

1                   **Improvement of k-epsilon Turbulence Model for CFD Simulation of Atmospheric**  
2                   **Boundary Layer around a High-rise Building Using Stochastic Optimization and Monte**  
3                   **Carlo Sampling Technique**  
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6                   **Mohammadreza Shirzadi<sup>a</sup>, Parham A. Mirzaei<sup>b,1</sup>, Mohammad Naghashzadegan<sup>a</sup>**

7                                   <sup>a</sup>Engineering Department, University of Guilan, Rasht, Iran

8                                   <sup>b</sup>Architecture and Built Environment Department, University of Nottingham, Nottingham, UK  
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10                   **Abstract**

12                   The accuracy of the computational fluid dynamics (CFD) to model the airflow around the  
13                   buildings in the atmospheric boundary layer (ABL) is directly linked to the utilized turbulence model.  
14                   Despite the popularity and their low computational cost, the current Reynolds Averaged Navier-Stokes  
15                   (RANS) models cannot accurately resolve the wake regions behind the buildings. The default values  
16                   of the RANS models' closure coefficients in CFD tools such as ANSYS CFX, ANSYS FLUENT,  
17                   PHOENIX, and STAR CCM+ are mainly adapted from other fields and physical problems, which are  
18                   not perfectly suitable for ABL flow modeling. This study embarks on proposing a systematic approach  
19                   to find the optimum values for the closure coefficients of RANS models in order to significantly  
20                   improve the accuracy of CFD simulations for urban studies. The methodology is based on stochastic  
21                   optimization and Monte Carlo Sampling technique. To show the capability of the method, a test case  
22                   of airflow around an isolated building placed in a non-isothermal unstable ABL was considered. The  
23                   recommended values for this case study in accordance with the optimization method were thus found  
24                   to be  $1.45 \leq C_{\varepsilon 1} \leq 1.5$ , of  $2.7 \leq C_{\varepsilon 2} \leq 3$ , and  $0.12 \leq C_{\mu} \leq 0.15$ . The default value of  $\sigma_k = 1$  is  
25                   suggested to be acceptable while the value of  $\sigma_{\varepsilon}$  is obtained through a correlation. The error of the  
26                   estimated reattachment length behind the building decreased from 170% for the default values to 28%  
27                   for the modified values.

28                   **Keywords:** CFD, Turbulence, Optimization, Microclimate, Monte Carlo Sampling, Atmospheric  
29                   Boundary Layer  
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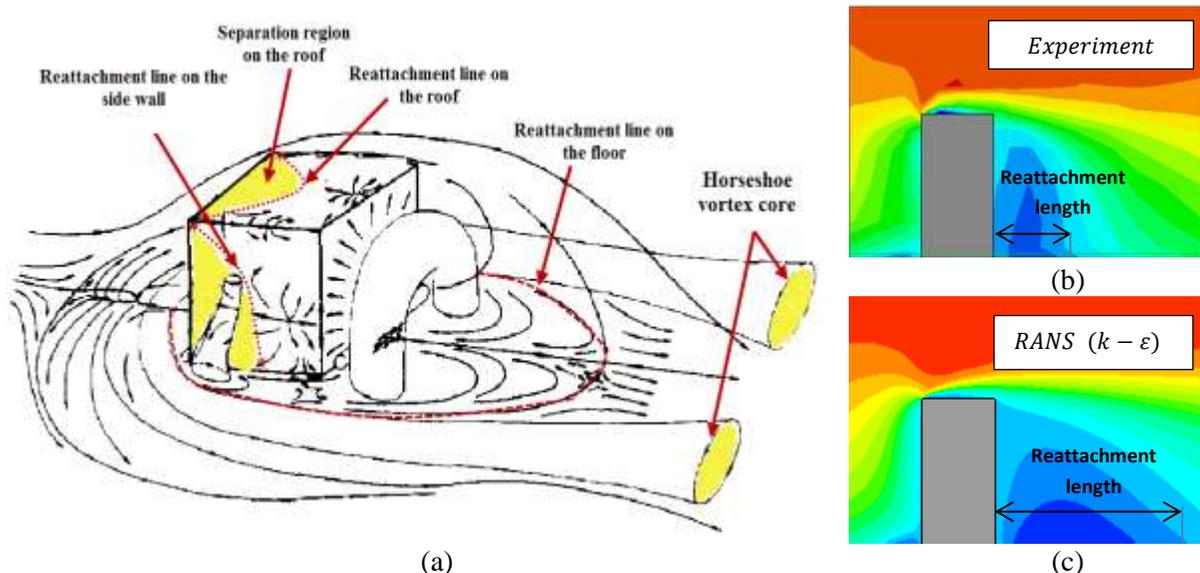
<sup>1</sup> Corresponding author: University Park, Nottingham, NG2RD, UK  
Tel.: +44 0115 95 14129; fax: +44 0115 951 3159  
Email: [Parham.Mirzaei\\_Ahranjani@nottingham.ac.uk](mailto:Parham.Mirzaei_Ahranjani@nottingham.ac.uk)

$\rho$	Density	$U_H$	Inflow mean streamwise velocity at building height $H$
$t$	Time	$H$	Building height
$x_i$	Component of space coordinate	$\alpha$	Power-law exponent
$U_i$	Component of the mean velocity vector	$q$	Hit rate
$\tau_{ij}$	Viscous stress tensor	$N$	Number of data points (48)
$S_{M_i}$	Body forces	$O_i$	Observed value
$\mu_t$	Turbulent viscosity	$P_i$	Predicted value
$\delta_{ij}$	Kronecker Delta function	$FAC2$	The fraction of the predictions within a factor of 2 of the observations
$k$	Turbulent kinetic energy	$X_f$	Reattachment length behind the building
$g_i$	Gravity vector	$X_r$	Reattachment length on the roof
$C_\mu$	$k - \varepsilon$ model constant	$u_i$	Fluctuating velocity component in the turbulent flow
$\mu$	Molecular viscosity	$\sigma_k$	$k - \varepsilon$ model constant
$\varepsilon$	Turbulent dissipation rate	$\sigma_\varepsilon$	$k - \varepsilon$ model constant
$P_k$	Shear production term in $k$ -equation	$C_{\varepsilon 2}$	$k - \varepsilon$ model constant
$P_{kb}$	Buoyant production term in $k$ -equation	$\theta_H$	Temperature at building height (11°C)
$C_{\varepsilon 1}$	$k - \varepsilon$ model constant	$\Delta\theta$	$\theta_f - \theta_H$
$\theta_f$	Floor temperature (45 °C)		

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38 **1. Introduction and literature review**

39 Airflow modeling in built environment has a significant potential to help urban planners,  
40 architects and engineers in the design stages of buildings and cities (Capeluto et al, 2003; Murakami,  
41 2006; Wong et al, 2011). In particular, an accurate modeling can bring about desired outcomes such as  
42 the improvement of the pedestrian-level wind comfort (Haghighat and Mirzaei, 2011; Mirzaei and  
43 Haghighat, 2012; Richards et al, 2002; Tsang et al, 2012), reduction of the pollution dispersion  
44 (Mirzaei and Haghighat, 2010, 2011; Yamada, 2004), minimizing the building energy consumption  
45 (Allegrini et al, 2015; Evins et al, 2014; Yi and Feng, 2013), utilizing wind energy for modern  
46 applications (Mirzaei and Rad, 2013), and mitigation of the urban heat island (Magli et al, 2015;  
47 Mirzaei, 2015). Among different techniques for analyzing airflow in outdoor climates such as wind  
48 tunnel experiments and on-site measurements, Computational Fluid Dynamics (CFD) emerged as a  
49 reliable and cost effective method to simulate the wind condition around buildings. Atmospheric  
50 boundary layer airflow around the buildings, as displayed in Fig.1(a), includes complex phenomena,  
51 such as separation, reattachment, large-scale turbulence and unsteady vortex shedding (Rodi, 1997);  
52 hence turbulence modeling has a significant impact on the accuracy of the CFD models. Despite many  
53 years of researches, CFD modeling of turbulent flow around buildings still remains a challenging issue  
54 (Lateb et al, 2016). Even for a simple cubic form of an isolated building, there is a noticeable  
55 disagreement between the experimental results and CFD predictions (see Fig.1(b) and Fig.1(c)).



56 Figure 1 (a) Flow visualization around an isolated building (Hunt et al, 1978). Streamwise velocity  
 57 distribution around an isolated building for (b) experiment by Yoshie et al (2011), (c) RANS turbulence mode

58 Early works presented in (Lakehal and Rodi, 1997; Murakami, 1993; Murakami et al, 1990;  
 59 Tamura et al, 1997) examined different turbulence models to predict the airflow around a generic bluff  
 60 body via focusing on the pressure distribution and separation of flow over the roof. In an attempt to  
 61 investigate the problem of the airflow modeling in urban areas, a working group for CFD modeling of  
 62 the wind environment around a building was organized by the Architectural Institute of Japan  
 63 (Shirasawa et al, 2003). Tominaga et al (2004) presented the result of a cross comparison of the  
 64 airflow around a single high-rise building in the lower part of the atmospheric boundary layer (ABL).  
 65 Also, they performed numerical simulation of a building complex in an actual urban region. Different  
 66 software and turbulence models were examined in their study for two test cases of 2:1:1 and 4:4:1  
 67 shaped building models based on the experiments from Yan and Kazuki (1998). Their results showed  
 68 that the standard  $k - \epsilon$  model mainly fails to produce the reverse flow over the roof, but revised  
 69 models (e.g. LK  $k - \epsilon$  (Kato, 1993), RNG  $k - \epsilon$  (Yakhot and Orszag, 1986), MMK  $k - \epsilon$  (Tsuchiya et  
 70 al, 1997) could more accurately predict the flow pattern. However, the standard  $k - \epsilon$  model and all  
 71 revised models overestimated the reattachment length behind the building.

72 A similar finding was presented by Yoshie et al (2007), Tominaga and Stathopoulos (2010),  
 73 Vardoulakis et al (2011), and Gousseau et al (2011) which emphasized the inaccuracy of the Reynolds  
 74 Averaged Navier Stokes (RANS) turbulence models in reproducing the weak wind regions behind  
 75 buildings and also in overestimating the reattachment length behind the building. In another study by  
 76 Köse and Dick (2010), it was shown that the poor accuracy of the RANS turbulence models for  
 77 prediction of the airflow around the buildings in ABL is accompanied with a low accuracy in  
 78 estimating the mean surface pressure over the building in comparison with LES models.

79 In a recent study by Tominaga (2015), the accuracy of the unsteady Reynolds-averaged Navier-  
 80 Stokes (URANS) turbulence modeling for an isolated building was investigated. He concluded that the  
 81 URANS simulation based on the  $k - \omega$  SST turbulence model is able to simulate the unsteady  
 82 fluctuations behind the building and providing a better velocity field in this region as well; however,  
 83 the model generally overestimates the separation in the corners.

84 The previous literature clearly demonstrates that the linear two-equation RANS turbulence models  
85 provide poor results for the airflow prediction around an isolated building, compared with the URANS  
86 and LES models. However, high complexity of the URANS and LES models in specifying accurate  
87 boundary condition, proper mesh size and time scale, in addition to their inherent high computational  
88 cost keep their potential application as a reliable and fast solution for many realistic engineering  
89 problems very limited. Despite the development of several methods for improving the RANS  
90 turbulence models, e.g. *RNG k - ε* (Yakhot et al, 1992) and Realizable *k - ε* (Shih et al, 1995), their  
91 application for the airflow modeling around buildings in ABL is limited due to their poor accuracy in  
92 resolving the flow in the weak wind regions.

