

# MODELLING AND DESIGN OF COLD-FORMED S960 STEEL BRACE-ROTATED TUBULAR T- AND X-JOINTS

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## Abstract

This paper presents detailed numerical investigation and design of cold-formed S960 steel grade brace-rotated (BR) tubular T- and X-joints. The BR tubular joint is one of the novel bird-beak tubular joint configurations, where the rotation of brace member(s) enhances joint resistance and aesthetic appearance. The numerical investigation was performed through finite element (FE) analysis. The tests carried out by the authors were used to develop accurate FE models of BR T- and X-joints, which in turn precisely replicated the joint resistances, load vs deformation curves and failure modes of test specimens. With an aim to broaden the data size, a comprehensive FE parametric study was performed using the verified FE models. The nominal resistances predicted from the literature and European code were compared to the joint failure resistances of 211 BR T- and X-joints specimens, including 192 FE specimens investigated in this study. The BR T- and X-joint specimens were failed by two failure modes, namely chord face failure (F) mode and a combination of chord face and chord side wall failure mode, i.e. combined failure (F+S) mode. It has been shown that the existing design provisions are unsuitable for the design of cold-formed S960 steel grade BR T- and X-joints investigated in this study. Hence, using three design approaches, accurate, less dispersed, reliable and user-friendly design equations are proposed in this study to estimate the joint failure resistances of cold-formed S960 steel grade BR T- and X-joints.

*Keywords: Brace-rotated joints; Cold-formed steel; Design provisions; FE analysis; High strength steel; Tubular joints.*

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## 30 1. Introduction

31 Brace-rotated (BR) tubular joints are obtained by rotating the brace members of conventional  
32 RHS-to-RHS joints about their centroidal axes, where RHS represents both square and rectangular  
33 hollow sections. The rotation of brace about the centroidal axis increases its effective width, which  
34 in turn enhances the joint resistance without further increasing the material and fabrication costs. In  
35 addition to the flat connecting end of brace member, the overall welding operation of a BR joint is  
36 relatively easier than that of RHS-to-RHS joints. Moreover, brace rotation provides relatively less  
37 hindrance for wind and wave loads compared to conventional RHS-to-RHS joint. These merits of  
38 brace rotation promote the application of these joints in structures subjected to different types of  
39 loading, including topsides and jackets of offshore structures, agricultural equipment, booms and jibs  
40 of cranes, wheels, bridges, towers, trusses, spatial structures, stadiums, buildings, prefabricated  
41 modular structures and so on. A wide range of analytical, experimental and numerical investigations  
42 were carried out on different types of conventional tubular joints in the last six decades. Design rules  
43 were subsequently proposed to predict the static resistances of conventional tubular joints made of  
44 normal strength steel (in this study, referred to steels with steel grades lower than or equal to S460).  
45 In order to extend the applicability of design rules for high strength steel (HSS) (in this study, referred  
46 to steels with steel grades higher than S460), the design rules are required to be multiplied by the  
47 recommended material factors ( $C_f$ ).

48 HSS hollow section members are in high demand in various civil engineering projects due to  
49 high strength-to-weight ratio, reduced handling cost and reduced erection time. However, the lack of  
50 adequate research work and design recommendations are the primary reasons hampering the  
51 widespread use of HSS tubular members. However, some studies have recently been conducted to  
52 investigate the static behaviour of cold-formed high strength steel (CFHSS) tubular T- and X-joints  
53 [1-7]. To the best of the authors' knowledge, only three studies are available for the BR joints in the  
54 literature [1,2,8]. The BR configuration with SHS braces was first studied by Bae et al. [8] through  
55 both analytical and experimental methods. In total, 21 tests were carried out by Bae et al. [8] to  
56 investigate the ultimate resistances of BR T-joints made of S235 steel grade SHS members. Design  
57 rules were then proposed for predicting the ultimate resistances of the investigated BR T-joints.

58 Pandey and Young [1,2] conducted experimental investigations on cold-formed S960 steel grade BR  
59 T- and X-joints, where BR joints were fabricated using both square and rectangular hollow sections  
60 (SHS and RHS) brace members. The brace-rotation angle ( $\omega$ ) in Bae et al. [8] was limited to 45°,  
61 however,  $\omega$  ranged from 27° to 63° in Pandey and Young [1,2]. The static resistances of cold-  
62 formed high strength steel (CFHSS) BR T- and X-joints undergoing compression loads were  
63 investigated by Pandey and Young [1,2]. In order to develop a comprehensive understanding of the  
64 static behaviour of CFHSS BR T- and X-joints, a detailed numerical investigation was performed in  
65 this study. The test [1,2] and numerical resistances were compared with the nominal resistances  
66 predicted from design rules given in Bae et al. [8] as well as with the nominal resistances predicted  
67 from RHS-to-RHS and circular hollow section (CHS)-to-RHS design rules given in EC3 [9]. It has  
68 been demonstrated that the existing design rules were unsuitable for the range of BR T- and X-joints  
69 investigated in this study. As a result, accurate, less dispersed and reliable design equations are  
70 proposed, using three design approaches, to predict the joint failure resistances ( $N_f$ ) of cold-formed  
71 S960 steel grade BR T- and X-joints. The joint failure resistance ( $N_f$ ) of BR T- and X-joints has been  
72 defined as the load corresponding to the first occurrence of ultimate resistance (i.e. peak load) ( $N_{max}$ )  
73 and the load at 3% chord connecting face indentation (i.e.  $0.03b_0$ ) in the load ( $N$ ) vs chord face  
74 indentation ( $u$ ) curve.

75

## 76 2. Brief description of experimental investigations

77 The joint failure resistances ( $N_f$ ) and ultimate resistances ( $N_{max}$ ) of cold-formed BR T- and X-  
78 joints made of S960 steel grade were investigated by Pandey and Young [1,2]. Axial compression  
79 loads were applied on BR T- and X-joints test specimens through brace members. The chord ends of  
80 BR T-joint test specimens were supported on rollers through bearing blocks. On the other hand, for  
81 BR X-joint test specimens, top brace end was fixed and vertical displacement was allowed at the  
82 bottom brace end. The braces and chords were made of S960 steel grade RHS members. The thermo-  
83 mechanically controlled processed plates of S960 steel grade were cold-formed to obtain hollow

84 section members. A fully robotic metal active gas welding process was used to weld brace and chord  
 85 members. In total, 19 tests were conducted, including 10 BR T-joints and 9 BR X-joints. Moreover,  
 86 chord ends were not welded to end plates and freely deformed during the tests. Fig. 1(a) presents  
 87 various notations for BR T-joint, which are also valid for BR X-joint. The static behaviour of BR T-  
 88 and X-joints primarily depend on non-dimensional geometric ratios, including  $\beta' (=b'_1/b_0)$ ,  $2\gamma (=b_0/t_0)$ ,  
 89  $\tau (=t_1/t_0)$  and  $h_0/t_0$ . The symbols  $b$ ,  $h$ ,  $t$  and  $R$  stand for cross-section width, depth, thickness and  
 90 external corner radius of RHS member, respectively. The subscripts 0 and 1 denote chord and brace  
 91 members, respectively.

92 In the test programs [1,2],  $\beta'$  varied from 0.53 to 0.88,  $2\gamma$  varied from 25.3 to 38.8,  $h_0/t_0$  varied  
 93 from 25.4 to 38.9 and  $\tau$  varied from 0.67 to 1.28. The lengths of brace members ( $L_1$ ) of BR T- and X-  
 94 joints were determined as  $2\sqrt{b_1^2 + h_1^2}$  mm. On the other hand, the lengths of chord members ( $L_0$ ) of  
 95 BR T- and X-joints were determined as  $h'_1 + 3h_0 + 180$  mm and  $h'_1 + 4h_0$  mm, respectively. The  
 96 symbols  $b'_1$  and  $h'_1$  represent effective width and depth of brace cross-section, respectively. For  
 97 SHS brace,  $b'_1$  and  $h'_1$  are equal to  $\sqrt{b_1^2 + h_1^2} - 0.83R_1$ . However, for RHS brace,  $b'_1$  and  $h'_1$  are  
 98 equal to  $2\max[b_1, h_1] \sin \omega - 0.83R_1$  and  $\sqrt{b_1^2 + h_1^2} - 0.83R_1$ , respectively. The measured static yield  
 99 strengths of tubular members ranged from 952 to 1059 MPa, while the measured static yield strength  
 100 of welding filler material was 965 MPa. The BR T- and X-joint test specimens were failed by two  
 101 failure modes, namely chord face failure (F) mode and a combination of chord face and chord side  
 102 wall failure mode, i.e. combined failure (F + S) mode. The test results were obtained in the form of  
 103  $N$  vs  $u$  curves, where  $N$  and  $u$  respectively denote static load and chord face indentation. The testing  
 104 machine was paused for 120 seconds at two different locations in each test. The load drops captured  
 105 during the pauses were used to convert the test curves into static curves. Consequently, the obtained  
 106 test results were free from the influence of the applied loading rate.

107

### 108 3. Numerical program

### 109 3.1. Finite element models of brace-rotated (BR) T- and X-joints

#### 110 3.1.1. Introduction

111 One of the popular finite element software, ABAQUS [10], was used to perform the numerical  
112 investigation in this study. The static (general) analysis procedure given in ABAQUS [10] was used  
113 as the solver. As the induced strains in the finite element (FE) models during the applied loads were  
114 unidirectional (i.e. no load reversal), the isotropic strain hardening law was selected for the analysis.  
115 The von-Mises yield criterion is generally the default criterion used to predict the onset of yielding  
116 in most metals, except for porous metals. Therefore, the yielding onsets of FE models in this study  
117 were based on the von-Mises yield theory. In the FE analyses, the growth of the time step was kept  
118 non-linear to reduce the overall computation time. Furthermore, the default Newton-Raphson method  
119 was used to find the roots of non-linear equilibrium equations. In addition to the accuracy associated  
120 with the Newton-Raphson method, one of the popular benefits of using this numerical technique is  
121 its quadratic convergent approach, which in turn significantly increases the convergence rate of non-  
122 linear problems.

123 The material non-linearity was considered in the FE models by assigning the measured values  
124 of static stress-strain curves of flat and corner regions of RHS members in the plastic material  
125 definition part of the FE models. However, prior to the inclusions of experimentally obtained  
126 constitutive material curves in the FE models, they were first converted into static curves, and then  
127 transformed into true stress-strain curves. On the other hand, the geometric non-linearities in FE  
128 models were included by enabling the non-linear geometry parameter (\*NLGEOM), which in turn  
129 allow FE models to undergo large displacement during the analyses. Furthermore, various factors,  
130 including through-thickness division, contact interactions, mesh seed spacing, corner region  
131 extension and element types, were also studied and discussed in the following sub-sections of this  
132 paper. The labelling of parametric BR T- and X-joint FE specimens was kept identical to the label  
133 system used in the test programs [1,2]. The values of  $\omega$  adopted in the FE parametric study are  
134 shown in Fig. 1(b).

#### 135 3.1.2. Material properties, element type and mesh seed spacing

136 The test specimens [1,2] were fabricated from tubular members that belonged to the identical  
137 batch of tubes used by Pandey and Young [11]. Additionally, Pandey and Young [12] investigated the  
138 material properties of welding filler material. The details pertaining to the material properties of  
139 welding filler material and tubular members can be referred to Pandey and Young [11,12]. The  
140 inclusions of static stress-strain curves in FE models helped averting the effect of loading rate from  
141 FE results. The true stress-strain curves of welding filler material as well as flat and corner portions  
142 of RHS members were assigned to the corresponding parts of the FE specimens. In this study, the  
143 influence of cold-working in RHS members was included in FE models by assigning wider corner  
144 regions. Various distances for corner extension in RHS members were considered in the sensitivity  
145 analyses, and finally, the corner portions were extended by  $2t$  into the neighbouring flat portions,  
146 which was in agreement with other studies conducted on CFHSS tubular members and joints (Ma et  
147 al. [13,14] and Pandey et al. [15,16]). Except for the welds, all other parts of the FE models were  
148 developed using the C3D20 element. On the other hand, the C3D10 element was used to model the  
149 weld parts due to their complicated shapes. The weld parts were freely meshed using the free-mesh  
150 algorithm, while brace and chord parts were meshed using the structure-mesh algorithm. The use of  
151 solid elements helped in making realistic fusions between tubular and weld parts of BR T- and X-  
152 joints FE models.

153 Convergence studies were conducted using different mesh sizes, and finally, chord and brace  
154 members were seeded at 4 mm and 7 mm intervals, respectively, along both longitudinal and  
155 transverse directions. Moreover, the seeding intervals of weld parts reciprocated the seeding spacings  
156 of their respective brace parts. In order to ensure the smooth transfer of stresses between the flat  
157 regions of RHS cross-section, the corner regions of RHS cross-section were split into ten elements.  
158 FE analyses were also conducted to examine the influence of divisions along the wall thickness of  
159 RHS members. The results of these FE analyses demonstrated the trivial influence of wall thickness  
160 divisions on the load vs chord face indentation curves of the investigated BR T- and X-joints. The  
161 use of the C3D20 element as well as the small wall thickness of test specimens, led to such  
162 observations. It is worth noting that a similar observation was also noticed in other studies (Pandey  
163 et al. [15,16] and Crockett [17]). Thus, for the validations of BR T- and X-joints FE models, the wall

164 thicknesses of tubular members were kept unsplit.

### 165 3.1.3. Weld modelling and contact interactions

166 Fillet welds were modelled around the junctions of BR T- and X-joints. According to the  
167 prequalified weld details of tubular joints given in AWS D1.1M [18], the weld leg sizes of the fillet  
168 welds were designed as 1.5 times the minimum of brace and chord wall thickness. The welds were  
169 modelled using the average values of measured weld sizes, which are reported in Pandey and Young  
170 [1,2]. The inclusions of weld geometries and weld material properties appreciably improved the  
171 overall accuracies of BR T- and X-joints FE models. In addition, modelling of weld parts facilitated  
172 in realistic load transfer between brace and chord members, which in turn helped in obtaining actual  
173 joint behaviour. The selection of the C3D10 element maintained optimum stiffness around the joint  
174 perimeter due to its ability to take complicated shapes. A total of two types of contact interactions  
175 was defined in BR T- and X-joints FE models. First, contact interaction between brace and chord  
176 members of BR T- and X-joints FE models. Second, contact interaction between chord members and  
177 bearing blocks of BR T-joint FE models. In addition, a tie constraint was also established between  
178 weld and tubular members of BR T- and X-joints FE models. Both contact interactions were  
179 established using the built-in surface-to-surface contact definition.

180 The contact interaction(s) between brace and chord members of BR T- and X-joints FE models  
181 was kept frictionless, while a frictional penalty equal to 0.3 was imposed on the contact interaction  
182 between chord member and bearing blocks of BR T-joint FE models. Along the normal direction of  
183 these two contact interactions, a ‘hard’ contact pressure overclosure was used. In addition, finite  
184 sliding was permitted between the interaction surfaces. For contact interactions and tie constraint, the  
185 surfaces were connected to each other using the ‘master-slave’ algorithm technique. This technique  
186 permits the separation of fused surfaces under tension, however, it does not allow penetration of fused  
187 surfaces under compression. This technique of fusion between various parts of FE models has been  
188 successfully used in several other investigations (Pandey et al. [15,16]; Lan et al. [19]; Li and Young  
189 [20]; Li and Young [21,22]). For the brace-chord interaction, the cross-section surface of the brace  
190 connected to the chord member was assigned as the ‘master’ region (relatively less deformable),

191 while the chord connecting surface was assigned as the ‘slave’ region (relatively more deformable).  
192 For the chord-bearing block interaction, the chord member was assigned as the ‘slave’ region, while  
193 the bearing block was assigned as the ‘master’ region. For the weld-tubular member tie connection,  
194 the weld surfaces were assigned as the ‘master’ regions, while the connecting brace and chord  
195 surfaces were assigned as the ‘slave’ regions.

#### 196 3.1.4. Boundary conditions and load applications

197 The boundary conditions in BR T- and X-joints FE models were assigned through reference  
198 points. Three reference points were created for the BR T-joint FE model, including one top reference  
199 point (TRP) and two bottom reference points (BRP-1 and BRP-2). The TRP replicated the fixed  
200 boundary condition of the top brace end, while BRP-1 and BRP-2 replicated the boundary conditions  
201 of rollers positioned at both chord ends. As shown in Fig. 2, the TRP was created at the cross-section  
202 centre of the top brace end, while BRP-1 and BRP-2 were created at 20 mm below the centre of the  
203 bottom surfaces of bearing blocks. The TRP, BRP-1 and BRP-2 were then coupled to their  
204 corresponding surfaces using the built-in kinematic coupling type. In order to exactly replicate the  
205 boundary conditions of the BR T-joint test setup, all degrees of freedom (DOF) of TRP were  
206 restrained. On the other hand, for BRP-1 and BRP-2, except for the translations along the vertical  
207 and longitudinal directions of the BR T-joint FE specimen as well as the rotation about the transverse  
208 direction of the chord member, all other DOF of BRP-1 and BRP-2 were also restrained. In addition,  
209 all DOF of other nodes of BR T-joint FE specimen were kept unrestrained for both rotation and  
210 translation.

211 With regard to the BR X-joint FE model, the top and bottom reference points (TRP and BRP)  
212 were created at the cross-section centres of the top and bottom brace members, as shown in Fig. 3.  
213 Subsequently, TRP and BRP were coupled to their respective brace end cross-section surfaces using  
214 the kinematic coupling type. In order to exactly replicate the boundary conditions of the BR X-joint  
215 test setup, all DOF of TRP were restrained. However, except for the translation along the vertical  
216 direction of the BR X-joint specimen, all other DOF of BRP were also restrained. Moreover, all DOF  
217 of other nodes of the BR X-joint FE specimen were kept unrestrained for both rotation and translation.



218 Using the displacement control method, compression load was then applied at the bottom reference  
219 points of the BR T- and X-joints FE models. In addition, the size of the step increment was kept small  
220 to obtain smooth load vs chord face indentation curves. Following this approach, the boundary  
221 conditions and load application in FE models were identical to the test programs [1,2].

