

# Modelling the Stiffness Development in Asphalt Concrete to Obtain Fatigue Failure Criteria

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## Abstract:

The study of fatigue is critical to good usability and durability of asphalt pavements. Inaccurate calculation of the fatigue failure criteria can cause incorrect evaluation of the fatigue performance, leading to inaccurate prediction of the pavement performance and poor maintenance planning. This paper develops a stiffness change tendency method (SCTM) that can be used to model the stiffness evolution in asphalt concrete and determine critical laboratory fatigue failure points to characterise different fatigue damage stages. A logistic model was selected to represent the relationship between the log of the stiffness ( $E$ ) and the log of the number of cycles ( $N$ ) obtained from two-point bending (2PB) fatigue tests. The measured stiffness reduction versus loading curves were determined for a range of asphalt mixtures in unaged, aged and moisture damaged conditions by testing at various temperatures and strains. By analysing the derivatives of the logistic model, it is possible to identify three transition points associated with fatigue progression. There is good agreement between the laboratory data and the logistic model proposed, confirming that the logistic model is a good approximation to the stiffness reduction curves. The number of loading cycles associated with the first two transition points in the SCTM ( $N_{P1}$  and  $N_{P2}$ ) were compared to the value of  $N_I$  and  $N_{fm}$  obtained from the energy ratio (ER) method and the ratio of dissipated energy change (RDEC) method, respectively. There are no statistically differences between the SCTM and two energy-based methods, proving that P1 can be viewed as the number of cycles to micro-crack initiation and propagation, and P2 can be defined as the macro-crack generation point (the true failure point). Three different mixtures are subjected to four-point bending (4PB) fatigue tests to demonstrate the applicability of the SCTM with different bitumen types, mixture grades and test methods. The SCTM provides a method to model stiffness development, obtain different fatigue failure criteria and characterise different fatigue damage stages, which could be useful in a simulation of pavement deterioration.

**Keywords:** asphalt mixtures, fatigue, stiffness loss, dissipated energy, pavement performance

### 33 **Highlights**

- 34 • The 2PB test is used to develop a stiffness change tendency method (STCM).
- 35 • The stiffness development is modelled by the STCM.
- 36 • The value of the transition points is compared against traditional energy-based methods.
- 37 • The SCTM is applicable for different bitumen grades, mixture grades and test methods

### 38 **1 Introduction**

39 Asphalt pavements are the main type of pavement structure used in road networks worldwide. With increasing  
40 vehicle numbers and higher traffic loads, there is a continuous need to improve the performance of asphalt  
41 pavements to maintain serviceability. One of the main distresses experienced by asphalt pavements is fatigue  
42 damage resulting from gradual weakening of pavements as a function of cumulative traffic loading cycles.  
43 Considerable research has been undertaken to explore methods to reduce the effect of fatigue damage and prolong  
44 the pavement life, critical to this is the accurate identification of the key stages during the fatigue performance  
45 deterioration and the appropriate selection of failure criteria [1-5].

46 Within the literature, various approaches have been used to define fatigue damage and analyse fatigue performance.  
47 Traditionally, fatigue failure is defined using a phenomenological approach with the fatigue failure criterion being  
48 defined as the number of loading cycles corresponding to a 50% stiffness reduction [6]. The traditional fatigue  
49 failure criterion is easy for calculation and prediction of the fatigue life, but results in the loss of information  
50 regarding fatigue behaviour characteristics. It is realized that a typical fatigue curve (stiffness versus number of  
51 cycles under strain-controlled mode) is presented by a “S” shaped curve with four distinct regions (internal heating,  
52 micro-crack formation, crack formation and propagation, and sample breakdown) separated by three transition  
53 points [7-8]. It is then suggested that the first and second transition points are possibly associated with  
54 microcracking and macrocracking in the material, respectively [9]. But the method of how to determine the  
55 transition points on the stiffness reduction curve, and how to relate the transition points to different fatigue  
56 cracking processes are still not very clear.

57 A range of energy-based approaches have subsequently been proposed as improved methods to characterise  
58 fatigue damage. The cumulative dissipated energy (CDE) approach and initial dissipated energy (IDE) approach  
59 are two of the earliest examples of this [8,10]; however, only a proportion of the energy dissipation happening in  
60 asphalt materials during the loading-unloading process is due to fatigue damage. Further methodologies to isolate  
61 the energy dissipation due to damage and viscoelasticity, including the ratio of dissipated energy change (RDEC),  
62 dissipated pseudo-strain energy (DPSE) and rate of DPSE approaches, were subsequently developed [2,11-12].  
63 Different from the traditional method, energy-based approaches are independent of test conditions and mode of  
64 loading when nonlinearity in material response is not significant [13-14]. More importantly, by calculating the  
65 plot of dissipated energy versus number of cycles, three stages that were clearly separated by two transition points  
66 were found. The transition points were then used as fatigue criteria that were defined and calculated using the  
67 energy-based approaches. Pronk and Hopman [15] used the energy ratio (ER) to define the number of cycles at a  
68 critical fatigue point ( $N_1$ ) where the relationship between the ER and the number of cycles becomes nonlinear, for

69 controlled strain fatigue tests. Pronk [16] then suggested that  $N_1$  could be defined as the number of cycles at which  
70 cracks are considered to initiate. Ghuzlan and Carpenter [11] used the RDEC to define macroscopic failure ( $N_{fm}$ ),  
71 which is related to crack propagation. The RDEC method was refined by Shen and Carpenter to calculate the  
72 value of  $N_{fm}$  [17]. Although these energy-based approaches greatly improve the ability to determine the  
73 appropriate failure criteria by relating criteria to crack initiation and propagation, they have a limitation in that the  
74 defined failure criteria cannot be directly connected to the stiffness development (strain-stress change).

75 This paper combines the benefits of both approaches (phenomenological and energy-based) by developing a  
76 method that can be used to model the stiffness evolution in asphalt concrete, determine critical laboratory fatigue  
77 failure points to characterise different fatigue damage stages, and relate the fatigue failure points to different  
78 fatigue cracking processes. This aim is achieved by selecting a logistic model to represent the relationship between  
79 the log of the stiffness ( $E$ ) and the log of the number of cycles ( $N$ ) obtained from two-point bending (2PB) fatigue  
80 tests to form a stiffness change tendency method (SCTM). The measured stiffness reduction versus loading curves  
81 were determined for a range of asphalt mixtures in unaged, aged and moisture damaged conditions by testing at  
82 various temperatures and strains. By analysing the derivatives of the logistic model, it is possible to identify three  
83 transition points associated with fatigue progression. The number of loading cycles associated with the first two  
84 transition points in the SCTM ( $N_{p1}$  and  $N_{p2}$ ) were compared to the value of  $N_1$  and  $N_{fm}$  obtained from two  
85 energy-based approaches (the ER and RDEC approaches), aiming to relate  $N_{p1}$  and  $N_{p2}$  to crack initiation and  
86 propagation. Finally, three different mixtures are subjected to four-point bending (4PB) fatigue tests to  
87 demonstrate the applicability of the SCTM in more bitumen grades, mixture grades and test methods.

## 88 2 Materials and Experimental Methods

### 89 2.1 Raw Materials and Mixture Design

90 Three asphalt concretes, namely the dense aggregate gradation bituminous mixture with NMAs of 14 mm (DBM  
91 -14) with a 40/60 penetration grade bitumen (Pen40/60), the dense aggregate gradation bituminous mixture with  
92 NMAs of 10 mm (DBM-10) with a crumb rubber modified bitumen (CRMB) and stone mastic asphalt with NMAs  
93 of 10 mm (SMA-10) + Pen40/60 were used for the comparison of the fatigue failure criteria of different asphalt  
94 concretes with different NMAs or bitumen types. The physical properties of Pen40/60 and CRMB are shown in  
95 Table 1. The aggregate gradings [18] for these three asphalt concretes are shown in Figure 1. Asphalt mixtures  
96 were compacted by a roller compactor with mixing and compaction procedures of the asphalt mixtures chosen  
97 from BS EN 12697-35: 2016 [19] and BS EN 12697-33: 2019 [20]. Hot mixture is poured into a pre-heated square  
98 mould ( $305 \times 305 \text{ mm}^2$ ) and compacted to the required slab thickness.

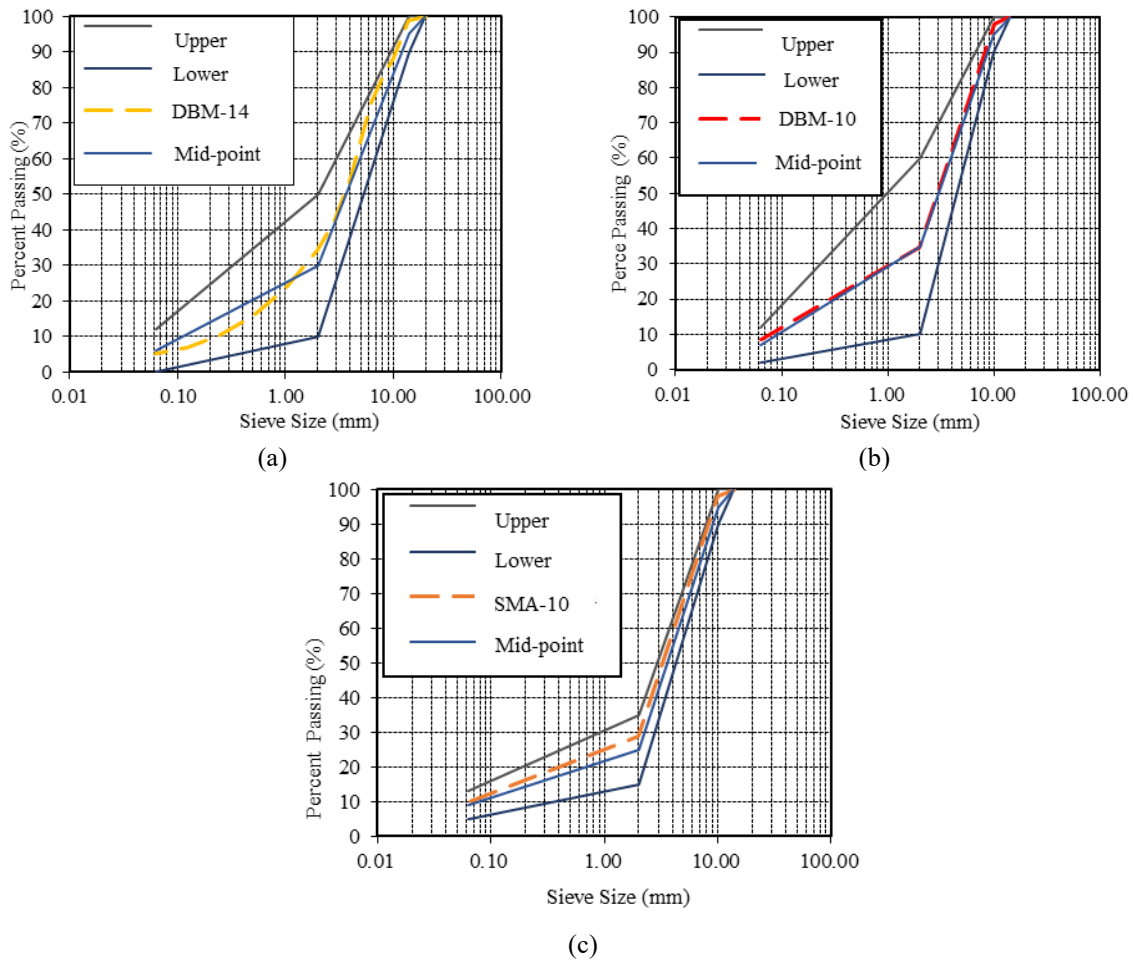
99 For the 2PB test, the DBM -14 + Pen40/60 mixture was used with granite aggregates and the bitumen content was  
100 5% by the weight of the mixture. For the 4PB test, three different mixtures including DBM-14 + CRMB, DBM-  
101 10 + Pen 40/60 and SMA-10 + Pen 40/60 were tested. The hot mixture is poured into pre-heated moulds of  
102 dimensions  $400 \text{ mm} \times 300 \text{ mm}$  and compacted to the required slab thickness of 80 mm to obtain the desired air

103 void content (7%). The target bitumen content was 5.5% for the DBM and 6.1% for the SMA. Prismatic  
 104 rectangular beams were cut out of these slabs with specified dimensions. Regarding the dimensions, the tolerance  
 105 requirements of  $380 \pm 6$  mm in length,  $50 \pm 2$  mm in height, and  $63 \pm 2$  mm in width as required by the AASHTO  
 106 T321 and ASTM D7460-10 were met.