93 Moreover, another limitation of the current RANS family models refers to their semi-empirical  
94 coefficients, which are mainly adapted from the fundamental and classical flow problems, e.g.  
95 homogenous decaying turbulence, free shear flow, and fully developed channel flow. The value of  
96 these coefficients collected in the work carried out by Launder and Spalding (1974) are shown in  
97 [Table 1](#). These values are used in most CFD tools such as ANSYS CFX, ANSYS FLUENT,  
98 PHOENIX, and STAR CCM+ as default parameters. However, experimental measurements performed  
99 in different studies show a slight difference in values for these coefficients. For instance, Mohamed  
100 and LaRue (1990) suggested a value of  $C_{\varepsilon 2} = 1.77$  which is lower than the default value of 1.92.  
101 Experimental and numerical analyses by Kim et al (1987) demonstrate that the variation of  $C_{\mu}$  for a  
102 channel flow in areas far from the wall ( $y^+ > 50$ ) is between 0.06 to 0.095, resulting in an average  
103 value of  $C_{\mu} = 0.09$ . The value of  $C_{\mu}$  for a temporal-mixing layer was reported between 0.07 and 0.11  
104 (Pope, 2001). In an experimental work by Tavoularis and Karnik (1989), different values for the ratio  
105  $\frac{C_{\varepsilon 2}-1}{C_{\varepsilon 1}-1}$ , ranging from 1.33 to 1.75, were observed for different shear flows. Once default values of  $C_{\varepsilon 1}$   
106 and  $C_{\varepsilon 2}$  are used, the ratio gets 2.09, which is noticeably different from the reported experimental  
107 values (Edeling et al, 2014a). All these studies imply that there is a noticeable uncertainty in these  
108 coefficients and as demonstrated in (Edeling et al, 2014b), best flow-independent values for these  
109 coefficients are unlikely to exist. As described in (Pope, 2001), the default values of the closure  
110 coefficients in the standard *k - ε* model are obtained from a compromise so as to enable the model to  
111 perform for a variety of the airflow problems.

112 In [Table 1](#), a number of studies associated with the effect of the closure coefficients for different  
113 physical problems are summarized. In an early work conducted by Duynkerke (1988), a set of  
114 modified closure coefficients for the standard *k - ε* model was suggested based on a comparison  
115 between the RANS model and a measurement study and LES model over a flat terrain for neutral and  
116 stable atmospheric boundary layer conditions. He used Panofsky and Dutton (1984) data and  
117 calculated  $C_{\mu} = 0.033$ , which is lower than its default value of 0.09. He also proposed values of  
118  $C_{\varepsilon 1} = 1.46$  and  $C_{\varepsilon 2} = 1.85$ , which are close to their default values of 1.44 and 1.92, respectively. For  
119 Von Karman constant equal to 0.4, he has also obtained  $\sigma_{\varepsilon} = 2.38$ , which is greater than its default  
120 value of 1.3 used in most of the CFD solvers. For  $\sigma_k$ , the default value of 1 was assumed. In a similar  
121 work by Detering and Etling (1985), a modification on the  $\varepsilon$  equation constants of the *k - ε* model  
122 was adapted for mesoscale atmospheric boundary layer modeling above a flat and complex terrain.

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Table 1 The value of closure coefficients for different flow problems

Ref	Physical model	Closure coefficients
Launder and Spalding (1974)	Free turbulent flows	$C_{\varepsilon 1} = 1.44, C_{\varepsilon 2} = 1.92, C_{\mu} = 0.09,$ $\sigma_{\varepsilon} = 1.3, \sigma_k = 1$
Mohamed and LaRue (1990)	Grid-generated turbulence	$C_{\varepsilon 2} = 1.77$
Kim et al (1987)	Fully developed channel flow	$0.06 \leq C_{\mu} \leq 0.095$
Pope (2001)	Fully developed channel flow in log-law region	$\sigma_{\varepsilon} = \frac{\kappa^2}{C_{\mu}^{1/2}(C_{\varepsilon 2} - C_{\varepsilon 1})}$
Duynkerke (1988)	Neutral and Stable ABL	$C_{\varepsilon 1} = 1.46, C_{\varepsilon 2} = 1.83, C_{\mu} = 0.033,$ $\sigma_{\varepsilon} = 2.38, \sigma_k = 1$
Detering and Etling (1985)	Neutral and Stable ABL	$C_{\varepsilon 1} = 1.13, C_{\varepsilon 2} = 1.9, \sigma_{\varepsilon} = 1.29, \sigma_k = 0.74$
Glover et al (2011)	Idealized street canyon	$C_{\varepsilon 1} = 1, C_{\varepsilon 2} = 2.2, C_{\mu} = 0.12, \sigma_{\varepsilon} = 0.42, \sigma_k = 0.462$
Edeling et al (2014b)	Wall-bounded flow with different favorite and adverse pressure gradient	Case dependent
Guillas et al (2014)	Idealized street canyon	$C_{\varepsilon 1} = 1, C_{\varepsilon 2} = 2.2, C_{\mu} = 0.12, \sigma_{\varepsilon} = 0.42, \sigma_k = 0.462$
Zahid Iqbal and Chan (2016)	High-raised cross-shaped buildings	$C_{\varepsilon 1} = 1, C_{\varepsilon 2} = 1.92, C_{\mu} = 0.12,$ $\sigma_{\varepsilon} = 0.5, \sigma_k = 0.53$

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126 Due to the inherent uncertainty in the value of the closure coefficients (Mohamed and LaRue,  
127 1990; Tavoularis and Karnik, 1989), some studies, therefore, considered these coefficients as uncertain  
128 variables; investigated the sensitivity of the RANS model outputs to the variability of the closure  
129 coefficients. For example, Dunn et al (2011), Glover et al (2011), Todd and Robert (2011), Cheung et  
130 al (2011), Edeling et al (2014b) and Guillas et al (2014) investigated the uncertainty in relation to the  
131 closure coefficients of the  $\varepsilon$  equation, and discussed the applicability of statistical analysis for  
132 improving the accuracy of RANS models. Dunn et al (2011) studied the uncertainty in relation to the  
133  $k - \varepsilon$  coefficients using the Latin Hypercube Sampling (LHS) method through considering different  
134 forms of probability density function (PDF) for the closure coefficients. They demonstrated that the  
135 highest uncertainty of the flow parameters occurs in the recirculating region and near the reattachment  
136 point. Furthermore, a Bayesian calibration approach was introduced in (Cheung et al, 2011) in which  
137 coefficients of Spalart–Allmaras model (Spalart and Allmaras, 1992) were calibrated for a set of  
138 incompressible CFD models over a flat plate. Experimental data for the velocity profile and wall shear  
139 stress were used in the calibration process. In a similar work, Edeling et al (2014b) performed 13  
140 separate Bayesian calibrations using the experimental velocity profile for 13 different pressure  
141 gradients. They used a two-dimensional compressible boundary layer program instead of a full RANS  
142 code in order to reduce the runtime and avoid surrogate model. Their results showed a noticeable  
143 variation of coefficient posteriors for the considered range of the flow for  $C_{\varepsilon 2}$  and  $C_{\mu}$ .

144 In another work performed by Guillas et al (2014), a Bayesian calibration of the  $k - \varepsilon$  closure  
145 coefficients for a flow in a street canyon was presented. They calibrated a CFD RANS model against a  
146 series of wind tunnel experiments (Kastner-Klein et al, 2001), and considered the turbulent kinetic

147 energy distribution between regular street canyons as the quantity of interest. Uniform priors for  
148 closure coefficients, including  $C_\mu, C_{\varepsilon 1}, C_{\varepsilon 2}$ , and  $\sigma_k$ , were considered in their method and it was  
149 concluded that the  $C_\mu$  values higher than 0.12 have the highest probability to better match the  
150 experimental data. For  $\sigma_k$ , values close to 0.5 were reported to be favorable. [Solazzo \(2008\)](#) reported a  
151 similar trend and showed that the lower values for  $\sigma_k$  and  $\sigma_\varepsilon$  than their default values can result in a  
152 better distribution of  $k$  inside the street canyon, and thus improve the accuracy of the  $k - \varepsilon$  model for  
153 such applications. In a recently published work by [Zahid Iqbal and Chan \(2016\)](#), a numerical and  
154 experimental analysis for the pedestrian wind environment around a group of high-rise cross-shaped  
155 buildings was presented. They used the closure coefficients proposed by [Guillas et al \(2014\)](#) and  
156 performed two experimental test cases to modify these coefficients. Their modified values (see [Table](#)  
157 [1](#)) showed a better agreement with the experimental results relative to the default values for the  
158 standard  $k - \varepsilon$  model.

159 This article aims to propose a systematic way to find the optimized values for the closure  
160 coefficients of RANS family turbulence models to improve the accuracy of CFD simulations for  
161 microclimate and urban studies. The methodology is based on a stochastic optimization approach and  
162 the Monte Carlo Sampling (MCS) technique, which is later applied to a case study to demonstrate the  
163 capability of the developed approach. Although the stochastic optimization and MCS method have  
164 been widely used for the reliability-based design and robust optimization of complex systems ([Shah et](#)  
165 [al, 2015](#); [Tang and Périaux, 2012](#)), their application for calibration of the closure coefficients for ABL  
166 flow modeling is a novel approach. The proposed method in this study requires fewer samples (CFD  
167 simulations) than the previous calibration methods based on the Bayesian approaches. The case study  
168 considered in this article is the airflow around a high-rise building in a non-isothermal ABL in which  
169 optimized closure coefficients for the  $k - \varepsilon$  model were investigated. A constant value for the  
170 turbulent Prandtl number was taken into consideration during the optimization. The experimental data  
171 of the airflow behind a high-rise building in an unstable non-isothermal turbulent flow by [Yoshie et al](#)  
172 [\(2011\)](#) were used in the calibration process to define various validation metrics. Using the MCS  
173 technique and stochastic optimization, a set of new closure coefficients will be obtained and  
174 accordingly they can improve the accuracy of the turbulence model. Numerical data for velocity in the  
175 wake region behind the building will be considered as the objectives of the optimization technique.

## 176 **2. Methodology**

177 The main objective of this study is to propose a systematic way to improve the accuracy of the  
178 RANS models in microclimate studies; it is achieved through modifying the closure coefficients of  
179 turbulence models using a stochastic optimization approach. To this end, a parametric sensitivity  
180 analysis will be performed at the first step to investigate the impact of the model coefficients on the  
181 accuracy of the CFD model. In the next step, the model coefficients will be inserted into an  
182 optimization module as a set of uncertain variables, and eventually, the best range of the coefficients  
183 will be calculated so accurately that the highest agreement between the experiment and CFD results  
184 can be achieved.

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187 **2.1 Optimization procedure**

188 Stochastic optimization approaches can be used in models in which exact data are unknown, but  
 189 bounded by a set of realization or scenarios (Goerigk and Schöbel, 2016). This is the case in RANS  
 190 turbulence models where the numerical values of the closure coefficients are chosen through  
 191 combination of heuristic and empirical decision making (Schaefer et al, 2016). Thus, RANS  
 192 coefficients can be considered as epistemic uncertainty variables with a uniform probability density  
 193 function (PDF) to provide an equal probability for all the values in the interval to be an optimum  
 194 candidate (Guillas et al, 2014). The concept of stochastic optimization used in this study, known as a  
 195 robust optimization method, is described in (Van der Velden and Koch, 2010).