### 222 3.1.5. Weld heat affected region (WHAR)

223 The heat transferred to parent tubular members during the welding process has a considerable  
224 influence on the overall behaviour of hollow section joints [15,16]. The design rules in international  
225 standards/guidelines (EC3 [9]; AISC 360 [23]; ISO 14346 [24]; IIW [25]; CIDECT [26]) are identical  
226 for HSS produced from different methods, namely by adding alloying elements and by various heat  
227 treatment techniques. However, it has been reported in some recent studies [27-30] that HSS  
228 produced by different methods exhibited different extents of softening around the welds.  
229 Investigations carried out by Stroetmann et al. [27], Javidan et al. [28] and Amraei et al. [29,30]  
230 reported 16% to 32% reductions in the ultimate strengths of S960 steel grade parent materials around  
231 the welds. The material properties of weld heat affected region (WHAR) of S960 steel grade tubular  
232 members with wall thickness ranged from 3 to 6 mm were investigated by Pandey and Young [5]. A  
233 14% to 32% reduction in the ultimate strengths of the parent metals was reported by Pandey and  
234 Young [5] in the first 6 mm distance of the WHAR. The definition of WHAR for tubular joints was  
235 proposed by Pandey et al. [15], as shown in Fig. 4. For BR T- and X-joints FE models, the spreads  
236 of WHAR are shown in Figs. 2 and 3, respectively. In addition, a simplified strength reduction ( $S_{rl}$ )  
237 model was proposed by Pandey et al. [15] for S900 and S960 steel grades tubular joints to integrate  
238 the material properties of WHAR in FE models, as illustrated in Fig. 5. The proposed strength  
239 reduction model was successfully used to perform the numerical investigation and design of CFHSS  
240 T- and TF-joints (Pandey et al. [15,16]). Therefore, it was also included in this investigation, and  
241 accordingly, material properties were assigned to the WHAR of BR T- and X-joints FE models. The  
242 adoption of WHAR appreciably improved the accuracies of FE models, and thus, the numerical  
243 results.

### 244 3.2. Validations of BR T- and X-joints FE models

245 The numerical modelling techniques described in the preceding section of this paper were used  
246 to develop BR T- and X-joints FE models. The test results of BR T- and X-joints reported in Pandey  
247 and Young [1,2] were used to validate their corresponding FE models. The validations were  
248 performed by duly comparing the  $N_f$ ,  $N_{max}$ ,  $N$  vs  $u$  curves and failure modes of test and FE specimens.  
249 The measured dimensions of tubular members and welds were used to develop all BR T- and X-joints  
250 FE models. In addition, measured material properties of tubular members, welds and WHAR were  
251 also included. The  $N_f$  and  $N_{max}$  of BR T- and X-joints test specimens were compared with those  
252 predicted from their corresponding FE models ( $N_{f,FE}$  and  $N_{max,FE}$ ), as shown in Tables 1 and 2,  
253 respectively. Referring to Table 1, when the joint failure resistances of BR T-joint ( $N_{f,T}$ ) test specimens  
254 were compared with the resistances predicted from BR T-joint FE models, the mean ( $P_m$ ) and  
255 coefficients of variation (COV) ( $V_p$ ) of the comparisons were 1.01 and 0.014, respectively. However,  
256 when the ultimate resistances of BR T-joint ( $N_{max,T}$ ) test specimens were compared with the FE  
257 resistances, the  $P_m$  and  $V_p$  of the comparisons were 1.00 and 0.017, respectively.

258 On the other hand, as shown in Table 2, when the joint failure resistances of BR X-joint ( $N_{f,X}$ )  
259 test specimens were compared with the resistances predicted from BR X-joint FE models, the  $P_m$  and  
260  $V_p$  of the comparisons were 1.01 and 0.023, respectively. However, when the ultimate resistances of  
261 BR X-joint ( $N_{max,X}$ ) test specimens were compared with the FE resistances, the  $P_m$  and  $V_p$  of the  
262 comparisons were 1.02 and 0.021, respectively. Likewise, the experimental investigation, the  $N_f$  of  
263 BR T- and X-joints FE specimens was determined by jointly considering the ultimate resistances and  
264 ultimate deformation limit (i.e.  $0.03b_0$ ) loads, whichever occurred earlier in the  $N$  vs  $u$  curves. In  
265 addition, the comparisons of  $N$  vs  $u$  curves between typical BR T- and X-joints test and FE specimens  
266 are shown in Figs. 6 and 7, respectively. Moreover, Figs. 8 and 9 present the comparisons of failure  
267 modes between typical BR T- and X-joints test and FE specimens, respectively. Therefore, from  
268 Tables 1-2 and Figs. 6-9, it can be concluded that the validated FE models precisely replicated the  
269 overall static behaviour of BR T- and X-joints investigated in this study.

### 270 3.3. Parametric FE modelling of BR T- and X-joints

#### 271 3.3.1. General

272 The data pool was widened by performing a comprehensive numerical parametric study using  
273 the validated BR T- and X-joints FE models. In total, 192 parametric FE analyses were performed in  
274 this study, including 96 BR T-joints and 96 BR X-joints. Table 3 presents the ranges and values of  
275 various critical parameters considered in the parametric study. All FE modelling techniques used in  
276 the validations of BR T- and X-joints were also employed in the parametric study. It is important to  
277 mention that, in this investigation, the  $N_f$  of all BR T- and X-joints parametric FE specimens were  
278 controlled by the ultimate deformation limit (i.e.  $0.03b_0$ ) criterion.

### 279 3.3.2. Specifications for parametric FE modelling of BR T- and X-joints

280 In the numerical investigation, the dimensions of tubular members included practical sizes.  
281 Overall, the values of cross-section width and depth of brace and chord members of parametric FE  
282 specimens ranged from 40 mm to 200 mm, while their wall thickness ranged from 2.5 mm to 12 mm.  
283 The exterior corner radii of brace and chord members ( $R_l$  and  $R_o$ ) conformed to the commercially  
284 produced HSS members (SSAB [31]). In this study,  $R_l$  and  $R_o$  were kept as  $2t$  for  $t \leq 6$  mm,  $2.5t$  for  
285  $6 < t \leq 10$  mm and  $3t$  for  $t > 10$  mm, which in turn also met the limits detailed in EN 10219-2 [32].  
286 The lengths of braces and chords of BR T- and X-joints FE specimens were determined using the  
287 formulae that were also used to design the test specimens [1,2], as mentioned in Section 2 of this  
288 paper. For meshing along the longitudinal and transverse directions of RHS members, seedings were  
289 approximately spaced at the minimum of  $b/30$  and  $h/30$ , where  $b$  and  $h$  stand for cross-section width  
290 and depth of the RHS member. Overall, the adopted mesh sizes of parametric FE specimens ranged  
291 from 3 mm to 10 mm. On the other hand, the seeding interval of weld parts of parametric FE  
292 specimens reciprocated the seeding interval of their corresponding brace parts. For precise replication  
293 of RHS curvatures, the corner regions of braces and chords were split into ten elements. Likewise, in  
294 the validation process, the corner regions of RHS members were extended by  $2t$  into their  
295 neighbouring flat portions. For FE specimens with  $t \leq 6$  mm, no divisions were made along the wall  
296 thicknesses of brace and chord members. However, when  $t > 6$  mm, the wall thicknesses of brace and  
297 chord members were divided into two layers. The design of fillet weld leg sizes for both BR T- and  
298 X-joints FE specimens was consistent with the design adopted in the test programs [1,2]. In the

299 parametric study, the material properties of flat and corner portions of RHS 150×150×6 were assigned  
300 to the flat and corner regions of brace and chord members of FE specimens. Besides, weld parts of  
301 all BR T- and X-joints parametric FE specimens were given the measured material properties of  
302 welding filler material. Table 4 presents the measured material properties of RHS 150×150×6 and  
303 welding filling material adopted in the parametric study, which include Young's modulus ( $E$ ), 0.2%  
304 proof stress and strain ( $\sigma_{0.2}$  and  $\varepsilon_{0.2}$ ), ultimate stress and strain ( $\sigma_u$  and  $\varepsilon_u$ ), fracture strain ( $\varepsilon_f$ ) and  
305 Ramberg-Osgood parameter ( $n$ ). On the other hand, the material properties and spread of WHAR  
306 were in accordance with the recommendations proposed by Pandey et al. [15].

### 307 3.3.3. Failure modes of BR T- and X-joints

308 The BR T- and X-joints test and FE specimens were failed by two failure modes, namely chord  
309 face failure (F) mode, and a combination of chord face and chord side wall failure mode, i.e.  
310 combined failure (F+S) mode. Overall, the BR T- and X-joints specimens were failed by the F mode  
311 when  $\beta' \leq 0.85$ . On the other hand, the F+S mode occurred for the BR T- and X-joints test and FE  
312 specimens when  $\beta' > 0.85$ . It is important to note that both these failure modes were defined  
313 corresponding to the  $N_f$ , which in turn was computed by jointly considering the ultimate and  $0.03b_0$   
314 limit loads. The test and parametric FE specimens were failed by the F mode, when the  $N_f$  was  
315 determined using only the ultimate deformation limit ( $0.03b_0$ ) load criterion. The applied loads of  
316 BR T- and X-joints specimens that failed by the F mode were monotonically increasing with the  
317 increase of chord face indentation. For BR T- and X-joints test and FE specimens that failed by the  
318 F+S mode, the load vs chord face indentation curves exhibited a visible peak load (i.e. ultimate  
319 resistance). Additionally, evident deformations of chord flange, chord webs and chord corner regions  
320 were noticed in the test and parametric FE specimens that failed by the F+S mode. Moreover, none  
321 of the test and FE specimens were failed by the global buckling of brace members.

322

## 323 4. Existing design provisions

324 The BR T- and X-joints are currently not covered in any international code and guideline. In

325 the literature, design rules are only available for S235 steel grade BR T-joint (Bae et al. [8]). The  
326 overall static performance of tubular T- and X-joints when subjected to axial compression loads  
327 through brace members are nearly similar. Therefore, in this investigation, the  $N_f$  of both BR T- and  
328 X-joints test and parametric FE specimens were evaluated against the nominal resistances predicted  
329 from the design rules proposed by Bae et al. [8]. Moreover, the BR joint configuration partially  
330 resembles to that of conventional RHS-to-RHS (due to orientation of chord) and CHS-to-RHS (due  
331 to orientation of brace) configurations. Thus, the  $N_f$  of BR T- and X-joints test and parametric FE  
332 specimens were also evaluated against the nominal resistances of RHS-to-RHS and CHS-to-RHS T-  
333 and X-joints design rules given in EC3 [9]. The measured dimensions and material properties of  
334 tubular members were used to calculate the nominal resistances. Under axial compression load, the  
335 chord members of BR T-joints were subjected to chord-in-plane bending. In this investigation, the  
336 effect of normal stresses developed due to chord-in-plane bending on the static resistances of BR T-  
337 joints was considered through the chord stress function ( $Q_f$ ). On the other hand, in this study, no  
338 preload was applied to the chord members of BR X-joints. Therefore, the value of  $Q_f$  for BR X-joints  
339 was set to unity in Eqs. (3) to (6). Furthermore, as design equations proposed by Bae et al. [8] were  
340 valid for S235 steel grade BR T-joints, thus, the nominal resistances predicted from Bae et al. [8]  
341 were multiplied by a material factor ( $C_f$ ) equal to 0.80 to facilitate their evaluations against the test  
342 and FE resistances of cold-formed S960 steel grade BR T- and X-joints.

#### 343 4.1. Bae et al. [8]

344 Bae et al. [8] proposed design equations (Eqs. (1) and (2)) to estimate the ultimate resistances  
345 of S235 steel grade BR T-joints subjected to compression loads through brace members. The  
346 proposed design equations are valid for  $0.38 \leq \beta' \leq 1.0$  and  $16.7 \leq 2\gamma \leq 33.3$ .

347 *Chord face failure ( $\beta' \leq 0.85$ )*

$$N_{Bae} = \frac{f_{y0} t_0^2}{4} \left[ 10 + \frac{4(1 + \beta')}{(1 - \beta')} \right] \quad (1)$$

348 *Chord web failure ( $\beta' = 1.0$ )*

$$N_{Bae} = 2f_k t_0 (0.89b_1) \quad (2)$$

349 In order to extend the applicability of Eqs. (1) and (2) for CFHSS BR joints investigated in  
 350 this study, a material factor ( $C_f$ ) equal to 0.80 should be included in Eqs. (1) and (2). The nominal  
 351 resistances determined after including the  $C_f$  factor in Eqs. (1) and (2) are represented by  $N_{Bae}^{\wedge}$ .

#### 352 4.2. EC3 [9]

353 The design rules given in EC3 [9] are applicable for tubular joints with steel grades up to S700.  
 354 However, a material factor ( $C_f$ ) is required to be multiplied to the design rules when steel grade  
 355 exceeds S355. When steel grade ranged from 550 to 700 MPa, the value of material factor ( $C_f$ ) is  
 356 equal to 0.80. Furthermore, EC3 [9] has explicitly recommended the value of partial safety factor for  
 357 tubular joints ( $\gamma_{M5}$ ), which is equal to 1.0. The design equations for chord face failure and chord side  
 358 wall failure modes are shown below:

#### 359 RHS-to-RHS T- and X-joints:

360 *Chord face failure ( $\beta \leq 0.85$ ):*

$$N_{E,RR}^{\wedge} = \frac{C_f}{\gamma_{M5}} Q_f \frac{f_{y0} t_0^2}{\sin \theta_1} \left( \frac{2\eta}{(1-\beta)\sin \theta_1} + \frac{4}{\sqrt{1-\beta}} \right) \quad (3)$$

361 *Chord side wall failure ( $\beta = 1.0$ ):*

$$N_{E,RR}^{\wedge} = \frac{Q_f}{\gamma_{M5}} \frac{f_b t_0}{\sin \theta_1} \left( \frac{2h_1}{\sin \theta_1} + 10t_0 \right) \quad (4)$$

#### 362 CHS-to-RHS T- and X-joints:

363 *Chord face failure ( $\beta' \leq 0.85$ ):*

$$N_{E,CR}^{\wedge} = \frac{\pi}{4} \frac{C_f}{\gamma_{M5}} Q_f \frac{f_{y0} t_0^2}{\sin \theta_1} \left( \frac{2(h_1'/b_0)}{(1-\beta')\sin \theta_1} + \frac{4}{\sqrt{1-\beta'}} \right) \quad (5)$$

364 *Chord side wall failure ( $\beta' = 1.0$ ):*

$$N_{E,CR}^{\wedge} = \frac{\pi}{4} \frac{Q_f}{\gamma_{M5}} \frac{f_b t_0}{\sin \theta_1} \left( \frac{2h_1'}{\sin \theta_1} + 10t_0 \right) \quad (6)$$

365 In Eqs. (1) to (6), the term  $f_{y0}$  represents the yield strength of chord member,  $f_k$  and  $f_b$  represent  
 366 buckling stress of chord member as per EC3 [33];  $\gamma_{M5}$  is the partial safety factor of tubular joints as  
 367 per EC3 [9] and  $\theta_l$  represents the angle between brace and chord members in degrees.

368

## 369 5. Reliability analysis

370 In order to examine the reliability of existing and proposed design equations, a reliability study  
 371 was performed as per AISI S100 [34]. The Eq. (7) was used to calculate the reliability index ( $\beta_0$ ). In  
 372 this investigation, a lower bound value of 2.50 was taken as the target  $\beta_0$ . Therefore, when  $\beta_0 \geq 2.50$ ,  
 373 the design equation was treated as reliable in this study.

$$\beta_0 = \frac{\ln(C_\phi M_m F_m P_m / \phi)}{\sqrt{V_M^2 + V_F^2 + C_P V_P^2 + V_Q^2}} \quad (7)$$

374 A dead load (DL)-to-live load (LL) ratio of 0.20 was used to compute the calibration coefficient  
 375 ( $C_\phi$ ) in Eq. (7). For the material factor, the mean value and COV were respectively symbolised by  
 376  $M_m$  and  $V_M$ . For the fabrication factor, the mean value and COV were respectively symbolised by  $F_m$   
 377 and  $V_F$ . Referring to AISI S100 [34], the  $M_m$  and  $V_M$  were adopted as 1.10 and 0.10, respectively.  
 378 Additionally,  $F_m$  and  $V_F$  were adopted as 1.00 and 0.10, respectively. The resistance factor required  
 379 to convert the nominal resistance to design resistance was denoted by  $\phi$ . The mean value of ratios  
 380 of test and FE resistances-to-nominal resistances predicted from literature and code was denoted by  
 381  $P_m$ , while the corresponding COV was denoted by  $V_P$ . The correction factor ( $C_P$ ) proposed by AISI  
 382 S100 [34] was also used in Eq. (7) to incorporate the effect of the amount of data under consideration.  
 383 Besides,  $V_Q$  symbolised the COV of load effects. In order to evaluate the reliability levels of EC3 [9]  
 384 design provisions, the DL and LL were combined as 1.35DL + 1.5LL as per EN [35], and thus, the  
 385 calculated value of  $C_\phi$  was 1.463. Further, to examine the reliability levels of the design equation  
 386 proposed by Bae et al. [8] as well as for the proposed design rules, the DL and LL were combined as  
 387 1.2DL + 1.6LL as per ASCE 7 [36], and the calculated value of  $C_\phi$  was 1.521.