107 Table 1: Properties of bitumen.

Properties	Pen 40/60	CRMB	Test methods
Specific gravity	1.030	1.040	BS 2000-549:2007 [21]
Penetration (25°C, 5 s, 100 g)/0.1 mm	43	33	BS EN 1426:2015 [22]
Softening point/°C	51.1	59.6	BS EN 1427:2015 [23]
Viscosity, 135°C/Pa.s	0.441	1.78	BS EN 13302:2018 [24]

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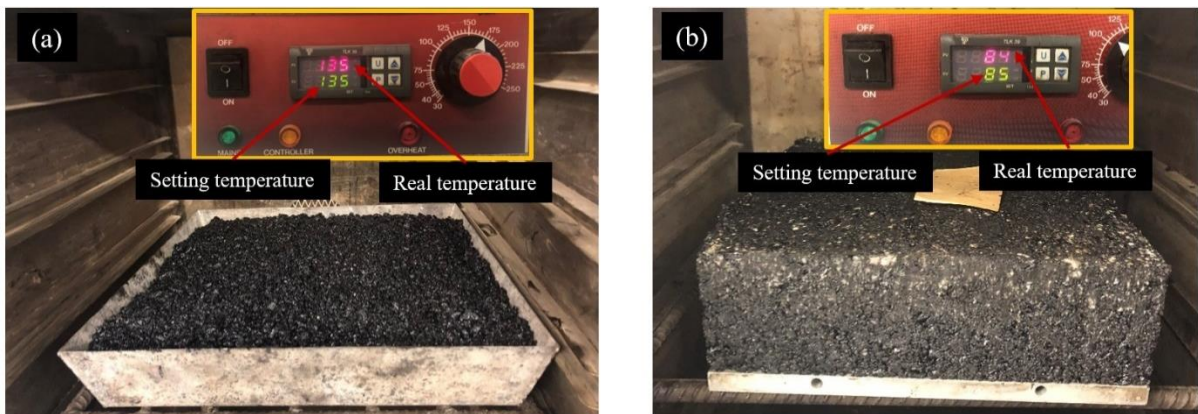


109 Figure 1: Gradations of (a) DBM-14, (b) DBM-10 and (c) SMA-10.

110 2.2 Specimen Fabrication and Test Methods

111 2.2.1 Thermal Oxidative Ageing Tests

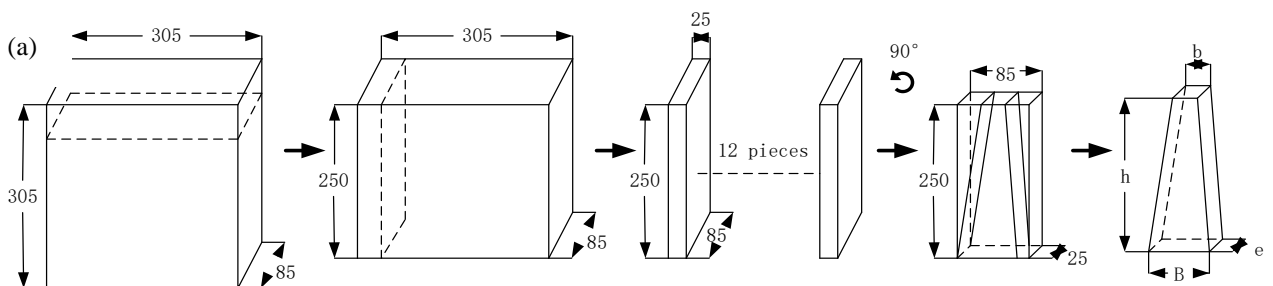
112 The test method for short-term ageing of the asphalt mixture was to put loose material in an oven at  $135 \pm 1 \text{ }^\circ\text{C}$   
113 for 4 hours. For long-term ageing, the compacted specimens were put in an oven at  $85 \pm 3 \text{ }^\circ\text{C}$  for 5 days according  
114 to TS 12697-52:2017 [25]. The oven setting is presented in Figure 2. After the ageing period, the oven is turned  
115 off and left to cool to room temperature and the specimens are then removed. All the specimens were placed in a  
116 temperature-controlled cabinet at  $20 \text{ }^\circ\text{C}$  for at least 6 hours before testing. The materials after short-term and long-  
117 term ageing can be seen in Figures 2 (a) and (b), respectively.



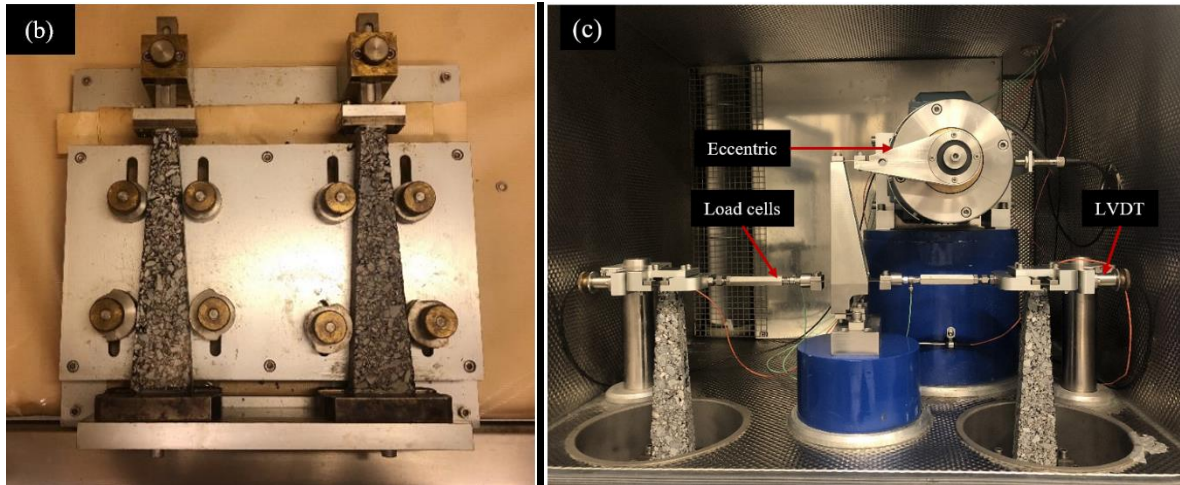
118  
119 Figure 2: Oven setting and the materials after (a) short-term ageing and (b) long-term ageing.

120 2.2.2 Two Point Bending (2PB) Fatigue Test

121 Fatigue tests were carried out using the 2PB equipment. The test consists of applying a continuous sinusoidal  
122 waveform at the top of a trapezoidal specimen. To prepare the samples for 2PB, trapezoidal specimens were  
123 manufactured from slabs with dimensions: 305 mm length, 305 mm width and 85 mm depth. Twelve specimens  
124 were produced from one slab using the process shown in Figure 3 (a). The dimensions required for a 14 mm size  
125 DBM are taken from BS EN 12697-24: 2018 [26]. The samples were glued to metal plates using epoxy resin to  
126 enable them to be fitted to the test apparatus (see Figure 3(b)). The bottom base is fixed so that the specimen is  
127 mounted as a vertical cantilever, allowing a sinusoidal displacement to be applied to the top of the specimen.



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130 Figure 3: Sample preparation and setting of 2PB: (a) schematic diagram of preparation of trapezoidal specimens;  
 131 (b) gluing procedure of samples and (c) setting procedure of samples.

132 The 2PB tests were conducted under strain-controlled mode using sinusoidal loading. The specimens are subjected  
 133 to different strain levels - 150, 200 and 250  $\mu\epsilon$  at different temperatures - 0, 5, 10 and 20  $^{\circ}\text{C}$  at a frequency of 15  
 134 Hz. All samples were conditioned for 4 hours before the start of the tests. According to the standard [26], the  
 135 sample stiffness at the 100<sup>th</sup> cycle is defined as the initial stiffness. In this study, the failure criteria were studied  
 136 at different reductions of the initial stiffness value. Therefore, the test was set to stop once the stiffness reach 10%  
 137 of the initial stiffness value. The equipment tests two specimens at each strain level. Figure 3 (c) shows the test  
 138 equipment and the samples ready for testing.

139 2.2.3 Four Point Bending (4PB) Fatigue Test

140 The 4PB test was applied to prove the applicability of the SCTM in another test method. The test consists of  
 141 applying a continuous sinusoidal waveform at a constant frequency of 15 Hz at the top of a prismatic specimen  
 142 by means of two loading points. The specimen is restrained at four points by means of four clamps. Free translation  
 143 and rotations are allowed at all load and reactions points. Deformation is calculated by means of three Linear  
 144 Variable Differential Transformers (LVDTs) located at three different points (Figure 4). The beams were tested  
 145 under strain-controlled mode at a strain level of 270  $\mu\epsilon$ . The thermostatic chamber and the loading equipment  
 146 were set at the test temperature, 10  $^{\circ}\text{C}$  for at least 3 hours prior to the start of the experiment. The fatigue test was  
 147 conducted according to BS EN 12697-24: 2012.



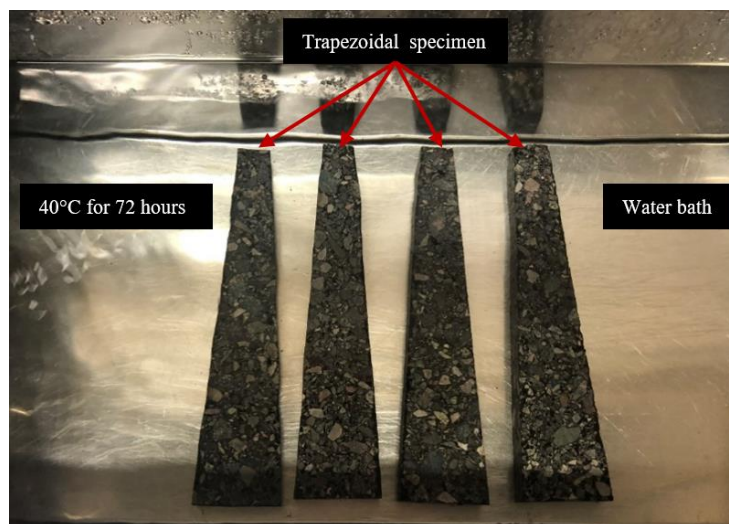
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Figure 4: Setup and sketch of 4PB test.