196 The brief description of the formulation of stochastic optimization can be mathematically stated as  
 197 finding a set of design variables  $X$  that (Koch et al, 2004):

$$\begin{aligned} \text{Minimize:} & \quad f(\mu_y(X), \sigma_y(X)) \\ \text{Subject to:} & \quad g_i(\mu_y(X), \sigma_y(X)) \leq 0 \\ & \quad X_L \leq X \leq X_U \end{aligned} \tag{1}$$

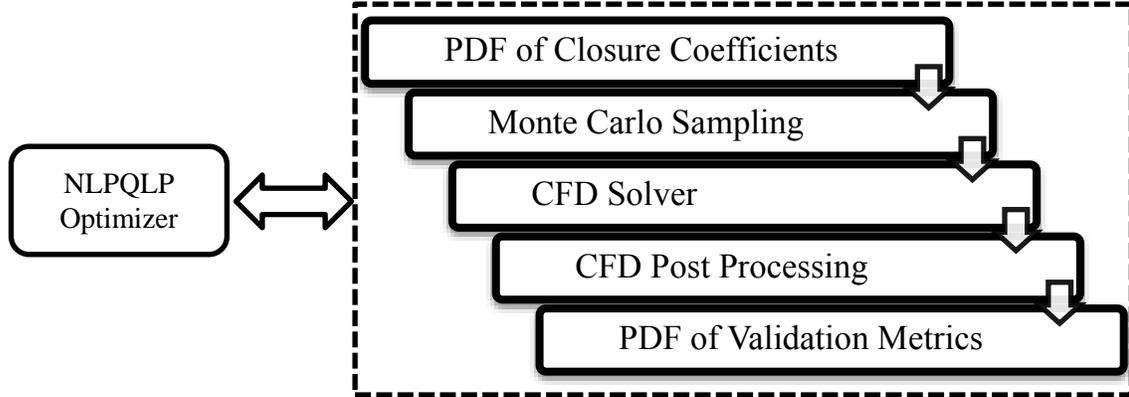
198 where  $X_L$  and  $X_U$  are the lower and upper limits for input parameter  $X$ . In this formulation, the output  
 199 constraint  $g_i$  is expressed in terms of mean value and standard deviation. A weighted sum approach  
 200 was used to define the objective function, which includes a term for mean value variation relative to  
 201 the target and a term to minimize the response variation (Koch et al, 2004):

$$F = \sum_{i=1}^l \left[ \frac{w_{1i}}{s_{1i}} (\mu_{y_i} - M_i)^2 + \frac{w_{2i}}{s_{2i}} \sigma_{y_i}^2 \right] \tag{2}$$

202 where  $w_{1i}$  and  $w_{2i}$  are the weighting factors, and  $s_{1i}$  and  $s_{2i}$  are the scale factors related to each term.  
 203 The weighting factors determine the importance of each objective while the scaling factors are used to  
 204 normalize the objectives.  $M_i$  stands for the target of the output response  $i$  and  $l$  is the total number of  
 205 output responses. The statistical variability of output responses (i.e.  $\mu_{y_i}$  and  $\sigma_{y_i}$ ), which are required  
 206 by the stochastic optimization formulation, can be estimated using the Monte Carlo simulation (MCS)  
 207 technique.

208 In Fig. 2, a schematic of the optimization process for calibrating the closure coefficients is shown.  
 209 By coupling the Monte Carlo sampling technique and CFD model, input variables (closure  
 210 coefficients) randomly vary in accordance with their given PDFs. CFD model will be repeatedly run to  
 211 characterize the statistical parameters of the output values (i.e. validation metrics), including their  
 212 mean and standard deviation values. By integrating the Monte Carlo sampling into an optimizer, not  
 213 only can the best mean value of the desired outputs (validation metrics) be calculated, but it is also  
 214 possible to minimize the standard deviation of the output values so as to reduce the effects of  
 215 uncertainty of the input variables on the output response. Nonlinear Programming with Non-Monotone  
 216 and Distributed Line Search (NLPQLP) optimization method (Schittkowski, 2006), a well suited  
 217 method for highly non-linear design spaces, was used for the optimization purpose. A descriptive  
 218 sampling technique (Tari and Dahmani, 2006) was used for MCS, which is more efficient than the  
 219 conventional simple random sampling method (Koch et al, 2004). 50 samples were considered for the  
 220 MCS during each optimization iteration. Different objective functions can be defined for the

221 optimization process, including the means and standard deviation of validation metrics, which depend  
 222 on the availability of the experimental data for each specific case.



223 **Figure 2** Schematic of the stochastic optimization of the closure coefficients

224 **2.2 Mathematical modeling**

225 The 3D steady Reynolds averaged Navier-Stokes (RANS) equations were used to simulate the  
 226 airflow around the building. These equations can be derived by substituting mean and fluctuating  
 227 components of the airflow variables into the Navier-Stokes equations (CFX, 2011):

$$\frac{\partial(\rho U_j)}{\partial x_j} = 0 \tag{3}$$

$$\frac{\partial}{\partial x_j}(\rho U_i U_j) = -\frac{\partial P}{\partial x_i} + \frac{\partial}{\partial x_j}(\tau_{ij} - \rho \overline{u_i u_j}) + S_{M_i} \tag{4}$$

228 where  $U_i$  is the average velocity and  $u_i$  is the fluctuating velocity.  $\tau_{ij}$  is the viscous stress tensor  
 229 (including both normal and shear components of the stress tensor) and  $S_{M_i}$  is the sum of body forces.  
 230 The Boussinesq model was used in this study. Temperature field was also calculated by solving the  
 231 energy equation while eddy diffusivity was used to model turbulent energy fluxes (CFX, 2011):

$$\frac{\partial}{\partial x_j}(\rho U_j h_{total}) = \frac{\partial}{\partial x_j} \left( \lambda \frac{\partial T}{\partial x_j} + \frac{\mu_t}{Pr_t} \frac{\partial h}{\partial x_j} \right) + \frac{\partial}{\partial x_j} [U_i (\tau_{ij} - \rho \overline{u_i u_j})] + U_j S_{M_j} \tag{5}$$

232 where  $\lambda$  is the thermal conductivity of air and  $Pr_t$  is the turbulent Prandtl number, which has a  
 233 constant value of 0.9.  $U_j S_{M_j}$  represents the work due to the external momentum source.  $h_{total}$  is the  
 234 total enthalpy and is related to the static enthalpy ( $h$ ) by:

$$h_{total} = h + \frac{1}{2} U^2 \tag{6}$$

235 Air was considered to be incompressible, which is reasonable for atmospheric boundary layer  
 236 (ABL) flows (Richards and Norris, 2011); the air density, specific heat capacity at constant pressure,  
 237 and thermal expansion coefficient were considered to be  $1.185 \text{ kg/m}^3$ ,  $1004.4 \text{ J/kg K}$ , and  
 238  $0.003356 \text{ 1/K}$ . The temperature was calculated from the static enthalpy as follows:

$$h - h_{ref} = C_p(T_{static} - T_{ref}) \quad (7)$$

239 where  $T_{ref} = 25^\circ\text{C}$  is the reference temperature and  $h_{ref}$  is the reference enthalpy which is zero at the  
240 reference temperature.

241 In this study the  $k - \varepsilon$  turbulence model with the Kato-Launder modification (Kato and Launder,  
242 1993) was used, which is based on the eddy viscosity hypothesis in which Reynolds stresses can be  
243 related to the mean velocity gradients and eddy (turbulent) viscosity by the gradient diffusion  
244 hypothesis as follows:

$$-\rho \overline{U_i U_j} = \mu_t \left( \frac{\partial U_i}{\partial x_j} + \frac{\partial U_j}{\partial x_i} \right) - \frac{2}{3} \delta_{ij} \rho k \quad (8)$$

245 where  $\mu_t$  is the eddy viscosity or turbulent viscosity, which can be defined as below:

$$\mu_t = C_\mu \rho \frac{k^2}{\varepsilon} \quad (9)$$

246 For the  $k - \varepsilon$  model, values of  $k$  and  $\varepsilon$  come directly from their differential transport equations  
247 (Mori et al, 1995):

$$\frac{\partial \rho U_j k}{\partial x_j} = \frac{\partial}{\partial x_j} \left[ \left( \mu + \frac{\mu_t}{\sigma_k} \right) \frac{\partial k}{\partial x_j} \right] + P_k - \rho \varepsilon + P_{kb} \quad (10)$$

$$\frac{\partial \rho U_j \varepsilon}{\partial x_j} = \frac{\partial}{\partial x_j} \left[ \left( \mu + \frac{\mu_t}{\sigma_\varepsilon} \right) \frac{\partial \varepsilon}{\partial x_j} \right] + \frac{\varepsilon}{k} (C_{\varepsilon 1} P_k - C_{\varepsilon 2} \rho \varepsilon + C_{\varepsilon 1} P_{\varepsilon b}) \quad (11)$$

248 where  $P_k$  is the production of turbulence due to shear, which is modified by Kato and Launder (1993):

$$P_k = \rho C_\mu \varepsilon S \Omega \quad (12)$$

249 where  $S$  and  $\Omega$  are respectively the dimensionless strain and vorticity parameters, which are calculated  
250 as below:

$$S = \frac{k}{\varepsilon} \sqrt{\frac{1}{2} \left( \frac{\partial U_i}{\partial x_j} + \frac{\partial U_j}{\partial x_i} \right)^2} \quad (13)$$

$$\Omega = \frac{k}{\varepsilon} \sqrt{\frac{1}{2} \left( \frac{\partial U_i}{\partial x_j} - \frac{\partial U_j}{\partial x_i} \right)^2} \quad (14)$$

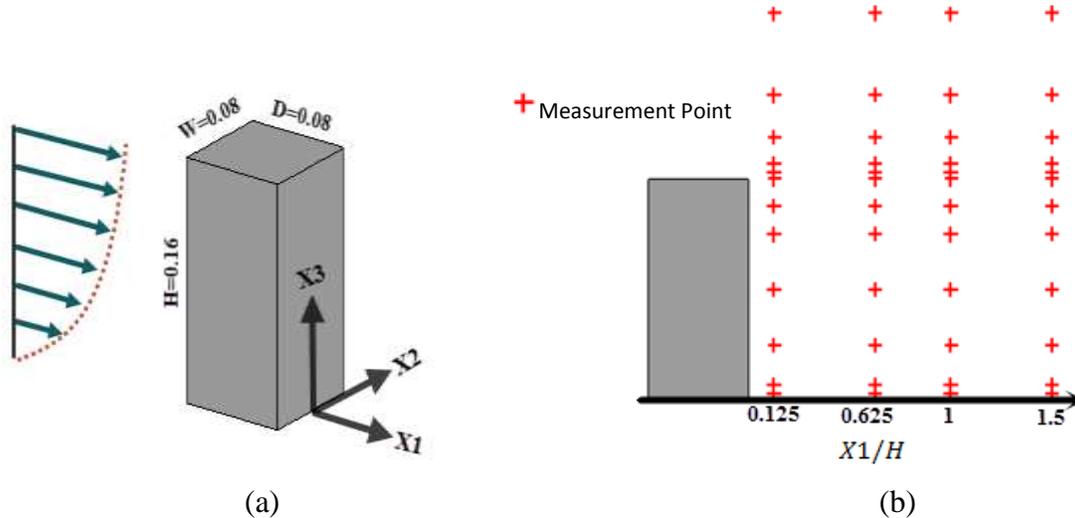
251  $P_{kb}$  and  $P_{\varepsilon b}$  are buoyancy turbulence production and dissipation terms, respectively:

$$P_{kb} = \frac{\mu_t}{\sigma_p} \beta g_i \frac{\partial T}{\partial x_i} \quad (15)$$

$$P_{\varepsilon b} = \max(0, P_{kb}) \quad (16)$$

252 where  $\sigma_p = 0.9$  is the turbulent Schmidt Number and  $\beta$  is the thermal expansion coefficient. Values of  
253 the closure coefficients, according to (Launder and Spalding, 1974), are predefined as the default  
254 values for most of the popular CFD tools as below:

$$C_\mu = 0.09, C_{\varepsilon 1} = 1.44, C_{\varepsilon 2} = 1.92, \sigma_k = 1, \sigma_\varepsilon = 1.3 \quad (17)$$



255 Figure 3 Schematic of Yoshie et al (2011) experiment: (a) bluff body dimensions, (b) measurement points

256 **3. CFD Simulation**

257 The RANS equations were solved using the commercial software ANSYS CFX, which uses an  
 258 element-based finite volume discretization method.