## 389 6. Comparisons of joint failure resistances with nominal resistances

390 The comparisons of  $N_f$  of BR T- and X-joints test and FE specimens with nominal resistances  
391 are shown in Tables 5 and 6, respectively. The comparisons are also graphically shown in Figs. 10 to  
392 13, 15 and 16. Table 5 presents the comparisons of  $N_{f,T}$  of BR T-joint test and parametric FE  
393 specimens with nominal resistances predicted from Bae et al. [8] and EC3 [9]. The comparisons  
394 results proved that the design rules proposed by Bae et al. [8], RHS-to-RHS and CHS-to-RHS T-  
395 joints design rules of EC3 [9] satisfactorily predicted the  $N_{f,T}$  of cold-formed S960 steel grade BR T-  
396 joints. However, the predictions were very dispersed and the design equations were found to be  
397 unreliable. Fig. 10 graphically presents the comparisons of  $N_{f,T}$  of BR T-joint test and parametric FE  
398 specimens with nominal resistances predicted from Bae et al. [8] and CHS-to-RHS T-joint design  
399 rule of EC3 [9]. The comparisons of  $N_{f,X}$  of BR X-joint test and parametric FE specimens with  
400 nominal resistances predicted from Bae et al. [8] and EC3 [9] are presented in Table 6. The  
401 predictions of the design rules proposed by Bae et al. [8] were found to be satisfactory and very  
402 dispersed but unreliable for the  $N_{f,X}$  of CFHSS BR X-joints. On the contrary, the comparisons of  
403 predictions of RHS-to-RHS and CHS-to-RHS X-joints design rules of EC3 [9] with the  $N_{f,X}$  of BR  
404 X-joints were found to be slightly unconservative, largely dispersed and unreliable. Fig. 11  
405 graphically presents the comparisons of  $N_{f,X}$  of BR X-joint test and parametric FE specimens with  
406 nominal resistances predicted from Bae et al. [8] and CHS-to-RHS X-joint design rule of EC3 [9].

407 The design equations proposed by Bae et al. [8] were developed for S235 steel grade BR T-  
408 joints. In addition, only SHS members were used as braces of BR T-joints in Bae et al. [8]. Overall,  
409 the design equations (Eqs. (1) and (2) of this paper) satisfactorily predicted the  $N_f$  of cold-formed  
410 S960 steel grade BR T- and X-joints, as reflected from the values of  $P_m$  shown in Tables 5 and 6.  
411 However, the predictions were very scattered, and thus, the design rules became unreliable. One of  
412 the possible reasons for highly scattered predictions could be due to the assumption of yield lines  
413 propagation at  $45^\circ$  from brace corners, which is primarily valid for SHS braces with  $\omega = 45^\circ$ . In this  
414 investigation, RHS members were also used as the braces of BR T- and X-joints and the values of



415  $\omega$  ranged from  $15^\circ$  to  $63^\circ$ . In addition, one of the important geometric parameters,  $2\gamma$  ( $=b_0/t_0$ ),  
416 which accounts for the slenderness of chord flat region, was not included in Eq. (1). Moreover, the  
417 stress-strain curve of S960 steel significantly deviates from that of mild steel (steel grades up to S355).  
418 The prolonged elasticity, absence of yield plateau, different extent of strain hardening, and low  
419 ultimate-to-yield strength ratio can change the response of HSS tubular joints, especially in the  
420 deformation and propagation of chord face yield line patterns and development of chord face  
421 membrane action, compared to the mild steel counterparts [15,37]. For small to medium values of  $\beta$   
422 ratio (i.e.  $\beta \leq 0.75$ ), normal strength steel T- and X-joints are expected to undergo relatively larger  
423 chord connecting face deformation compared to corresponding HSS counterparts at the same load  
424 level. For HSS T- and X-joints with small to medium values of  $\beta$  ratio (i.e.  $\beta \leq 0.75$ ), and especially  
425 for large values of  $2\gamma$  ratio, the current  $0.03b_0$  deformation limit seems not sufficient to develop plastic  
426 hinges in the chord connecting face. Therefore, the strength of HSS material from the proportional  
427 limit to yield strength could not be effectively utilised owing to the existing  $0.03b_0$  deformation limit  
428 criterion [15].

429

## 430 **7. Proposed design rules**

431 In order to estimate the  $N_f$  of cold-formed S960 steel grade BR T- and X-joints, design rules  
432 are proposed in this study using three design approaches. Under the first approach, named as  
433 proposal-1, new design equations are proposed to predict the  $N_f$  of CFHSS BR T- and X-joints. Under  
434 the second approach, named as proposal-2, the  $N_f$  of CFHSS BR T- and X-joints are predicted by  
435 applying a correction factor on the current CHS-to-RHS T- and X-joints design rule (Eq. (5)) given  
436 in EC3 [9]. Under the third approach, named as proposal-3, a design equation has been proposed  
437 using a simplified yield line model to predict the  $N_f$  of CFHSS BR T- and X-joints investigated in  
438 this study. Furthermore, as welds were modelled in all parametric FE specimens, the effects of weld  
439 and associated WHAR were implicitly included in the proposed design equations. In order to  
440 calculate the design resistances ( $N_d$ ), the proposed nominal resistances ( $N_{pn1}$ ,  $N_{pn2}$  and  $N_{pn3}$ ) in the  
441 following sub-sections of this paper shall be multiplied by their correspondingly recommended

442 resistance factors ( $\phi$ ), i.e.  $N_d = \phi (N_{pn1} \text{ or } N_{pn2} \text{ or } N_{pn3})$ . The design rules proposed in this study are  
 443 valid for  $0.20 \leq \beta \leq 0.67$ ,  $0.26 \leq \beta' \leq 0.88$ ,  $16.6 \leq 2\gamma \leq 40$ ,  $0.50 \leq \tau \leq 1.28$ ,  $15^\circ \leq \omega \leq 63^\circ$  and  
 444  $\theta_l = 90^\circ$ .

#### 445 7.1. Proposal-1 (Unified design equation)

446 The parameters  $\beta'$ ,  $2\gamma$ ,  $h_0/t_0$  and  $\tau$  demonstrated a considerable influence on the static behaviour  
 447 of BR T- and X-joints. Thus, new design equations (i.e. Eqs. (8) and (9)) are proposed to estimate the  
 448  $N_f$  of cold-formed S960 steel grade BR T- and X-joints by duly considering the effect of important  
 449 geometric parameters as well as the  $P_m$  and  $V_p$  of the overall comparison.

450 For BR T-joint:

$$N_{pn1} = \frac{f_{y0} t_0^2 e^{2\beta'} (\tau + 0.7)}{[0.6 + 0.01(2\gamma)] \left[ 0.5 + 0.02 \left( \frac{h_0}{t_0} \right) \right]} \quad (8)$$

451 For BR X-joint:

$$N_{pn1} = \frac{f_{y0} t_0^2 e^{2.3\beta'} (0.6\tau + 0.7)}{[0.4 + 0.017(2\gamma)] \left[ 0.5 + 0.02 \left( \frac{h_0}{t_0} \right) \right]} \quad (9)$$

452 As shown in Table 5, the  $P_m$  and  $V_p$  of the proposed design equation for BR T-joint (i.e. Eq. (8))  
 453 are 1.00 and 0.149, respectively. On the other hand, referring to Table 6, the  $P_m$  and  $V_p$  of the proposed  
 454 design equation for BR X-joint (i.e. Eq. (9)) are 1.04 and 0.160, respectively. For both Eqs. (8) and  
 455 (9),  $\phi$  equal to 0.80 was recommended, resulting in  $\beta_0$  equal to 2.52 and 2.57, respectively. Thus,  
 456 Eqs. (8) and (9) must be multiplied by  $\phi$  equal to 0.80 to get their corresponding design resistances  
 457 ( $N_d$ ). The comparisons of  $N_f$  of BR T- and X-joints test and FE specimens with nominal resistances  
 458 predicted from Bae et al. [8], CHS-to-RHS design rule of EC3 [9] and proposed design equations  
 459 under proposal-1 (Eqs. (8) and (9)) are graphically presented in Figs. 10 and 11, respectively.  
 460 Compared to the existing design provisions, the predictions from Eqs. (8) and (9) are relatively more  
 461 accurate, less dispersed and reliable for the  $N_f$  of CFHSS BR T- and X-joints.

462 The formats of the proposed new design equations, i.e. Eqs. (8) and (9), are identical. Therefore,

463 an attempt has been made to propose a unified design equation to predict the  $N_f$  of cold-formed S960  
 464 steel grade BR T- and X-joints. The proposed unified design equation, as shown in Eq. (10), is valid  
 465 for  $0.26 \leq \beta' \leq 0.88$ . The values of coefficients (A to G) are given in Table 7.

$$N_{pm1} = f_{y0} t_0^2 \frac{e^{A\beta'} (B\tau + C)}{\left[ D + E(2\gamma) \right] \left[ F + G \left( \frac{h_0}{t_0} \right) \right]} \quad (10)$$

## 466 7.2. Proposal-2 (Simplified design equations)

467 Under proposal-2, a correction factor based on geometric parameter  $2\gamma$  was applied on the  
 468 current CHS-to-RHS T- and X-joints design rules given in EC3 [9], as shown in Eqs. (11) and (12),  
 469 to predict the  $N_f$  of cold-formed S960 steel grade BR T- and X-joints.

470 For BR T-joint:

$$N_{pm2} = [1.39 - 0.02(2\gamma)] N_{E,CR}^{\wedge} \quad (11)$$

471 For BR X-joint:

$$N_{pm2} = [1.52 - 0.025(2\gamma)] N_{E,CR}^{\wedge} \quad (12)$$

472 The term  $N_{E,CR}^{\wedge}$  in Eqs. (11) and (12) can be obtained from Eq. (5). As shown in Table 5, the  
 473  $P_m$  and  $V_p$  of the proposed design equation for BR T-joint (i.e. Eq. (11)) are 1.05 and 0.182,  
 474 respectively. On the other hand, referring to Table 6, the  $P_m$  and  $V_p$  of the proposed design equation  
 475 for BR X-joint (i.e. Eq. (12)) are 1.06 and 0.187, respectively. For both Eqs. (11) and (12),  $\phi$  equal  
 476 to 0.80 was recommended, resulting in  $\beta_0$  equal to 2.51. Thus, Eqs. (11) and (12) must be multiplied  
 477 by  $\phi$  equal to 0.80 to get their corresponding design resistances ( $N_d$ ). The comparisons of  $N_f$  of BR  
 478 T- and X-joints test and FE specimens with nominal resistances predicted from Bae et al. [8], CHS-  
 479 to-RHS design rule of EC3 [9] and proposed design equations under proposal-2 (Eqs. (11) and (12))  
 480 are graphically presented in Figs. 12 and 13, respectively. Compared to the existing design provisions,  
 481 the predictions from Eqs. (11) and (12) are relatively more accurate, less dispersed and reliable for  
 482 the  $N_f$  of CFHSS BR T- and X-joints.

## 483 7.3. Proposal-3 (Yield line model)

484 A simplified yield line model based on the deformed shape of chord connecting face(s) of BR  
 485 T- and X-joint test specimens [1,2] is proposed in this study, as shown in Fig. 14. The yield line theory  
 486 is based on the principle of virtual work. Accordingly, the work done by the external forces is equal  
 487 to the internal work done by the yield lines. In the proposed model, the yield lines propagate along  $\alpha$   
 488 (degrees) from the brace corners, and after reaching the chord corners, the yield lines further deviate  
 489 by  $\lambda$  (degrees). Using the principle of virtual work, the design equation to predict the nominal  
 490 resistances of BR T- and X-joints can be derived as follows:

$$491 \quad \text{Total external work done } (W_e) = \text{External force } (N) \times \text{deformation } (\delta) = N\delta$$

$$492 \quad \text{Total internal work done } (W_i) = \sum_{i=1}^n (M_p \theta_i) l_i$$

493 where  $M_p$  denotes plastic moment per unit length of the yield line and equal to  $f_{y0}t_0^2/4$ ,  $\theta_i$  represents  
 494 the absolute rotation of the  $i^{\text{th}}$  yield line and  $l_i$  stands for the actual length of the yield line under  
 495 consideration. Using the symmetry of the proposed yield line model as well as for the sake of  
 496 simplicity, only the left hand side of the model was used to derive the internal work. Referring to Fig.  
 497 14, the lengths of yield lines from 1 to 8 are equal to  $l_1 = b_0/2 \sin[90 - (\lambda - \alpha)]$ ;  $l_2 = x/\cos \alpha$ ;  
 498  $l_3 = b_0/2 \tan(\lambda - \alpha) + x \tan \alpha$ ;  $l_4 = 2x \tan \alpha$ ;  $l_5 = b_1$ ;  $l_6 = h_1$ ;  $l_7 = h_1 \cos \omega + (b_0/2)/\tan(\lambda - \alpha)$ ;  
 499 and  $l_8 = x/\cos \alpha$ , where  $x = b_0(1 - \beta')/2$ . Thus, the total internal work can be calculated as follows:

$$W_i = 2M_p \sum_{i=1}^8 \theta_i l_i = 2M_p \left( \frac{\delta}{p} l_1 + \frac{\delta}{l_2} l_2 + \frac{\delta}{l_3} l_3 + \frac{\delta}{x} l_4 + \frac{\delta}{p} l_5 + \frac{\delta}{q} l_6 + \frac{\delta}{q} l_7 + \frac{\delta}{l_8} l_8 \right) \quad (13)$$

500 where  $p$  and  $q$  respectively represent the average distances of the yield lines  $l_1$  and  $l_7$  from the brace  
 501 member, and expressed as follows:

$$p = \frac{p_1 + p_2}{2} = \frac{1}{2} (l_2 \sin \lambda + l_3 \sin [90 - (\lambda - \alpha)]) \quad (14)$$

$$q = \frac{q_1 + q_2}{2} = \frac{1}{2} (l_3 \sin \psi + l_8 \sin [90 - \alpha + \omega]) \quad (15)$$

502 By applying a virtual unit displacement, i.e.  $\delta = 1$ , and substituting the values of  $l_1$  to  $l_8$  and  $M_p$ ,  
 503 Eq. (13) can be simplified as:

$$W_i = 2 \left[ \frac{f_{y0} t_0^2}{4} \left( 3 + 2 \tan \alpha + \frac{\frac{b_0}{2 \sin [90 - (\lambda - \alpha)]} + b_1}{p} + \frac{h_1 (1 + \cos \omega) + \frac{b_0}{2} \tan (\lambda - \alpha)}{q} \right) \right] \quad (16)$$

504 In Eq. (15), the angle  $\psi$  can be determined as  $\psi = \tan^{-1} \left[ (b_0/2) / \{l_3 + (h_1 \cos \omega - x \tan \alpha)\} \right]$ . A  
 505 sensitivity analysis was conducted by adopting different values of  $\alpha$  and  $\lambda$ . Overall, the joint  
 506 resistances of BR T- and X-joints test and FE specimens correlated well with  $\alpha = 40^\circ$  and  $\lambda = 50^\circ$ , as  
 507 shown by the values of  $P_m$  and  $V_p$  in Tables 5 and 6. Therefore, on substituting,  $\alpha = 40^\circ$  and  $\lambda = 50^\circ$   
 508 in Eq. (16) and equating external and internal work done, the following equation can be obtained.

$$N = \frac{f_{y0} t_0^2}{2} \left\{ 4.7 + \left( \frac{b_0 + 2b_1}{2p} \right) + \left( \frac{h_1 (1 + \cos \omega) + 0.1b_0}{q} \right) \right\} \quad (17)$$

509 It can be noticed that Eq. (17) cannot include one of the important geometric parameters,  $2\gamma$   
 510 ( $=b_0/t_0$ ), which accounts for the slenderness of the chord connecting face(s). Therefore, a reduction  
 511 factor based on  $2\gamma$  was applied to Eq. (17) to finally derive the nominal resistance equation for cold-  
 512 formed S960 steel grade BR T- and X-joints, as follows:

$$N_{pn3} = [0.58 - 0.007(2\gamma)] \left[ f_{y0} t_0^2 \left\{ 4.7 + \left( \frac{b_0 + 2b_1}{2p} \right) + \left( \frac{h_1 (1 + \cos \omega) + 0.1b_0}{q} \right) \right\} \right] \quad (18)$$

513 where

$$p = \frac{b_0}{20} (10 - 9\beta') \quad (19)$$

$$q = \frac{b_0}{20} \left[ (5 - 4\beta') \sin \psi + \frac{b_0}{3} (1 - \beta') \sin (\omega + 50) \right] \quad (20)$$

$$\psi = \tan^{-1} \left( \frac{b_0}{2h_1 \cos \omega + \frac{b_0}{5}} \right) \quad (21)$$

514 The comparisons of  $N_f$  of BR T- and X-joints test and FE specimens with corresponding  
 515 nominal resistances predicted from Eq. (18) are shown in Tables 5 and 6, respectively. As shown in  
 516 Table 5, the  $P_m$  and  $V_p$  of Eq. (18) for BR T-joints are 1.02 and 0.215, respectively. On the other hand,  
 517 referring to Table 6, the  $P_m$  and  $V_p$  of Eq. (18) for BR X-joints are 1.01 and 0.232, respectively. For  
 518 Eq. (18),  $\phi$  equal to 0.70 was recommended, resulting in  $\beta_0$  equal to 2.66 and 2.53 for BR T- and

519 X-joints, respectively. Thus, Eq. (18) must be multiplied by  $\phi$  equal to 0.70 to get the corresponding  
520 design resistances ( $N_d$ ). The comparisons of  $N_f$  of BR T- and X-joints test and FE specimens with  
521 nominal resistances predicted from Bae et al. [8], CHS-to-RHS design rule of EC3 [9] and Eq. (18)  
522 are graphically presented in Figs. 15 and 16, respectively. Compared to the existing design provisions,  
523 the predictions from Eq. (18) are relatively more accurate, less dispersed and reliable for the  $N_f$  of  
524 CFHSS BR T- and X-joints.

525 For BR T- and X-joints, the distributions of the comparison ratios of the  $N_f$  of test and FE  
526 specimens-to-nominal resistances predicted from Bae et al. [8], EC3 [9] and design equations  
527 proposed in this study under proposal-1 are shown in Figs. 17 and 18, respectively.