150 2.2.4 Moisture Damage Tests

151 The sensitivity test used in this research is that described in BS EN 12697-23:2017 [27]. The vacuum procedure  
 152 was conducted under a residual pressure of  $6.7 \pm 0.3$  kPa for  $30 \pm 5$  minutes before the specimens were placed in  
 153 a water bath. A thermostatically controlled water bath is capable of maintaining the prescribed conditioning  
 154 temperature in the vicinity of the specimens. The bath is equipped with a perforated shelf, placed on spacers above  
 155 the bottom. The capacity of the bath is such that the test specimens can be stored with the upper surface more than  
 156 20 mm below the water surface. The specimens were put in a water bath at  $40 \pm 2$  °C for a period of  $72 \pm 2$  hours,  
 157 as shown in Figure 5.



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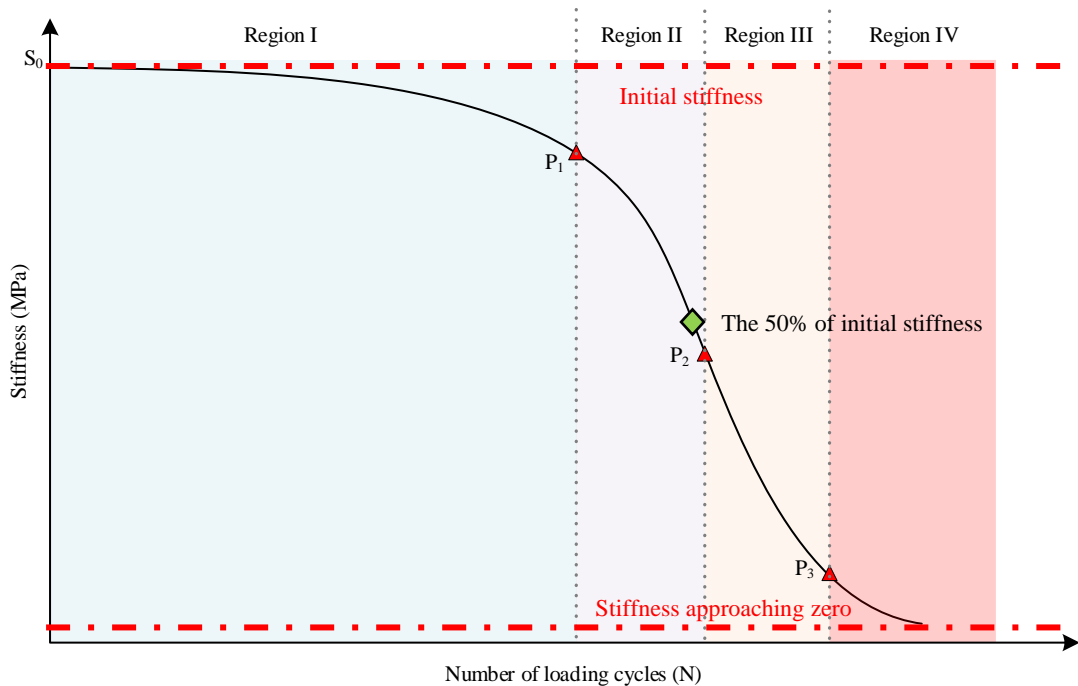
Figure 5: Water bath for moisture conditioning of specimens.

160 2.3 Stiffness Change Tendency Method (SCTM)

161 A typical plot of the “S” shape stiffness reduction curve is shown in Figure 6. Four stages are defined by

162 researchers based on three transition points (P1, P2 and P3), relating to different damage progression [8-9, 28].  
 163 Prior to P1, the asphalt stiffness is high and decreases in a relatively stable manner from the initial stiffness. After  
 164 P1, the reduction in stiffness rapidly increases until the second transition point P2, at P2 the rate of loss of stiffness  
 165 is maximum. Following P2 the reduction in stiffness decreases and the stiffness approaches zero after P3. In  
 166 regions III and IV, the road would probably already have failed. The SCTM is developed to model stiffness change  
 167 and determine transition points.

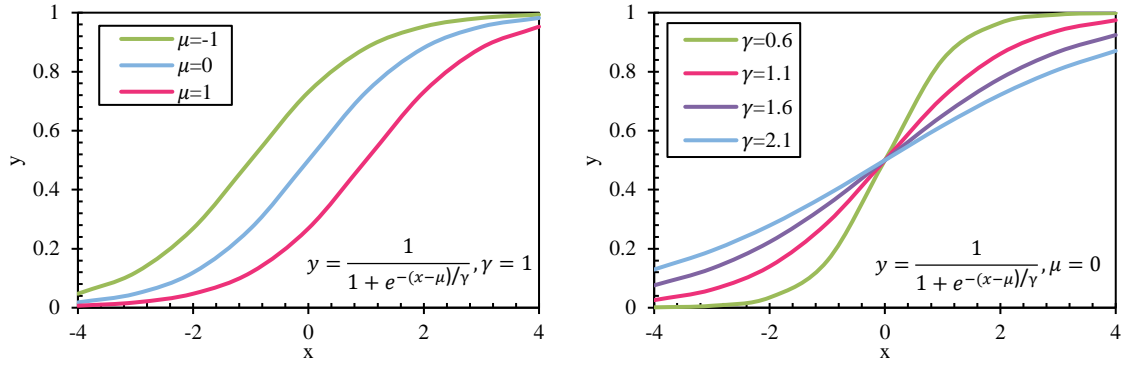
168 The typical logistic curves and the base logistic equation are shown in Figure 7; two asymptotes 0 and 1 can be  
 169 seen. Changing the value of  $\mu$  shifts the function in the x direction; changing the value of  $\gamma$  impacts the gradient  
 170 of the function [29]. It can be seen that the shape of the typical curves in Figure 7 are similar to the stiffness  
 171 reduction curve in Figure 6. Therefore, the logistic model is used to determine the relationship between the log of  
 172 the stiffness ( $E$ ) and the log of the number of cycles ( $N$ ) by transforming the base logistic equation to obtain  
 173 Equation 1. The constants A, B and C are determined using regression methods in the SAS software. Two  
 174 asymptotes A1 and A2 can be seen in Figure 8. A1 can be seen to represent the initial stiffness when the sample  
 175 deformation is highly elastic. A2 represents the stiffness as the number of cycles tends to infinity and the sample  
 176 fails. The second derivative of the logistic model represents the changing rate of the gradient of the function,  
 177 therefore it can be used to calculate the exact transition points P1, P2 and P3 [30-31]. These are represented by  
 178 the maximum, minimum and equilibrium of the second derivative of Equation 1 ( $\lg E''$ ), as shown in Figure 8.



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180 Figure 6: The four regions of a stiffness reduction curve, based on three transition points.





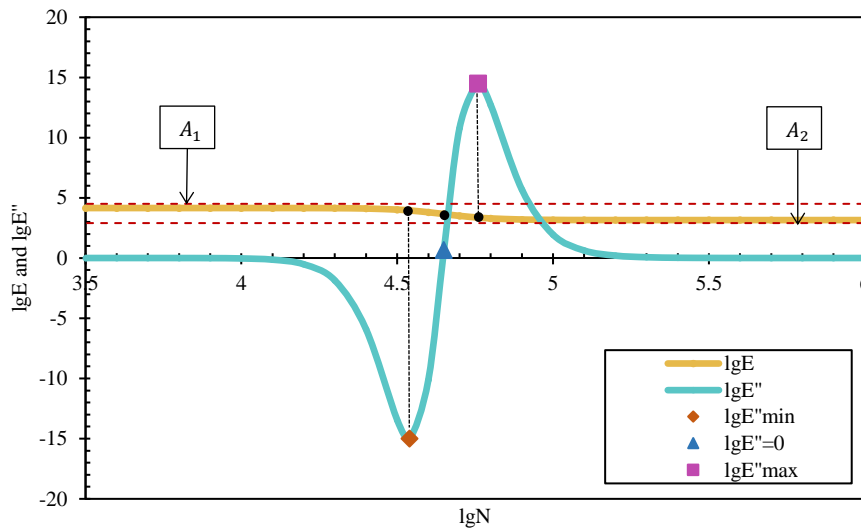
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Figure 7: The typical curves of the logistic regression.

$$\lg E = A - \frac{(\lg N)^B}{C^B + (\lg N)^B} \quad (1)$$

183 where  $E$  is the stiffness (MPa),  $N$  is the number of cycles,  $A$ ,  $B$  and  $C$  are constants.



184

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Figure 8: Fitting logistic equation and its second derivative.

#### 186 2.4 Energy-based Approaches

187 The ER and RDEC approaches are applied to validate the results obtained from the SCTM. The data collected at  
 188 10 °C under 150  $\mu\epsilon$  is chosen as an example; the phenomenological stiffness ( $E$ ) versus the number of cycles ( $N$ )  
 189 on logarithmic scales is applied to show the position of  $N_{f50}$  as a comparison (Figure 12).

190 The basis of the ER and RDEC approaches is to calculate the dissipated energy cycle per unit volume ( $W$ ), shown  
 191 in Equation 2 [32]. It is known that an asphalt mixture is a viscoelastic material, therefore its stress-strain curve  
 192 under repeated loading is non-linear. The loading and unloading paths do not overlap, instead they form loops,

193 known as hysteresis loops [3]. The area of the hysteresis loop is defined as the dissipated energy [2].

$$W = \pi \sigma \varepsilon \sin \varphi, \quad (2)$$

194 where  $\sigma$  is the stress level in each cycle;  $\varepsilon$  is the strain level in each cycle and  $\varphi$  is the phase angle in each cycle.

195 For asphalt mixtures under strain-controlled loading mode, the dissipated energy evolution will follow three stages  
 196 connected by two transitions [17], as shown in Figure 9. The first stage is considered as a no damage or minimal  
 197 damage region. The ER approach was used to calculate the value of the first transition point ( $N_1$ ).  $N_1$ , is defined  
 198 as the point of crack initiation [4], at this point the ER changes from a good linear relationship to a significantly  
 199 noisier relationship, as shown in Figure 10. ER is the ratio between the initial dissipated energy ( $W_0$ ) multiplied  
 200 by the cycle number ( $n$ ) and the dissipated energy in the  $n^{th}$  cycle ( $W_n$ ), shown in Equation 3 [8].

$$ER = \frac{nW_0}{W_n} = \frac{n(\pi\sigma_0\varepsilon_0 \sin \varphi_0)}{(\pi\sigma_n\varepsilon_n \sin \varphi_n)}, \quad (3)$$

201 where  $\sigma_0$ ,  $\varepsilon_0$  and  $\varphi_0$  are stress, strain and phase angle in the initial cycle and  $\sigma_n$ ,  $\varepsilon_n$  and  $\varphi_n$  are stress, strain and  
 202 phase angle in the  $n^{th}$  cycle.

203 After the first transition point, the dissipated energy curve enters the second stage (the plateau stage) where the  
 204 dissipated energy drops at a relatively constant rate, until the third stage where the dissipated energy curve has a  
 205 significantly different slope (Figure 9). The transition point between the second and the third stage is related to  
 206 macrocrack propagation [11]. Ghuzlan and Carpenter [11] used the RDEC approach to define  $N_{fm}$  as the  
 207 macroscopic failure, according to Figure 11 and Equation 4. The RDEC approach was refined by Shen and  
 208 Carpenter [17] to calculate  $N_{fm}$ , the cycle number corresponding to the maximum  $\Delta A$  (Equations 5-7).

$$RDEC = \frac{W_{n+1} - W_n}{W_n}, \quad (4)$$

209 where RDEC is the ratio of the dissipated energy change per load cycle,  $W_n$  is the dissipated energy produced in  
 210 load cycle  $n$  and  $W_{n+1}$  is the dissipated energy produced in load cycle  $n + 1$ .

$$\Delta A = |AT - SDE|, \quad (5)$$

$$SDE = \sum_0^n DE, \quad (6)$$

$$AT = \frac{(DE_0 + DE_i) \times n}{2}, \quad (7)$$

211 where  $n$  is the cycle number,  $DE_0$  is the initial dissipated energy and  $DE_i$  is the dissipated energy at cycle  $i$ . SDE  
 212 is the area under the dissipated energy curve,  $AT$  is the trapezoidal area and  $\Delta A$  is the difference between SDE and  
 213  $AT$ , as shown in Figure 9.