259 **3.1 Description of the wind tunnel experiment for unstable ABL**

260 As seen in Fig.3, the experimental data for the closure coefficients optimization were taken from  
 261 Yoshie et al (2011) in which a detailed experimental analysis on airflow and gas dispersion was  
 262 conducted around a high-rise building in a non-isothermal ABL. The target building had a dimension  
 263 of  $W \times D \times H = 0.08(m) \times 0.08(m) \times 0.16(m)$ , which was placed in an atmospheric wind tunnel at  
 264 Tokyo Polytechnic University. The surface of the wind tunnel had a uniform temperature of  $45.3^{\circ}C$   
 265 while the air velocity and temperature at the inlet were reported  $U_H = 1.37 \frac{m}{s}$  and  
 266  $\theta_H = 11^{\circ}C$ , respectively.

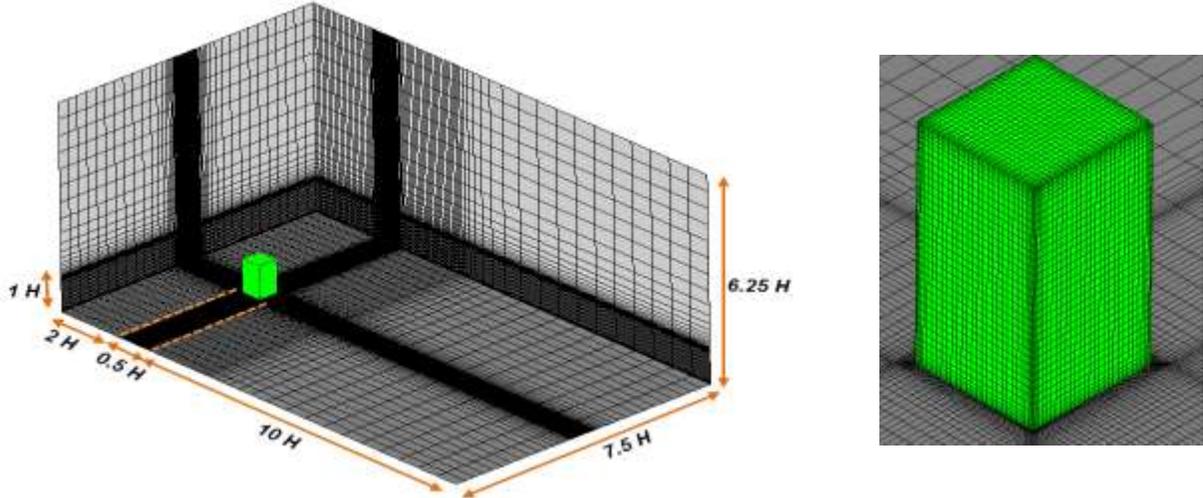
267 **3.2 Computational domain, grid, and boundary conditions**

268 A rectangular computational domain, as shown in Fig.4, was considered for the isolated building  
 269 case based on the recommendations by AIJ guidelines (Tominaga et al, 2008) and similar studies  
 270 (Mirzaei and Carmeliet, 2013). The domain width, length, and height were  $1.2(m) \times 2(m) \times 1(m)$ .  
 271 ICEM CFD meshing package was used to create structured hexahedral mesh applying the blocking  
 272 technique. A grid-sensitivity analysis was conducted for three different mesh numbers with 229,401;  
 273 396,864; and 686,585 cells as coarse, medium and fine mesh configurations. Results showed a very  
 274 negligible difference, less than 1%, between the prediction of the velocity profile in the wake region  
 275 for the medium and fine meshes; hence the medium mesh configuration was selected for the study.  
 276 Number of the cells around the building block was  $30 \times 30 \times 45$ . An O-grid block with first-layer  
 277 size of  $1.3 \times 10^{-4}(m)$  was used around the building, which resulted to an average  $y^+ \approx 1$  for the  
 278 solid surfaces. No-slip boundary condition was considered for all solid walls and a constant  
 279 temperature boundary condition was applied to the ground surface. All solid walls were treated as  
 280 smooth walls. Symmetric wall boundary condition was considered for the lateral boundaries while a  
 281 free-slip wall boundary condition was assumed for the top boundary surface. Zero static pressure was  
 282 applied at the outlet plane. Inlet boundary condition for the vertical velocity, temperature and turbulent

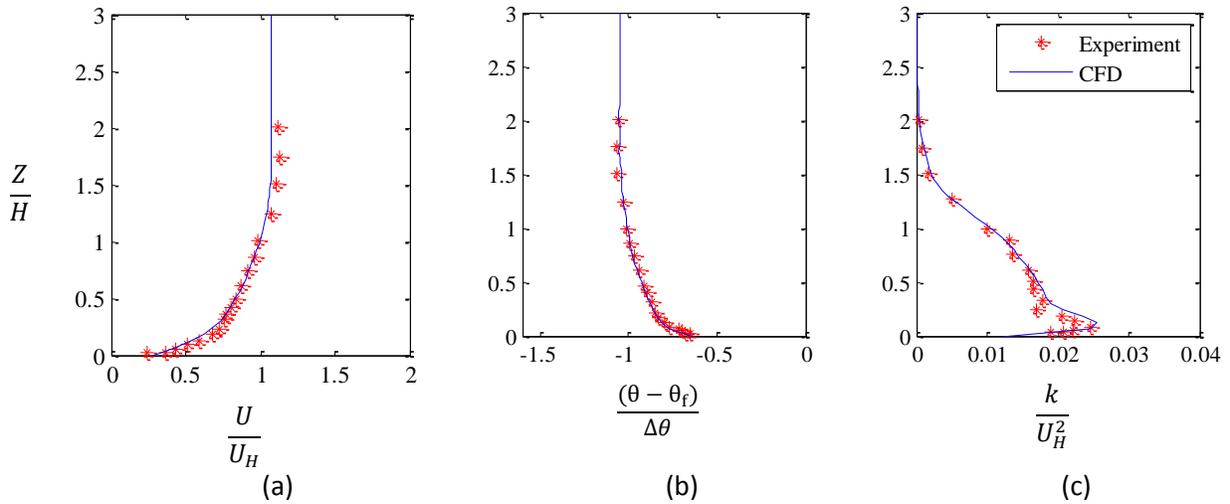
283 kinetic energy profiles were also obtained directly from Yoshie et al (2011) experiment. Turbulent  
 284 kinetic energy dissipation rate  $\varepsilon(z)$  was also approximated from the below equation (Yoshie et al,  
 285 2011):

$$\varepsilon(z) = \overline{u_1 u_3} \frac{\partial U_1}{\partial x_3} - g_3 \beta \overline{u_3 \theta'} \quad (18)$$

286 where  $\overline{u_3 \theta'}$  is the turbulent heat flux obtained from the experiment. Vertical distribution of the time  
 287 averaged streamwise velocity, turbulent kinetic energy, and temperature are depicted in Fig. 5.



288 Figure 4 Computational domain and grid arrangement



289 Figure 5 Inflow boundary condition for (Yoshie et al, 2011): (a) velocity, (b) temperature, (c) turbulent  
 290 kinetic energy

291 **3.3 Solver setting**

292 Pressure-velocity coupling was based on the Rhie-Chow interpolation proposed by Rhie and  
 293 Chow (1983) while a co-located grid layout was further used. The High Resolution Scheme was used  
 294 for the discretization of the advection terms while tri-linear shape functions were used to evaluate the

295 spatial derivatives for all the diffusion terms. For the near-wall treatment, scalable wall function based  
 296 on the modification of the [Launder and Spalding \(1974\)](#) was used. The CFD solver iterations have  
 297 been continued until reaching RMS residual of less than  $10^{-5}$  for continuity, velocity components,  
 298 energy,  $k$  and  $\varepsilon$  equations.

#### 299 4. Results

300 In this section, results of the proposed systematic approach for a test case of the defined non-  
 301 isothermal ABL flow around a building based on the  $k - \varepsilon$  model will be presented. At first, results of  
 302 a sensitivity analysis on the CFD model's response to the closure coefficients variation are presented.  
 303 After that, the main outcomes of the optimization methodology are discussed.

##### 304 4.1 Sensitivity analysis of the CFD model response to the variation of the closure coefficients

305 In order to find the effect of the closure coefficients variation on the response of the CFD model, a  
 306 parametric sensitivity analysis has been initially conducted. Results of the parametric sensitivity  
 307 analysis were then used to identify the influential parameters for being later used in the statistical  
 308 optimization. As shown in ([Dunn et al, 2011](#); [Guillas et al, 2014](#)), the highest uncertainty of flow  
 309 parameters occurred in the recirculating region and near the reattachment point after the leeward side  
 310 within the street canyon. Hence, in the case study, velocity data at 48 points in the wake region along  
 311 four streamwise positions, i.e.  $\frac{x_1}{H} = 0.125$ ,  $\frac{x_1}{H} = 0.625$ ,  $\frac{x_1}{H} = 1$ , and  $\frac{x_1}{H} = 1.5$ , were selected as the  
 312 target points for calculation of the validation metrics (see [Fig.3 \(b\)](#)).

313 Two validation metrics were adapted in this study to quantify the agreement between the  
 314 experimental and numerical results. These metrics are namely the hit rate  $q$  and the fraction of the  
 315 predictions within a factor of two of the observations ( $FAC2$ ) defined as follows ([Tominaga, 2015](#)):

$$q = \frac{1}{N} \sum_{i=1}^N n_i \quad \text{if } \left| \frac{P_i - Q_i}{P_i} \right| \leq D_q \text{ or } |P_i - Q_i| \leq W_q \quad n_i = 1 \quad \text{else } n_i = 0 \quad (19)$$

$$FAC2 = \frac{1}{N} \sum_{i=1}^N n_i \quad \text{if } 0.5 \leq \frac{P_i}{Q_i} \leq 2 \quad n_i = 1 \quad \text{else } n_i = 0 \quad (20)$$

316 where  $Q_i$  and  $P_i$  are the observed (measured) and predicted (computed) values of a given variable,  
 317 respectively, and  $N$  is the number of data points. The thresholds for  $q$  are recommended  $D_q = 0.25$   
 318 and  $W_q = 0.03$  for streamwise velocity ([Gousseau et al, 2013](#); [Tominaga, 2015](#)). For a complete  
 319 agreement between the experimental and numerical results, the value of  $q$  and  $FAC2$  should be 1. To  
 320 perform the parametric sensitivity study, four coefficients of the  $k - \varepsilon$  turbulence model, i.e.  $C_{\varepsilon 1}$ ,  $C_{\varepsilon 2}$ ,  
 321  $C_{\mu}$ , and  $\sigma_k$ , were linearly altered while for each variable, a number of 20 uniformly distributed samples  
 322 were selected among its interval. The value of  $\sigma_{\varepsilon}$  was calculated using the [eq. \(21\)](#) for each set of the  
 323 closure coefficients. In regard to the previous studies in literature, a range of closure coefficients was  
 324 considered as depicted in [Table 2](#).

$$\sigma_{\varepsilon} = \frac{\kappa^2}{C_{\mu}^{1/2} (C_{\varepsilon 2} - C_{\varepsilon 1})} \quad (21)$$

325

326

Table 2 Standard values and range of the closure coefficients for parametric study