528

## 529 **8. Conclusions**

530 The static resistances of cold-formed steel brace-rotated (BR) tubular T- and X-joints were  
531 numerically investigated in this study. The braces of BR T- and X-joints were made of square and  
532 rectangular hollow sections (SHS and RHS), however, chords were only made of SHS. The nominal  
533 0.2% proof stress of tubular members was 960 MPa. The rotation angle ( $\omega$ ) of brace members  
534 ranged from 15° to 63°. The test results reported in Pandey and Young [1,2] were used to develop  
535 accurate finite element (FE) models of BR T- and X-joints. An extensive FE parametric study was  
536 subsequently performed, which comprised 96 BR T-joints and 96 BR X-joints. The welds and  
537 associated weld heat affected regions were included in all FE parametric models, which appreciably  
538 improved the accuracy of numerical results. In this study, the joint failure resistances ( $N_f$ ) of all BR  
539 T- and X-joints FE specimens were controlled by the  $0.3b_0$  ultimate deformation limit criterion.

540 The BR T- and X-joints test and FE specimens were failed by two failure modes, namely chord  
541 face failure (F) mode and a combination of chord face and chord side wall failure mode, i.e. combined  
542 failure (F+S) mode. The design rules given in Bae et al. [8] and EC3 [9] are found to be unsuitable  
543 for the design of BR T- and X-joints investigated in this study. As a result, accurate, less dispersed  
544 and reliable design equations are proposed, by three design approaches, to predict the joint failure  
545 resistances ( $N_f$ ) of cold-formed S960 steel grade BR T- and X-joints. In the first approach, a unified  
546 design equation has been proposed. In the second approach, design equations are proposed by

547 applying correction factors on the existing CHS-to-RHS design rule given in EC3 [9]. However, in  
548 the third approach, the design equation is developed using a simplified yield line model. The design  
549 equations proposed in this study are valid for  $0.20 \leq \beta \leq 0.67$ ,  $0.26 \leq \beta' \leq 0.88$ ,  $16.6 \leq 2\gamma \leq 40$ ,  $0.50$   
550  $\leq \tau \leq 1.28$ ,  $15^\circ \leq \omega \leq 63^\circ$  and  $\theta_l = 90^\circ$ .

551

## 552 **Acknowledgement**

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554 Council of the Hong Kong Special Administrative Region, China (PolyU 15218720).

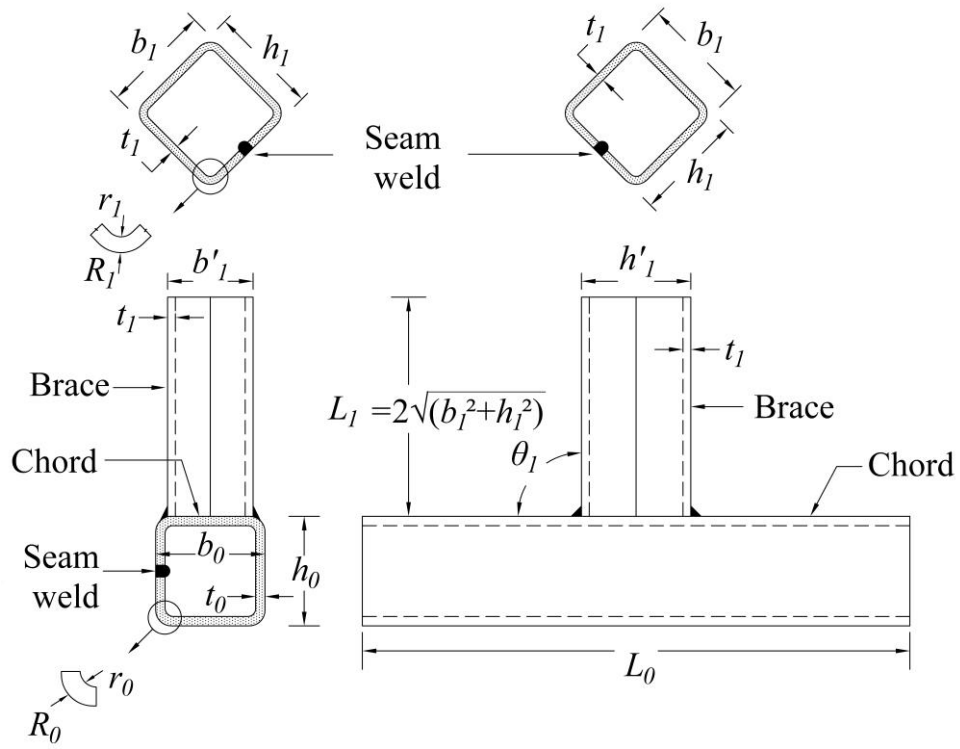
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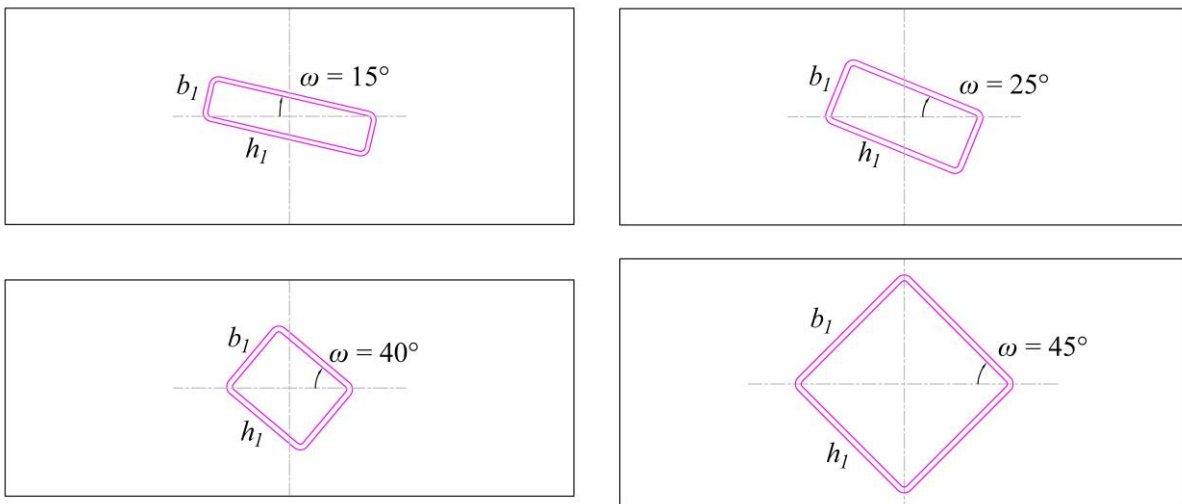
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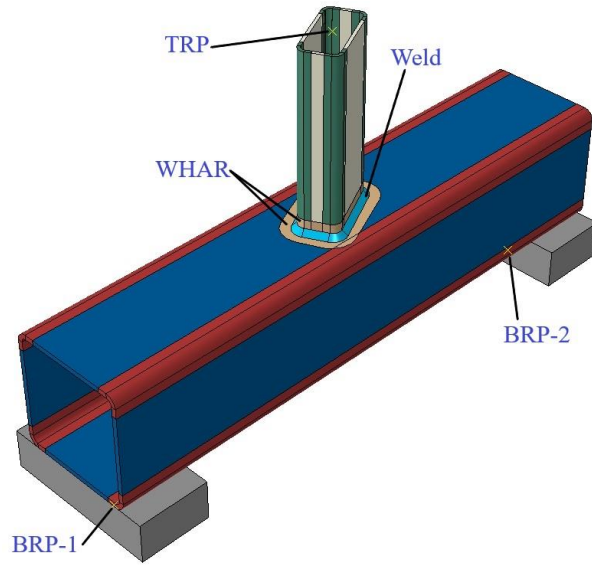


(a) Definitions of notations for BR T-joint (also valid for BR X-joint).

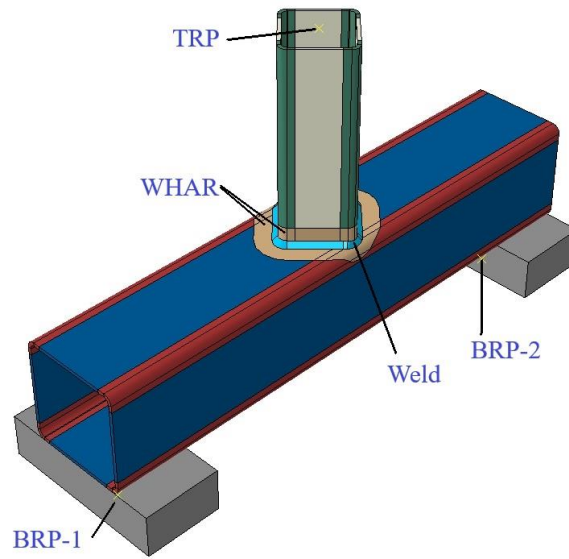


(b) Orientations of brace member adopted in the parametric study.

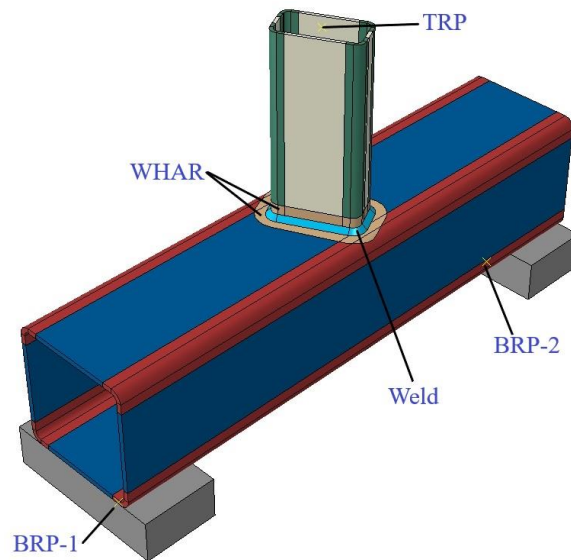
Fig. 1. Notations and brace rotation angles of BR T- and X-joints.



(a) BR T-joint FE model with  $\omega = 27^\circ$ .

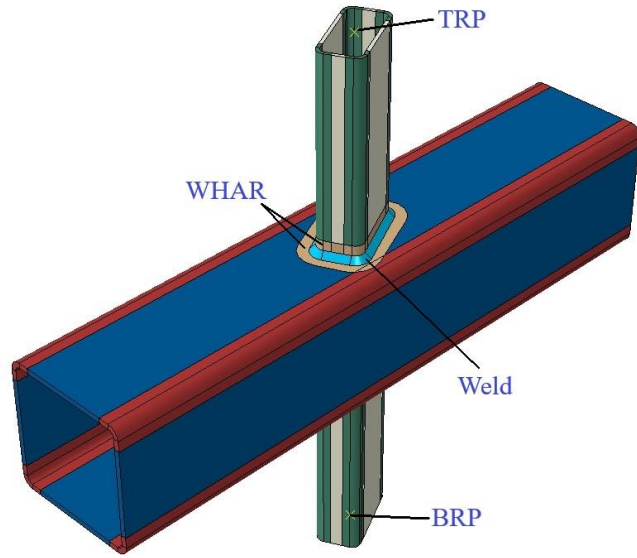


(b) BR T-joint FE model with  $\omega = 45^\circ$ .

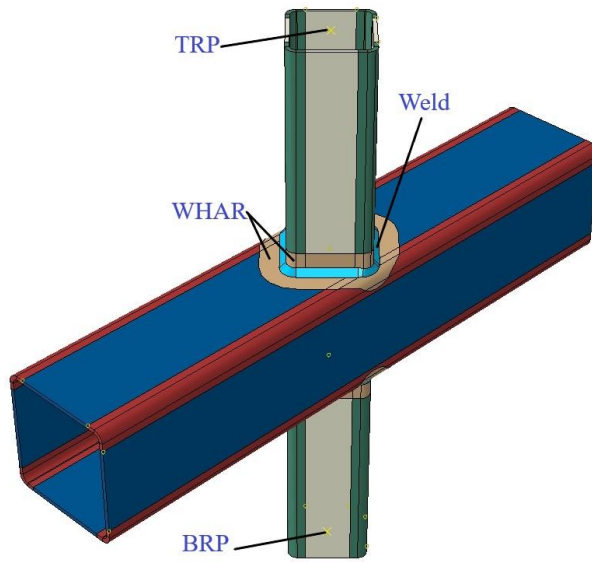


(c) BR T-joint FE model with  $\omega = 63^\circ$ .

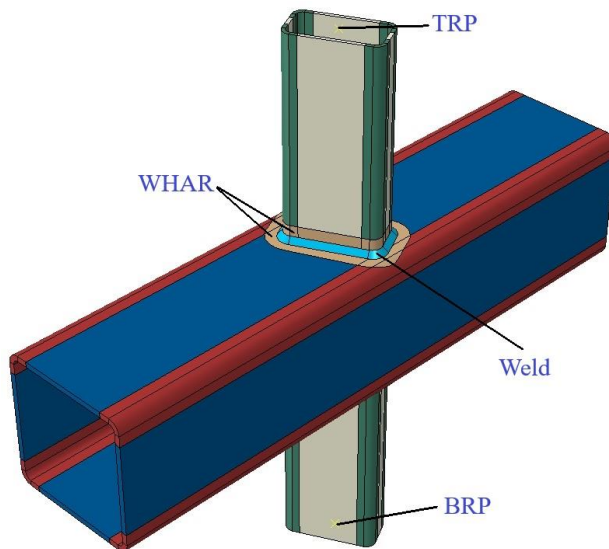
Fig. 2. Typical FE models of BR T-joints.



(a) BR X-joint FE model with  $\omega = 27^\circ$ .



(b) BR X-joint FE model with  $\omega = 45^\circ$ .



(c) BR X-joint FE model with  $\omega = 63^\circ$ .

Fig. 3. Typical FE models of BR X-joints.

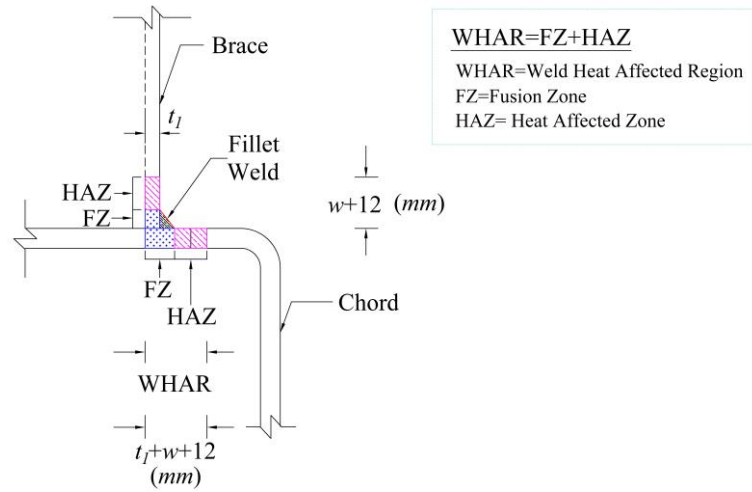


Fig. 4. Definition of weld heat affected region (WHAR) [15].

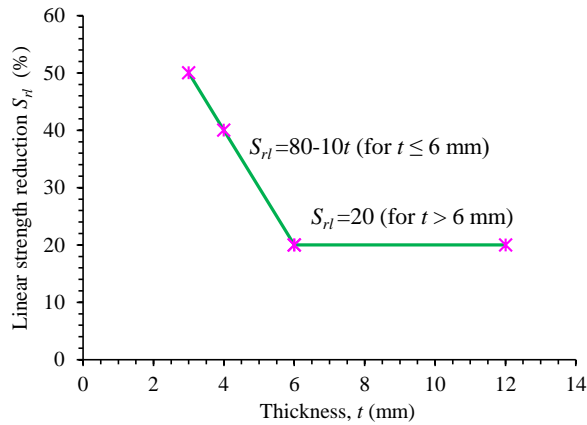


Fig. 5. Strength reduction model for WHAR of S900 and S960 steel grades tubular joints [15].

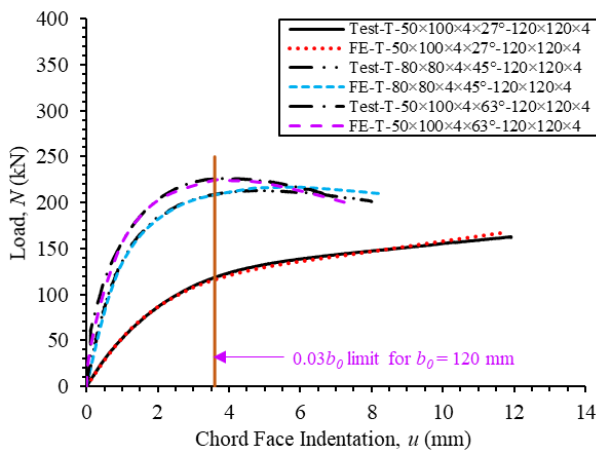


Fig. 6. Comparisons of test and FE load vs chord face indentation curves for BR T-joints.