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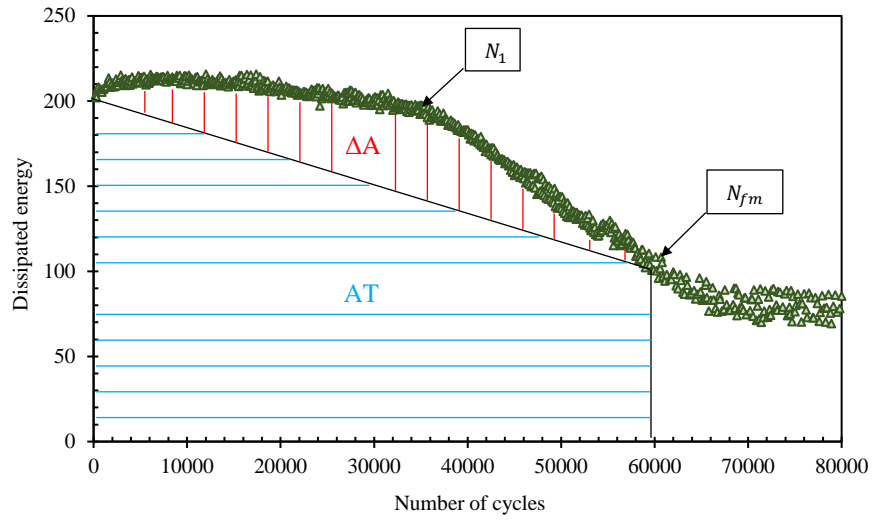
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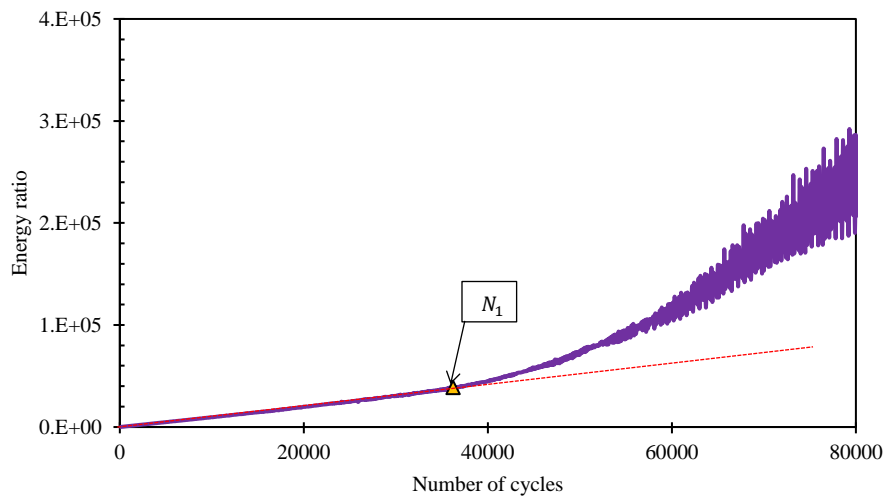
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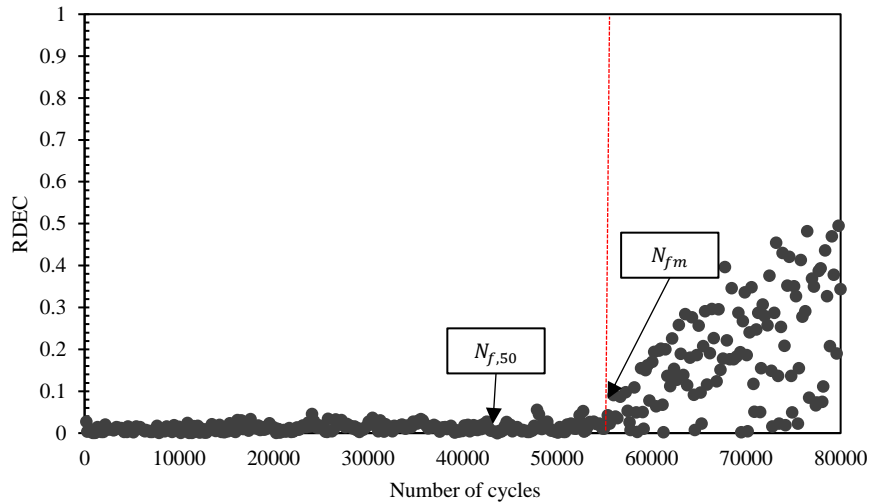
Figure 9: Dissipated energy versus number of load cycles (10°C, 150µε).



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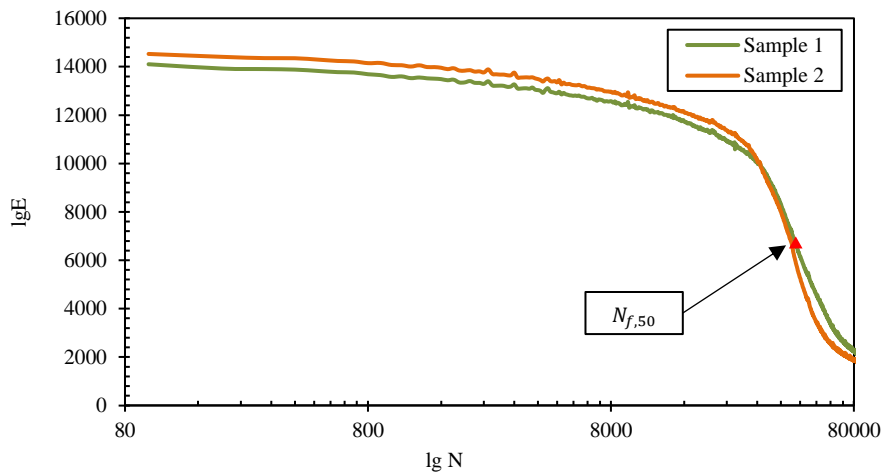
Figure 10: ER versus number of load cycles (10 °C, 150 µε).



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Figure 11: RDEC versus number of load cycles (10 °C, 150  $\mu\epsilon$ ).



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Figure 12:  $\lg E$  versus  $\lg N$  (10 °C, 150  $\mu\epsilon$ ).

### 229 3 Results and Discussion

#### 230 3.1 Stiffness Change Tendency Method (SCTM)

231 The test design mainly considers the applicability of the SCTM for different conditions, including aged and  
 232 moisture damaged, different test methods, different test temperatures and strains, and different types of mixtures,  
 233 as shown in Table 2. Samples were subjected to four repeated tests per test condition; with the results being  
 234 averaged. Based on the data points (shown in blue in Figures 13-14) obtained from the tests, a logistical model of  
 235 the form in Equation 1 is fitted to the data (shown in red Figures 13-14) with the constants A, B and C being  
 236 determined using SAS software. The constants determined using the SAS software are shown in Table 2 alongside  
 237 their  $R^2$  values. The  $R^2$  values show that in all cases there is good agreement between the laboratory data and the  
 238 fitted logistic model, confirming that the logistic model it is a suitable model to model the stiffness reduction  
 239 curve. Visual comparison of Figures 13-14 shows the equation does not quite match at the beginning of some

240 curves. The initial drop in stiffness at the beginning of the stiffness reduction curves, which may be partly due to  
 241 temperature increase, are not considered by the equation. The derivatives of the logistic model are shown in Table  
 242 3 alongside the Stiffness Ratio ( $SR = E/S_0$ ). This illustrates that the stiffness ratio at the three transition points  
 243 is similar across all samples. P1 occurs between 48% and 68% of the initial stiffness ( $S_0$ ). The stiffness ratio at P2  
 244 ranges between 22% and 35% of  $S_0$ , and at P3 ranges between 12% and 17% of  $S_0$ .

245 Based on the fitted logistic model, three transition points (P1, P2 & P3 as shown in Figures 13-14) associated with  
 246 fatigue progression were determined to characterise the different fatigue damage stages. The stiffness reduction  
 247 in region I (prior to P1) is observed to be slow. The stiffness drops sharply during region II (between P1 and P2),  
 248 after region II the rate of loss of stiffness decreases until P3.

249 The values of the stiffness and corresponding number of cycles at each failure point is affected by factors such as  
 250 temperature, ageing and moisture damage. Seen from Table 3 (sample sets 1-4), under the same strain level, at the  
 251 higher the temperature, the later the failure point occurs.  $E_{p1}$ ,  $E_{p2}$  and  $E_{p3}$  happen at 52%, 22% and 12% of  $S_0$  at  
 252 20°C, respectively, while at 0 °C, they happen at 68%, 31% and 17% of  $S_0$ , respectively. Similarly, seen from  
 253 Figures 13 (a) and (b), the higher the temperature, the greater the corresponding number of cycles at the failure  
 254 point.  $N_{p1}$ ,  $N_{p2}$  and  $N_{p3}$  at 20 °C are 55000, 72500 and 93400, while at 0°C, these are 22400, 41100 and 72500,  
 255 respectively. Figures 13 (c) and (d) show that fatigue performance deteriorates after ageing and moisture damage  
 256 compared to undamaged condition. Moisture damage has a greater impact on fatigue performance with aged  
 257 samples experiencing 35500 cycles while moisture damage samples experienced 19100 cycles at 15% initial  
 258 stiffness. However, the value of P1 for the moisture damaged samples occurs at a lower stiffness value compared  
 259 to the aged samples.

260 The fitting results are presented in Figure 14 alongside the data obtained for the 4PB tests. As with the 2PB, a  
 261 good likeness is observed and the  $R^2$  values of all fitted models are close to unity, demonstrating that the SCTM  
 262 is applicable to be used to model stiffness development of different asphalt mixtures and different test methods.  
 263 The 4PB fatigue tests of the three bituminous mixtures in Figure 14 were stopped at 20% of the initial stiffness of  
 264 the samples, however, the value of  $N_{p3}$  can still be predicted using the SCTM. This proves that the SCTM is able  
 265 to predict fatigue failure points and remaining fatigue life as long as part of the fatigue test data is known, removing  
 266 the need for time intensive laboratory testing.