	$C_{\varepsilon 1}$	$C_{\varepsilon 2}$	$\sigma_k$	$C_\mu$
Standard value	1.44	1.92	1	0.09
Ranges	1:1.5	1.5:3	0.8:1.4	0.05:0.15

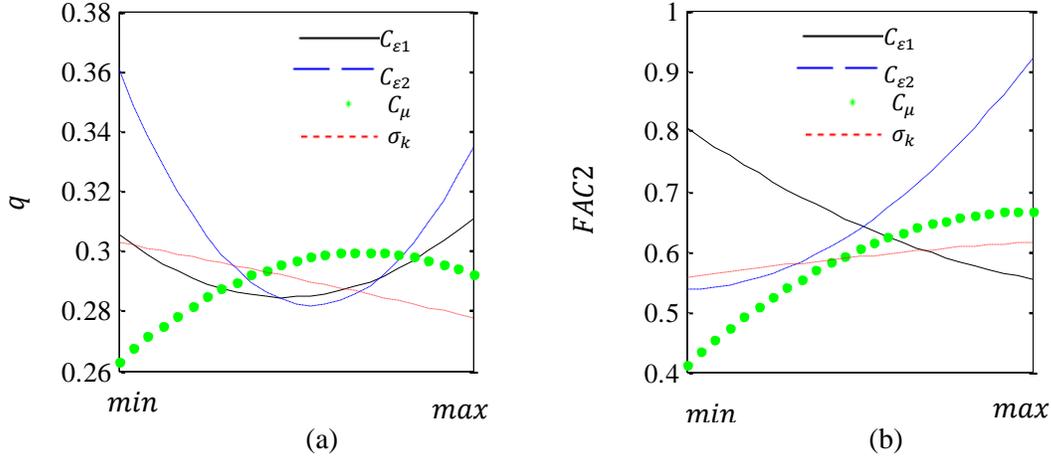


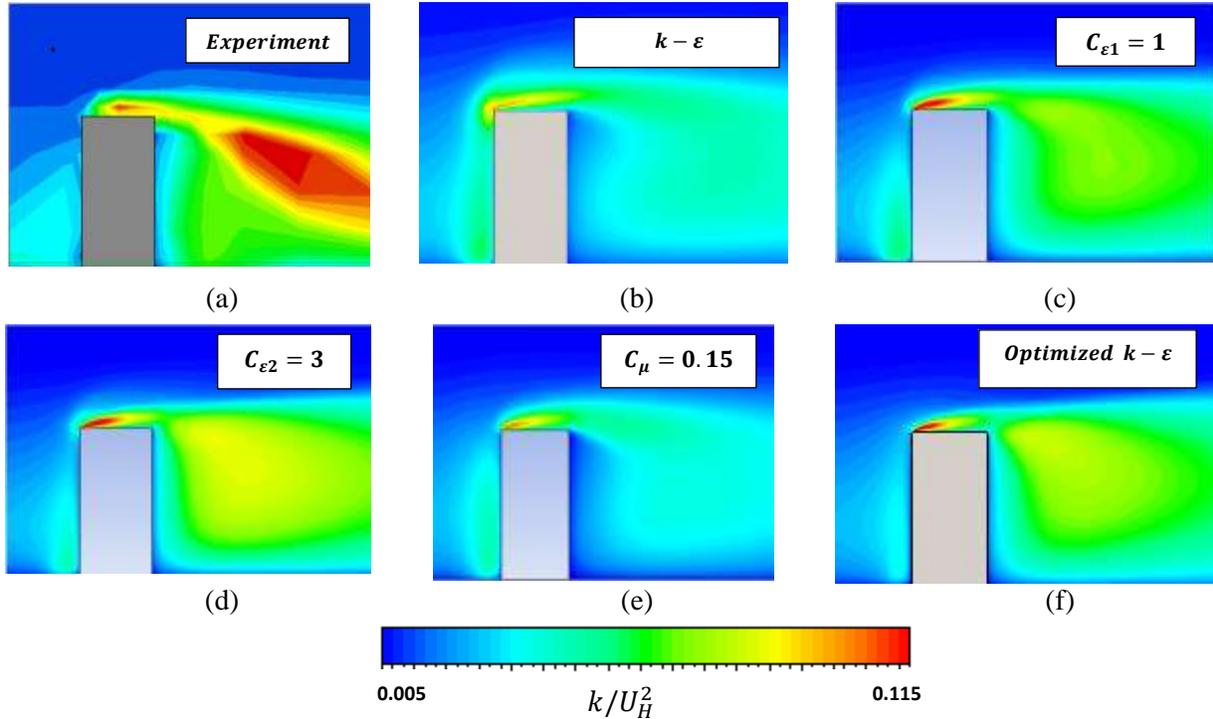
Figure 6 Variation of the validation metrics for the streamwise velocity component for isolated building case study: (a) hit rate  $q$ , (b)  $FAC2$

328  
329

330 In Fig. 6, the variation of the validation metrics for the streamwise velocity against the closure  
 331 coefficients is plotted. It can be seen that  $C_{\varepsilon 2}$  and  $C_\mu$  have a noticeable impact on both validation  
 332 metrics, but  $\sigma_k$  shows a lower impact. Considering the variation of  $FAC2$ , it reveals that the lower  
 333 values of  $C_{\varepsilon 1}$ , namely values around 1, provide a higher accurate results in terms of streamwise  
 334 velocity distribution. In contrast, the higher values of  $C_{\varepsilon 2}$ , namely  $C_{\varepsilon 2} \approx 3$ , show a better agreement  
 335 with the experiment. Same trend can be seen for  $C_\mu$  where higher value for  $FAC2$  is obtained for the  
 336 higher values of  $C_\mu$ , ranging between 0.11 and 0.15. Both validation metrics seem to be less sensitive  
 337 to  $\sigma_k$  for this data set. Nonlinear variation of  $q$  and  $FAC2$  shows the necessity of using an optimization  
 338 technique to systematically find optimal coefficients. It can be concluded that the default values for the  
 339 closure coefficients, as shown in (Edeling et al, 2014b; Guillas et al, 2014), are not accurate for the  
 340 considered test case with a strong wake region.

341 To demonstrate the effect of the closure coefficients on the turbulent kinetic energy distribution at  
 342 the wake region behind the building, contours of  $k/U_H^2$  are depicted in Fig. 7 obtained from the default  
 343 value and three other cases of the closure coefficient values in addition to the experimental results by  
 344 Yoshie et al (2011). It can be seen that for the reference case, which corresponds to the case with  
 345 default value for the closure coefficients, the level of the turbulent kinetic energy inside the wake  
 346 region behind the building is considerably low. For default closure coefficients, not only is the large  
 347 mixing process behind the building underestimated, but the generation of  $k$  over the roof is also under-  
 348 predicted. For the case with  $C_{\varepsilon 1}=1$ , distribution level of  $k$  inside the wake region is noticeably  
 349 increased. Same improvement in the distribution of  $k$  inside the wake region is observed for the case  
 350 specified with  $C_{\varepsilon 2} = 3$ . A minor improvement can be also seen for the case with  $C_\mu = 0.15$ . For the  
 351 cases with  $C_{\varepsilon 1}=1$  and  $C_{\varepsilon 2} = 3$ , the position of the formation of the high turbulent kinetic energy over  
 352 the roof has changed in a way that is much closer to the experiment in comparison to the case with  
 353 default coefficient values. Over the roof area, the average of  $k$  is noticeably increased and a more

354 agreement with the experiment is found. Improving the prediction accuracy of the  $k$  distribution both  
 355 inside the wake region behind the building and the separation region over the roof leads to a better  
 356 estimation of the reattachment lengths in these regions.



357 Figure 7 Contours of the turbulent kinetic energy: (a) experiments by Yoshie et al (2011), (b)  $k - \epsilon$ , (c)  
 358  $C_{\epsilon 1} = 1$ , (d)  $C_{\epsilon 2} = 3$ , (e)  $C_{\mu} = 0.15$ , (f) optimized coefficients

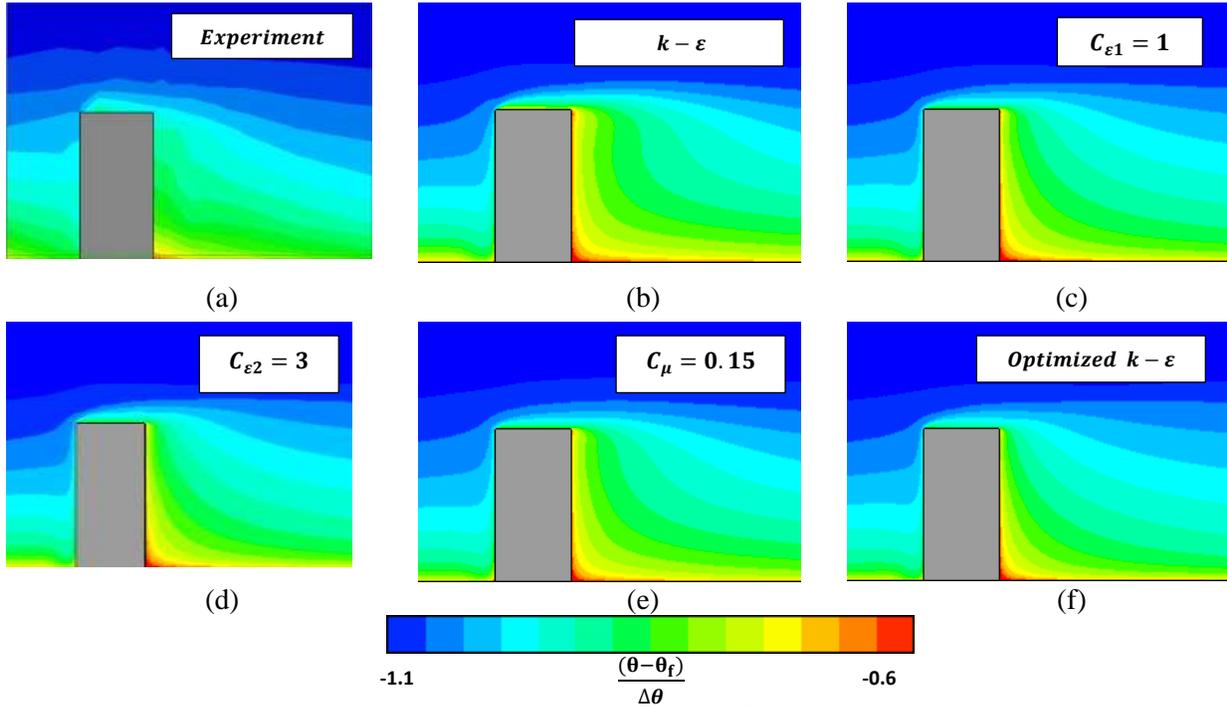
359 Predicted values for the roof reattachment length ( $X_r$ ) and the floor reattachment length ( $X_f$ ) are  
 360 presented in Table 3. The experimental value for the reattachment length at the floor is estimated to be  
 361  $X_f = 0.096 (m)$ . This value for the reference case with the default coefficients is  $X_f = 0.260(m)$   
 362 while it is  $X_f = 0.138 (m)$  when  $C_{\epsilon 2} = 3$ . In the case of  $C_{\mu} = 0.15$ , the  $k$  distribution increased in  
 363 relation to the reference case, but its increase is lower than that of altered  $C_{\epsilon 1}$  and  $C_{\epsilon 2}$ . This resulted in  
 364 a longer reattachment length of  $X_f = 0.201(m)$ . The shortest roof reattachment length is predicted for  
 365  $C_{\epsilon 1}=1$  followed by the case for  $C_{\epsilon 2} = 3$ . This value is not reported in the experiment, but it can be  
 366 estimated to be around  $X_r \approx 0.045 (m)$ .