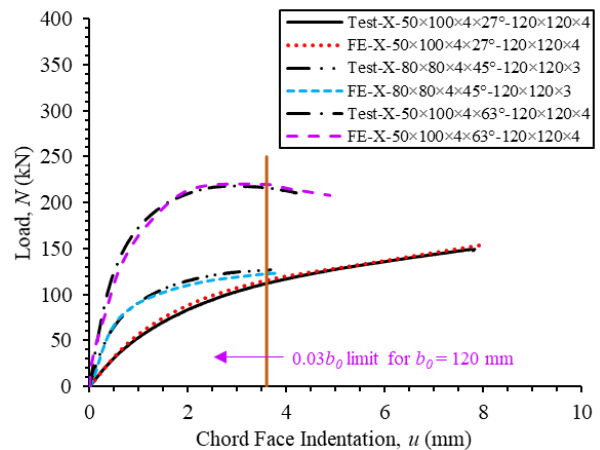
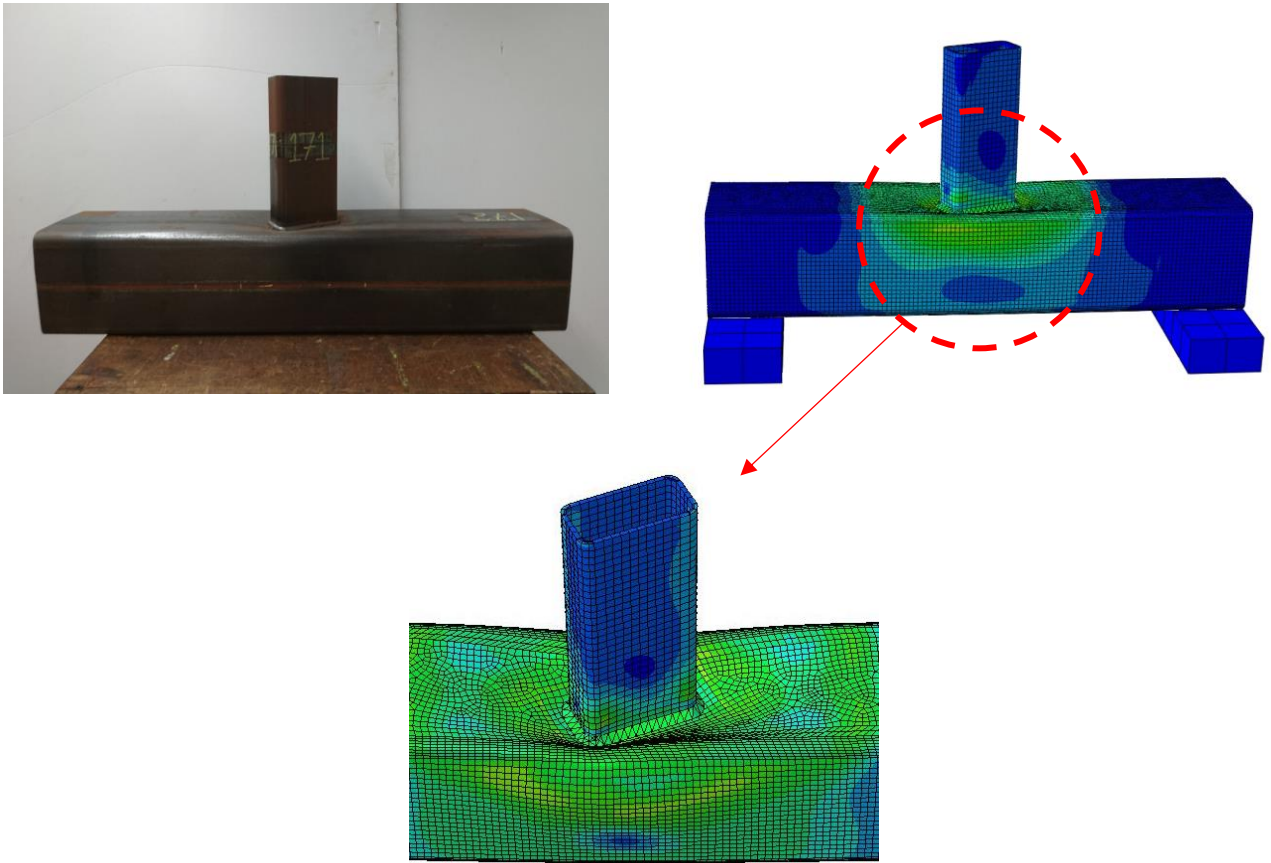
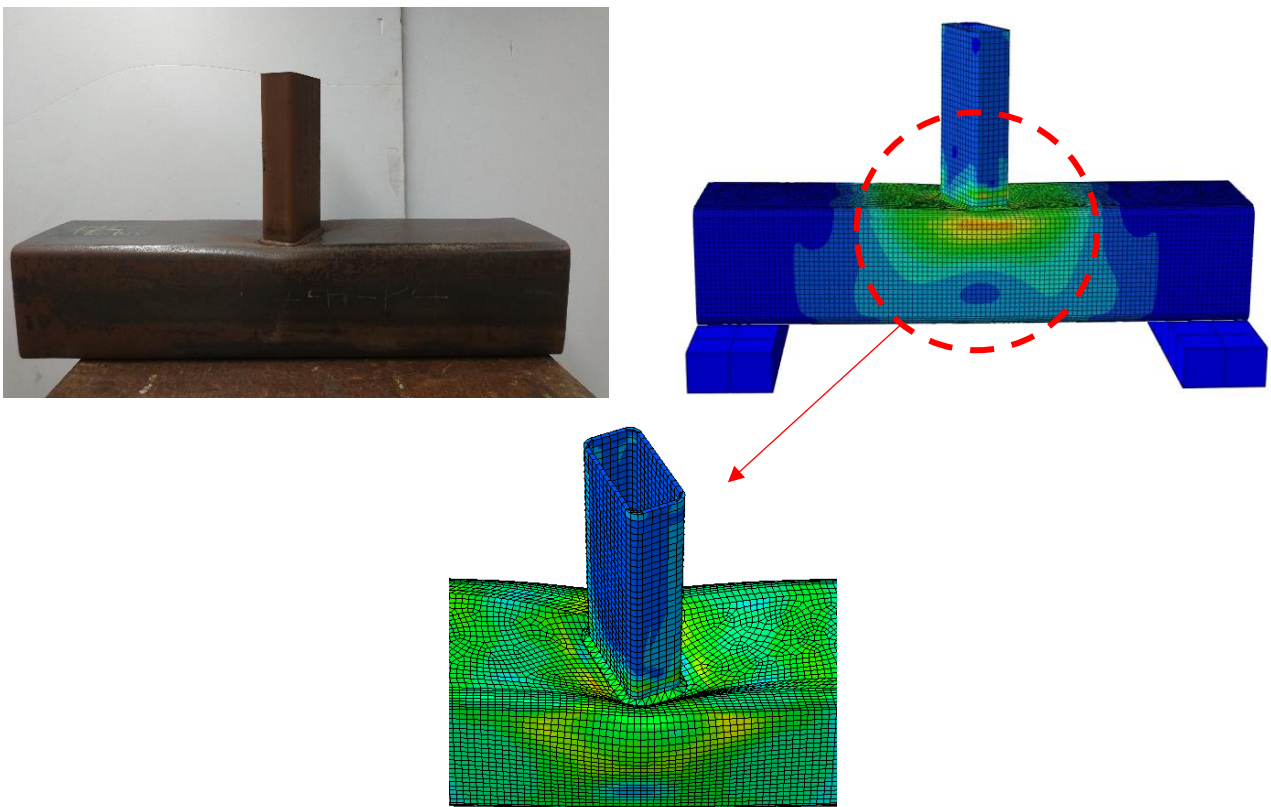


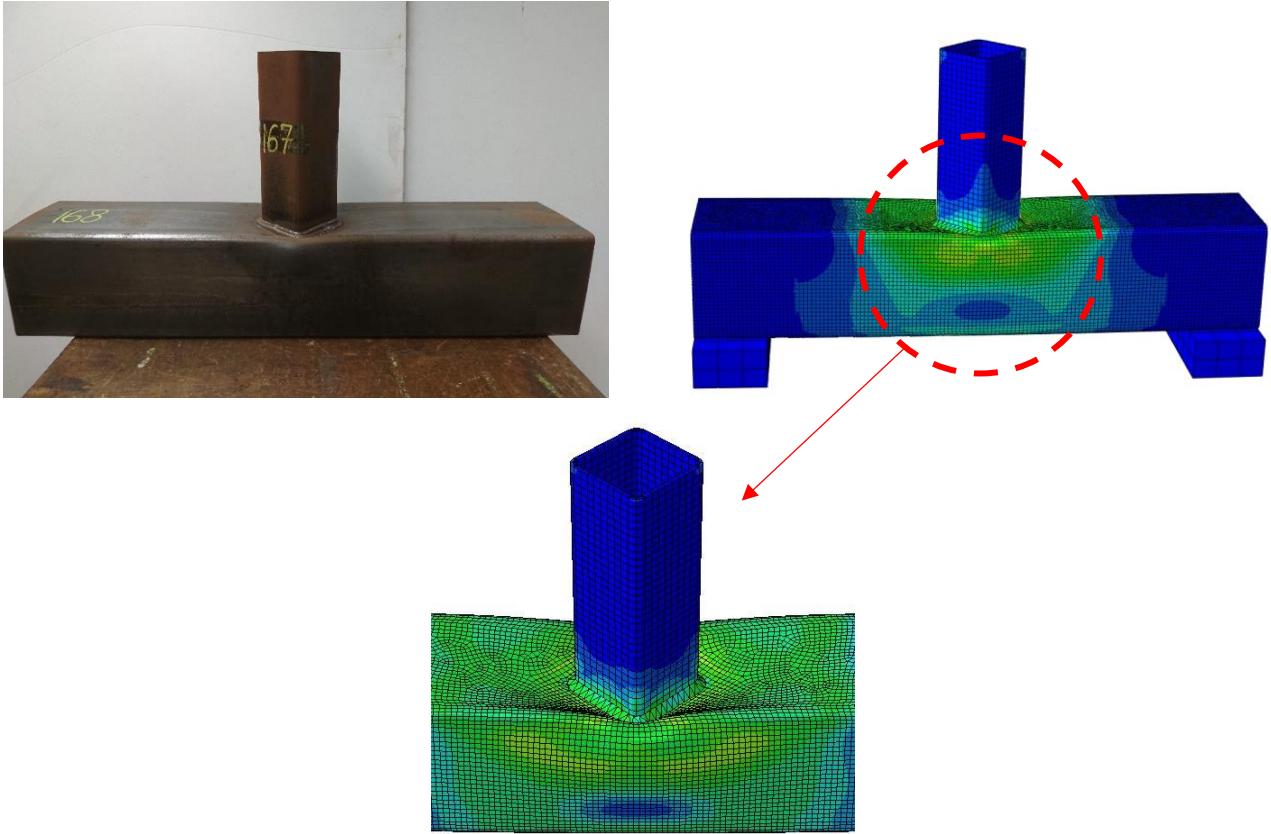
Fig. 7. Comparisons of test and FE load vs chord face indentation curves for BR X-joints.



(a) Test vs FE comparison for chord face failure (F) mode of T-50×100×4×27°-150×150×6 ( $\beta' = 0.55$ ).

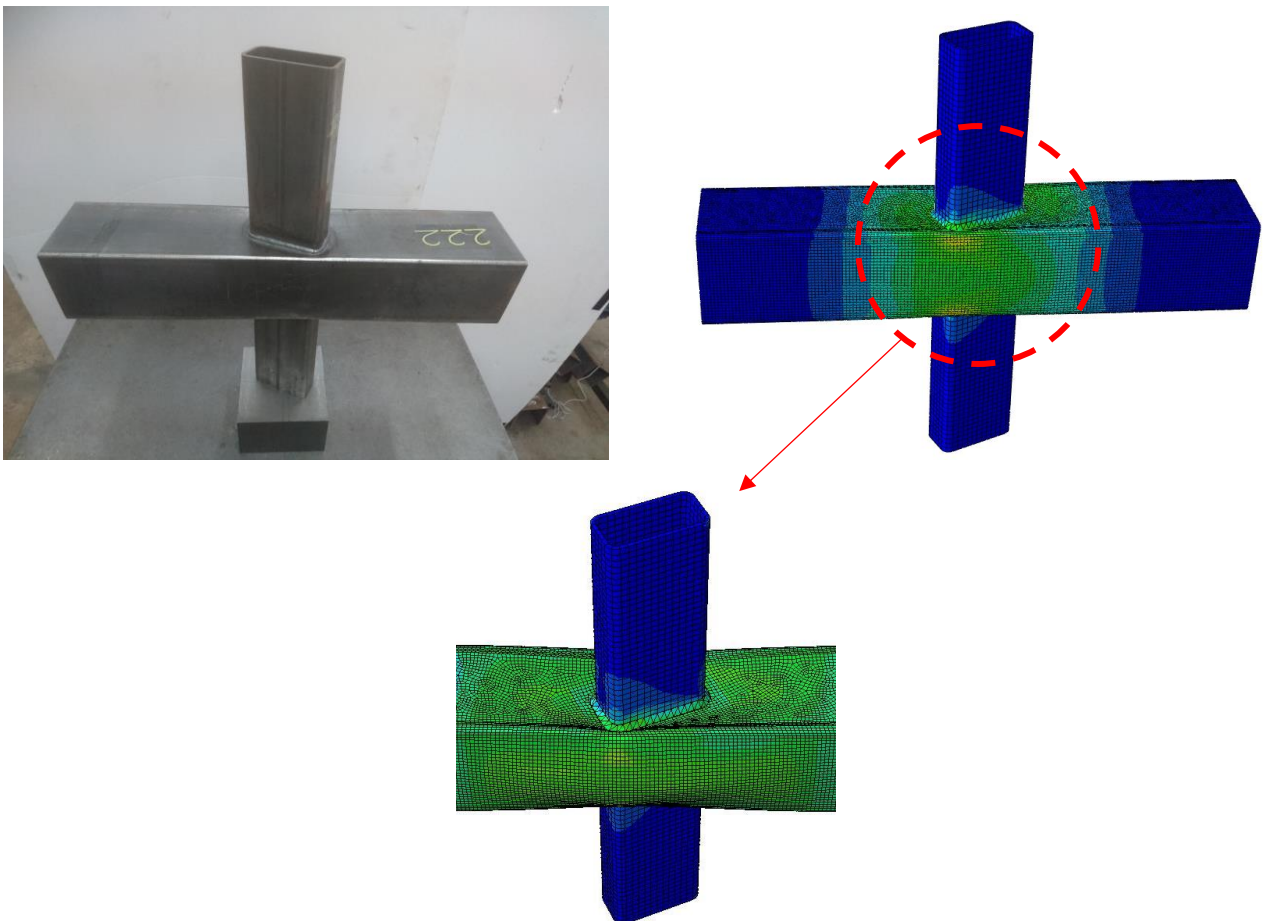


(b) Test vs FE comparison for chord face failure (F) mode of T-50×100×4×63°-150×150×6 ( $\beta' = 0.70$ ).

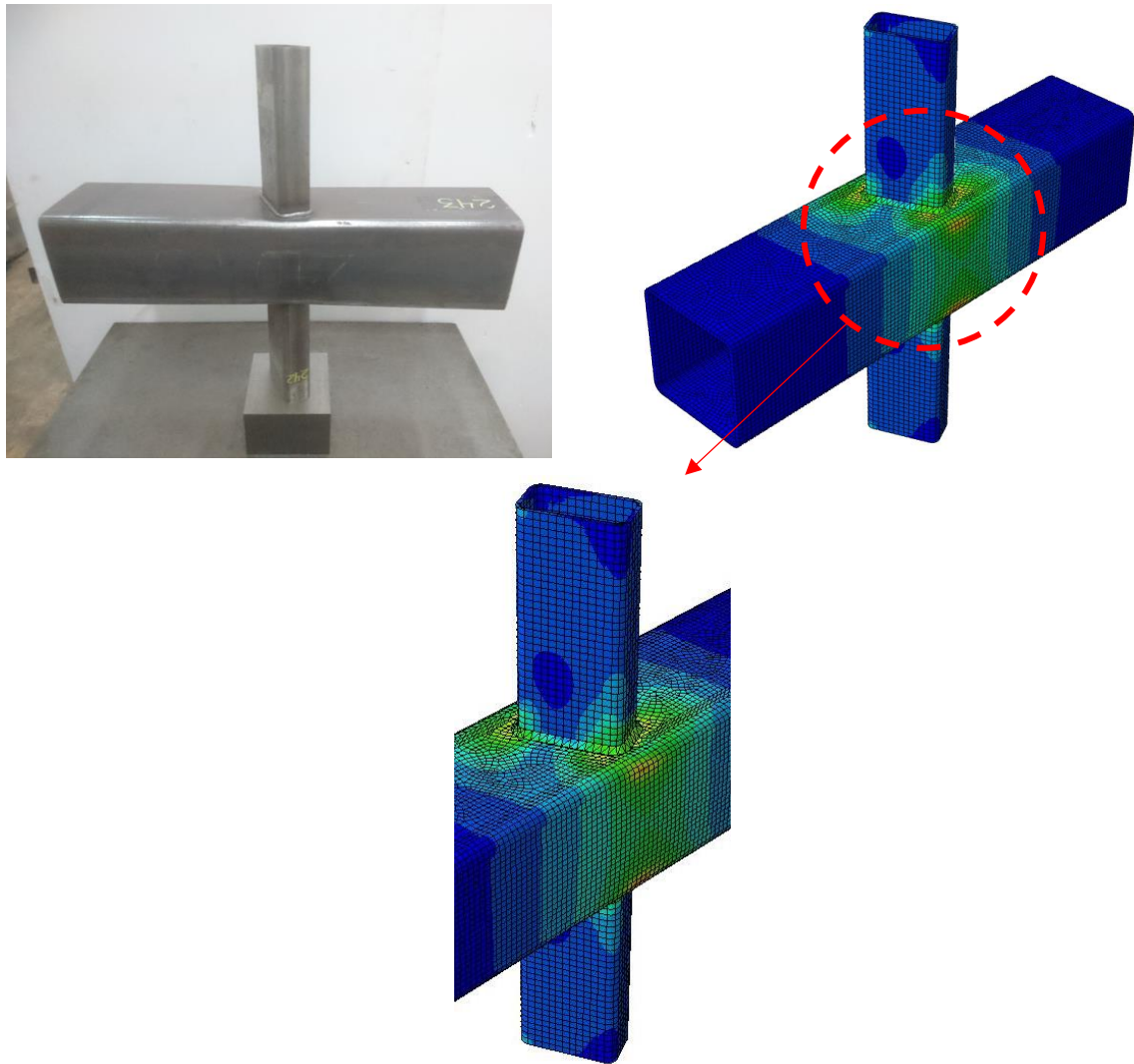


(c) Test vs FE comparison for combined failure (F+S) mode of T-80×80×4×45°-120×120×4 ( $\beta' = 0.87$ ).

Fig. 8. Test vs FE failure modes comparisons for BR T-joints.

