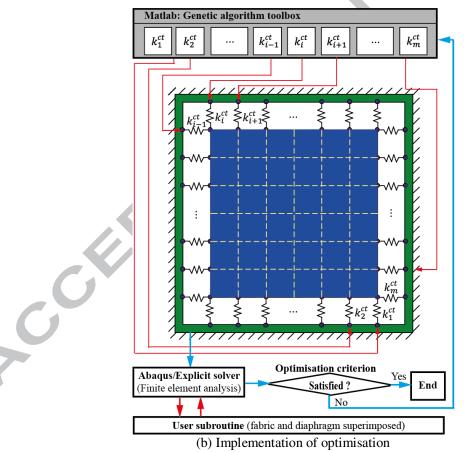


Figure 4: Flow chart for Step I: optimisation of riser position.

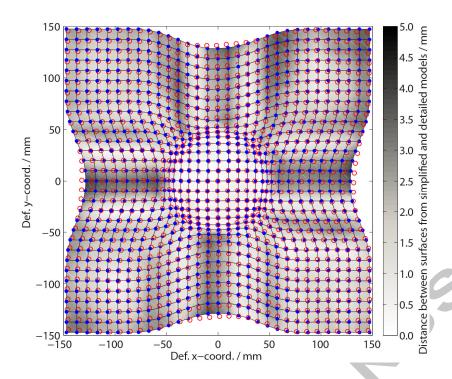


Figure 5: Comparison of grids obtained from simplified (red circles) and detailed (blue solid dots) FE models. The contour shows the distance between the two grids, which represents the error between the two modelling approaches, presented as a grey-scale value.

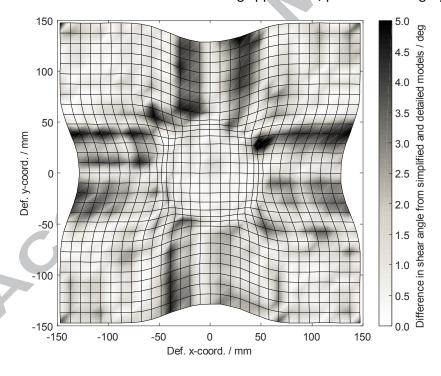


Figure 6: The difference in shear angle obtained from the simplified model against the one obtained from the detailed model. The contour shows the difference in shear angle, presented as a grey-scale value.

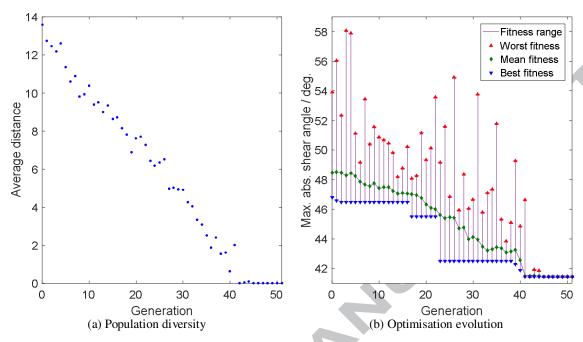


Figure 7: GA stability and convergence analyses for riser position optimisation in Step I.

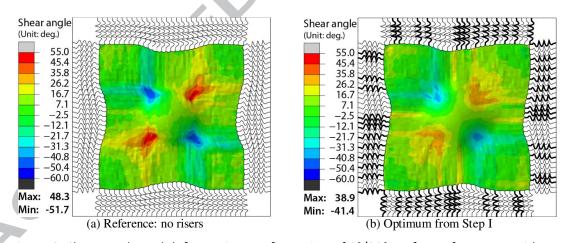


Figure 8: Shear angle and deformation configuration of 0°/90° preforms from DDF with optimised riser arrangement against reference case with no risers (using a simplified FE model). Springs represented by thick lines indicate that a riser is required in that corresponding location, while springs represented by thin lines indicate that no riser is required.