367 Table 3 Comparison of the reattachment length on roof ( $X_r$ ) and reattachment length behind the building  
 368 ( $X_f$ )

	<i>Experiment</i>	$k - \epsilon$	$C_{\epsilon 1} = 1$	$C_{\epsilon 2} = 3$	$C_{\mu} = 0.15$	$\sigma_k = 1.4$	<i>Optimized</i> $k - \epsilon$
$X_r$	NA	0.061	0.023	0.024	0.029	0.043	0.016
$X_f$	0.096	0.260	0.159	0.138	0.201	0.228	0.123

369 In Fig. 8, contours of the temperature distribution around the building for different closure  
 370 coefficients, which proved to have a positive effect on the validation metrics, are displayed and  
 371 compared with the experimental data. It is important to note that a fixed turbulent Prandtl number was  
 372 considered for all simulations. In the case of default closure coefficients, due to the poor mixing

373 behind the building, the temperature diffusion inside the wake region is noticeably lower than that of  
 374 the experimental observation. For the cases with  $C_{\varepsilon 1} = 1$  and  $C_{\varepsilon 2} = 3$ , thanks to the higher diffusion  
 375 of the momentum inside the wake region, the temperature distribution becomes more realistic and a  
 376 very close agreement with the experimental data can be obtained. For the case with  $C_{\mu} = 0.15$   
 377 temperature distribution has insignificant improvement due to the lower diffusion of the momentum  
 378 inside the wake region. Results of the parametric sensitivity study show that among the considered  
 379 closure coefficients for the considered flow condition, all the coefficients except  $\sigma_k$  have a significant  
 380 impact on the accuracy of the  $k - \varepsilon$  model in terms of velocity, turbulent kinetic energy, and  
 381 temperature distribution. Hence,  $C_{\varepsilon 1}$ ,  $C_{\varepsilon 2}$  and  $C_{\mu}$  were selected as the input variables for the stochastic  
 382 optimization to find a suitable set of closure coefficients.

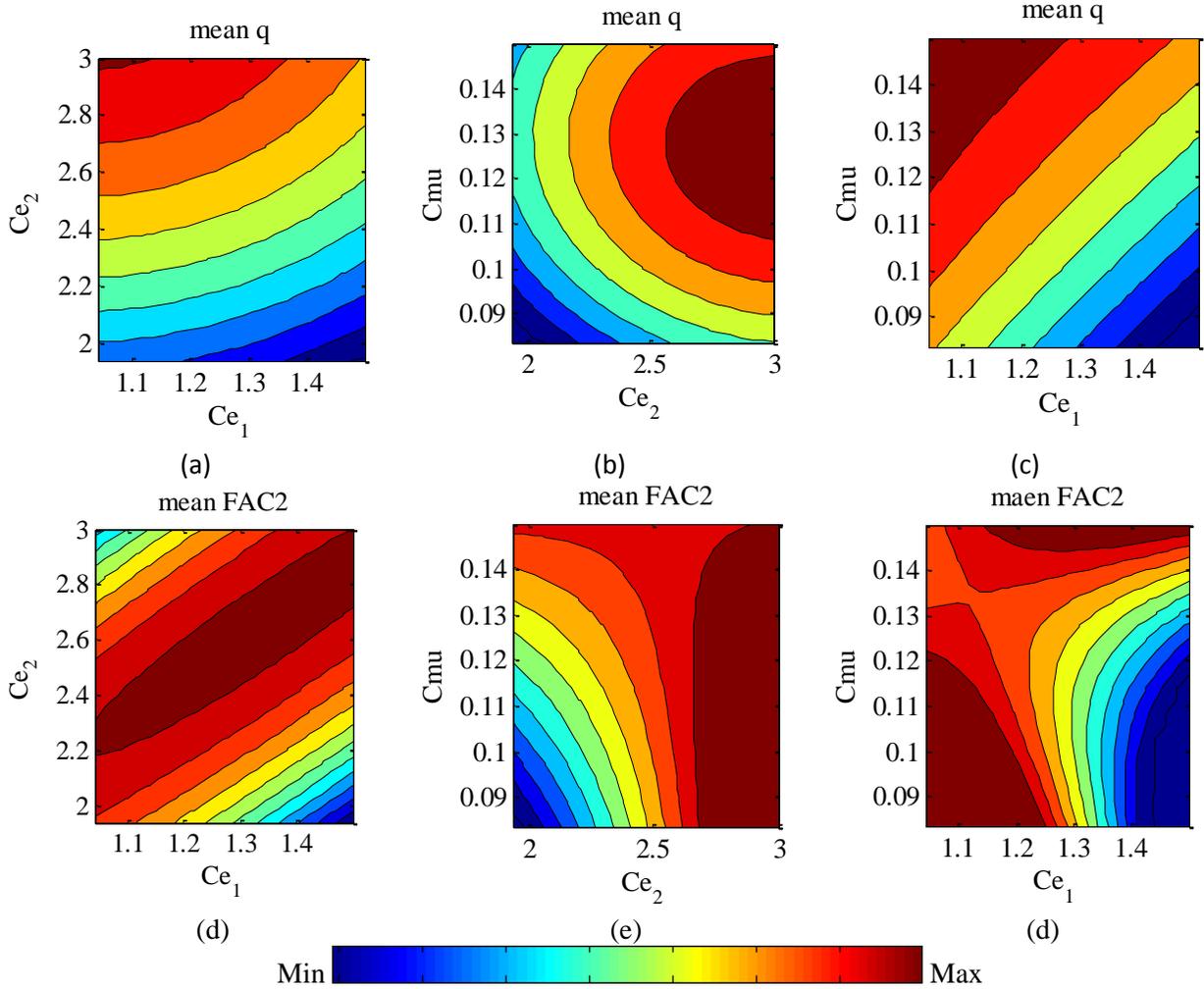


383 Figure 8 Distribution of the non-dimensional temperature  $\frac{(\theta - \theta_f)}{\Delta\theta}$ : (a) experiments by Yoshie et al (2011),  
 384 (b)  $k - \varepsilon$ , (c)  $C_{\varepsilon 1} = 1$ , (d)  $C_{\varepsilon 2} = 3$ , (e)  $C_{\mu} = 0.15$ , (f) optimized coefficients

#### 385 4.2 Optimization results

386 Based on the results of the parametric study, a stochastic optimization using the Monte Carlo  
 387 sampling technique was performed to find out a modified set of closure coefficients, providing CFD  
 388 results with a higher agreement with the experimental data in terms of the validation metrics defined in  
 389 [eq. \(19\)](#) and [eq. \(20\)](#). In the stochastic optimization process, all input variables, including  $C_{\varepsilon 1}$ ,  $C_{\varepsilon 2}$  and  
 390  $C_{\mu}$ , were treated as the random or uncertainty variables with a uniform PDF ranged in accordance with  
 391 the values in [Table 2](#).  $\sigma_k$  was not considered in the optimization as it has a low impact on the  
 392 validation metrics according to the sensitivity parametric study ([Fig. 6](#), [Fig. 7](#) and [Fig. 8](#)) and it was set  
 393 to its default value of 1. Probability density function of  $\sigma_{\varepsilon}$  was obtained during the optimization  
 394 iterations based on [eq. \(21\)](#). The maximum iteration for the optimization loop was set to 100 while a  
 395 termination accuracy of  $10^{-6}$  was considered for optimization convergence. The objective functions

396 of the both validation metrics, i.e.  $FAC2$  and  $q$ , were considered to be maximized to reach an ideal  
 397 value of 1, which can be interpreted as the best agreement between the CFD simulations and  
 398 experiment. An equal importance was considered for the mean and the standard deviation values of the  
 399 validation metrics ( $FAC2$  and  $q$ ), hence a weighing factor of 1 was considered for  $w_{1_i}$  and  $w_{2_i}$  in eq.  
 400 (2). The maximum value for  $FAC2$  and  $q$  is 1 and thus the values of the scaling factors  $s_{1_i}$  and  $s_{2_i}$   
 401 were set to 1 for all objectives in eq. (2).



402 Figure 9 Variation of the mean value of the validation metrics for the streamwise velocity component and  
 403 the closure coefficients during the optimization process

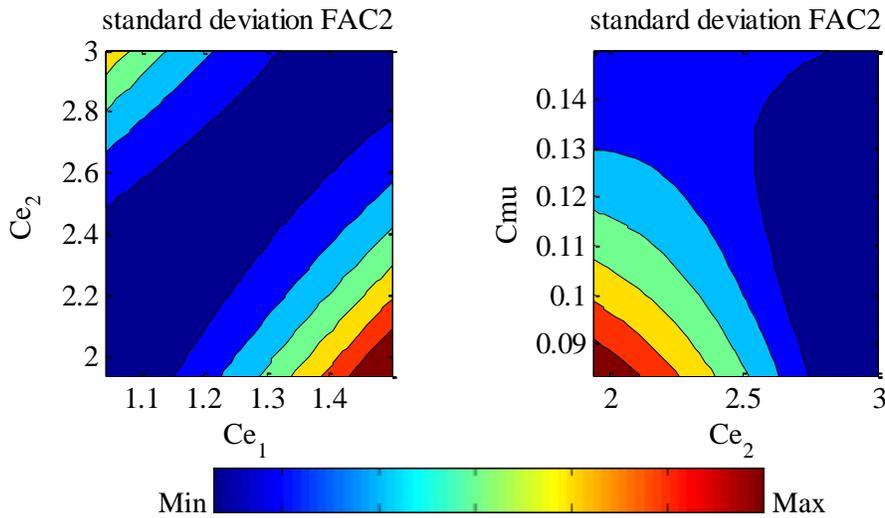
404 Four objective functions were considered for the optimization process, including the mean values  
 405 of  $q$  and  $FAC2$  and their standard deviations for the streamwise velocity component at 48  
 406 measurement points shown in Fig. 3 (b). The optimization process have been executed for 250 CFD  
 407 simulations. Variation of the mean value of the validation metrics for the streamwise velocity  
 408 component and the closure coefficients during the optimization process is shown in Fig.9. Contours of  
 409 hit rate  $q$  show that the highest agreement between the CFD and the experiment occurs for  $C_{\epsilon_1}$  values  
 410 in the range of  $1.1 \leq C_{\epsilon_1} \leq 1.3$  and for  $C_{\epsilon_2}$  values in the range of  $2.6 \leq C_{\epsilon_2} \leq 3$ . The most suitable  
 411 value of  $C_{\mu}$ , which results in high hit rate values, is found for  $0.12 \leq C_{\mu} \leq 0.15$ . In terms of the

412 second validation metric (*FAC2*), a quite similar result is obtained. For  $1.1 \leq C_{\varepsilon 1} \leq 1.5$  and  $2.7 \leq$   
 413  $C_{\varepsilon 2} \leq 3$ , *FAC2* has the highest value. The mean value of *FAC2* is acceptable for  $C_{\mu}$  ranges between  
 414 0.12 and 0.15. In general, it can be concluded that the highest probability of having a very close  
 415 agreement between CFD results of the  $k - \varepsilon$  model with those of the experimental analysis of non-  
 416 isothermal airflow around a high-rise building in terms of the mean values of  $q$  and *FAC2* occurs for  
 417 the closure coefficients in the ranges of  $1.1 \leq C_{\varepsilon 1} \leq 1.5$ ,  $2.7 \leq C_{\varepsilon 2} \leq 3$ , and  $0.12 \leq C_{\mu} \leq 0.15$ . It is  
 418 noteworthy to mention that the value of  $\sigma_k$  is assumed as its default value of 1 while the value of  $\sigma_{\varepsilon}$   
 419 can be calculated using [eq. \(21\)](#), which results in  $0.32 \leq \sigma_{\varepsilon} \leq 0.56$ .