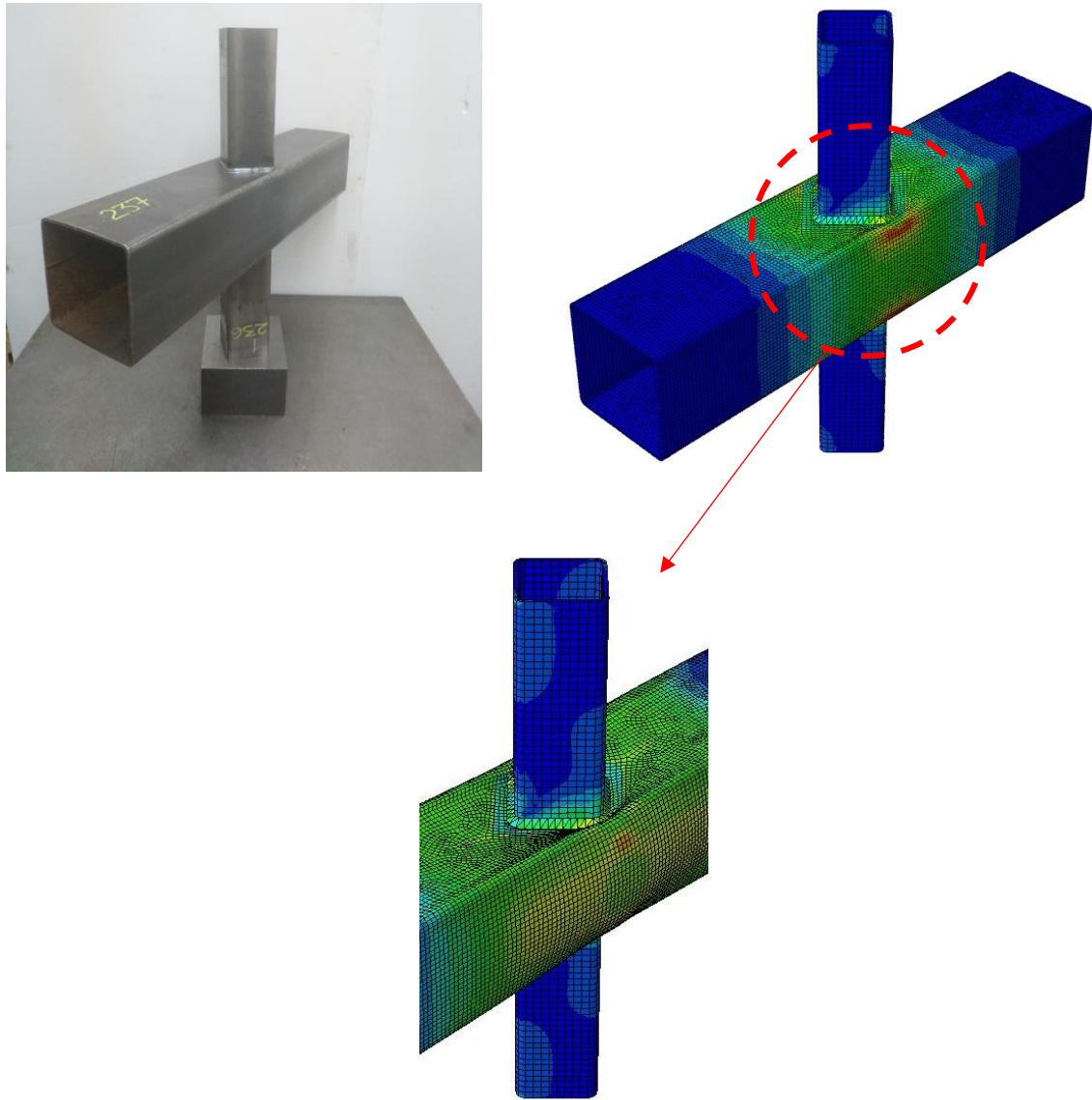


(a) Test vs FE comparison for chord face failure (F) mode of X-50×100×4×27°-120×120×3 ( $\beta' = 0.68$ ).



(b) Test vs FE comparison for chord face failure (F) mode of X-50×100×4×63°-150×150×6 ( $\beta' = 0.70$ ).





(c) Test vs FE comparison for combined failure (F+S) mode of X-80×80×4×45°-120×120×3 ( $\beta' = 0.87$ ).

Fig. 9. Test vs FE failure modes comparisons for BR X-joints.

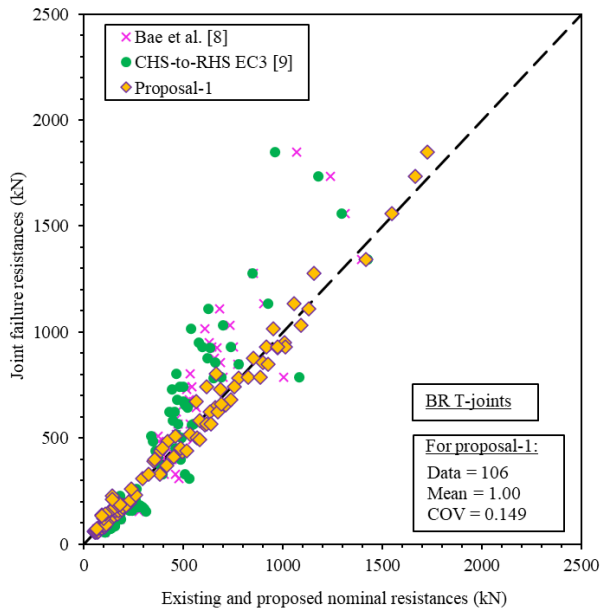


Fig. 10. Comparisons of test and FE joint failure resistances with existing and proposed (Proposal-1) nominal resistances for BR T-joints.

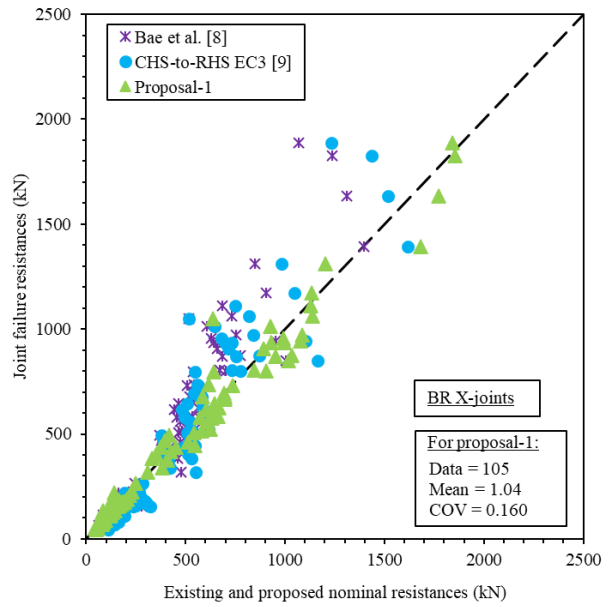


Fig. 11. Comparisons of test and FE joint failure resistances with existing and proposed (Proposal-1) nominal resistances for BR X-joints.

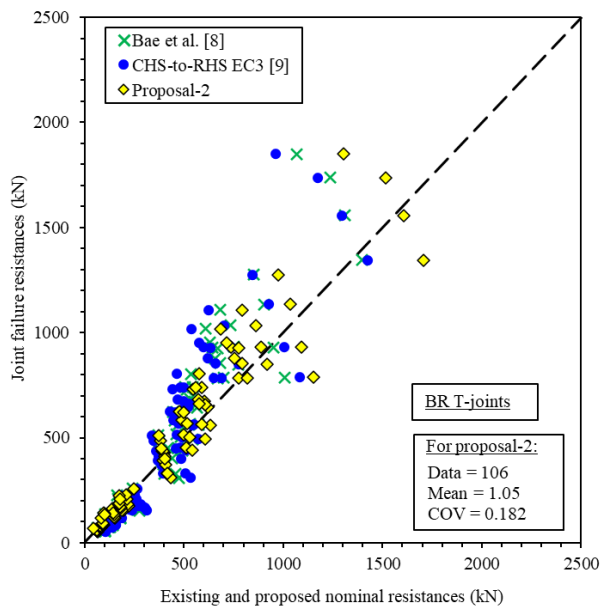


Fig. 12. Comparisons of test and FE joint failure resistances with existing and proposed (Proposal-2) nominal resistances for BR T-joints.

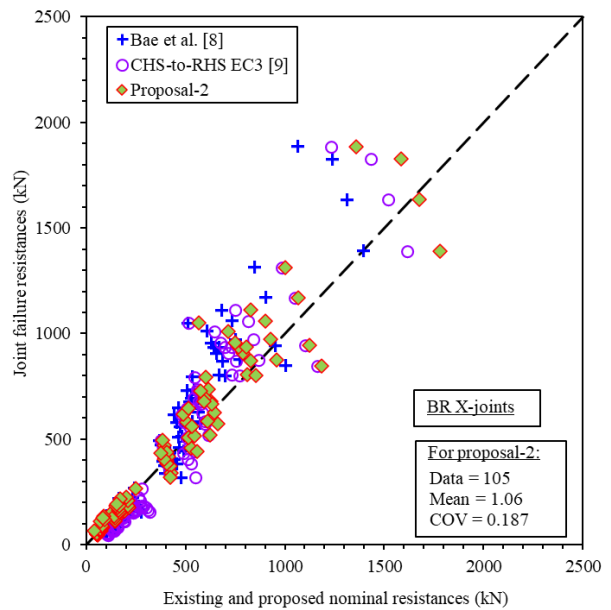


Fig. 13. Comparisons of test and FE joint failure resistances with existing and proposed (Proposal-2) nominal resistances for BR X-joints.

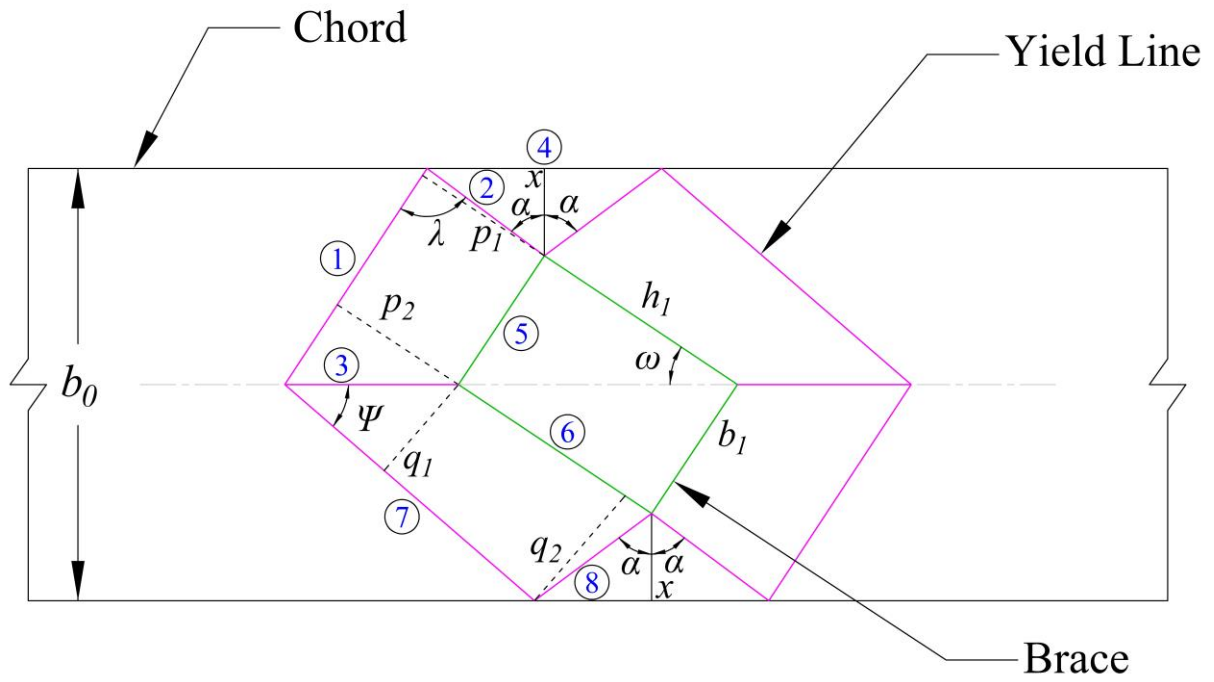


Fig. 14. Simplified yield line model of BR T- and X-joints.

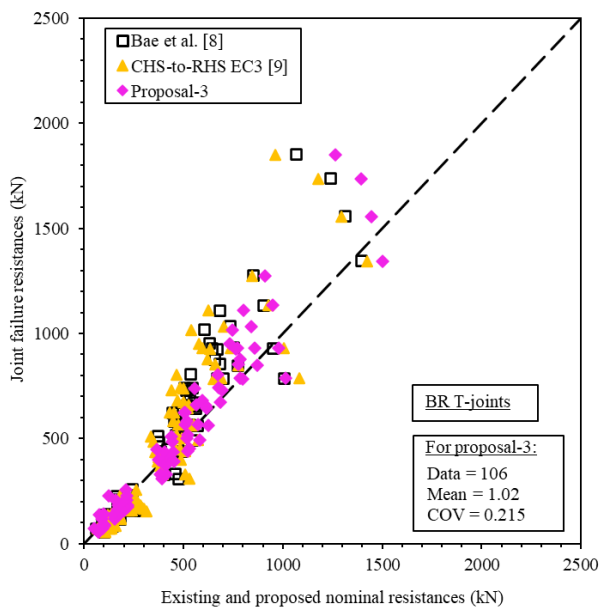


Fig. 15. Comparisons of test and FE joint failure resistances with existing and proposed (Proposal-3) nominal resistances for BR T-joints.

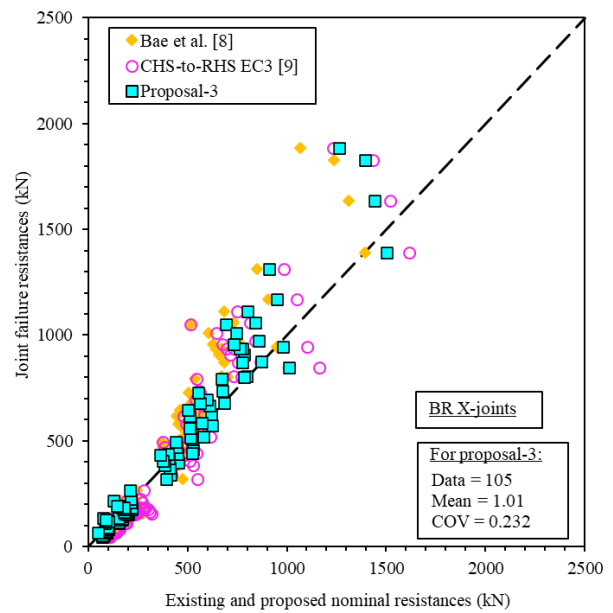
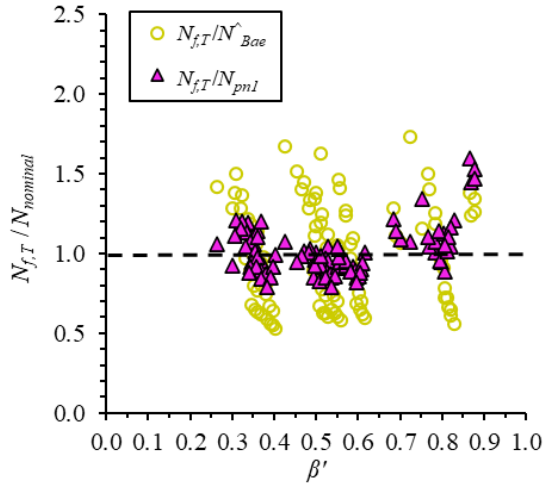
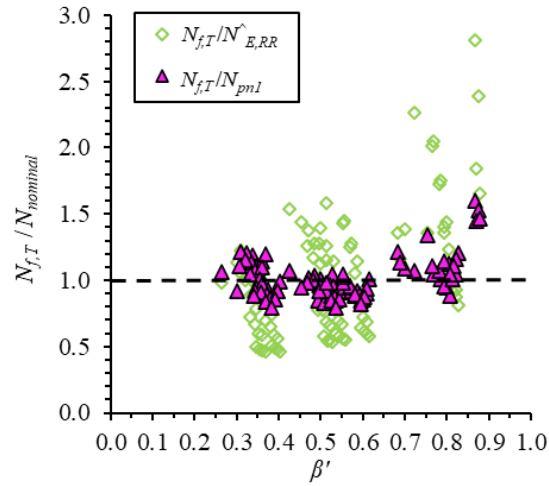


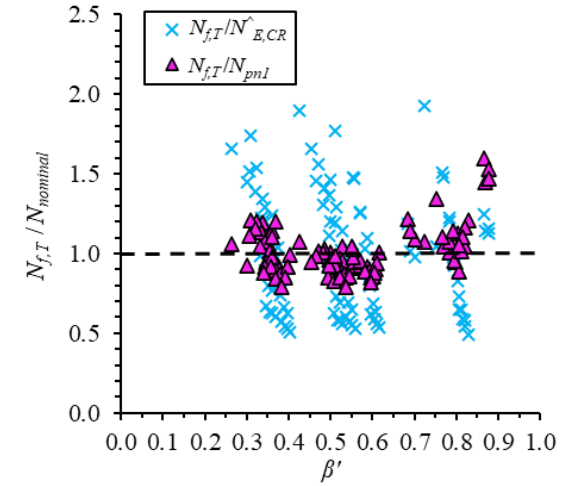
Fig. 16. Comparisons of test and FE joint failure resistances with existing and proposed (Proposal-3) nominal resistances for BR X-joints.