267 Table 2: Fitting results for the Logistic model.

Sample set	Test Method	Bitumen	Mixture type	Temperature (°C)	Strain ( $\mu\epsilon$ )	Condition	Logistic model constants			$R^2$
							A	B	C	
1	2PB	Pen 40/60	DBM-14	0	200	unaged	4.32	23.55	4.63	0.94
2	2PB	Pen 40/60	DBM-14	5	200	unaged	4.23	46.4	4.27	0.88

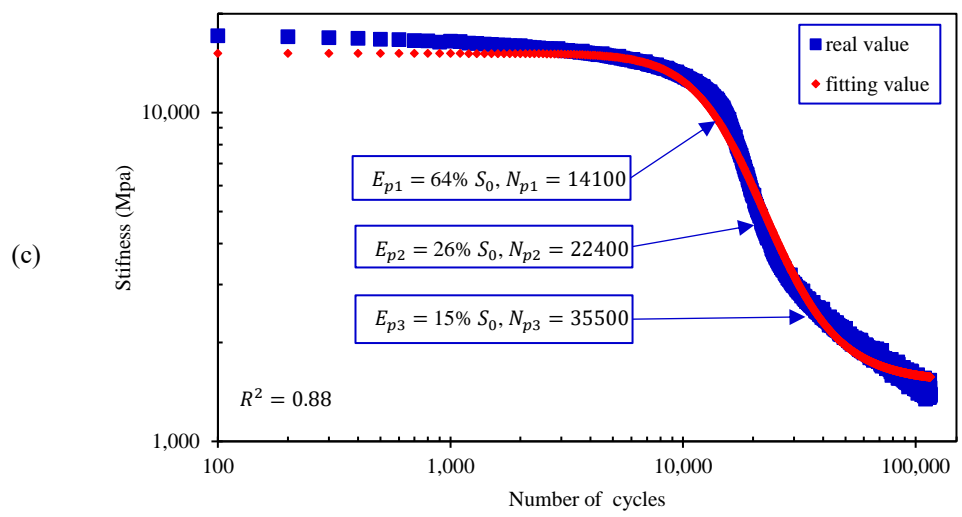
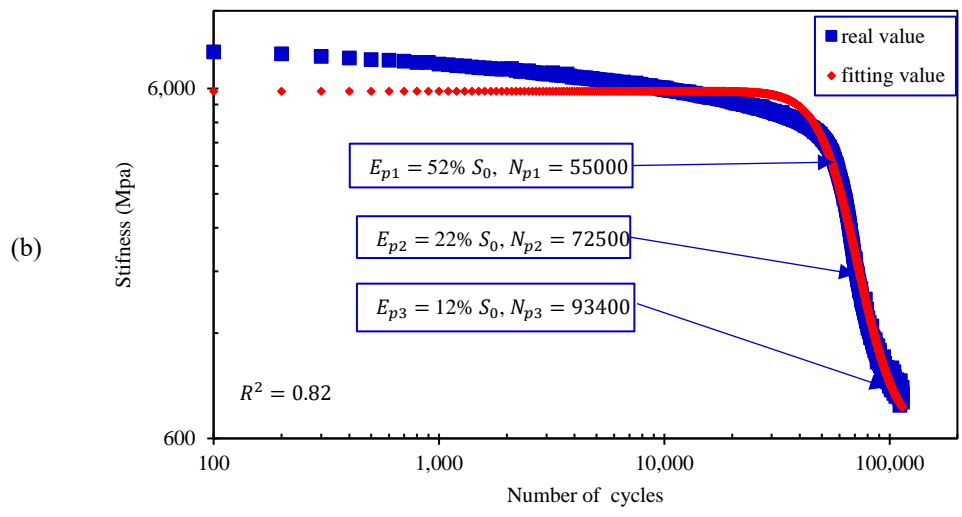
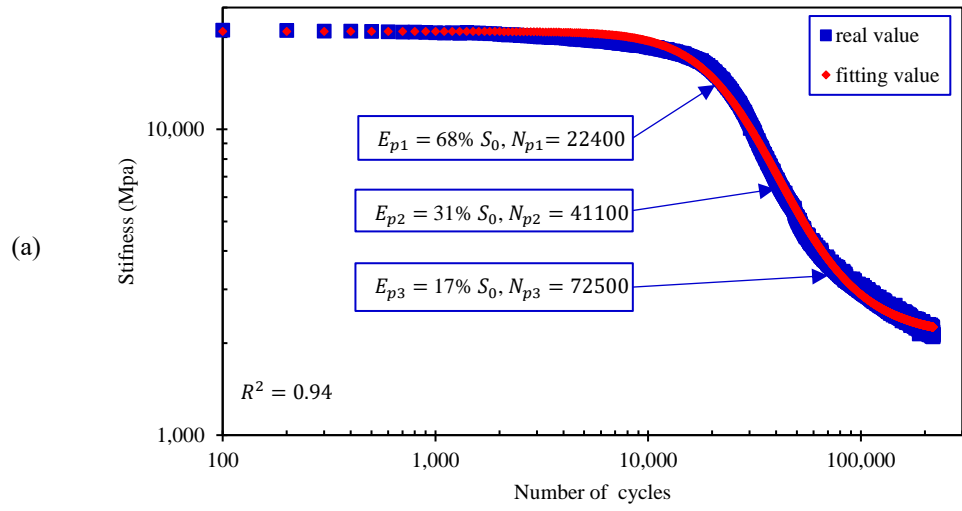
3	2PB	Pen 40/60	DBM-14	10	200	unaged	4.15	57.57	4.65	0.86
4	2PB	Pen 40/60	DBM-14	20	200	unaged	3.77	56.1	4.86	0.82
5	2PB	Pen 40/60	DBM-14	10	150	unaged	4.1	37.96	4.77	0.91
6	2PB	Pen 40/60	DBM-14	10	250	unaged	4	57.41	4.33	0.80
7	2PB	Pen 40/60	DBM-14	10	200	moisture damaged	4.06	39.91	4.15	0.86
8	2PB	Pen 40/60	DBM-14	10	200	aged	4.18	27.74	4.36	0.88
9	4PB	Pen 40/60	DEM-10	10	270	unaged	4.06	19.10	3.90	0.88
10	4PB	CRMB	DEM-10	10	270	unaged	4.11	16.42	3.85	0.87
11	4PB	Pen 40/60	SMA-10	10	270	unaged	4.08	22.00	3.93	0.89

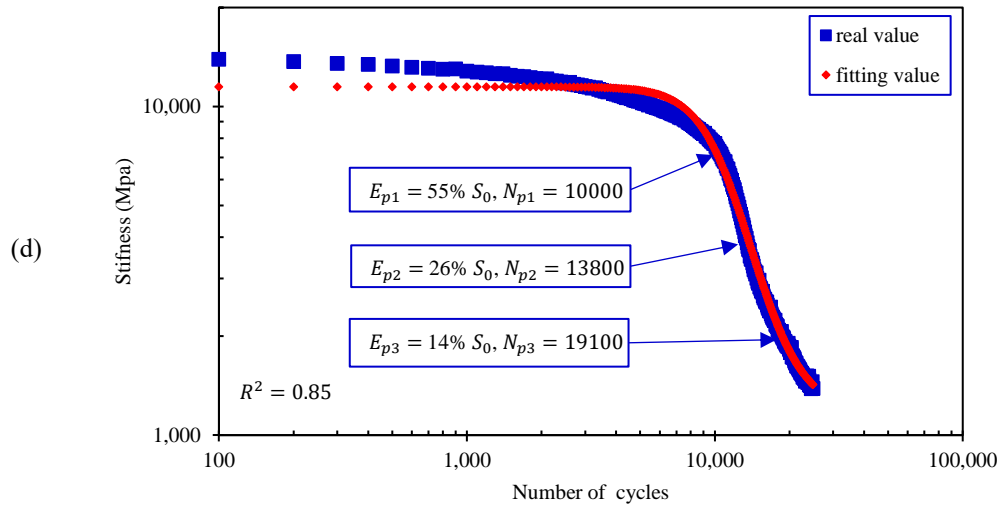
268

Table 3: Results of second derivative function and the corresponding stiffness ratio (SR).

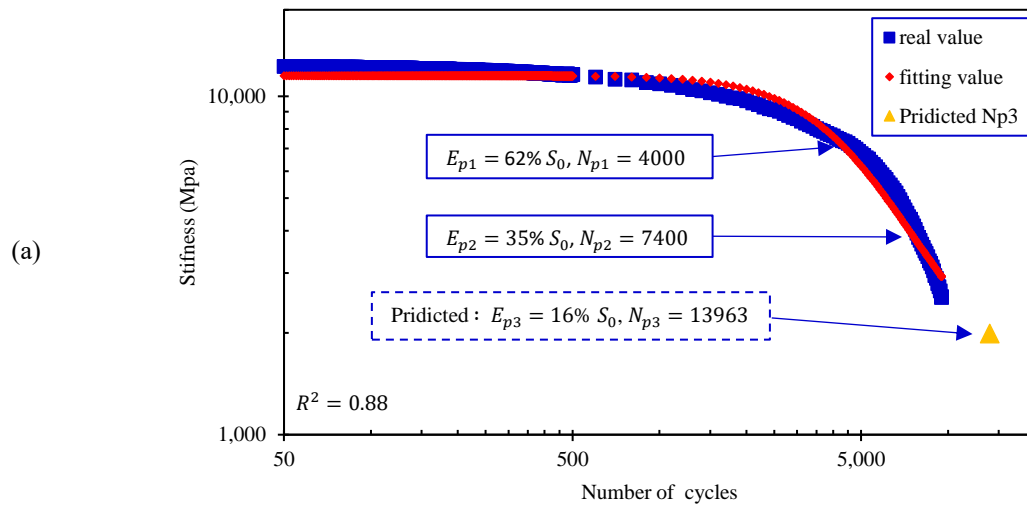
Sample set	$lgE''_{min}$	$lgE'' = 0$	$lgE''_{max}$	$E_{p1}$ (MPa)	$E_{p2}$ (MPa)	$E_{p3}$ (MPa)	$S_0$ (MPa)	SR <sub>p1</sub>	SR <sub>p2</sub>	SR <sub>p3</sub>
1	4.35	4.61	4.86	14447	6632	3686	21099	68%	31%	17%
2	4.15	4.27	4.38	11189	5015	2825	18868	59%	27%	15%
3	4.54	4.65	4.76	9367	4225	2215	17105	55%	25%	13%
4	4.74	4.86	4.97	3965	1700	935	7633	52%	22%	12%
5	4.6	4.76	4.93	8340	4333	2039	14102	59%	31%	15%
6	4.23	4.33	4.43	6512	3608	1646	13667	48%	26%	12%
7	4.00	4.14	4.28	10900	4395	2592	17126	55%	26%	14%
8	4.15	4.35	4.55	7727	3654	1933	13931	64%	26%	15%
9	3.59	3.87	4.15	7623	4327	1986	12234	62%	35%	16%
10	3.52	3.83	4.14	8018	4169	2202	13343	60%	31%	17%
11	3.68	3.91	4.15	8196	4287	2063	12649	65%	34%	16%