420 As described earlier, not only were the mean values of the validation metrics considered in the  
 421 stochastic optimization process, but their standard deviations were also included in the objective  
 422 function to reduce the impact of the uncertainty of the closure coefficients on the validation metrics.  
 423 [Fig. 10](#) shows contours of the standard deviation of *FAC2* for the streamwise velocity. It can be seen  
 424 that in the specified ranges, where the mean values of the validation metrics are optimum, their  
 425 standard deviations are also in their minimum values. Finally, the optimum values of the closure  
 426 coefficients, resulted in the highest mean value for the validation metrics with the lowest standard  
 427 deviation, can be expressed as follows:

$$C_{\varepsilon 1} = 1.489, C_{\varepsilon 2} = 2.801, C_{\mu} = 0.146, \sigma_{\varepsilon} = 0.373, \sigma_k = 1 \quad (22)$$

428

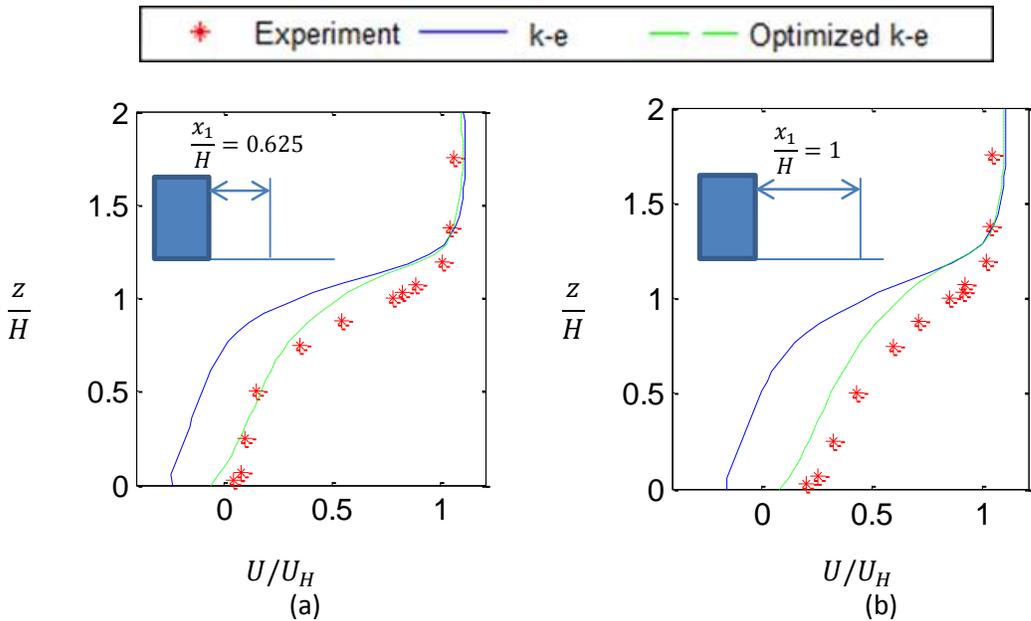


429 [Figure 10](#) Plot of the standard deviation of the *FAC2* for streamwise velocity in the optimization process

430 The mean values of validation metrics  $q$  and *FAC2* for streamwise velocity increased from 0.31  
 431 and 0.54 to 0.47 and 0.91 for default coefficients and optimized coefficients, respectively. The  
 432 standard deviation of  $q$  and *FAC2* were also found to be 0.05 and 0.03 for the optimized coefficients.  
 433 In general, using the modified closure coefficients in the  $k - \varepsilon$  formulation results in a higher  
 434 momentum mixing and turbulent kinetic energy inside the wake region behind the building. This was  
 435 achieved by altering the production and dissipation terms in  $k$  and  $\varepsilon$  equations. The increase of the  
 436 momentum diffusion is related to the value of  $C_{\mu}$ , rising from 0.09 to 0.14, and increase of the TKE  
 437 level inside the wake region. For the case considered in this study, when the modified closure

438 coefficients were used, the average values of the momentum diffusion,  $k$  diffusion,  $k$  production term,  
 439  $\varepsilon$  production term, and  $\varepsilon$  dissipation term over the measurement points in the wake region grew about  
 440 40%, 51%, 52%, 32% and 34%, respectively.

441 In order to observe the effect of the optimized closure coefficients on the airflow distribution  
 442 around the building, results of the CFD simulation with the optimized closure coefficients are  
 443 presented and discussed. In Fig. 11 (a) and Fig. 11(b), vertical distribution of the streamwise velocity  
 444  $\frac{U}{U_H}$  at two locations behind the building, i.e.  $\frac{x_1}{H} = 0.625$  and  $\frac{x_1}{H} = 1$ , are depicted for the reference  
 445 CFD model with default closure coefficients as well as the optimized CFD model with the new set of  
 446 the closure coefficients. The results are also compared to those reported in Yoshie et al (2011). For the  
 447 reference case with the default coefficients, the reverse flow in the wake region is overestimated due to  
 448 the poor momentum mixing behind the building. For the case with optimized closure coefficients, a  
 449 significant improvement in the prediction accuracy of the velocity distribution in the wake region can  
 450 be clearly observed, which results from a better momentum mixing. The reattachment length predicted  
 451 for the default closure coefficients, as reported in Table 3, is  $X_f = 0.260(m)$ , which is much longer  
 452 than that of the experiment with the value of  $X_f = 0.096(m)$ . The predicted reattachment length  
 453 behind the building for the optimized coefficients is  $X_f = 0.123(m)$ , appearing closer to the value of  
 454 the measurement.

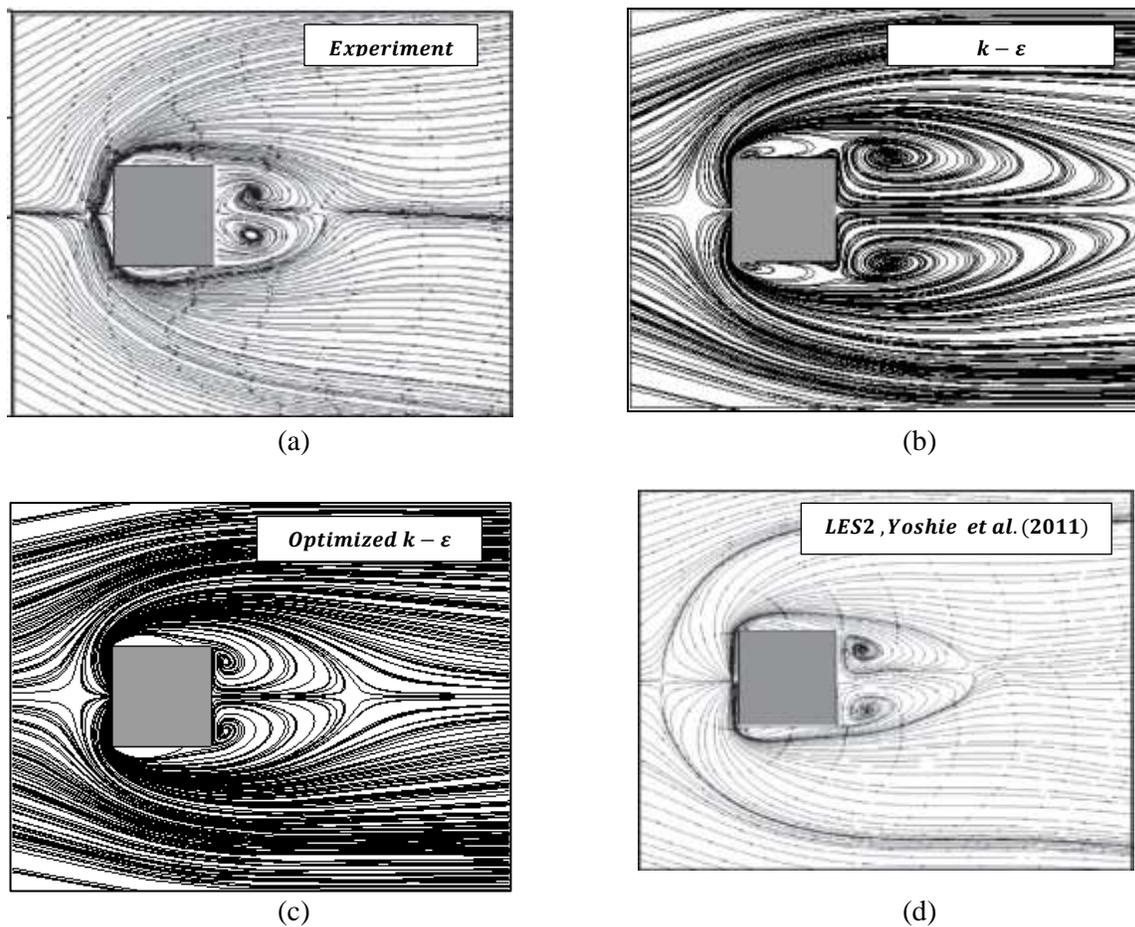


455 Figure 11 Vertical distribution of the streamwise velocity  $\frac{U}{U_H}$  in the wake region behind the building at:  
 456 (a)  $\frac{x_1}{H} = 0.625$ , (b)  $\frac{x_1}{H} = 1$

457 In the case of modified coefficients, distribution of turbulent kinetic energy along with the  
 458 diffusion of TKE and its production term inside the wake region behind the building have been  
 459 increased noticeably in comparison with the results obtained by the default coefficients (see Fig. 7).  
 460 However, comparison between the experimentally measured turbulent kinetic energy (Fig. 7(a)) and  
 461 those predicted by modified RANS model (Fig. 7(f)), shows that the CFD model significantly under-  
 462 predicts the  $k$  distribution behind the building. It refers to the fact that the steady RANS models are

463 inherently incapable of calculating the unsteady nature of the turbulent kinetic energy because of the  
464 large-scale fluctuations behind the building.

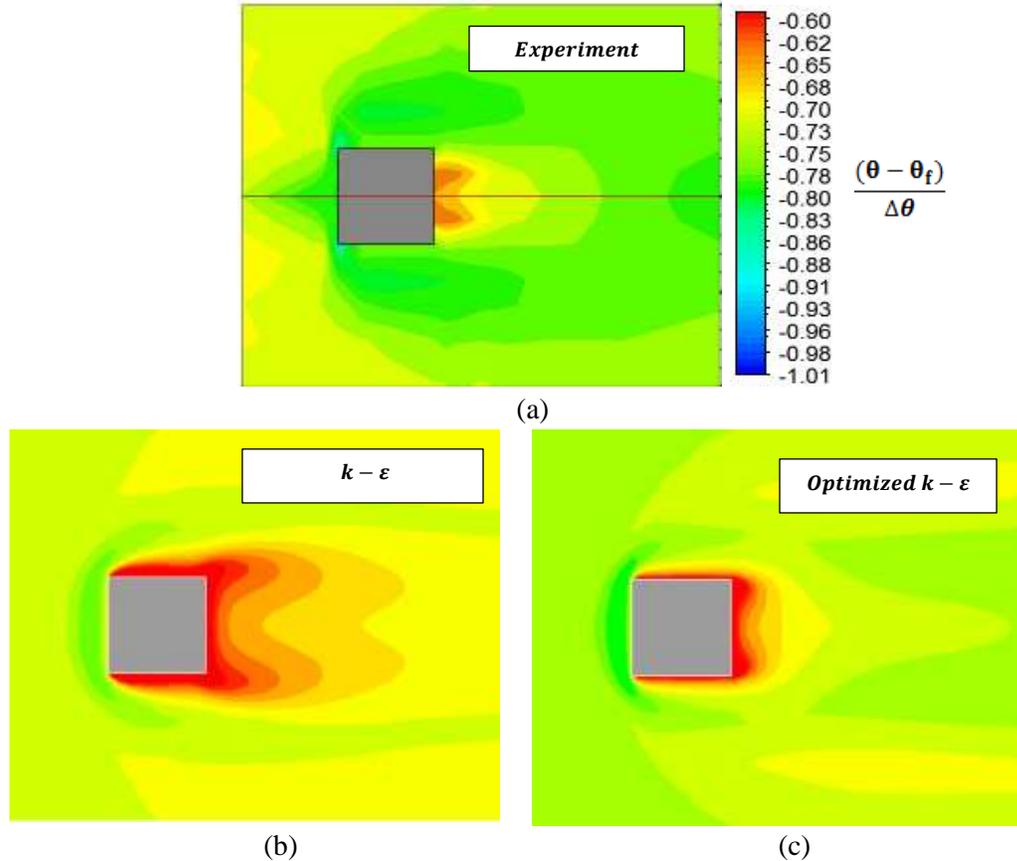
465 In Fig. 12, the horizontal distribution of the streamlines is depicted for the  $k - \varepsilon$  model using  
466 default and optimized closure values. These streamlines are further compared with the results of the  
467 experimental and LES models presented in Yoshie et al (2011). A long recirculation region can be  
468 seen for the  $k - \varepsilon$  model with the default coefficients. However, for the case with modified closure  
469 coefficients, the length of the recirculating region considerably decreases. The results hence show  
470 more agreement with the experimental data and LES; namely a more accurate, but computationally  
471 expensive model.