(a) For Bae et al. [8]

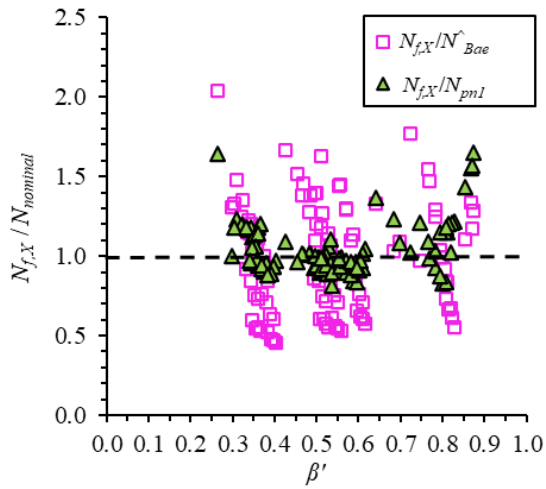


(b) For RHS-to-RHS T-joint of EC3 [9]

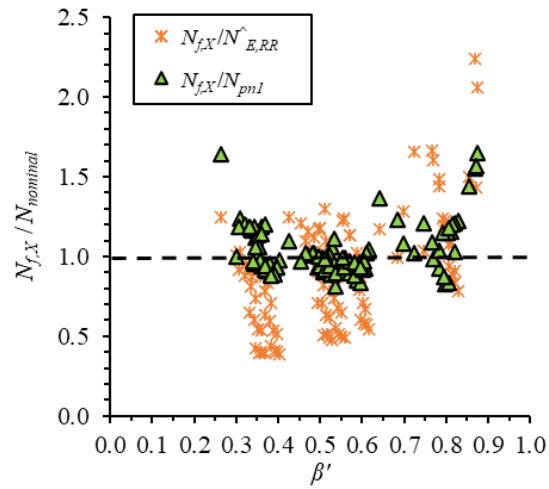


(c) For CHS-to-RHS T-joint of EC3 [9]

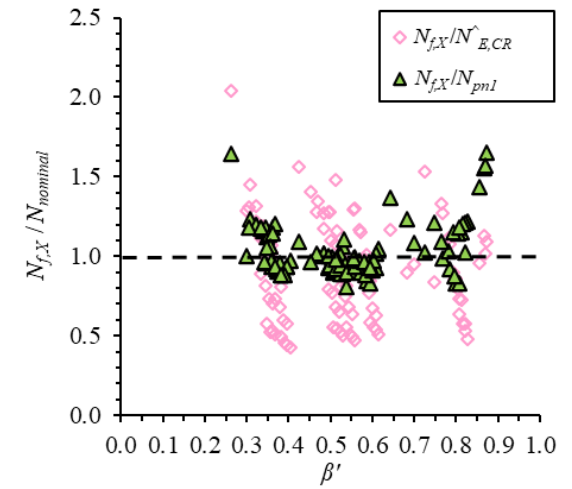
Fig. 17. Distributions of joint failure resistance ( $N_{f,T}$ ) comparisons ratios for BR T-joints.



(a) For Bae et al. [8]



(b) For RHS-to-RHS X-joint of EC3 [9]



(c) For CHS-to-RHS X-joint of EC3 [9]

Fig. 18. Distributions of joint failure resistance ( $N_{f,X}$ ) comparisons ratios for BR X-joints.

Table 1. Test vs FE resistance comparisons for BR T-joints.

Specimens	$\beta'$	Test Joint Failure Resistances (kN) <sup>#</sup>	Test Joint Ultimate Resistances (kN) <sup>#</sup>	Numerical Joint Failure Resistances (kN)	Numerical Joint Ultimate Resistances (kN)	Comparisons	
$T-b_1 \times h_1 \times t_1 \times \omega - b_0 \times h_0 \times t_0$	$\frac{b_1}{b_0}$	$N_{f,T}$	$N_{max,T}$	$N_{f,FE}$	$N_{max,FE}$	$\frac{N_{f,T}}{N_{f,FE}}$	$\frac{N_{max,T}}{N_{max,FE}}$
T-50×100×4×27°-150×150×6	0.55	185.7	-	186.2	-	1.00	-
T-50×100×4×27°-150×150×6-R	0.55	187.8	-	188.1	-	1.00	-
T-50×100×4×63°-150×150×6	0.70	258.1	-	257.8	-	1.00	-
T-50×100×4×27°-120×120×4	0.68	119.4	-	116.1	-	1.03	-
T-50×100×4×63°-120×120×4	0.87	226.2	226.50	223.9	224.1	1.01	1.01
T-50×100×4×27°-120×120×3	0.69	70.8	-	68.5	-	1.03	-
T-50×100×4×63°-120×120×3	0.88	137.9	138.40	137.2	138.71	1.01	1.00
T-80×80×4×45°-120×120×4	0.87	209.4	213.30	209.1	216.8	1.00	0.98
T-80×80×4×45°-140×140×4	0.75	142.0	-	138.1	-	1.03	-
T-80×80×4×45°-120×120×3	0.88	134.6	137.10	134.2	134.1	1.00	1.02
					Mean ( $P_m$ )	1.01	1.00
					COV ( $V_p$ )	0.014	0.017

Note: “ - ” denotes not applicable; <sup>#</sup>data obtained from Pandey and Young [1].

Table 2. Test vs FE resistance comparisons for BR X-joints.

Specimens	$\beta'$	Test Joint Failure Resistances (kN) <sup>#</sup>	Test Joint Ultimate Resistances (kN) <sup>#</sup>	Numerical Joint Failure Resistances (kN)	Numerical Joint Ultimate Resistances (kN)	Comparisons	
$X-b_1 \times h_1 \times t_1 \times \omega - b_0 \times h_0 \times t_0$	$\frac{b_1}{b_0}$	$N_{f,X}$	$N_{max,X}$	$N_{f,FE}$	$N_{max,FE}$	$\frac{N_{f,X}}{N_{f,FE}}$	$\frac{N_{max,X}}{N_{max,FE}}$
X-50×100×4×27°-150×150×6	0.53	184.4	-	182.5	-	1.01	-
X-50×100×4×63°-150×150×6	0.70	266.9	-	265.7	-	1.00	-
X-50×100×4×27°-120×120×4	0.64	112.8	-	115.6	-	0.98	-
X-50×100×4×63°-120×120×4	0.87	218.4	218.4	219.5	219.9	0.99	0.99
X-50×100×4×27°-120×120×3	0.68	64.9	-	65.0	-	1.00	-
X-50×100×4×63°-120×120×3	0.87	136.3	136.3	136.0	136.10	1.00	1.00
X-80×80×4×45°-120×120×4	0.85	191.7	197.6	182.5	190.70	1.05	1.04
X-80×80×4×45°-140×140×4	0.75	120.1	-	116.1	-	1.03	-
X-80×80×4×45°-120×120×3	0.87	126.5	127.0	122.9	123.3	1.03	1.03
					Mean ( $P_m$ )	1.01	1.02
					COV ( $V_p$ )	0.023	0.021

Note: “ - ” denotes not applicable; <sup>#</sup>data obtained from Pandey and Young [2].

Table 3. Ranges of critical parameters used in parametric study.

Parameters	Validity Ranges
$\beta (b_1/b_0)$	[0.20 to 0.67]
$\beta' (b'_1/b_0)$	[0.26 to 0.88]
$2\gamma (b_0/t_0)$	[16.6 to 40]
$\tau (t_1/t_0)$	[0.50 to 1.28]
$\omega$	[15° to 63°]
$\theta_1$	90°

Table 4. Mechanical properties of tubular member and weld adopted in parametric study.

Materials	Measured Mechanical Properties						
	$E$ (GPa)	$\sigma_{0.2}$ (MPa)	$\epsilon_{0.2}$ (%)	$\sigma_u$ (MPa)	$\epsilon_u$ (%)	$\epsilon_f$ (%)	$n$
SHS/RHS (150×150×6) [11]	208.5	1059.1	0.71	1145.7	1.48	9.37 <sup>a</sup>	5.31
Weld Material [12]	202.7	965.2	0.68	1023.4	5.41	17.15 <sup>b</sup>	8.13

Note: <sup>a</sup> fracture strain based on 50 mm gauge length; <sup>b</sup> fracture strain based on 25 mm gauge length.

Table 5. Comparisons between test and FE resistances with existing and proposed nominal resistances for BR T-joints.

Specimens	$\beta'$	Joint Failure Resistances (kN)	Comparisons					
			$\frac{N_{f,T}}{N_{Bae}^{\wedge}}$	$\frac{N_{f,T}}{N_{E,RR}^{\wedge}}$	$\frac{N_{f,T}}{N_{E,CR}^{\wedge}}$	$\frac{N_{f,T}}{N_{pn1}}$	$\frac{N_{f,T}}{N_{pn2}}$	$\frac{N_{f,T}}{N_{pn3}}$
T- $b_1 \times h_1 \times t_1 \times \omega$ - $b_0 \times h_0 \times t_0$	$\frac{b_1}{b_0}$	$N_{f,T}$						
T-40×150×6×15°-200×200×12	0.34	671.9	1.22	0.99	1.34	1.19	1.12	0.98
T-40×150×7.8×15°-200×200×12	0.32	740.7	1.36	1.11	1.54	1.20	1.26	1.09
T-40×150×9.6×15°-200×200×12	0.31	804.0	1.50	1.22	1.74	1.21	1.39	1.20
T-60×150×12×15°-200×200×12	0.26	730.2	1.42	0.99	1.66	1.06	1.34	1.05
T-40×150×5×15°-200×200×10	0.35	390.2	1.01	0.79	1.06	1.10	0.99	0.86
T-40×150×6.5×15°-200×200×10	0.33	438.0	1.15	0.90	1.22	1.13	1.13	0.97
T-40×150×8×15°-200×200×10	0.32	484.5	1.29	1.00	1.39	1.15	1.27	1.09
T-40×150×10×15°-200×200×10	0.31	510.9	1.38	1.06	1.51	1.11	1.37	1.16
T-40×150×3.33×15°-200×200×6.66	0.36	134.1	0.77	0.58	0.77	1.14	0.94	0.78
T-40×150×4.33×15°-200×200×6.66	0.35	137.9	0.80	0.60	0.80	1.06	0.97	0.81
T-40×150×5.33×15°-200×200×6.66	0.34	152.5	0.89	0.67	0.90	1.07	1.09	0.90
T-40×150×6.66×15°-200×200×6.66	0.33	164.3	0.97	0.72	0.99	1.04	1.19	0.98
T-40×150×2.5×15°-200×200×5	0.37	61.1	0.62	0.46	0.61	1.20	1.01	0.78
T-40×150×3.25×15°-200×200×5	0.36	62.3	0.64	0.47	0.62	1.10	1.03	0.80
T-40×150×4×15°-200×200×5	0.36	62.9	0.65	0.48	0.63	1.01	1.05	0.81
T-40×150×5×15°-200×200×5	0.35	65.4	0.68	0.50	0.67	0.94	1.11	0.85
T-60×60×6×45°-200×200×12	0.37	561.3	0.98	0.91	1.03	0.93	0.89	0.90
T-60×60×7.8×45°-200×200×12	0.36	643.6	1.14	1.06	1.23	0.97	1.04	1.04
T-60×60×9.6×45°-200×200×12	0.34	656.6	1.18	1.09	1.29	0.92	1.08	1.07

T-60×60×12×45°-200×200×12	0.30	682.5	1.28	1.14	1.45	0.92	1.20	1.14
T-60×60×5×45°-200×200×10	0.38	329.6	0.82	0.74	0.83	0.87	0.79	0.80
T-60×60×6.5×45°-200×200×10	0.37	371.8	0.94	0.84	0.97	0.89	0.91	0.91
T-60×60×8×45°-200×200×10	0.36	412.7	1.06	0.94	1.10	0.91	1.03	1.01
T-60×60×10×45°-200×200×10	0.34	437.0	1.14	1.01	1.20	0.88	1.11	1.08
T-60×60×3.33×45°-200×200×6.66	0.40	116.0	0.64	0.56	0.62	0.92	0.77	0.74
T-60×60×4.33×45°-200×200×6.66	0.39	120.8	0.67	0.59	0.66	0.87	0.81	0.78
T-60×60×5.33×45°-200×200×6.66	0.38	132.7	0.75	0.65	0.74	0.87	0.90	0.86
T-60×60×6.66×45°-200×200×6.66	0.37	142.1	0.81	0.70	0.80	0.84	0.98	0.93
T-60×60×2.5×45°-200×200×5	0.40	54.2	0.53	0.46	0.51	0.99	0.85	0.76
T-60×60×3.25×45°-200×200×5	0.40	55.9	0.55	0.48	0.53	0.92	0.88	0.79
T-60×60×4×45°-200×200×5	0.39	56.9	0.56	0.49	0.54	0.85	0.91	0.80
T-60×60×5×45°-200×200×5	0.38	59.0	0.59	0.50	0.57	0.79	0.95	0.84
T-60×130×6×25°-200×200×12	0.50	783.6	1.17	1.12	1.20	1.01	1.01	0.98
T-60×130×7.8×25°-200×200×12	0.48	877.2	1.34	1.28	1.41	1.03	1.16	1.12
T-60×130×9.6×25°-200×200×12	0.47	930.1	1.45	1.38	1.56	1.02	1.26	1.20
T-60×130×12×25°-200×200×12	0.42	1017.2	1.68	1.54	1.89	1.07	1.49	1.37
T-60×130×5×25°-200×200×10	0.51	452.7	0.96	0.88	0.94	0.93	0.88	0.86
T-60×130×6.5×25°-200×200×10	0.50	516.0	1.11	1.02	1.11	0.96	1.03	0.99
T-60×130×8×25°-200×200×10	0.48	582.4	1.28	1.17	1.30	1.00	1.19	1.13
T-60×130×10×25°-200×200×10	0.47	624.1	1.40	1.26	1.45	0.98	1.31	1.23
T-60×130×3.33×25°-200×200×6.66	0.52	158.1	0.74	0.66	0.69	0.98	0.85	0.79
T-60×130×4.33×25°-200×200×6.66	0.51	163.7	0.77	0.68	0.73	0.91	0.89	0.82
T-60×130×5.33×25°-200×200×6.66	0.51	183.9	0.88	0.77	0.84	0.94	1.02	0.93
T-60×130×6.66×25°-200×200×6.66	0.49	184.4	0.90	0.78	0.86	0.85	1.04	0.94
T-60×130×2.5×25°-200×200×5	0.53	73.7	0.61	0.54	0.56	1.05	0.93	0.80
T-60×130×3.25×25°-200×200×5	0.52	75.7	0.63	0.55	0.58	0.97	0.97	0.83
T-60×130×4×25°-200×200×5	0.52	74.4	0.62	0.54	0.58	0.87	0.96	0.82
T-60×130×5×25°-200×200×5	0.51	79.1	0.67	0.58	0.63	0.83	1.04	0.88
T-75×90×6×40°-200×200×12	0.53	786.0	1.12	1.14	1.14	0.95	0.96	1.00
T-75×90×7.8×40°-200×200×12	0.51	854.4	1.25	1.26	1.29	0.95	1.08	1.10
T-75×90×9.6×40°-200×200×12	0.50	925.7	1.38	1.39	1.46	0.95	1.20	1.21
T-75×90×12×40°-200×200×12	0.45	951.9	1.51	1.44	1.65	0.95	1.33	1.30
T-75×90×5×40°-200×200×10	0.54	439.0	0.89	0.87	0.86	0.85	0.81	0.84
T-75×90×6.5×40°-200×200×10	0.52	502.9	1.04	1.01	1.02	0.89	0.95	0.97
T-75×90×8×40°-200×200×10	0.51	566.8	1.19	1.15	1.19	0.92	1.10	1.11
T-75×90×10×40°-200×200×10	0.50	622.2	1.34	1.28	1.37	0.92	1.24	1.24
T-75×90×3.33×40°-200×200×6.66	0.55	156.6	0.70	0.67	0.65	0.91	0.80	0.79
T-75×90×4.33×40°-200×200×6.66	0.54	163.4	0.74	0.70	0.69	0.86	0.84	0.83
T-75×90×5.33×40°-200×200×6.66	0.53	181.4	0.83	0.78	0.78	0.87	0.95	0.93
T-75×90×6.66×40°-200×200×6.66	0.52	196.0	0.91	0.84	0.86	0.85	1.05	1.02
T-75×90×2.5×40°-200×200×5	0.56	74.2	0.58	0.55	0.53	0.99	0.89	0.81
T-75×90×3.25×40°-200×200×5	0.55	75.8	0.60	0.56	0.55	0.91	0.92	0.84
T-75×90×4×40°-200×200×5	0.55	77.9	0.62	0.58	0.57	0.86	0.95	0.87
T-75×90×5×40°-200×200×5	0.54	80.3	0.65	0.60	0.60	0.79	1.00	0.90
T-90×90×6×45°-200×200×12	0.59	847.6	1.10	1.14	1.09	0.92	0.92	0.97
T-90×90×7.8×45°-200×200×12	0.57	931.6	1.24	1.27	1.26	0.92	1.05	1.09
T-90×90×9.6×45°-200×200×12	0.56	1033.3	1.41	1.45	1.47	0.95	1.20	1.23
T-90×90×12×45°-200×200×12	0.51	1108.6	1.62	1.58	1.77	0.98	1.40	1.38
T-90×90×5×45°-200×200×10	0.59	492.6	0.90	0.90	0.86	0.85	0.81	0.85
T-90×90×6.5×45°-200×200×10	0.58	565.1	1.06	1.05	1.03	0.89	0.96	0.99
T-90×90×8×45°-200×200×10	0.57	660.6	1.27	1.25	1.25	0.96	1.15	1.17
T-90×90×10×45°-200×200×10	0.55	740.0	1.46	1.43	1.48	0.98	1.33	1.34
T-90×90×3.33×45°-200×200×6.66	0.61	173.4	0.70	0.68	0.64	0.90	0.78	0.78
T-90×90×4.33×45°-200×200×6.66	0.60	183.5	0.75	0.72	0.69	0.86	0.84	0.84
T-90×90×5.33×45°-200×200×6.66	0.59	208.0	0.86	0.83	0.80	0.89	0.97	0.96
T-90×90×6.66×45°-200×200×6.66	0.58	229.5	0.97	0.92	0.91	0.89	1.10	1.07
T-90×90×2.5×45°-200×200×5	0.62	84.1	0.59	0.58	0.53	1.00	0.89	0.83
T-90×90×3.25×45°-200×200×5	0.61	87.2	0.62	0.60	0.56	0.94	0.93	0.86