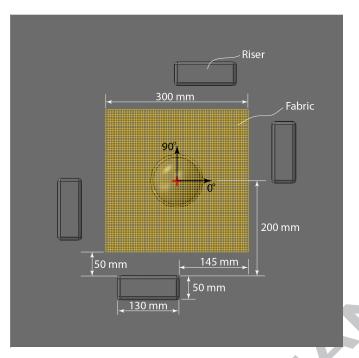


Figure 9: Optimum riser arrangement determined from Step I for 0°/90° preforms by DDF.

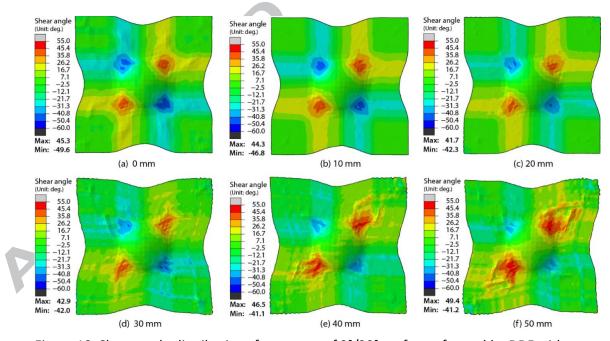


Figure 10: Shear angle distributions for a range of $0^{\circ}/90^{\circ}$ preforms formed by DDF with different riser heights for the optimised pattern.

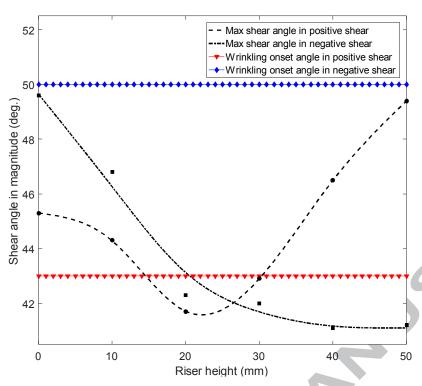


Figure 11: Evolution of maximum shear angle in both positive and negative shear regions for $0^{\circ}/90^{\circ}$ preforms when placing risers at different heights at the optimised positions.

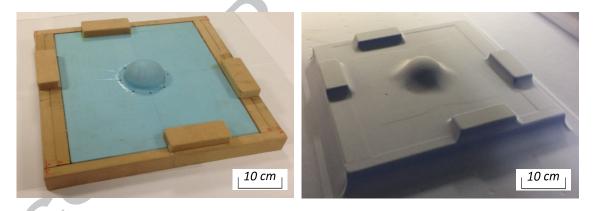


Figure 12: Application of risers to reduce wrinkling defects using a laboratory scale diaphragm forming machine installed at the University of Nottingham.

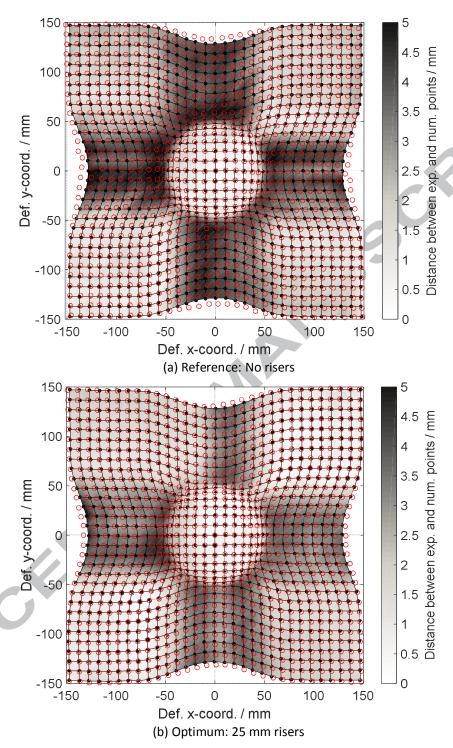


Figure 13: Comparison of experimental (red circles) and numerical (black solid dots) grids. The contour shows the distance between the grid intersections obtained from experiment and simulation using the detailed FE model, presented as a grey-scale value.



Figure 14: Validation of wrinkling defect removal in DDF using (a) reference case without risers and (b) optimum riser arrangement (Original grid spacing = 10mm).