269

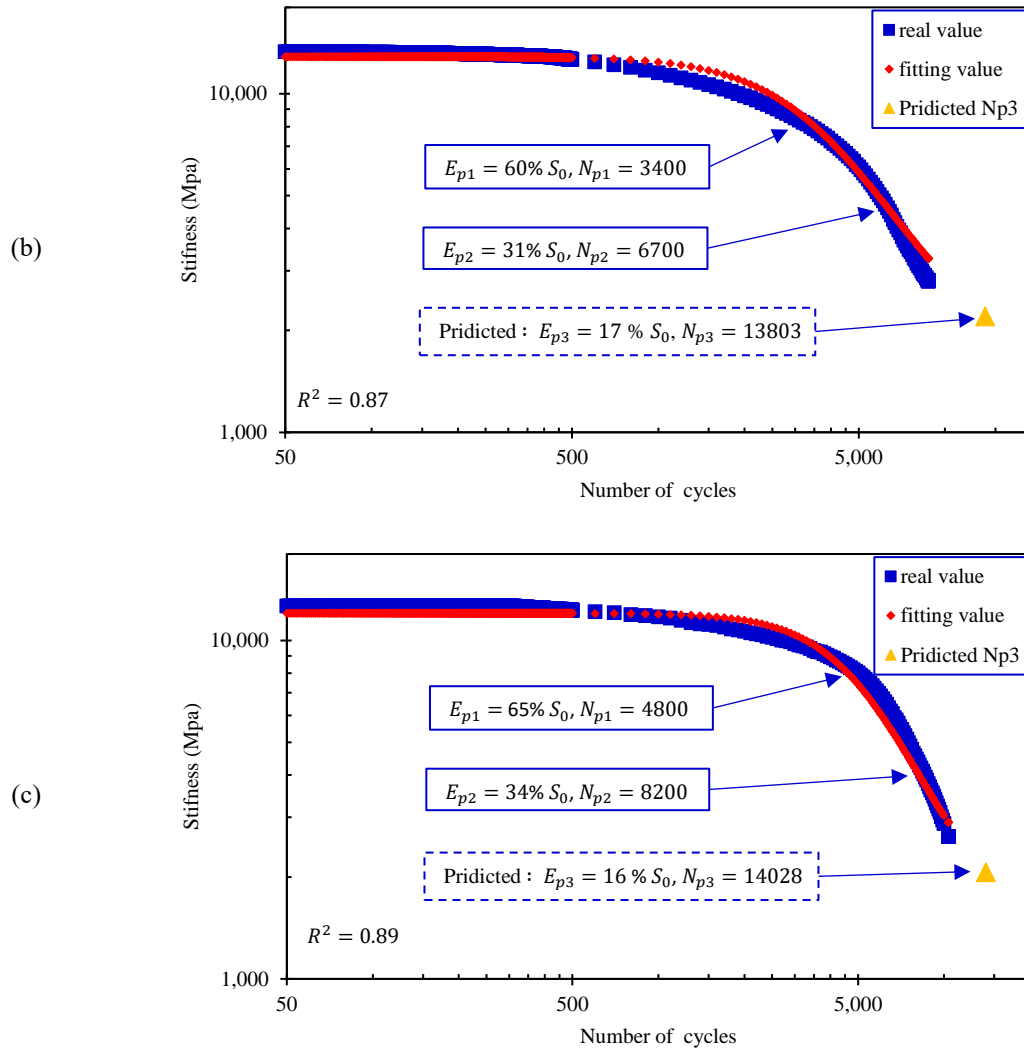




270 Figure 13: Fitting results under different temperatures, strains and conditions of 2PB tests: (a) 0 °C and 200 με;  
 271 (b) 20 °C and 200 με; (c) 10 °C and 200 με aged; (d) 10 °C and 200 με moisture damaged.

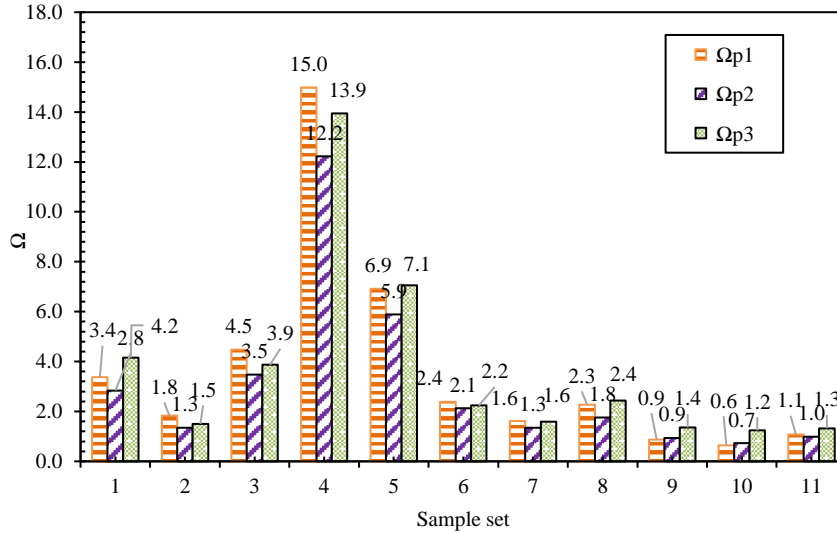






272 Figure 14: Fitting results of different asphalt mixtures at 10 °C under 270  $\mu\epsilon$  of 4PB tests (a) DBM-  
 273 14+Pen40/60; (b) DBM-10+ CRMB; (c) SMA-10+Pen40/60.

274 Unlike the traditional 50% initial stiffness, the transition point happens randomly in the stiffness reduction curve,  
 275 therefore, it is necessary to evaluate the fatigue performance with regards to the unit stiffness loss. The quotient  
 276 of the number of cycles and the stiffness loss,  $\Omega = \frac{N}{(S_0 - E)^2}$ , was calculated to determine the number of cycles lost  
 277 per unit stiffness. In other words,  $\Omega$  expresses how easily stiffness is lost and can be used to evaluate fatigue  
 278 performance; the more cycles required for a unit stiffness loss, the better the fatigue performance. As shown in  
 279 the Figure 15, for most sample sets,  $\Omega_{p2}$  is the smallest, which indicates that stiffness loss is greater at P2, the  
 280 stiffness reduction curve also shows that the stiffness decreases quickest near P2. Generally,  $\Omega$  decreased as the  
 281 temperature increased and the strain decreased. For 2PB tests, fatigue performance under 200  $\mu\epsilon$  at 20 °C is the  
 282 best as  $\Omega$  is the largest, whereas fatigue performance after moisture damage is the worst and has the smallest  $\Omega$   
 283 value.  $\Omega$  value of 4PB tests are smaller than that of 2PB tests, fatigue performance of SMA-10+Pen40/60 is the  
 284 best as  $\Omega$  is the largest.



285

286

Figure 15: The number of cycles with respect to the stiffness loss,  $\Omega$ , at P1, P2 and P3.

287

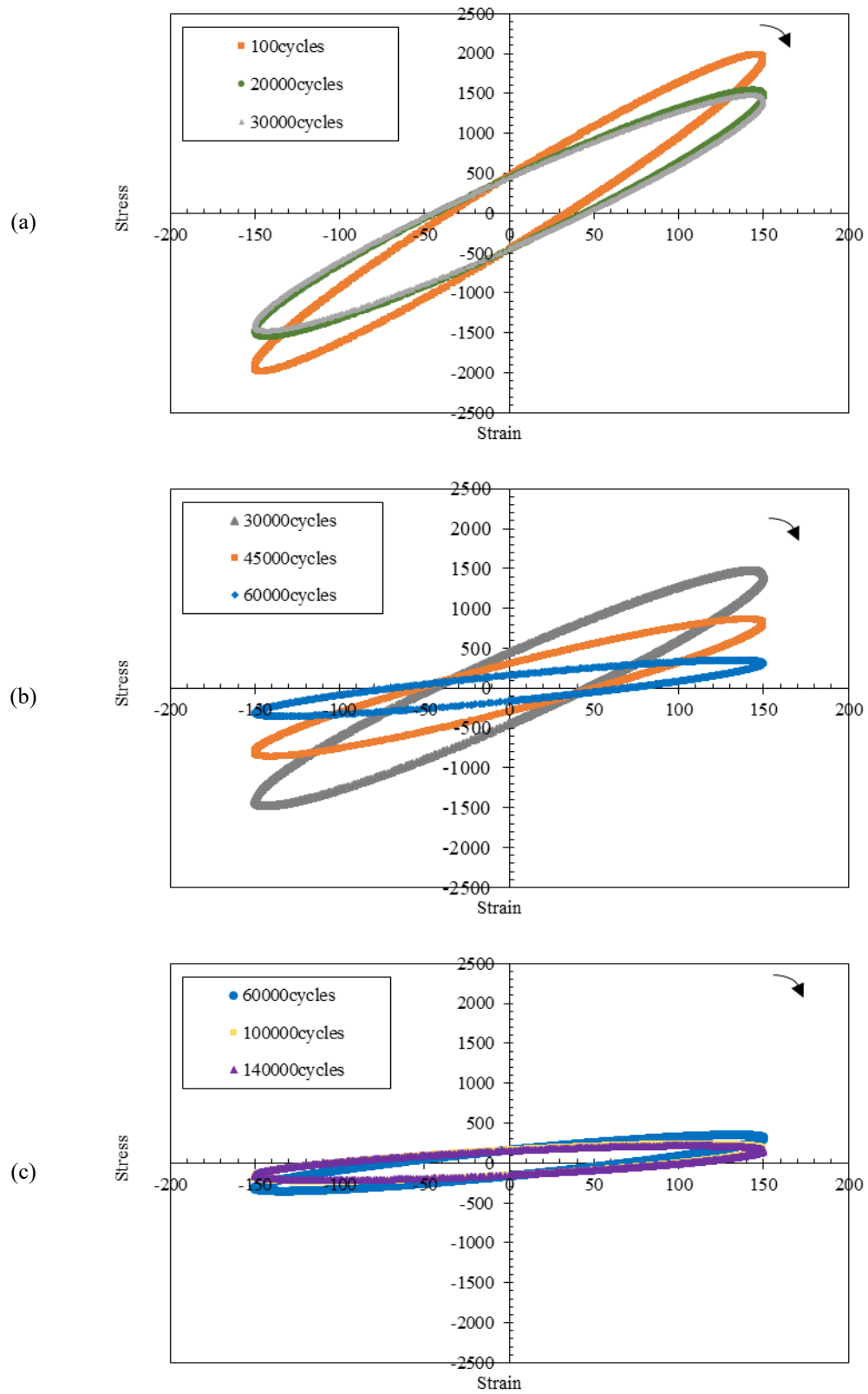
### 3.2 Energy-based Fatigue Failure Criteria