T-90×90×4×45°-200×200×5	0.60	89.7	0.65	0.62	0.59	0.88	0.97	0.89
T-90×90×5×45°-200×200×5	0.59	93.1	0.68	0.64	0.62	0.82	1.03	0.94
T-120×120×6×45°-200×200×12	0.80	1344.2	0.96	1.44	0.94	0.95	0.79	0.90
T-120×120×7.8×45°-200×200×12	0.78	1557.5	1.19	1.75	1.20	1.01	0.97	1.08
T-120×120×9.6×45°-200×200×12	0.77	1736.4	1.40	2.05	1.48	1.04	1.15	1.25
T-120×120×12×45°-200×200×12	0.72	1850.0	1.73	2.26	1.93	1.07	1.42	1.47
T-120×120×5×45°-200×200×10	0.81	787.3	0.78	1.12	0.73	0.89	0.68	0.78
T-120×120×6.5×45°-200×200×10	0.79	929.5	0.98	1.36	0.92	0.96	0.85	0.95
T-120×120×8×45°-200×200×10	0.78	1133.7	1.25	1.73	1.22	1.08	1.09	1.20
T-120×120×10×45°-200×200×10	0.77	1275.5	1.50	2.01	1.51	1.10	1.31	1.40
T-120×120×3.33×45°-200×200×6.66	0.82	308.3	0.65	0.93	0.58	1.05	0.71	0.78
T-120×120×4.33×45°-200×200×6.66	0.81	330.0	0.72	1.00	0.65	1.01	0.79	0.86
T-120×120×5.33×45°-200×200×6.66	0.80	400.5	0.91	1.23	0.82	1.12	0.99	1.07
T-120×120×6.66×45°-200×200×6.66	0.79	450.1	1.07	1.40	0.98	1.14	1.17	1.24
T-120×120×2.5×45°-200×200×5	0.83	154.3	0.56	0.81	0.49	1.21	0.82	0.84
T-120×120×3.25×45°-200×200×5	0.82	165.2	0.61	0.87	0.54	1.16	0.90	0.92
T-120×120×4×45°-200×200×5	0.82	172.4	0.66	0.91	0.59	1.11	0.97	0.98
T-120×120×5×45°-200×200×5	0.81	182.5	0.73	0.97	0.65	1.05	1.06	1.06
T-50×100×4×27°-150×150×6	0.55	185.7*	1.04	0.97	0.99	1.03	1.06	1.01
T-50×100×4×27°-150×150×6-R	0.55	187.8*	1.05	0.97	1.00	1.04	1.07	1.02
T-50×100×4×63°-150×150×6	0.70	258.1*	1.07	1.38	0.98	1.09	1.05	1.23
T-50×100×4×27°-120×120×4	0.68	119.4*	1.28	1.36	1.19	1.22	1.45	1.35
T-50×100×4×63°-120×120×4	0.87	226.2*	1.38	2.81	1.24	1.60	1.31	1.80
T-50×100×4×27°-120×120×3	0.69	70.8*	1.11	1.16	1.01	1.14	1.57	1.38
T-50×100×4×63°-120×120×3	0.88	137.9*	1.34	2.39	1.16	1.53	1.39	1.84
T-80×80×4×45°-120×120×4	0.87	209.4*	1.24	1.84	1.14	1.45	1.16	1.33
T-80×80×4×45°-140×140×4	0.75	142.0*	1.15	1.36	1.05	1.34	1.45	1.44
T-80×80×4×45°-120×120×3	0.88	134.6*	1.26	1.66	1.12	1.47	1.33	1.45
Mean ( $P_m$ )			0.99	1.02	0.99	1.00	1.05	1.02
COV ( $V_p$ )			0.311	0.437	0.365	0.149	0.182	0.215
Resistance factor ( $\phi$ )			1.00	1.00	1.00	0.80	0.80	0.70
Reliability index ( $\beta_0$ )			1.25	0.96	1.04	2.52	2.51	2.66

Note: \* data obtained from Pandey and Young [1].

Table 6. Comparisons between test and FE resistances with existing and proposed nominal resistances for BR X-joints.

Specimens	$\beta'$	Joint Failure		Comparisons					
		Resistances (kN)							
$X-b_1 \times h_1 \times t_1 \times \omega - b_0 \times h_0 \times t_0$	$\frac{b_1}{b_0}$	$N_{f,X}$	$\frac{N_{f,X}}{N_{Bae}^{\wedge}}$	$\frac{N_{f,X}}{N_{E,RR}^{\wedge}}$	$\frac{N_{f,X}}{N_{E,CR}^{\wedge}}$	$N_{pn1}$	$N_{pn2}$	$N_{pn3}$	
X-40×150×6×15°-200×200×12	0.34	677.8	1.23	0.88	1.19	1.16	1.08	0.99	
X-40×150×7.8×15°-200×200×12	0.32	734.9	1.35	0.95	1.32	1.20	1.19	1.09	
X-40×150×9.6×15°-200×200×12	0.31	794.4	1.48	1.03	1.45	1.24	1.32	1.18	
X-60×150×12×15°-200×200×12	0.26	1050.4	2.04	1.24	2.04	1.64	1.85	1.52	
X-40×150×5×15°-200×200×10	0.35	395.7	1.02	0.74	0.99	1.12	0.97	0.87	
X-40×150×6.5×15°-200×200×10	0.33	437.9	1.15	0.81	1.11	1.17	1.09	0.97	
X-40×150×8×15°-200×200×10	0.32	471.0	1.25	0.88	1.22	1.20	1.20	1.06	
X-40×150×10×15°-200×200×10	0.31	493.6	1.33	0.92	1.31	1.18	1.28	1.12	
X-40×150×3.33×15°-200×200×6.66	0.36	127.4	0.73	0.53	0.70	1.18	0.91	0.74	
X-40×150×4.33×15°-200×200×6.66	0.35	130.8	0.76	0.55	0.73	1.13	0.95	0.77	
X-40×150×5.33×15°-200×200×6.66	0.34	144.9	0.85	0.61	0.82	1.18	1.06	0.86	
X-40×150×6.66×15°-200×200×6.66	0.33	155.6	0.92	0.65	0.89	1.18	1.16	0.93	
X-40×150×2.5×15°-200×200×5	0.37	52.7	0.53	0.39	0.51	1.20	0.98	0.67	

X-40×150×3.25×15°-200×200×5	0.36	53.9	0.55	0.40	0.53	1.14	1.02	0.69
X-40×150×4×15°-200×200×5	0.36	53.4	0.55	0.40	0.53	1.06	1.01	0.69
X-40×150×5×15°-200×200×5	0.35	57.5	0.59	0.43	0.57	1.06	1.11	0.74
X-60×60×6×45°-200×200×12	0.37	574.8	1.00	0.84	0.96	0.91	0.87	0.92
X-60×60×7.8×45°-200×200×12	0.36	626.3	1.11	0.91	1.07	0.94	0.97	1.01
X-60×60×9.6×45°-200×200×12	0.34	667.2	1.20	0.97	1.16	0.96	1.05	1.09
X-60×60×12×45°-200×200×12	0.30	694.2	1.31	1.01	1.29	1.00	1.16	1.16
X-60×60×5×45°-200×200×10	0.38	339.5	0.85	0.71	0.81	0.89	0.79	0.82
X-60×60×6.5×45°-200×200×10	0.37	373.5	0.94	0.78	0.90	0.92	0.89	0.91
X-60×60×8×45°-200×200×10	0.36	416.9	1.07	0.87	1.03	0.98	1.01	1.02
X-60×60×10×45°-200×200×10	0.34	438.8	1.14	0.92	1.11	0.97	1.08	1.09
X-60×60×3.33×45°-200×200×6.66	0.40	109.5	0.60	0.52	0.57	0.94	0.74	0.70
X-60×60×4.33×45°-200×200×6.66	0.39	113.6	0.63	0.54	0.60	0.91	0.78	0.73
X-60×60×5.33×45°-200×200×6.66	0.38	126.9	0.71	0.60	0.68	0.95	0.88	0.82
X-60×60×6.66×45°-200×200×6.66	0.37	133.9	0.76	0.63	0.73	0.94	0.95	0.87
X-60×60×2.5×45°-200×200×5	0.40	46.5	0.45	0.39	0.43	0.97	0.82	0.65
X-60×60×3.25×45°-200×200×5	0.40	47.9	0.47	0.40	0.44	0.93	0.86	0.67
X-60×60×4×45°-200×200×5	0.39	48.4	0.48	0.41	0.45	0.89	0.87	0.68
X-60×60×5×45°-200×200×5	0.38	52.2	0.52	0.44	0.50	0.88	0.95	0.74
X-60×130×6×25°-200×200×12	0.50	804.4	1.20	0.99	1.10	0.95	0.99	1.01
X-60×130×7.8×25°-200×200×12	0.48	906.8	1.38	1.12	1.27	1.02	1.15	1.16
X-60×130×9.6×25°-200×200×12	0.47	936.2	1.46	1.16	1.34	1.01	1.22	1.21
X-60×130×12×25°-200×200×12	0.42	1011.1	1.66	1.25	1.56	1.09	1.42	1.36
X-60×130×5×25°-200×200×10	0.51	459.9	0.98	0.82	0.89	0.90	0.87	0.87
X-60×130×6.5×25°-200×200×10	0.50	507.7	1.10	0.90	1.00	0.94	0.98	0.97
X-60×130×8×25°-200×200×10	0.48	580.5	1.28	1.03	1.17	1.02	1.15	1.12
X-60×130×10×25°-200×200×10	0.47	616.2	1.39	1.10	1.28	1.02	1.26	1.21
X-60×130×3.33×25°-200×200×6.66	0.52	153.9	0.72	0.62	0.65	0.99	0.85	0.77
X-60×130×4.33×25°-200×200×6.66	0.51	158.1	0.75	0.63	0.68	0.95	0.89	0.80
X-60×130×5.33×25°-200×200×6.66	0.51	176.1	0.84	0.70	0.77	0.99	1.00	0.89
X-60×130×6.66×25°-200×200×6.66	0.49	177.0	0.86	0.71	0.79	0.93	1.03	0.91
X-60×130×2.5×25°-200×200×5	0.53	67.4	0.55	0.48	0.50	1.06	0.97	0.73
X-60×130×3.25×25°-200×200×5	0.52	69.0	0.57	0.49	0.52	1.01	1.00	0.75
X-60×130×4×25°-200×200×5	0.52	72.0	0.60	0.51	0.55	0.99	1.06	0.79
X-60×130×5×25°-200×200×5	0.51	71.5	0.61	0.51	0.55	0.91	1.06	0.79
X-75×90×6×40°-200×200×12	0.53	801.5	1.14	1.01	1.04	0.89	0.94	1.02
X-75×90×7.8×40°-200×200×12	0.51	870.6	1.27	1.10	1.16	0.91	1.05	1.12
X-75×90×9.6×40°-200×200×12	0.50	934.8	1.40	1.18	1.28	0.94	1.16	1.22
X-75×90×12×40°-200×200×12	0.45	955.4	1.52	1.20	1.41	0.97	1.28	1.30
X-75×90×5×40°-200×200×10	0.54	443.6	0.90	0.81	0.81	0.81	0.80	0.85
X-75×90×6.5×40°-200×200×10	0.52	514.1	1.06	0.93	0.96	0.89	0.95	0.99
X-75×90×8×40°-200×200×10	0.51	560.6	1.18	1.02	1.08	0.92	1.06	1.10
X-75×90×10×40°-200×200×10	0.50	646.8	1.40	1.17	1.28	1.00	1.25	1.29
X-75×90×3.33×40°-200×200×6.66	0.55	159.1	0.71	0.65	0.64	0.95	0.83	0.80
X-75×90×4.33×40°-200×200×6.66	0.54	166.3	0.75	0.68	0.68	0.93	0.88	0.85
X-75×90×5.33×40°-200×200×6.66	0.53	180.7	0.83	0.74	0.75	0.95	0.97	0.93
X-75×90×6.66×40°-200×200×6.66	0.52	191.7	0.89	0.78	0.81	0.94	1.05	0.99
X-75×90×2.5×40°-200×200×5	0.56	67.5	0.53	0.49	0.48	0.99	0.91	0.74
X-75×90×3.25×40°-200×200×5	0.55	69.0	0.55	0.50	0.49	0.94	0.95	0.76
X-75×90×4×40°-200×200×5	0.55	70.5	0.56	0.51	0.51	0.90	0.98	0.78
X-75×90×5×40°-200×200×5	0.54	75.5	0.61	0.55	0.55	0.90	1.06	0.85
X-90×90×6×45°-200×200×12	0.59	875.3	1.13	1.02	1.01	0.85	0.91	1.00
X-90×90×7.8×45°-200×200×12	0.57	972.0	1.29	1.13	1.16	0.89	1.05	1.13
X-90×90×9.6×45°-200×200×12	0.56	1060.8	1.45	1.24	1.30	0.93	1.18	1.26
X-90×90×12×45°-200×200×12	0.51	1111.0	1.63	1.30	1.48	0.98	1.34	1.38
X-90×90×5×45°-200×200×10	0.59	520.7	0.95	0.87	0.85	0.83	0.83	0.90
X-90×90×6.5×45°-200×200×10	0.58	584.1	1.10	0.98	0.98	0.88	0.96	1.02
X-90×90×8×45°-200×200×10	0.57	677.3	1.30	1.14	1.16	0.97	1.14	1.20
X-90×90×10×45°-200×200×10	0.55	730.0	1.44	1.23	1.30	0.99	1.27	1.32

X-90×90×3.33×45°-200×200×6.66	0.61	177.0	0.71	0.67	0.63	0.93	0.82	0.80
X-90×90×4.33×45°-200×200×6.66	0.60	186.6	0.76	0.70	0.68	0.91	0.88	0.85
X-90×90×5.33×45°-200×200×6.66	0.59	208.2	0.86	0.79	0.77	0.96	1.00	0.96
X-90×90×6.66×45°-200×200×6.66	0.58	224.0	0.95	0.85	0.85	0.96	1.10	1.05
X-90×90×2.5×45°-200×200×5	0.62	81.2	0.57	0.55	0.51	1.05	0.97	0.80
X-90×90×3.25×45°-200×200×5	0.61	85.0	0.61	0.57	0.54	1.02	1.03	0.84
X-90×90×4×45°-200×200×5	0.60	86.3	0.62	0.58	0.55	0.97	1.06	0.86
X-90×90×5×45°-200×200×5	0.59	89.5	0.66	0.60	0.58	0.93	1.12	0.90
X-120×120×6×45°-200×200×12	0.80	1391.2	1.00	1.22	0.86	0.83	0.78	0.93
X-120×120×7.8×45°-200×200×12	0.78	1635.0	1.25	1.44	1.08	0.92	0.98	1.13
X-120×120×9.6×45°-200×200×12	0.77	1825.3	1.47	1.60	1.27	0.99	1.15	1.31
X-120×120×12×45°-200×200×12	0.72	1885.9	1.77	1.66	1.53	1.02	1.39	1.49
X-120×120×5×45°-200×200×10	0.81	848.1	0.84	1.07	0.73	0.83	0.72	0.84
X-120×120×6.5×45°-200×200×10	0.79	943.8	0.99	1.19	0.86	0.88	0.84	0.96
X-120×120×8×45°-200×200×10	0.78	1171.1	1.29	1.48	1.12	1.03	1.10	1.24
X-120×120×10×45°-200×200×10	0.77	1313.2	1.55	1.66	1.33	1.09	1.31	1.44
X-120×120×3.33×45°-200×200×6.66	0.82	318.9	0.67	0.91	0.58	1.03	0.75	0.81
X-120×120×4.33×45°-200×200×6.66	0.81	383.1	0.84	1.09	0.72	1.15	0.94	1.00
X-120×120×5.33×45°-200×200×6.66	0.80	404.8	0.92	1.15	0.79	1.15	1.03	1.08
X-120×120×6.66×45°-200×200×6.66	0.79	435.4	1.04	1.24	0.89	1.15	1.16	1.19
X-120×120×2.5×45°-200×200×5	0.83	154.6	0.56	0.78	0.48	1.22	0.93	0.84
X-120×120×3.25×45°-200×200×5	0.82	164.8	0.61	0.83	0.53	1.21	1.02	0.92
X-120×120×4×45°-200×200×5	0.82	174.5	0.67	0.88	0.58	1.20	1.11	0.99
X-120×120×5×45°-200×200×5	0.81	184.9	0.74	0.94	0.64	1.18	1.22	1.07
X-50×100×4×27°-150×150×6	0.53	184.4*	1.07	0.89	0.97	1.11	1.09	1.03
X-50×100×4×63°-150×150×6	0.70	266.9*	1.09	1.29	0.95	1.08	1.07	1.25
X-50×100×4×27°-120×120×4	0.64	112.8*	1.33	1.17	1.17	1.37	1.57	1.36
X-50×100×4×63°-120×120×4	0.87	218.4*	1.33	2.24	1.13	1.55	1.27	1.69
X-50×100×4×27°-120×120×3	0.68	64.9*	1.03	0.99	0.90	1.23	1.62	1.27
X-50×100×4×63°-120×120×3	0.87	136.3*	1.29	2.06	1.09	1.65	1.53	1.80
X-80×80×4×45°-120×120×4	0.85	191.7*	1.10	1.50	0.96	1.44	1.24	1.32
X-80×80×4×45°-140×140×4	0.75	120.1*	0.97	1.04	0.84	1.21	1.29	1.19
X-80×80×4×45°-120×120×3	0.87	126.5*	1.17	1.43	1.01	1.57	1.48	1.41
Mean ( $P_m$ )			0.99	0.90	0.90	1.04	1.06	1.01
COV ( $V_p$ )			0.345	0.394	0.352	0.160	0.187	0.232
Resistance factor ( $\phi$ )			1.00	1.00	1.00	0.80	0.80	0.70
Reliability index ( $\beta_0$ )			1.16	0.79	0.85	2.57	2.51	2.53

Note: \* data obtained from Pandey and Young [2].

Table 7. Values of coefficients for BR T- and X-joints unified design equation.

Joint Types	Coefficients						
	A	B	C	D	E	F	G
BR T-joint	2	1	0.7	0.6	0.01	0.5	0.02
BR X-joint	2.3	0.6	0.7	0.4	0.017	0.5	0.02