472 Figure 12 Horizontal distribution of the streamlines near the ground ( $\frac{x_3}{H} = 0.025$ ): (a) experiment by  
473 Yoshie et al (2011), (b) default  $k - \varepsilon$  closure coefficients, (c) modified  $k - \varepsilon$  closure coefficients, (d) LES-2  
474 from (Yoshie et al, 2011)

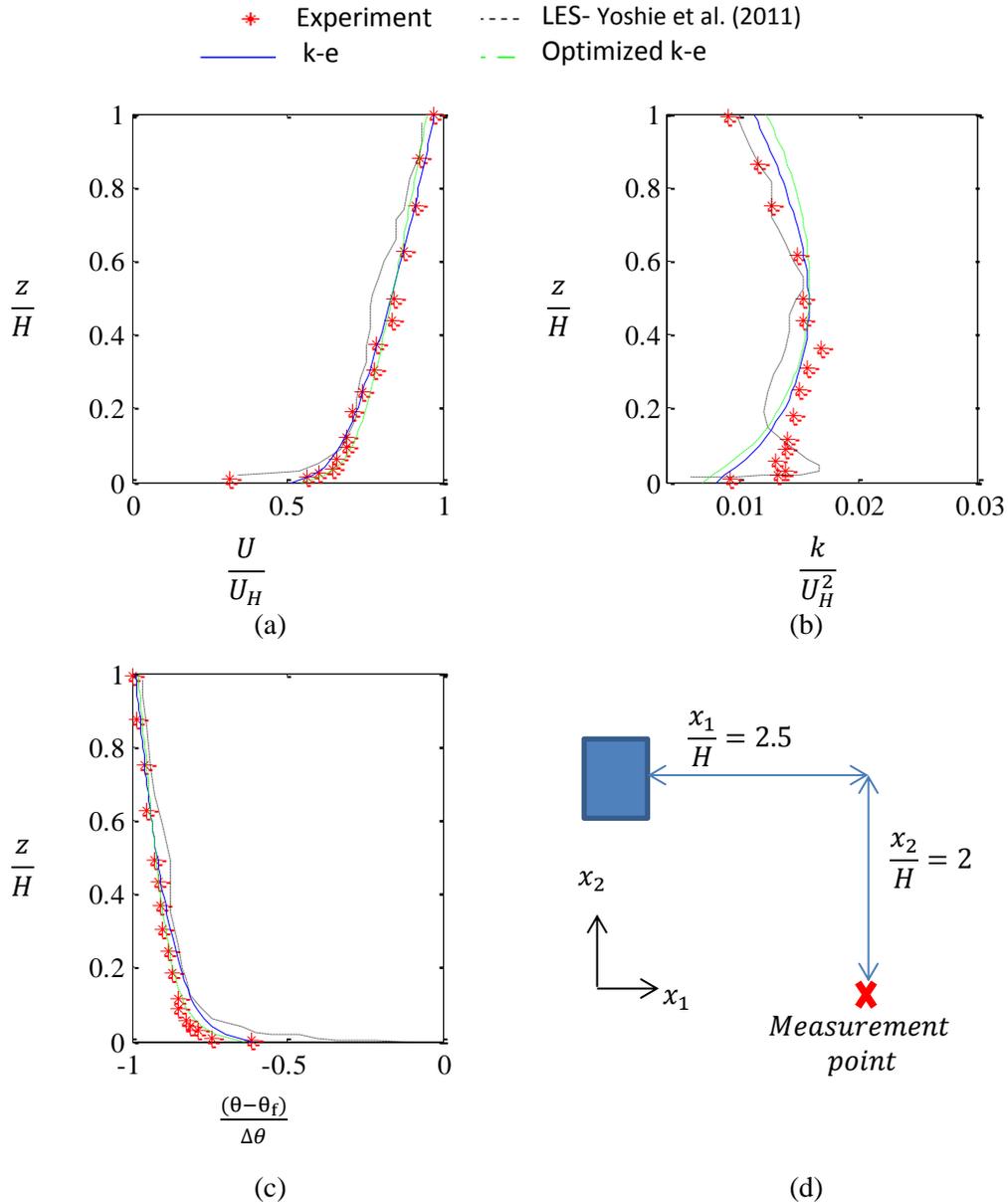
475 Contours of the temperature distribution on the same position are also illustrated in Fig. 13. For  
476 the case using the default closure coefficients, not only is the level of temperature for the ground  
477 surface predicted to be in a higher range than the experiment, but a different temperature pattern is  
478 further estimated in the wake region behind the building. For the optimized coefficients, however, the  
479 temperature level over the ground surface is closer to the experiment while the temperature  
480 distribution behind the building is spread shorter than that of the case with the default coefficients. The

481 higher temperature observed in the CFD model around the building surface refers to the implemented  
 482 adiabatic boundary condition; whereas seemingly the building is not completely isolated from the  
 483 ground surface in the experiment. Even if considering the uncertainty in the near wall measurement,  
 484 the CFD results by the modified coefficients are more acceptable than those of the default coefficients.



485 Figure 13 Contours of the temperature distribution  $\frac{(\theta - \theta_f)}{\Delta\theta}$  near the ground surface: (a) experiment by Yoshie  
 486 et al (2011), (b) default  $k - \epsilon$  closure coefficients, (c) modified  $k - \epsilon$  closure coefficients

487 As merely those measurement points that are in the wake region behind the building are  
 488 considered in the optimization process, it is noteworthy investigating the distribution of flow  
 489 properties in a high speed region far from the building. In Fig. 14, vertical profiles of the streamwise  
 490 velocity, turbulent kinetic energy, and temperature along a vertical line placed far from the building at  
 491  $\frac{x_1}{H} = 2.5$  and  $\frac{x_2}{H} = 2$  are shown. Numerical results obtained by the modified coefficient are compared  
 492 with those obtained by default coefficients and experimental data. Also, results of a LES model  
 493 presented in (Yoshie et al, 2011) are plotted. It can be seen that the vertical profiles are very similar for  
 494 the modified and reference cases as well as the experiment. LES model estimated the turbulent kinetic  
 495 energy more accurately, which refers to the higher accuracy of LES in reproducing the unsteady  
 496 contribution of turbulent kinetic energy. It is noteworthy to mention that in the current optimization  
 497 process solely the mean value of the streamwise velocity component was considered; however the  
 498 accuracy of the modified  $k - \epsilon$  model in predicting the turbulent kinetic energy can be further  
 499 improved by incorporating the experimental value of  $k$  into the optimization process.



501 Figure 14 Vertical profiles of flow parameters along a vertical line far from the building in high speed  
 502 region at  $\frac{x_1}{H} = 2.5$  and  $\frac{x_2}{H} = 2$ : (a) streamwise velocity  $\frac{U}{U_H}$ , (b) turbulent kinetic energy  $\frac{k}{U_H^2}$ ,  
 503 (c) non-dimensional temperature  $\frac{(\theta - \theta_f)}{\Delta\theta}$ , (d) measurement position

504  
 505 **5. Conclusion**

506 Steady RANS models (including the  $k - \varepsilon$  model with Kato-Launder modification) based on the  
 507 two-equation turbulence models underestimate the momentum diffusion behind the building in the  
 508 weak wind regions. This results in estimating a large recirculating flow in the wake region and a long  
 509 reattachment length on the ground. Also, a poor accuracy for the temperature field around the  
 510 building, specifically in the wake region, is predicted with steady RANS models. Application of the

511 default closure coefficients of RANS turbulence models in the popular commercial CFD tools proved  
512 to be inaccurate for CFD modeling of the microclimate studies. A systematic approach is therefore  
513 proposed in this study in order to improve the accuracy of the RANS family turbulence models  
514 applying the stochastic optimization and Monte Carlo Sampling technique. In the optimization  
515 process, the closure coefficients were treated as a series of random variables with a given PDF to  
516 achieve the best agreement with the experimental data in accordance with the validation metrics.  
517 Effectiveness of the proposed methodology for the modification of the closure coefficients of the  $k - \varepsilon$   
518 model was shown for simulation scenario of an isolated building placed in a non-isothermal  
519 atmospheric boundary layer. In urban areas, because of both the presence of thermal radiation and low  
520 air velocity due to the sheltering effect, buoyancy effect is of high importance. A sensitivity analysis  
521 was initially conducted to investigate the impact of the  $k - \varepsilon$  closure coefficients on the accuracy of  
522 the CFD model in comparison with the results of the experimental analysis. The default values of the  
523 closure coefficients for the  $k - \varepsilon$  model used in the popular CFD tools such as ANSYS CFX, ANSYS  
524 FLUENT, PHOENIX and STAR CCM+ are  $C_\mu = 0.09, C_{\varepsilon 1} = 1.44, C_{\varepsilon 2} = 1.92, \sigma_k = 1$  and  $\sigma_\varepsilon = 1.3$ .  
525 However, the recommended values based on the optimization method were found to be  $1.45 \leq C_{\varepsilon 1} \leq$   
526  $1.5$  and  $2.7 \leq C_{\varepsilon 2} \leq 3$  and  $0.12 \leq C_\mu \leq 0.15$  while the default value of  $\sigma_k$  was suggested to be  
527 acceptable. Based on the numerical results, the modified closure coefficients showed a significant  
528 improvement in the accuracy of the CFD model in terms of the velocity, turbulent kinetic energy, and  
529 temperature distribution around the building as well as the reattachment length behind the building.  
530 The proposed methodology was applied to an isolated building, which is a classical problem in urban  
531 aerodynamic, but it can certainly be applied to urban models in dense areas with a group of buildings.  
532 Also, it is noteworthy saying, despite the significant improvement in the prediction accuracy achieved  
533 by the optimization method, the RANS turbulence models have inherent shortcomings concerning the  
534 gradient-diffusion hypothesis and also incapability to reproduce the large-scale fluctuations of flow  
535 parameters around the building. Our future work will focus on extending the application of the  
536 proposed systematic approach in this study to other CFD modeling examples for the airflow prediction  
537 in the urban studies in which we also consider the uncertainty of the turbulent Prandtl number in the  
538 energy equation as a calibrating parameter. Through the proposed method, one can find a modified set  
539 of the closure coefficients using the available experiment, and then apply the modified coefficients in  
540 the CFD model for design and analysis purposes.

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