288

#### 3.2.1 Dissipated Energy Approach

289

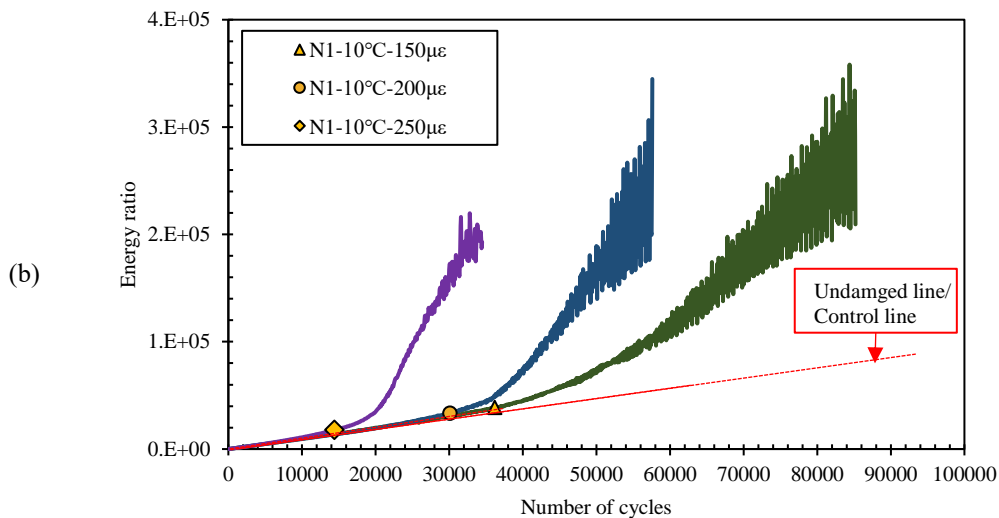
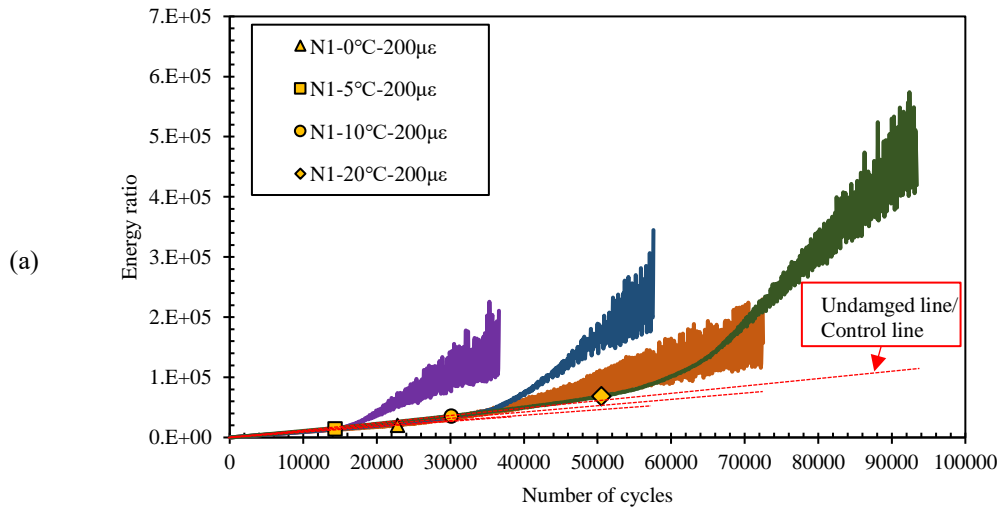
Hysteresis loops were calculated to explore the energy change law for different number of cycles in a strain-  
 290 controlled mode. The damage accumulates and the dissipated energy decreases as the number of loading cycles  
 291 is increased. The first stage is the no damage or minimal damage region, during which the material's stress-strain  
 292 hysteresis loop remains similar and there is no apparent change in dissipated energy as seen in Figure 16a. In the  
 293 second stage, the dissipated energy steadily decreases, which is reflected in the clockwise movement and decrease  
 294 in area of the stress-strain hysteresis loop (Figure 16b). Most fatigue failure criteria are defined at this stage as it  
 295 is where macro damage happens. In the third stage, the change in dissipated energy (the stress-strain hysteresis  
 296 loop) once again becomes minimal; however the area of the stress-strain hysteresis loop decreased significantly  
 297 compared to the stage 1 (Figure 16c).

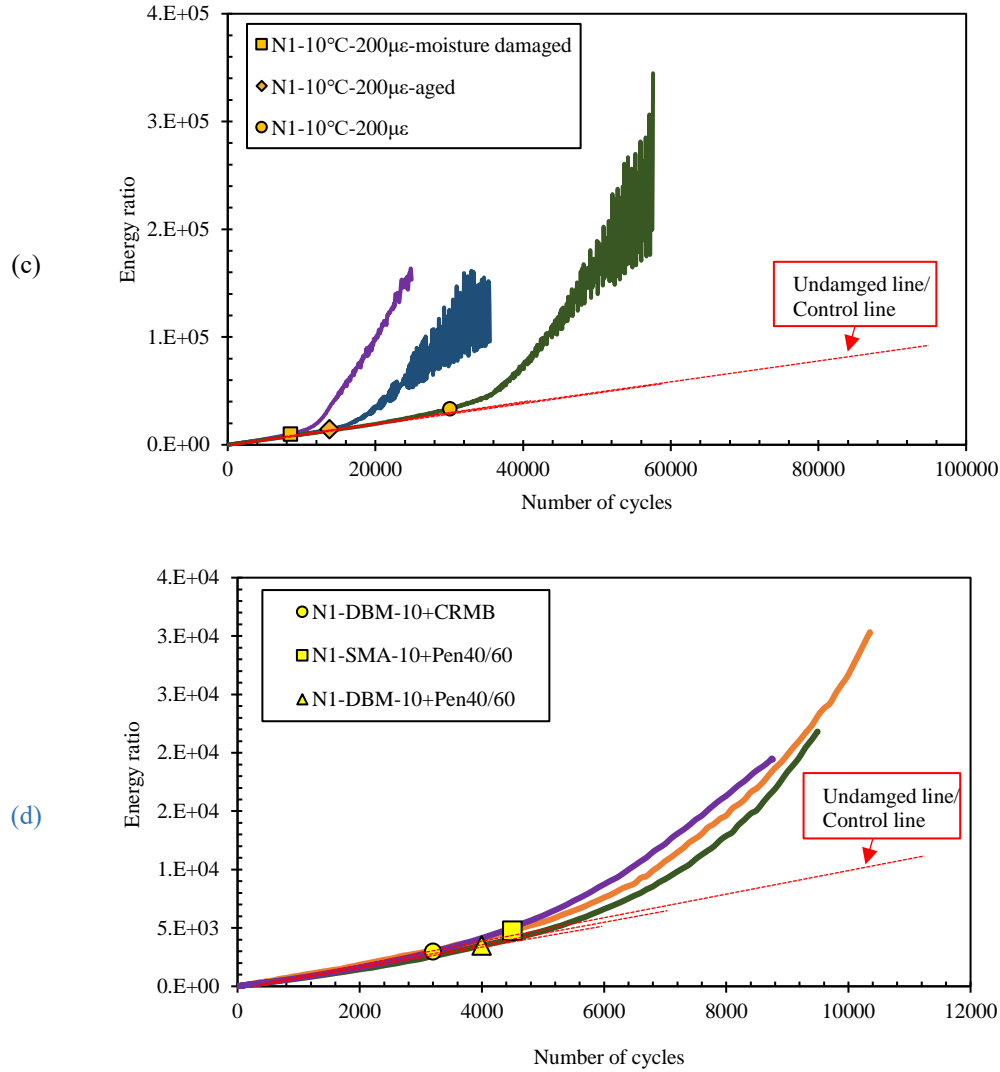


298 Figure 16: Stress-strain hysteresis loops change (10 °C, 150  $\mu\epsilon$ ) for controlled strain 2PB test: (a) first stage; (b)  
 299 second stage and (c) third stage.

300 3.2.2 Energy Ratio (ER) Approach

301 Seen in the plot of ER versus number of cycles in the strain-controlled mode (Figure 17), the ER is initially a  
 302 straight line with a rapid increase after a given number of cycles.  $N_1$  is the inflection point where the ER curve  
 303 deviates from the linear section (undamaged line in Figure 17). The value of  $N_1$  is affected by factors including  
 304 temperature, strain, ageing and moisture damage. It can be seen from Figure 17a-c that  $N_1$  generally increases as  
 305 the temperature increases and decreases with strain increase. Ageing and moisture damage caused different  
 306 reduction rates of  $N_1$ , but the impact of moisture damage is more significant. Figure 17d shows ER versus number  
 307 of load cycles for controlled strain 4PB test of different mixtures. The  $N_1$  of SMA-10+ Pen40/60 is the largest  
 308 whereas the  $N_1$  of DBM-10+ CRMB is the smallest, showing that the value of  $N_1$  is also affected by the bitumen  
 309 type and mixture type.  $N_1$  calculated under all conditions gave a similar undamaged control line thereby  
 310 confirming that the control line of  $ER$  vs  $N$  curves is independent of all the factors investigated.

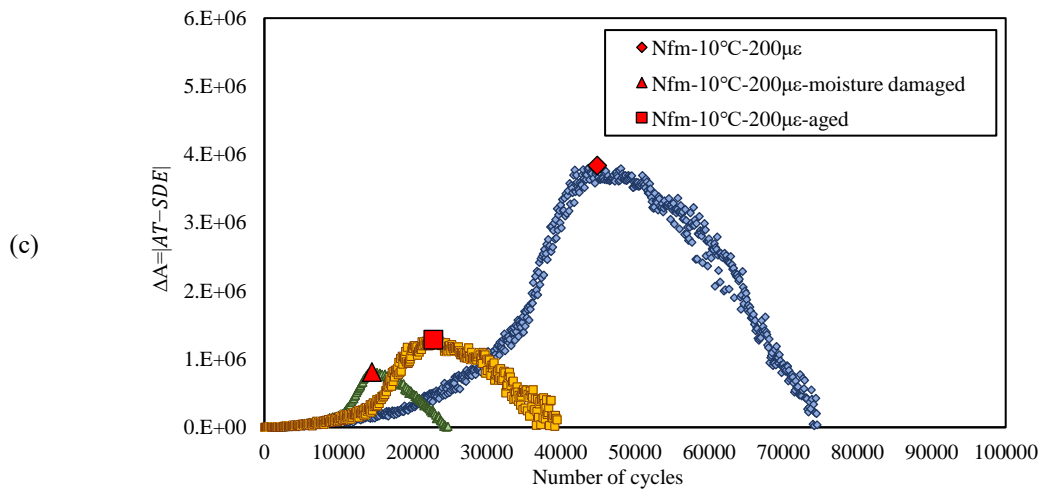
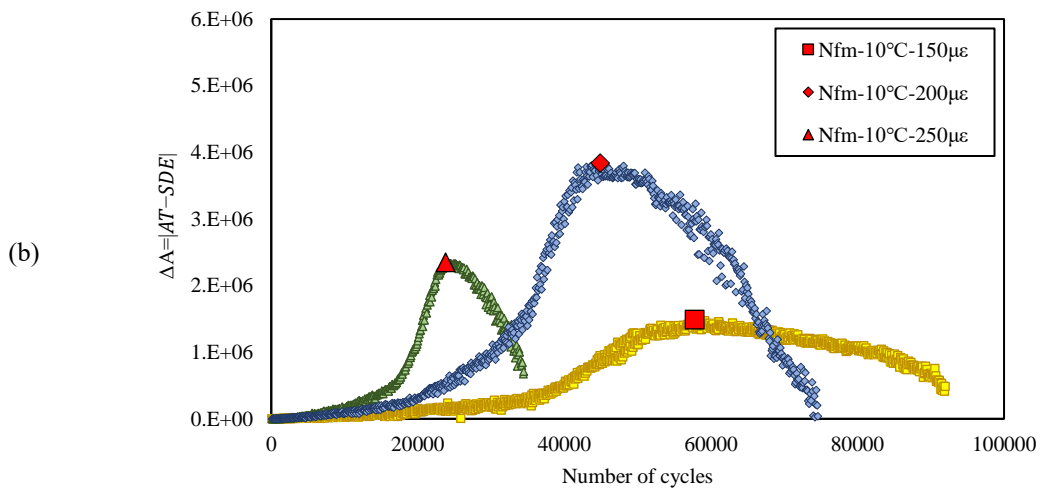
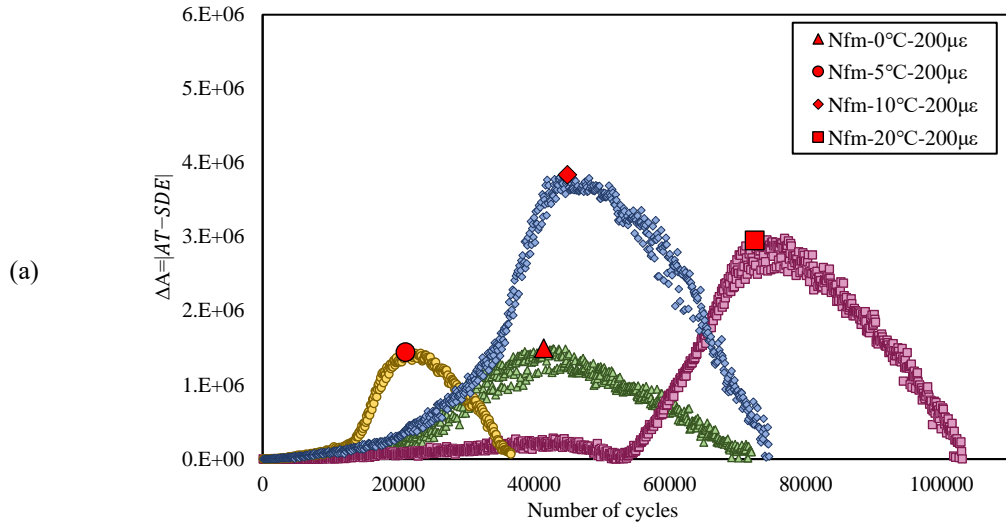


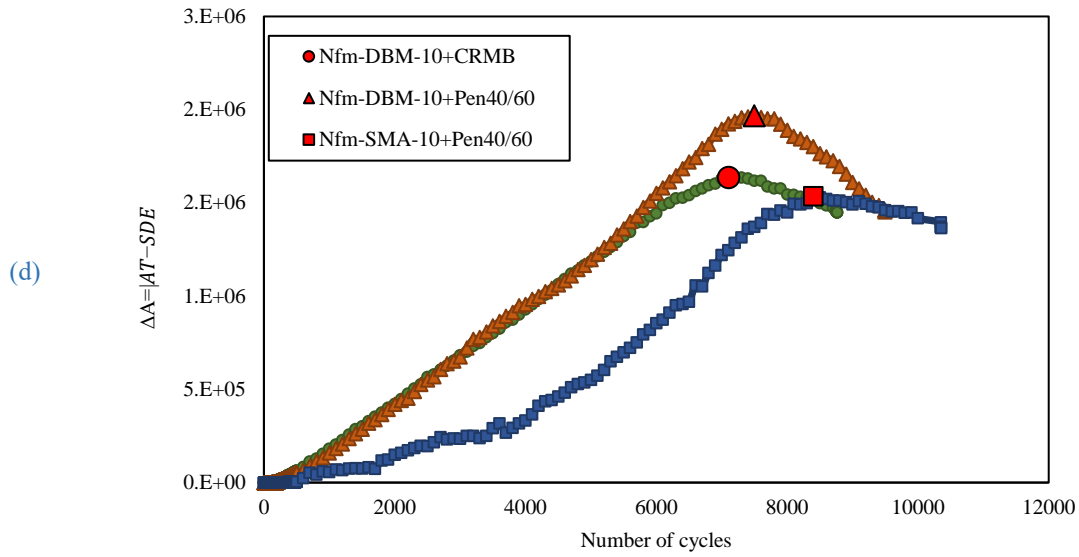


311 Figure 17: Energy ratio versus number of load cycles for controlled strain 2PB test: (a) at different temperatures;  
 312 (b) under different strains; (c) under different conditions; and for 4PB test: (d) of different mixtures.

313 3.2.3 Ratio of Dissipated Energy Change (RDEC) Approach

314 The results of the second transition point ( $N_{fm}$ ) are shown in Figure 18. Under strain-controlled loading mode,  
 315 the area  $\Delta A$  is initially very small but increases with loading cycles. There is a point where the area  $\Delta A$   
 316 is maximised and this point corresponds to  $N_{fm}$ , where macroscopic failure occurs, after which the area  $\Delta A$   
 317 decreases. The value of  $N_{fm}$  is also affected by the factors in the same way as  $N_1$ , it generally increases as the  
 318 temperature increases and decreases with strain increase. Ageing and moisture damage also causes the decrease  
 319 in  $N_{fm}$ , where the impact of moisture damage is more significant. The value of  $N_{fm}$  is also affected by the bitumen  
 320 type and mixture type. Unlike  $N_1$ , the dissipated energy decline rate at  $N_{fm}$  is maximum. At this point cracks are  
 321 propagating rapidly, which is considered to be the true point of failure.

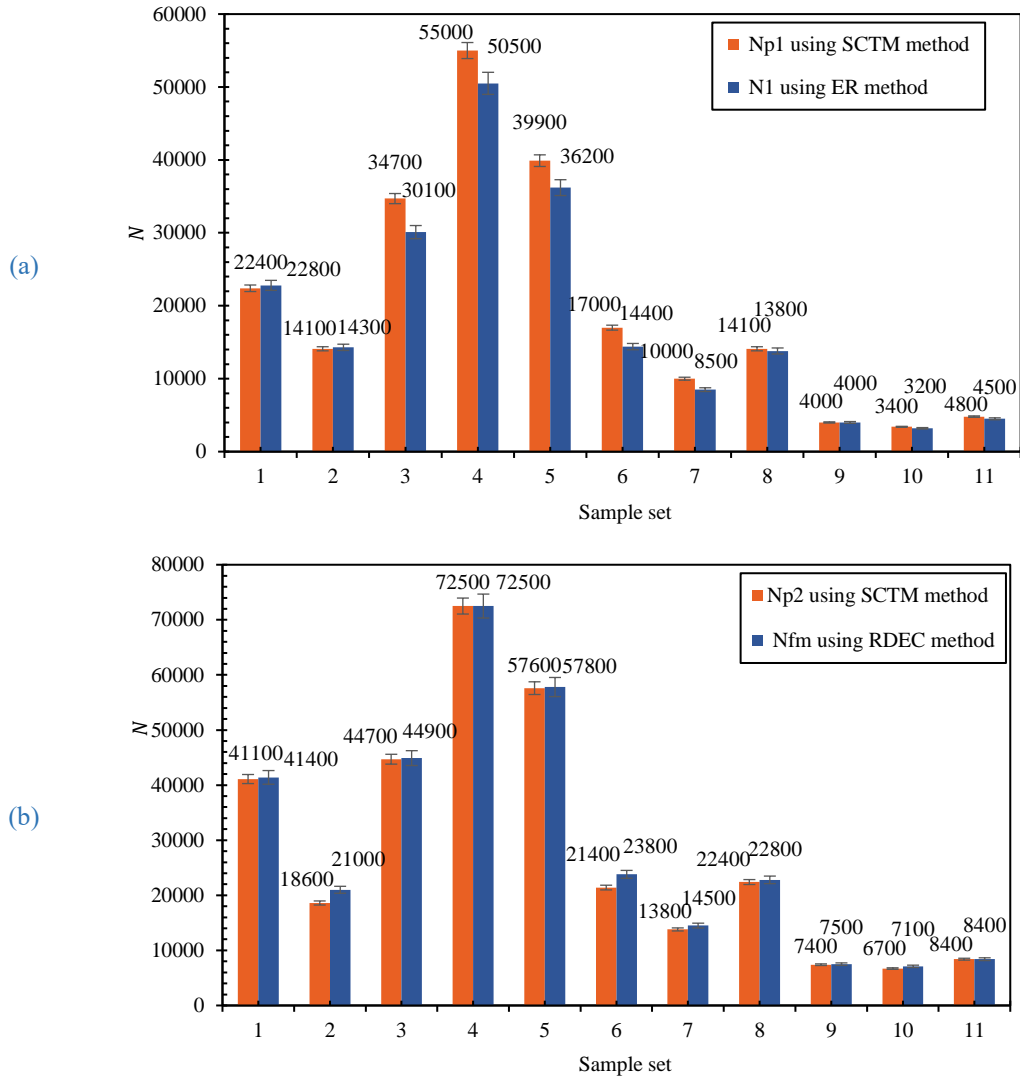




322 Figure 18:  $\Delta A$  versus number of cycles for controlled strain 2PB test: (a) At different temperatures, (b) Under  
 323 different strains; (c) Under different conditions and for 4PB test: (d) of different mixtures.

324 3.3 Validation of the SCTM

325 This research compares the number of loading cycles at P1 and P2 in the SCTM with the values of  $N_1$  and  $N_{fm}$   
 326 obtained from two different energy-based methods (the ER method and the RDEC method). The results obtained  
 327 from the two different energy-based methods and SCTM are shown in Figure 19. The figure shows that differences  
 328 between P1 and micro cracking point are generally greater than those between P2 and the macro-crack generation  
 329 point. For different conditions, including aged and moisture damaged, different test methods, different test  
 330 temperatures and strains, and different types of mixtures, the results calculated by the SCTM are similar to those  
 331 calculated by the energy-based methods. The one-way analysis of variance (ANOVA) is applied to show the  
 332 difference between the proposed method and the two energy methods. The results indicate that there are no  
 333 statistically significant differences between the SCTM and the two energy methods, as determined by one-way  
 334 ANOVA ( $F < F_{crit}$ ,  $P\text{-value} > 0.05$  for all sample sets), confirming that the SCTM is indeed appropriate to model  
 335 stiffness loss and be used as a means to calculate the fatigue failure criteria. P1 can be viewed as the number of  
 336 cycles to micro-crack initiation and propagation. The region prior to P1 is the micro-crack formation region. P2  
 337 can be defined as the macro-crack generation point; the true failure point is at P2.



338 Figure 19: Comparison of fatigue failure criteria gained from different methods: (a)  $N_{p1}$  and  $N_1$  ; and  
 339 (b)  $N_{p2}$  and  $N_{fm}$ .

340 **4 Conclusions**

341 In this paper a stiffness change tendency method (SCTM) is proposed to determine transition points in lgE v lgN  
 342 curves. These transition points are compared to the values of  $N_1$  and  $N_{fm}$  calculated using energy-based methods.  
 343 The main conclusions of the study are as follows:

- 344 1. There is good agreement between the laboratory data and the proposed logistic model, confirming that  
 345 the logistic model is a good approximation to model stiffness development.
- 346 2. It is illustrated that the stiffness ratio at the three transition points are similar across all samples. P1 occurs  
 347 between 48% and 68% of the initial stiffness ( $S_0$ ). The stiffness ratio at P2 ranges between 22% and 31%  
 348 of  $S_0$ , and at P3 ranges between 12% and 17% of  $S_0$ . Under the same strain level, the higher the



- 349 temperature, the later the failure point occurs.
- 350 3. The quotient of cycles with the respect to the stiffness loss,  $\Omega$ , at the different fatigue failure points can  
351 effectively evaluate the fatigue performance of asphalt concrete at different damage stages; it is affected  
352 by factors, such as temperature, strain level, ageing and moisture damage.
- 353 4. The number of loading cycles at the transition points P1 and P2 ( $N_{p1}$  and  $N_{p2}$ ) in the SCTM have a good  
354 likeness with the values of  $N_1$  and  $N_{fm}$  obtained in the energy-based methods. P1 can be viewed as the  
355 number of cycles to micro-crack initiation and propagation. The region prior to P1 is the micro-crack  
356 formation region. P2 can be defined as the macro-crack generation point; the true failure point is at P2.
- 357 5. The SCTM is feasible to be used to model stiffness development of different asphalt mixtures and test  
358 methods. The SCTM also has benefits of predicting fatigue failure points and remaining fatigue life as  
359 long as part of fatigue test data are known.

360 The SCTM combines the benefits of both approaches (phenomenological and energy-based). It provides a  
361 promising method of modelling stiffness development, determining the fatigue failure points to characterise  
362 different fatigue damage stages, evaluating fatigue performance, as well as, relating the fatigue failure points to  
363 different fatigue cracking processes, which could all be useful in a simulation of pavement deterioration in the  
364 future.

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## 367 **6 Conflicts of Interest**

368 The authors declare no conflict of interest.

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